# P-Delta Effect in Seismic Reinforced Concrete Portal Frames 

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Signature


## Dedication

To My Parents, My Brothers, and My Sisters

## IV

## Acknowledgments

Initially, give Allah all the praise and all the recognition. As none of us can control of anything not even ourselves, and everything happens by his order.

I must mention my parents at the beginning of my thanks for providing me with the unfailing support and continuous encouragement throughout my years of study. They have provided me with the study's environment which otherwise I could not complete this research on time. O Allah, I ask you by your mercy which envelopes all things, that you forgive them.

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Thanks also go to my nephew Mohammad and my niece Jwana for their love. May Allah bless them, keep them and give them peace.

أنا الموقع أدناه مقدم الرسالة التي تحمل عنوان

## P-Delta Effect in Seismic Reinforced Concrete Portal Frames

أقر بأن ما اشتملت عليه هذه الرسالة إنما هو نتاج جهدي الخاص، باستثناء ما تمت الإشارة اليه حيثما ورد، وأن هذه الرسالة ككل، أو أي جزء منها لم يقدم من قبل لنيل أية درجة أو لقب علمي أو بحثي للى أية مؤسسة علمية أو بحثية أخرى.

## Declaration

The work provided in this thesis, unless otherwise referenced, is the researcher's own work, and has not been submitted elsewhere for any other degree or qualification.

## Student's Name:

Signature:

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P-Delta Effect in Seismic Reinforced Concrete Portal Frames By<br>Suhib Mohammad Abu Qbeitah<br>Supervisor<br>Dr. Abdul R. Touqan


#### Abstract

P-Delta effect is a second-order nonlinear effect, which occurs in every structure when members are subjected to axial loads and bending moments and more eminent in high-rise buildings. ASCE 7 limits the base shear value using a certain period limit equation. The objective of this thesis is to prove that this limit if considered as a limit for the structure's period, which is not a code requirement; we will not run into problems of P-Delta to a certain number of storeys. Furthermore, although it is not a code requirement, the second objective of this thesis is to propose it as a code requirement.

Multi-storey buildings spread widely to accommodate the formidable number of people who need dwellings especially in metropolises. Thus, to study multi-storey effect on P-Delta analysis, P-Delta effect is investigated on six reinforced concrete portal frame multi-storey models from fivestorey to thirty-storey which abide by the proposed equation versus three models from five-storey to fifteen-storey which do not abide by the proposed equation. Seismic loads are applied on all models which are determined using modal response spectrum analysis compatible with IBC 2015 and ASCE 7 codes.


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As a final result, most probably, engineers will not run into problems with P-Delta if they build reinforced concrete portal frames until 25 storeys if the structure abides by the proposed equation and 8 storeys if the structure does not abide by the proposed equation for 10 percent increase between first-order and second-order analyses.

## Chapter One

## Introduction

## 1. 1 Preview

Beam-column is a structural member that is subjected to axial force and bending moment. In real structures, all members in a frame have both bending moments and axial forces. However, for simplicity, if the axial force effect is very small compared to the bending moment, it is more convenient to treat the member as a beam. Furthermore, if the bending moment effect is very small compared to axial force, it is more convenient to treat the member as a column. Therefore, beams or columns are special cases of beam-columns. The bending moments (and deflections) that are present in beam-columns consist of two types: primary bending moments (and deflections) which emerge from applied moments at the member and/or from transverse loads on the member; secondary bending moments (and deflections), which arise as a result of the axial force acting through the lateral displacement of the member, this phenomenon is called P-Delta effect. To account for P-Delta effect, a second-order analysis is needed. First-order analysis is referring to writing the equilibrium and kinematic relationships on the original geometry of the structure, whilst, second-order analysis is referring to writing equilibrium and kinematic relationships on the deformed configuration of the structure. Generally, second-order analysis needs an iterative procedure to obtain the desired solution. The
reason for that is the deformed configuration of the structure is not known and the solution needs to be repeated many times until reaching reasonable solution for the problem. The deformed configuration which is obtained from current phase is used as the basis for formulation of the subsequent phase of iterations.

There are two distinct types of P-Delta effect can be identified: the P- $\delta$ effect (member P-Delta effect), and the P- $\Delta$ effect (structure P-Delta effect). The P- $\delta$ effect has reference to the effects of the axial force in an individual member subjected to a deflection between its endpoints. P- $\Delta$ effect has reference to the effects of vertical loads acting on the laterally displaced structure. These secondary effects cause the member to deform more and induce additional stresses in the member. As a result, they have an undermining or destabilizing effect on the structure. So, in the seismic design of multi-storey structures allowance should be made for P-Delta effect.

This research deals with P-Delta effect in seismic portal frames, because Palestine and its conterminous countries are located in a highly seismic zone. Their seismicity derives from their location near the Dead Sea Fault, which has a length equals 1000 Km . It extends from Al Aqaba Gulf northward along the Wade Araba, Dead Sea and Tiberias to central Lebanon, and fades away in the south of Turkey. Palestine has a long history in earthquakes over millennium of years. It has documentation and information of the historical earthquakes for almost 2000 years. One of the oldest documentations about earthquakes in Palestine is in the Bible. No
region on earth has this long documentation and literature for earthquake events. Information about these ancient earthquakes can be found by archaeological findings. As a summary, Palestine has a long heritage of earthquakes which makes it a vulnerable region for major earthquakes at any time in the future.

Shear wall or dual systems are commonly used in Palestine to resist seismic loads, whereas moment resisting frames are less frequently used for such purposes. In this study, moment resisting frames are used. Moment resisting frames are rectilinear assemblages of columns and beams. Resistance to seismic loads is provided primarily by rigid frame action thatis, by developing bending moments and shear forces in frame members and joints. In moment resisting frames, the joints which connect beams and columns are designed to be rigid. This causes the columns and beams to bend during earthquakes.

### 1.2 Literature Review

Studies on the parameters the affect P-Delta analysis have been conducted on several fronts. Researchers have worked independently in understanding and quantifying these parameters as follows:
A. Gupta et al (2000) investigated the seismic response on three models of 3,9 , and 20 stories of ductile steel moment resisting frames located in regions with different seismicity levels (Los Angeles, Seattle and Boston), and subjected to seismic loads with different return periods (72,474 and

2475 years) and the seismic response was investigated. They concluded that the seismic response becomes very sensitive to building models when P-Delta was included.
U. Ashraf et al (2013) studied P-Delta effect in reinforced concrete portal frames with rigid joints. Their study was on 12 models with different heights, they observed that displacements change in exponential way when changing number of stories or structure's height under P-Delta effect. They concluded that P-Delta effect is necessary to be included in reinforced concrete high-rise buildings.
D. Yousuf et al (2013) studied the effect of global slenderness on P-Delta analysis in high-rise buildings. The structural analysis program STAAD Pro v8i was used to analyze 4 different models with different slenderness ratios and the percent variation between P-Delta analysis and first-order analysis is computed. It was concluded that due to wide displacement variation with increasing slenderness, P-Delta analysis is required for structures higher than 7 stories.
R. Vijayalakshmi et al (2017) presented a method of designing of P-Delta effects in high rise buildings. P-Delta effect for different number of stories from ten to forty stories was investigated. The load deflection curves and the results so obtained have been compared. The results of the analysis showed that P-Delta effect is more in the upper stories.
N. Pravin et al (2015) investigated the P-Delta effect on multi-storey buildings by taking three models (15, 20 and 25 storey). If the change in
bending moments is more than $10 \%$, P -Delta should be considered in design. Earthquake load is applied for zone III in ETABS software. The results show that it is essential to consider the P-Delta effect for 25-storey building. So buildings having height more than or equal to 75 m , should be designed considering P -Delta effect.
C. Konapure and P. Dhanshettei (2015) studied the P-Delta effect in multi-story buildings. Twelve models were analyzed using STAAD Pro V8i structural analysis program from 5 to 27 stories with 2-storey increment. Displacements, storey drifts, and column moment were computed. It was concluded that as number of stories increase, the P-Delta effect will become more predominant. Moreover, P-Delta effect is negligible up to 7-storey.

### 1.3 Objectives and Scope

The main objective of this dissertation is to prove that the period limit in ASCE 7 which limits the calculations of base shear (i.e. section 12.8.2, ASCE 7) has implicitly another meaning which is if the structure's period abides by this equation; the structure most probably will not run into problems of P-Delta. The ASCE 7 does not require satisfying this limit for the structure's fundamental period, so, the second objective of this thesis is to propose this limit as a code's requirement. This thesis aims to show that abiding by this limit will implicitly keep the structure away from P-Delta analysis to a certain number of storeys. Thus, if the structure is stiffened to
abide by this proposed equation; the structure will not run into P-Delta problems.

### 1.4 Methodology

Six reinforced concrete models which abide by the proposed equation (i.e. section 12.8.2, ASCE 7) with different number of storeys (5-storey, 10storey, 15-storey, 20-storey, 25-storey, and 30-storey) are analyzed on SAP2000 software by applying vertical loads and earthquake loads (i.e. response spectrum analysis) and the percent increase in second order moments compared to first order moments on all columns are determined and the critical column (i.e. the highest percent increase) is adopted. The previous analysis is repeated on the same models nevertheless are not abiding by the proposed equation. The relationships between percent increase and the number of storeys for the models which abide and defy by the proposed equation are drawn. The stability coefficients are also determined for all models and compared with the percent increase between first-order analysis and second-order analysis on the critical column. Previous objectives are investigated on reinforced concrete portal frames models using SAP2000, version 19.0 software with hand calculations when it is needed. International Building Code 2015 (IBC 2015) and Minimum Design Loads for Buildings and Other Structures (ASCE 7-10) are adopted in the entire course of this study.

### 1.5 Thesis's Structure

Chapter 2 expounds both member and structure P-Delta effects and reviews the methods which deal with P-Delta effect on an element or system levels, with clarifying the advantages and disadvantages of each method. Furthermore, it clarifies the main concepts that can be used in second-order analysis.

In Chapter 3, SAP2000, version 19.0.0 results in both static and dynamic analyses are verified. The verification on static results is made by computing the P-Delta effect on a portal frame by both SAP2000 software and hand calculations using approximate equations and the results from both methods are compared. The verification on dynamic results is made by computing the natural period for a portal frame by both SAP2000 software and hand calculations and the results are compared.

Chapter 4 deals with studying the effect of P-Delta analysis on a six models that abide by the proposed equation which increase number of storeys by five-storey increment from 5-storey to 30 -storey subjected to response seismic analysis loads versus studying the same effect on another three models which also increase number of storeys by five-storey increment from 5-storey to 15 -storey nevertheless they do not abide by the proposed equation. Moreover, the relationship between the percent increase between first-order analysis and second-order analysis and the number of storeys is drawn and tabulated. Furthermore, a comparison between the percent increase and stability coefficients is tabulated for both the abiding and defying models.

Chapter 5 summarizes the conclusions of this research and it suggests numerous recommendations for practicing engineers and future studies.

## Chapter Two

## Synthesis of Current Information

## 2. 1 Introduction

P-Delta effect, which is a second-order nonlinear effect, occurs in every structure where elements are subjected to axial load and bending moment and more prominent in tall buildings. P-Delta effect holds an essential role in analysis of the structure. The P-Delta effect is mainly dependent on the applied loads, building's material and structure geometry. Examples of geometry are the height, stiffness, and asymmetry of the structure. Building asymmetry leads to imbalance in distribution of mass, stiffness, etc. Buildings were constructed without considering the second order P Delta effect and appropriate characteristic for seismic resistance constitute the main source of risk during an earthquake.

Studies on P-Delta effects have been conducted on several fronts. Researchers have worked independently in understanding and quantifying P-Delta effects and in developing methods to account for P-Delta effects. Secondary moments generally produce injurious effects to slender compression members and so they must be taken into consideration in design. The methods in which the secondary moments are integrated in the analysis and design of structural members in reinforced concrete portal frames are gathered in this and the following sections.

Many approaches to account for both member and structure P-Delta effects are discussed in this chapter. General comments on their advantages and disadvantages are also made.

### 2.2 P- $\delta$ Effect

To brief this effect which is called member P-Delta effect, a beam-column is shown in Fig. 2.1 with joint translation restrained and rotation permissible. The end-moments $M_{J}$ and $M_{K}$ and the point-load $N$ will produce first-order moment $\mathrm{M}_{1}$ and first-order deflection $\delta_{1}$. When the beam-column's member is subjected to axial force P , the axial force will act on the first-order deflection to produce second-order moment $\mathrm{M}_{2}$ and secondary deflection $\delta_{2}$. These second-order moments and deflections have been caused by P- $\delta$ effect. Since P- $\delta$ effect will boost the instability of the member, it is mentioned here as the member instability effect.

The second-order effect can be accounted for by using one of the following methods of analysis:

Closed-form solution: The closed-form solution to account for member's P-Delta effect can be found by solving the governing differential equation which describes the behavior of beam-column after deformation in the elastic range with the proper boundary conditions as follows:

$$
\begin{equation*}
\mathrm{EI} \frac{\mathrm{~d}^{4} v}{\mathrm{dx}^{4}}+\mathrm{P} \frac{\mathrm{~d}^{2} v}{\mathrm{dx}^{2}}=\mathrm{w} \tag{2.1}
\end{equation*}
$$



Figure 2.1 P- $\delta$ Effect [3].

Where
$\mathrm{E}=$ material's modulus of elasticity
$\mathrm{I}=$ section's moment of inertia
$\mathrm{v}=$ member's deflection in y -direction
$\mathrm{x}=$ variable distance along the member
$\mathrm{w}=$ transverse distributed load
$\mathrm{P}=$ axial force on the member
The closed-form solution is possible for members without material nonlinearity. If yielding or inelasticity occurs in the member, the moment-curvature-thrust relationship becomes non-linear. In such cases, the use of formal mathematics for the solution of the governing differential equation becomes intractable and engineers must move to numerical methods to obtain solutions. In some cases, closed-form solution is still possible if simplified assumptions have been made with respect to stress-strain relationship, and the deflection shape of the member. Generally, the numerical approaches are the best procedure to follow.

Numerical methods to account for $\mathbf{P}-\boldsymbol{\delta}$ effect: For simplicity, the yielding of the material is ignored and the focus of this study is on the elastic range of the member. Numerical methods in contrast to closed-form solution can account for $\mathrm{P}-\delta$ effect in both elastic and inelastic range. In this section, four numerical methods to account for P- $\delta$ effect are discussed by the author: (1) the finite element approach, (2) the pseudo load approach, (3) the beam-columns approach, (4) moment amplification method. The following is a brief of each method:

Elastic analysis using finite element approach [37]: This approach gained a lot of approval in the engineering field. The finite element
approach uses the energy theorem. There are different ways by which finite element can be created. The most common one is displacementbased formulation. In this approach, the displacement fields for both the axial deformation and transverse deformation are assumed for the element. The stiffness matrix which relates the nodal displacements to the nodal forces is attained by minimizing the total potential energy according to the principle of stationary potential energy, which states that equilibrium is attained when the variation of the potential energy vanishes. The main advantage of this approach is that it can solve complicated problems which do not have a closed-form solution. Its deficiency is in the chosen of element size which affects convergence characteristics, since if the member is subjected to large axial force; it is suitable to use small element size. The second deficiency of this method is the displacement fields for both axial and transverse deformations which their assumption does not represent the real element behavior. This method is used in SAP 2000 software to account for P-Delta effect.

We are all aware that a member has a larger lateral stiffness when subjected to tension force and a lower lateral stiffness when subjected to compression force. This general type of behavior is caused by change in the geometric stiffness of the structure. It is obvious that this stiffness is a function of the load in the member whether positive (tension) or negative (compression). The fundamental equation for the geometric stiffness of a cable element is easy to derive. Consider the horizontal cable in Fig. 2.2 with length L and tension force $T$. If the cable is subjected to lateral displacements $V_{i}$ and $V_{j}$,
then forces $F_{i}$ and $F_{j}$, shall be produced. Note that it is assumed all forces and displacements are positive in the up direction. Furthermore, it is assumed that the displacements are small and do not change the tension in the cable.


Figure 2.2 Forces Acting on a Cable Element

Taking moments about point j in the deformed position, the following equilibrium equation can be written.

$$
\begin{equation*}
\mathrm{F}_{\mathrm{i}}=\frac{\mathrm{T}}{\mathrm{~L}}\left(\mathrm{v}_{\mathrm{i}}-\mathrm{v}_{\mathrm{j}}\right) \tag{2.2}
\end{equation*}
$$

The lateral forces can be expressed in the following matrix form:

$$
\left[\begin{array}{l}
F_{i} \\
F_{j}
\end{array}\right]=\frac{T}{L}\left[\begin{array}{cc}
1 & -1 \\
-1 & 1
\end{array}\right]\left[\begin{array}{l}
v_{i} \\
v_{j}
\end{array}\right]
$$

$$
\begin{equation*}
\text { Or symbolically } \mathrm{F}_{\mathrm{g}}=\mathrm{k}_{\mathrm{g}} \mathrm{v} \tag{2.3}
\end{equation*}
$$

Note that the geometric stiffness matrix is a function of element's length and the force in the element. The geometric stiffness for a beam element can be derived in the same way by adding bending properties in which the deformed shape is assumed to be a cubic function caused by rotations $\Phi_{\mathrm{i}}$ and $\Phi_{j}$ at the ends, additional moments $\mathrm{M}_{\mathrm{i}}$ and $\mathrm{M}_{\mathrm{j}}$ are developed. The force-displacement relationship is given by the following matrix:

$$
\left[\begin{array}{c}
F_{i} \\
M_{i} \\
F_{j} \\
M_{j}
\end{array}\right]=\frac{T}{30 L}\left[\begin{array}{cccc}
36 & 3 L & -36 & 3 L \\
3 L & 4 L^{2} & -3 L & -L^{2} \\
-36 & -3 L & 36 & -3 L \\
3 L & -L^{2} & -3 L & 4 L^{2}
\end{array}\right]\left[\begin{array}{c}
v_{i} \\
\phi_{i} \\
v_{j} \\
\phi_{j}
\end{array}\right]
$$

$$
\begin{equation*}
\text { Or symbolically } \mathrm{F}_{\mathrm{G}}=\mathrm{k}_{\mathrm{G}} \mathrm{~V} \tag{2.4}
\end{equation*}
$$

Elastic analysis using pseudo load approach [38]: Pseudo approach uses analog between beam-column and a beam. Previous discussed approach (finite element approach) accounts for second-order effect by changing the stiffness matrix throughout the course of analysis, but pseudo load approach accounts for second-order effect in pseudo loads. This means that the same stiffness matrix can be used in all approach's cycles. Another advantage of this method over finite element approach is that the finite element cares about mesh size which affects the results, on the other hand, in pseudo load approach; the analysis does not depend on the way in which the member is divided. The main concept of this method is that the differential equation which governs beam-column with distributed load,
end-moments and axial force is as discussed in Eq. 2.1, if the previous equation is rearranged, it can be written as follows:

$$
\begin{equation*}
\text { EI } \frac{\mathrm{d}^{4} v}{\mathrm{dx}^{4}}=\mathrm{w}-\mathrm{P} \frac{\mathrm{~d}^{2} v}{\mathrm{dx}^{2}} \tag{2.5}
\end{equation*}
$$

Where
$\frac{\mathrm{d}^{2} \mathrm{v}}{\mathrm{dx}^{2}}=$ the second derivative of member's deflection with respect to variable distance along the member

## $\mathrm{E}, \mathrm{I}, \mathrm{w}, \mathrm{P}, \mathrm{v}, \mathrm{x}$ are as defined before

For small displacements, we can write:

$$
\begin{equation*}
\frac{\mathrm{d}^{2} \mathrm{v}}{\mathrm{dx}}=-\frac{\mathrm{M}}{\mathrm{EI}} \tag{2.6}
\end{equation*}
$$

Where
$\mathrm{M}=$ the bending moment.

Eq. 2.5 and Eq. 2.6 give

$$
\begin{equation*}
\mathrm{EI} \frac{\mathrm{~d}^{4} \mathrm{v}}{\mathrm{dx}^{4}}=\mathrm{w}+\mathrm{M} \frac{\mathrm{P}}{\mathrm{EI}} \tag{2.7}
\end{equation*}
$$

The differential equation of equilibrium for a beam member is as follows:

$$
\begin{equation*}
E I \frac{d^{4} v}{d x^{4}}=w \tag{2.8}
\end{equation*}
$$

From the analog between Eq. 2.7 and Eq. 2.8, it can be concluded that, the difference between beam and beam column is the term $M \frac{P}{E I}$, so if the bending moment in each cycle is multiplied by the factor $\frac{P}{E I}$ to give the pseudo loads; the new moment can be determined. This process keeps
repeating itself and the results of each cycle are used in the successive cycle until the value of $M$ remains almost constant.

Elastic analysis using beam-columns approach [3]: In this method, the stiffness matrix which relates the member's forces and the member's displacements of a beam-column's member is created from the modified slope-deflection equations which take into account the effect of axial force on the member's bending stiffness. Second-order effect is accounted for in the stiffness matrix; this means that stiffness matrix is updated during the course of analysis. As a result, the solution can only be obtained by iterative procedure. The main disadvantages of this method are as follows: (1) its complexity, because in every case the stiffness matrix's formula needs to be modified. For example, if in-span loads apply at the member, the stiffness matrix formula must be changed, or if the member is not horizontally oriented, the stiffness matrix must be transformed into global coordinates, (2) the effect of curvature shortening (i.e. bowing effect) since if the bending moments are large, its effect on axial deformation must be included in the term EA/L in the stiffness matrix.

Moment amplification method [3]: In this method, the first-order moments obtained from a first-order analysis are magnified by amplification factors to account for member P- $\delta$ effect. The total moment (and deflection) is equal to the sum of primary and secondary moments. In this method, the form of secondary moment is assumed (e.g. half sine, parabola, etc.), and then after rearranging equations, the maximum moment
is derived, which equals to the first-order moment multiplied by an amplification factor. The following amplification factor is derived based on secondary moment in the form of half sine wave (see theory of beamcolumns Volume 1).

$$
\begin{equation*}
M_{\max }=\left(\frac{\mathrm{C}_{\mathrm{m}}}{1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{ek}}}}\right) \mathrm{M}_{\mathrm{lmax}}=\mathrm{B}_{1} \mathrm{M}_{\mathrm{lmax}} \tag{2.9}
\end{equation*}
$$

Where
$\mathrm{M}_{\text {max }}=$ maximum moment in the member

$$
C_{m}=1+\psi \frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}
$$

$\psi=\frac{\delta_{1} \mathrm{P}_{\mathrm{e}}}{\mathrm{M}_{1 \text { max }}}-1$
$\mathrm{P}=$ axial load
$P_{\mathrm{ek}}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}$ is the elastic buckling load considering end restraining conditions
$\mathrm{B}_{1}=\mathrm{P}-\delta$ moment amplification factor
$\mathrm{M}_{1 \text { max }}=$ maximum first-order moment
Table 2.1 summarizes $C_{m}$ and $\psi$ values for commonly beam-columns with transverse loading.
$\psi$ is a factor with values equal or less than zero. It equals zero when the member is subjected to distributed load and the joint translation is prevented, which makes the member to deflect in a single curvature which implies maximum value for $\mathrm{C}_{\mathrm{m}}$ factor which leads to maximum secondorder effect. When the value of $\psi$ increases in negative; this means the second-order effect will decrease (i.e. $\mathrm{C}_{\mathrm{m}}$ will decrease).

## $\mathbf{C}_{\mathrm{m}}$ factor for beam-column subjected to end moments

$\mathrm{C}_{\mathrm{m}}$ factor to beam-columns subjected to end moments is referred to the concept of equivalent moment factor; equations to account for P-Delta effect have been derived for the general case of a beam-column which is subjected to unequal end-moments. The maximum moment in the member may occur at the end of the member, or it may occur somewhere within its length. That's why the need came to use of the equivalent moment factor. The concept of equivalent moment factor is shown in Fig. 2.3. The end moments $M_{A}$ and $M_{B}$ are replaced by a couple of equal and opposite equivalent moment $\mathrm{M}_{\mathrm{Eq}}$. The magnitude of the equivalent moment is such that the maximum moment generated by it will be equal to that generated by $\mathrm{M}_{\mathrm{A}}$ and $\mathrm{M}_{\mathrm{B}}$.

The equivalent moment factor $C_{m}$ is a function of $M_{A}, M_{B}$ and axial force $P$, where $M_{A}$ is the smaller and $M_{B}$ is the larger. $M_{A} / M_{B}$ is positive when the member bends in double curvature (i.e. $\mathrm{M}_{\mathrm{A}}$ and $\mathrm{M}_{\mathrm{B}}$ rotate in the same direction) and negative when the member bends in single curvature (i.e. $\mathrm{M}_{\mathrm{A}}$ and $\mathrm{M}_{\mathrm{B}}$ rotate in opposite direction). Approximate expressions have been proposed that eliminate its dependency on axial force P as follows:

Table 2.1 Values of $\Psi$ and $\mathbf{C m}$ [4].

| Case | $\psi$ | $C_{m}$ |
| :---: | :---: | :---: |
| 1 | 0 | 1.0 |
| $2$ | -0.4 | 1-0.4 P/P ${ }_{8 k}$ |
| $3$ <br> - | -0.4 | $1-0.4 \mathrm{P} / \mathrm{P}_{\mathrm{ek}}$ |
|  | -0.2 | $1-0.2 \mathrm{P} / \mathrm{P}_{\mathrm{ek}}$ |
|  | -0.3 | $1-0.3 \mathrm{P} / \mathrm{P}_{\mathrm{ek}}$ |
| 6 | -0.2 | 1-0.2 P/P Pk |



Figure 2.3 Representation of the Concept of Equivalent Moment [2].

Massonnet expression (1959)
Massonnet proposed the following equation:

$$
\begin{equation*}
C_{m}=\sqrt{\left(0.3\left(\frac{M_{A}}{M_{B}}\right)^{2}-0.4\left(\frac{M_{A}}{M_{B}}\right)+0.3\right)} \tag{2.10}
\end{equation*}
$$

It is worth to be mentioned that $\mathrm{C}_{\mathrm{m}}$ is not function of the axial force in Massonnet expression. When the axial force is large, the Massonnet expression gives reasonable results. However, when the axial force is small, it gives unreasonable results.

## Austin expression (1961)

The Austin expression has the form

$$
\begin{equation*}
C_{m}=0.6-0.4\left(\frac{M_{A}}{M_{B}}\right) \quad \geq 0.4 \tag{2.11}
\end{equation*}
$$

Similar to the Massonnet expression, in Austin expression $\mathrm{C}_{\mathrm{m}}$ is not a function of the axial force in the member. In sway frames, the columns usually subjected to large axial force and bent in double curvature, which are the conditions at which the Austin expression gives conservative results. Furthermore, the Austin expression gives unconservative results if the axial force is small and if the columns bent in single curvature. Moreover, the condition that $\mathrm{C}_{\mathrm{m}}$ must be equal or larger than 0.4 has been removed because it is a very conservative condition.

## Duan-Sohal-Chen expression (1989)

As can be seen form Fig. 2.4, the Austin expression gives unconservative results for small axial forces (i.e. approximately $\mathrm{P} / \mathrm{P}_{\mathrm{e}} \leq 0.68$ from Fig 2.4) and if $M_{A} / M_{B}$ is negative (i.e. the member bents in singe curvature). To
handle this unwanted situation, Duan-Sohal-Chen (1989) have proposed amended expression which takes into account the axial force's value as follows:

$$
\begin{equation*}
C_{m}=1+0.25\left(\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}\right)-0.6\left(\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}\right)^{\frac{1}{3}}\left(\frac{\mathrm{M}_{\mathrm{A}}}{\mathrm{M}_{\mathrm{B}}}+1\right) \tag{2.12}
\end{equation*}
$$

The various expressions for $\mathrm{C}_{\mathrm{m}}$ are plotted in Fig. 2.5, as can be observed; the approximate expressions give to some extent a pretty approximation to the theoretical one. Because of its straightforwardness, the Austin expression has been adopted by numerous design codes.


Figure 2.4 Comparison between Exact Amplification Factor and Austin's Expression [3].


Figure 2.5 Comparison between Various Expressions of $\mathrm{C}_{\mathrm{m}}$ and Theoretical Expression [2].

### 2.3 P- $\Delta$ effect

As a result of applying a horizontal force $H$, the frame moves until reaching an equilibrium position as shown in Fig. 2.6. The corresponding deflection which is computed on the un-deformed configuration is called the firstorder deflection and is denoted by $\Delta_{1}$. If a vertical force P is applied; it drifts the frame further by interaction with the lateral displacement $\Delta_{1}$ caused by force H until reaching a new equilibrium position. The lateral deflection which coincides with the new equilibrium position is denoted by $\Delta$ (Fig. 2.7). The aforementioned incident by which the vertical axial force P interacts with the first-order deflection $\Delta_{1}$ which is caused by the lateral force H is called the structure P-Delta effect, which leads to increase in both deflection and overturning moment. For the reason that the additional deflection and overturning moment have prejudicial effects on the stiffness
and stability of the frame, they should be taken into account in analysis and design.

In order to calculate precisely the final deflection $\Delta$ and the second-order moment M taking into consideration the structure $\mathrm{P}-\Delta$ effect, a nonlinear static analysis based on the deformed configuration of the structure is needed. Second-order analysis usually requires an iterative procedure. To account for P- $\Delta$ effect, it can be used either closed-form solution or approximate solutions.

Closed-form solution: The differential equation on the deformed configuration of the frame can be written with proper boundary conditions. However, closed-form procedure is highly complex for the entire frame and the required limitations are rarely met in reality. Thus, the numerical solutions are the best approaches to follow.


Figure 2.6 Equilibrium Position before P- $\Delta$ Effect [3].


Figure 2.7 Equilibrium Position after $\mathrm{P}-\Delta$ Effect [3].
Numerical methods to account for $\mathbf{P}-\Delta$ effect: The previous explained approaches (i.e. finite element approach, pseudo load approach, and beamcolumn approach) still applicable to account for structure P- $\Delta$ effect. Two practical moment amplification methods are discussed in this section as follows:

Story magnifier method (Rosenblueth, 1965; Stevens, 1967; Cheong-SiatMoy, 1972)

This method has been proposed by Rosenblueth in 1965 and then modified and developed by Stevens in 1967 and Cheong-Siat-Moy in 1972. This method contains two assumptions which facilitate the method of solution. The first assumption is that each story behaves independently of other stories. Second assumption is that the second-order moment which has resulted from P- $\Delta$ effect is equivalent to moment caused by a lateral force. From previous assumptions, we can say that, the story sway moments are directly proportional to the lateral deflections of the story.

$$
\begin{equation*}
\mathrm{M}=\left(\frac{1}{1-\sum \mathrm{P} \Delta_{1} / \sum \mathrm{Hh}}\right) \mathrm{M}_{1 \text { sway }}=\mathrm{B}_{2} \mathrm{M}_{1 \text { sway }} \tag{2.13}
\end{equation*}
$$

Where
$\mathrm{M}=$ maximum end moment accounting for $\mathrm{P}-\Delta$ effect
$\mathrm{M}_{1 \text { sway }}=$ maximum primary moment due to swaying of the story
$\mathrm{B}_{2}=\mathrm{P}-\Delta$ moment amplification factor
$\Delta_{1}=$ First-order deflection
$\sum \mathrm{P}=$ summation of axial forces
$\sum \mathrm{H}=$ summation of lateral forces
From Eq. (2.13), it can be concluded that to account for $\mathrm{P}-\Delta$ effect, the first-order moment is computed and then multiplied by the moment amplification factor to account for second-order effect. This method gives acceptable results if the beams are stiff (i.e. inflection point exists in every column). The main disadvantage of this method is that it is a function of first-order deflection. Wherefore, to determine the second-order moment, a first-order analysis is needed.

Multiple-column magnifier method (Yura, 1971)
This method has been proposed by Yura in 1971. It has another name known as the modified effective length method. By assuming that the story sway moments are directly proportional to the lateral deflections of the story, the maximum end moments M accounting for the $\mathrm{P}-\Delta$ effect can be written:

$$
\begin{equation*}
\mathrm{M}=\left(\frac{1}{1-\sum \mathrm{P} / \sum \mathrm{P}_{\mathrm{ek}}}\right) \mathrm{M}_{1 \text { sway }}=\mathrm{B}_{2} \mathrm{M}_{1 \text { sway }} \tag{2.14}
\end{equation*}
$$

Where, $\mathrm{M}, \mathrm{M}_{1 \text { sway }}, \mathrm{B}_{2}, \sum \mathrm{P}$ are as defined before
$\mathrm{P}_{\mathrm{ek}}$ is elastic buckling load considering column end restraining conditions The advantage of this method is that its simplicity since it is not function of first-order deflection. The main disadvantage of this method is that if the $\mathrm{P}-\Delta$ effect is large, the story magnifier method will give preferable results.

### 2.4 Second-order analysis using solution algorithms:

In previous beam-column and finite element approaches, the account for nonlinear effect was by updating the stiffness matrix in every cycle. As the equilibrium configuration of the frame keeps changing, subsequently, the analysis can be carried out using a set of load increments. The states of equilibrium and kinematic relationships at the end of preceding load increment are used to formulate the stiffness relationships at the current load increment. Different schemes were made to determine the size of the load increment. As the size of the load increment has a strong impact on convergence of the problem and the solution time, it is assumed based on previous experiences. Two main used approaches are discussed in this section as follows: (1) load control method, (2) displacement control method. The main concept in both previous methods is to start with a known solution point on the equilibrium path to obtain another solution point on the path. The various solution schemes differ by the technique in which the load increment's size is determined and the manner in which iterations are carried out. By iterating the procedure, the whole loaddeflection curve can be traced. The solution after x iterations will be
converged, and the load and displacement are computed. The updated matrix can then be constructed and the solution steps to the next increment.

### 2.4.1 Load control method [3]

In this method, the load increment's size at every load step is determined as a percentage of total applied loads and this percentage is assumed based on the engineering decision and previous experiences. This method can be used in iterative or non-iterative procedures. If it is used in a non-iterative procedure, it is called simple incremental method, and if it used in an iterative procedure, it is called load control Newton-Raphson method.

### 2.4.1.1 Simple incremental method

As discussed before that the load increment's size is a percentage of total applied loads. The main concept of this method is that the stiffness matrix from preceding load step (i.e. i-1) and the load increment at load step $i$ are used to determine the displacement at the load step i . Thereafter, the displacement is added to previous load step's displacement to obtain a new cumulative displacement. Eventually, the stiffness matrix for load step i can be updated and the solution can go ahead to the next step. The advantage of this method is that it does not imply any iteration; consequently, it is a straightforward procedure to follow. Simultaneously, the main disadvantage of this method is that it does not contain any iteration to eliminate the drift-off-error between actual equilibrium path and the calculated equilibrium path as shown in Fig. 2.8.


Figure 2.8 Drift-off-Error in Simple Incremental Method [3].

### 2.4.1.2 Load control Newton-Raphson method

This method has the same concept as simple incremental method. However it has iterations to eliminate the drift-off-error. The iterations in every load step continue until drift-off-error becomes negligibly small as shown in Fig. 2.9. The advantage of this method is that the correction is made by subjecting the frame to unbalanced forces (i.e. difference between internal and external forces). The disadvantages of this method are : (1) the solution diverges at the limit point and (2) a number of iterations and errors may be required to determine the appropriate load increment (3) the stiffness matrix need to be updated in every iteration during each load step.


Figure 2.9 Load Control Newton-Raphson Method [3].

### 2.4.1.3 Modified Load control Newton-Raphson method

This method differs from load control Newton-Raphson method in that the same stiffness matrix is used for each load step and the stiffness matrix is updated occasionally which is considered an advantage for this method as shown in Fig. 2.10. Thereby, the computational time for the problem is decreased. The main deficiency of this method is that it has a slow rate of convergence.


Figure 2.10 Modified Newton-Raphson Method [3].

### 2.4.2 Displacement control method [39]

This method implies a constant displacement instead of a constant load as such the case of simple incremental method or load control-NewtonRaphson method as shown in Fig. 2.11. A displacement variable is prescribed, and then applying equations to determine the load displacement increment for current iteration based on previous stiffness matrix iteration, then, the load increment factor for current iteration within the same load step is determined, thereafter, the displacement increment which includes displacement increment associated with unbalanced forces can be updated. The process keeps repeating until convergence is achieved, then a new load step can be started. The main advantage of this method is that because it is
performed based on a constant displacement it is applicable for limit point. On the other hand, it has a deficiency in how to choose the proper control displacement which affects the convergence characteristics.


Figure 2.11 Displacement Control Method [3].

### 2.5 Moment-curvature-thrust relationship

The simplest cross section to obtain a closed-form solution for M-Ф-P relationship under uniaxial bending is a rectangular cross section. However, for general cross-sections under biaxial bending, numerical methods are used to obtain the moment-curvature-thrust relationship. For rectangular cross-section under uniaxial bending, the $\mathrm{M}-\Phi-\mathrm{P}$ relationship can be derived by using the integration for axial force and bending moment
on the cross section area, then depending on the stress distribution (i.e. elastic, primary plastic and secondary plastic stress distribution); the moment-curvature-thrust relationship can be obtained. If the axial deformation is large, $\mathrm{P}-\varepsilon-\mathrm{M}$ relationship (i.e. relationship between axial force P and axial deformation $\varepsilon$ for certain values of bending moments) is needed. Generally, beam-columns have large flexural deformation compared to axial deformation; consequently, $\mathrm{M}-\Phi-\mathrm{P}$ relationship is required. Fig 2.12 shows M-Ф-P relationship for rectangular cross section. It can be seen that, the moment capacity of the member decreases as the magnitude of axial force increases which is a foreseeable result.

The derivation of preceding $\mathrm{M}-\Phi-\mathrm{P}$ curves for rectangular cross-section is done by assuming that:
(1) Plane sections before bending remain plane after bending (i.e. EulerBernoulli beam theory).
(2) The geometry of the cross-section does not change after section is subjected to loads.
(3) The material follows elastic-perfectly plastic behavior in stressstrain relationship.

For general cross-sections which the material manifests a more complex stress-strain behavior, closed-form solutions for $\mathrm{M}-\Phi-\mathrm{P}$ relationship become difficult. So, the M-Ф-P relationships can be derived by computerbased numerical procedures.


Figure 2.12 M-Ф-P Curves for Rectangular Cross-Section [2].

### 2.6 Effective length factor $K$

The effective length factor $K$ is defined as a factor when multiplied by the member's length of end-restrained column; it gives the equivalent length of a pinned-pinned column as shown in Fig. 2.13. Furthermore, this equivalent length has the same buckling load as the end-restrained column. Since the moment at hinge equals zero, the effective length factor K is the distance between two adjacent points of zero curvature. Mathematically, the effective length factor K is given by the following formula:

$$
\begin{equation*}
\mathrm{K}=\sqrt{\frac{\mathrm{Pe}}{\mathrm{Pcr}}}=\sqrt{\frac{\pi^{2} \mathrm{EI}}{\mathrm{~L}^{2} \mathrm{Pcr}}} \tag{2.15}
\end{equation*}
$$

Where
$\mathrm{P}_{\mathrm{e}}=$ Euler buckling load
$\mathrm{P}_{\mathrm{cr}}=$ elastic buckling load.
$\mathrm{L}=$ unsupported length of the column.
E and I are as defined before.
Table 2.2 summarizes some theoretical and recommended values for K factor for isolated columns. As can be seen from this table, the recommended design values are greater than theoretical ones to be more conservative in design. With regards to columns in a framework, the effective length factor $K$ theoretically can be determined from stability functions analysis (i.e. eigenvalue analysis). However, in practice, such analysis is not practical. Wherefore, a simple approach is discussed by the author that was developed by Julian and Lawrence (1959). This procedure has been adopted by many design codes among them ACI 318, AISC, and AASHTO specifications. This method is rather simple and straightforward to obtain the effective length factor K by using the proper alignment chart. In order to clarify the idea, the determination of K -factor for member in both braced and in an unbraced frame is discussed separately. The design of a beam-column starts with the determination of the stiffness ratio of columns and girders $G_{A}$ and $G_{B}$, from which the effective length factor $K$ is determined.

35


Figure 2.13 Isolated Column [18].

Table 2.2 Theoretical and Recommended K Factor [18].

| Buckled shape <br> of column <br> shown by <br> dashed line |  | 0.7 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |

### 2.6.1 Braced frames

Braced frames are the frames in which the side way is prevented as shown in Fig. 2.14, and, therefore, the K-factor is never greater than 1.0. The structural model employed for determination of K-factor for a framed column braced against sideway is shown in Fig. 2.14.


Figure 2.14 Model for Braced Frame [18].

The assumptions used in this model are:

1. All members have a constant cross section and behave elastically.
2. The axial forces in the beams are negligible.
3. All columns buckle simultaneously.
4. All joints are rigid.
5. At buckling, the rotations at the near and far ends of the beams are equal in magnitude and opposite in direction.
6. The restraining moment which is provided by the beam at joint is distributed on columns based on their stiffnesses.

To determine the K-factor using monograph form in Fig. 2.14 (Julian and Lawrence, 1959), $G_{A}$ and $G_{B}$ at $A$ and $B$ ends of the column under consideration are determined according to Eq. (2.16).

$$
\begin{equation*}
\mathrm{G}=\frac{\sum\left(\frac{\mathrm{EI}}{\mathrm{~L}}\right) \text { Columns }}{\sum\left(\frac{\mathrm{EI}}{\mathrm{~L}}\right) \text { Beams }} \tag{2.16}
\end{equation*}
$$

Where,
$\mathrm{I}=$ section's moment of inertia.
$\mathrm{E}=$ material's modulus of elaticity.
$\mathrm{L}=$ member's length.
$\mathrm{G}=$ end restraint factor.
Subscripts A and $\mathrm{B}=$ the joints at the two ends of the column section being considered.

Figs. 2.15 and 2.17 are widely used alignment charts in engineering practice to obtain the effective length factor K for the column under consideration. The minimum value of $K$ (i.e. $K=0.5$ ) coincides with the
case when girders are stiff so that the column's end conditions close to fixed supports. On the other hand, the maximum value of K (i.e. $\mathrm{K}=1$ ) coincides with the case when girders are flexible so that the column's end conditions close to pin supports.


Figure 2.15 Alignment Chart for Braced Frame [18].

### 2.6.2 Unbraced frames

If a rigid frame depends solely on frame action to resist lateral forces, its side way is permitted as shown in Fig. 2.16. In this case, the K-factor is never smaller than 1.0 and its maximum value is infinity. The model for
the determination of K-factor for a framed column subjected to sideway is shown in Fig. 2.16. The assumptions used for this model are the same as those used for the model of the braced frame, except that, for this model, the fifth assumption is modified as follows: the rotations at the near and far ends of the girders are equal in magnitude and direction.


Figure 2.16 Model for Unbraced Frame [18].

To determine the K-factor using monograph form in Fig. 2.16 (Julian and Lawrence, 1959), $G_{A}$ and $G_{B}$ at A and B ends of the column under consideration are determined according to Eq. (2.16). The value of the
effective length factor $K$ is obtained as the intersection of the straight line with the K scale.

Note that, G equals zero when the column is connected rigidly to the footing. However, for practical design, a value of 1 is taken. Likewise, if the column is connected to the footing by a pinned support, $G$ is infinity, but for practical design, a value of 10 is taken. The use of $G$ equals 1 for a column which is rigidly attached to a properly designed footing and $G$ equals 10 for a hinged support reflects the fact that an ideally fixed or hinged support condition will never be run across in a real frame.


Figure 2.17 Alignment Chart for Unbraced Frame [18].

### 2.6.3 Modifications to alignment charts

In using previous alignment charts, the analyst must be aware of the assumptions that were used in developing these alignment charts. For example if the material is in inelastic range, Yura (1971) has developed a procedure to take into account the material nonlinearity effect on $K$-factor. Several modifications are briefed as follows:

Girders are not rigidly connected to columns: If the forth assumption in section 2.6.1 is violated which means that the end-conditions of restraining girder is not rigidly connected to column's end, the stiffness of the girder must be modified by the factor $\alpha_{k}$ in the determination of $G_{A}$ and $G_{B}$ which is developed by Duan et al (1996) as shown in Tables 2.3 and 2.4.

Table 2.3 Modification Factors $\boldsymbol{\alpha K}$ for Braced Frames [18].

| End conditions of restraining girder |  |  |
| :---: | :---: | :---: |
| Near end | Far end | $\alpha_{k}$ |
| Rigid | Rigid | 1.0 |
| Rigid | Hinged | 1.5 |
| Rigid | Semi-rigid | $\left(1+\frac{6 E_{g} I_{g}}{L g R_{k F}}\right) /\left(1+\frac{4 E_{g} I_{g}}{L g R_{k F}}\right)$ |
| Rigid | Fixed | 2.0 |
| Semi-rigid | Rigid | $1 /\left(1+\frac{4 E_{g} I_{g}}{L g R_{k N}}\right)$ |
| Semi-rigid | Hinged | $1.5 /\left(1+\frac{3 E_{g} I_{g}}{L g R_{k N}}\right)$ |
| Semi-rigid | Semi-rigid | $\left(1+\frac{6 E_{g} I_{g}}{L g R_{k F}}\right) / R^{*}$ |
| Semi-rigid | Fixed | $2 /\left(1+\frac{4 E_{g} I_{g}}{L_{g} R_{k N}}\right)$ |

Note: $R^{*}=\left(1+\frac{4 E g I_{g}}{L g R_{k N}}\right)\left(1+\frac{4 E g I_{g}}{L g R_{k F}}\right)-\left(\frac{E g I_{g}}{L g}\right)^{2} \frac{4}{R_{k N} R_{k F}}$

Table 2.4 Modification Factors $\alpha$ K for Unbraced Frames [18].

| End conditions of restraining girder |  |  |
| :---: | :---: | :---: |
| Near end | Far end | $\alpha_{k}$ |
| Rigid | Rigid | 1 |
| Rigid | Hinged | 0.5 |
| Rigid | Semi-rigid | $\left(1+\frac{2 E_{g} I_{g}}{L_{g} R_{k F}}\right) /\left(1+\frac{4 E_{g} I_{g}}{L g R_{k F}}\right)$ |
| Rigid | Fixed | $2 / 3$ |
| Semi-rigid | Rigid | $1 /\left(1+\frac{4 E_{g} I_{g}}{L_{g} R_{k N}}\right)$ |
| Semi-rigid | Hinged | $0.5 /\left(1+\frac{3 E_{g} I_{g}}{L_{g} R_{k N}}\right)$ |
| Semi-rigid | Semi-rigid | $\left(1+\frac{2 E_{g} I_{g}}{L g R_{k F}}\right) / R^{*}$ |
| Semi-rigid | Fixed | $(2 / 3) /\left(1+\frac{4 E_{g} I_{g}}{L g R_{k N}}\right)$ |
|  |  |  |

Note: $R^{*}=\left(1+\frac{4 E_{g} I_{g}}{L_{g} R_{k N}}\right)\left(1+\frac{4 E_{g} I_{g}}{L_{g} R_{k F}}\right)-\left(\frac{E_{g} I_{g}}{L_{g}}\right)^{2} \frac{4}{R_{k N} R_{k F}}$

The beams sections are not prismatic: When the column is restrained by a girder with tapered rectangular section as shown in Fig. 2.18, a modified factor has been developed by King et al (1993). The factor is multiplied by the girder's stiffness in determination of $G_{A}$ and $G_{B}$, which is a function of far end conditions and girder's tapering characteristics, $r$ and $a$.


Figure 2.18 Tapered Rectangular Girders [18].

Unsymmetrical frames: When the columns differ in moment of inertia, height, or loads. Updating for the effective length factor K is needed. Chu and Chow (1969) have proposed a modification factor $\beta$ as shown in Fig. 2.19. The formula has the form:

$$
\begin{equation*}
\mathrm{K}_{\text {adjusted }}=\beta \mathrm{K}_{\text {alignment chart }} \tag{2.17}
\end{equation*}
$$



Figure 2.19 Chart for Modification Factor $\beta$ in an Asymmetrical Frame [18].
Note that: every line corresponds to certain $\lambda$, where the upper line corresponds to $\lambda=1$ and the lower line corresponds to $\lambda=0$, and every line among upper and lower limits is at 0.1 increment.

### 2.7 Shear mode deformation in frames

Frames deflect in a shear mode which is similar to a beam with fixed support affected by support settlement as shown in Fig. 2.20. On the other hand, shear walls deflect in a cantilever mode which is similar to a cantilever beam with fixed support as shown in Fig. 2.21. If the frames were used solitary in the building to resist lateral loads, it will create bending moments in beams and columns to resist the applied shear force at each floor while the effect of overturning moment will be secondary and, in most, it will be negligible. The stories will stay level although the joints itself will rotate. Furthermore, if the shear walls were used solitary in the structure to resist lateral loads, it will generate bending moment at each level equal to the overturning moment at the same level. If both systems (i.e. frames and shear walls) were used to resist the lateral loads, each system will try to block the other from making its natural deflection and the final deflection shape will be between the two systems as shown in Fig. 2.22.

In the entire course of this study, reinforced concrete portal frames are used, which implies that, frames resist the seismic loads. Consequently, the typical deflection shape is shear mode deflection. As can be seen from shear mode deflection, the relative displacement will be maximum value at the ground columns, at the same time, the maximum axial force will be at the ground columns. Thus, the maximum P-Delta effect by sense in this special case is expected to be at the bottom floors. For this reason, the bending moment from both first order and second order analyses will be computed at all columns in all floors and the critical column will be adopted.


Figure 2.20 Shear Mode Deformation [9].

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Figure 2.21 Cantilever Mode Deformation [9].


Figure 2.22 Frame-Shear Wall Interaction [9].

## Chapter Three

## Verification on SAP2000 Results

### 3.1 Introduction

It is vital to verify any software before it is used in calculations to make sure that the software works properly. During the entire course of this thesis SAP2000, version 19.0 is used. Consequently, it is verified in this chapter for both static and dynamic analyses. With regarded to static analysis a 2-D reinforced concrete portal frame model with both vertical and horizontal applied loads is used. This model is built on SAP2000, version 19.0 software with the given dimensions. The structure's geometry, material, section properties and load patterns are defined. Gravity and horizontal loads are applied. The bending moment on a chosen column from P-Delta analysis is determined and compared with bending moment determined from approximate equations. With regarded to dynamic analysis a 2-D portal frame's structural period is determined by both hand calculations and SAP2000, version 19.0.0 software and the results are compared. To make sure that SAP2000 software works properly, the resulting percentages between hand calculations and SAP2000 results must be equal or lower than five percent.

### 3.2 Verification on static results

In this section, the results of non-linear static analysis (i.e. geometric nonlinearity, material nonlinearity is out of scope of this research) using

SAP2000 software are compared to hand calculations using moment amplification methods. A 2-D frame model is used in this case which is illustrated in Fig. 3.1. It consists of two spans in x-direction with span length equals six meters. Moreover, the frame consists of three stories with each story height equals four meters. All columns sections properties are $0.15 \mathrm{~m} \times 0.25 \mathrm{~m}$ in x -direction and y -direction respectively. The dimensions of beams are $0.20 \mathrm{~m} \times 0.20 \mathrm{~m}$ in depth and width. The loads divide into three types: distributed dead loads per unit length, distributed live loads per unit length and lateral loads. The distributed dead loads equal $10 \mathrm{kN} / \mathrm{m}$ and $5 \mathrm{kN} / \mathrm{m}$ on the first span and on the second span respectively. The distributed live loads equal $20 \mathrm{kN} / \mathrm{m}$ and $10 \mathrm{kN} / \mathrm{m}$ on the first span and on the second span respectively. Lateral loads increase from lower stories to the top stories with increment of 200 kN . The column at which bending moment's magnitude is calculated from both SAP2000 and hand calculations is denoted by AB . Fixed supports (i.e. rigid support) are used which means it can resist vertical and horizontal forces as well as moment.

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Figure 3.1 Frame Subjected to Horizontal and Vertical Loads

### 3.2.1 Hand calculations using moment amplification methods

### 3.2.1.1 P- $\Delta$ effect (structure P-Delta effect)

The frame is analyzed using SAP2000 software and the first-order analysis moments on column AB are shown in Fig. 3.2. Various moment amplification methods that can be used to account for P- $\Delta$ effect (e.g. story magnifier method, multiple-column magnifier method, etc.). Multiplecolumn magnifier method is chosen because it is a straightforward
procedure and it is not function of first-order deflection of the frame as defined in Eq. 2.11 as follows:


Figure 3.2 First-Order Moment Results for Column AB
$\mathrm{M}=\left(\frac{1}{1-\sum \mathrm{P} / \sum \mathrm{P}_{\mathrm{ek}}}\right) \mathrm{M}_{1 \text { sway }}=\mathrm{B}_{2} \mathrm{M}_{1 \text { sway }}$
Where
$\mathrm{M}=$ maximum end moment accounting for $\mathrm{P}-\Delta$ effect.
$\mathrm{M}_{1 \text { sway }}=$ maximum primary moment due to swaying of the story.
$\mathrm{B}_{2}=\mathrm{P}-\Delta$ moment amplification factor.
$\sum \mathrm{P}=$ summation of axial forces.
$\mathrm{P}_{\mathrm{ek}}$ is elastic buckling load considering column end restraining conditions.
$\sum \mathrm{P}=((20+10) \times 6+(10+5) \times 6) \times 3=810 \mathrm{kN}$
$\sum \mathrm{P}_{\mathrm{ek}}=\mathrm{P}_{\mathrm{ek} \text { middle }}+2 \times \mathrm{P}_{\text {ek edge }}$

## For middle:

$\mathrm{G}=\frac{\sum\left(\frac{\mathrm{EI}}{\mathrm{L}}\right) \text { Columns }}{\sum\left(\frac{\mathrm{EI}}{\mathrm{L}}\right) \text { Beams }}$
Where, $I$ is the moment of inertia, $E$ is the moduls of elasticity, and $L$ is the member's length.
$\mathrm{I}_{\text {Beam }}=\frac{0.2^{4}}{12}=1.33 \times 10^{-4} \mathrm{~m}^{4} \quad \mathrm{I}_{\text {Column }}=\frac{0.25 \times 0.15^{3}}{12}=7.03 \times 10^{-5} \mathrm{~m}^{4}$
$\mathrm{G}_{\mathrm{A}}=\frac{7.03 \times 10^{-5} \times 2 / 4}{1.33 \times 10^{-4} \times 2 / 6}=0.79$
$G_{B}=0$
$P_{\mathrm{ek}}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}$
Where, $I$ is the moment of inertia, E is the moduls of elasticity, L is the member's length, and K is the effective length factor.
$\mathrm{E}=24.8 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$
The effective length factor $K=1.14$ (From Fig. 2.16)
$\mathrm{P}_{\mathrm{ek} \text { middle }}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}=\frac{\pi^{2} \times 24.8 \times 10^{6} \times 7.03 \times 10^{-5}}{(1.14 \times 4)^{2}}=828 \mathrm{kN}$

## For edge:

$\mathrm{G}_{\mathrm{A}}=\frac{\sum\left(\frac{\mathrm{EI}}{\mathrm{L}}\right) \text { Columns }}{\sum\left(\frac{\mathrm{EI}}{\mathrm{L}}\right) \text { Beams }}=\frac{7.03 \times 10^{-5} \times 2 / 4}{1.33 \times 10^{-4} / 6}=1.58$
$G_{B}=0$

The effective length factor $\mathrm{K}=1.24$ (From Fig. 2.16).

$$
\begin{aligned}
& \mathrm{P}_{\text {ek edge }}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}=\frac{\pi^{2} \times 24.8 * 10^{6} \times 7.03 \times 10^{-5}}{(1.24 \times 4)^{2}}=700 \mathrm{kN} \\
& \sum \mathrm{P}_{\mathrm{ek}}=\mathrm{P}_{\mathrm{ek} \text { middle }}+2 \times \mathrm{P}_{\mathrm{ek} \mathrm{edge}}=827.6+2 \times 699.55=2230 \mathrm{kN} \\
& \mathrm{~B}_{2}=\left(\frac{1}{1-\sum \mathrm{P} / \sum \mathrm{P}_{\mathrm{ek}}}\right)=\frac{1}{1-810 / 2230}=1.57
\end{aligned}
$$

### 3.2.1.2 P- $\boldsymbol{\delta}$ Effect (member P-Delta effect)

The moment amplification method is used to account for $\mathrm{P}-\delta$ Effect as defined in Eq. 2.6. From Austin expression (Austin, 1961), approximate expression for $\mathrm{C}_{\mathrm{m}}$ factor is computed as defined in Eq. 2.8.

$$
C_{m}=0.6-0.4\left(\frac{M_{A}}{M_{B}}\right)
$$

Where, $M_{B}$ is the larger of the two end moments.

$$
\begin{aligned}
& \mathrm{M}_{\mathrm{A}}=\mathrm{M}_{\text {Dead }}+\mathrm{M}_{\mathrm{Live}}=4.97+2.49=7.46 \mathrm{kN} . \mathrm{m} \\
& \mathrm{M}_{\mathrm{B}}=\mathrm{M}_{\text {Dead }}+\mathrm{M}_{\mathrm{Live}}=8.49+4.25=12.80 \mathrm{kN} . \mathrm{m} \\
& C_{m}=0.6-0.4\left(\frac{7.46}{12.80}\right)=0.365 \\
& \mathrm{P} \approx 1.15((20+10) \times 3+(10+5) \times 3) \times 3=466 \mathrm{kN} \\
& \mathrm{G}_{\mathrm{A}}=0.79 \\
& \mathrm{G}_{\mathrm{B}}=0
\end{aligned}
$$

The effective length factor $\mathrm{K}=0.62$ (From Fig. 2.14).

$$
\begin{aligned}
& \mathrm{P}_{\mathrm{e}}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}=\frac{\pi^{2} \times 24.8 \times 10^{6} \times 7.03 \times 10^{-5}}{(0.62 \times 4)^{2}}=2800 \mathrm{kN} \\
& \mathrm{M}_{\max }=\left(\frac{C_{m}}{1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}}\right) \mathrm{M}_{1 \max }=\mathrm{B}_{1} \mathrm{M}_{1 \max } \\
& \mathrm{~B}_{1}=\frac{\mathrm{C}_{\mathrm{m}}}{1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}}=\frac{0.365}{1-\frac{465.75}{2800}}=0.43\left(\text { Take } \mathrm{B}_{1}=1\right) \\
& \mathrm{M}_{\mathrm{u}}=\mathrm{B}_{2} \mathrm{M}_{1 \text { sway }}+\mathrm{B}_{1} \mathrm{M}_{1 \text { non-sway }}=1.57 \times 857+1 \times(4.25+8.49)=1360 \mathrm{kN} . \mathrm{m}
\end{aligned}
$$

### 3.2.2 SAP2000 19 Results

The previous 2-D frame (Fig. 3.1) model is analyzed using SAP2000, version19.0.0 software by applying static non-linear load case with P-Delta geometric nonlinearity parameter as shown in Fig. 3.3. The P-Delta moment results for column AB are shown in Fig. 3.4 and the difference as percentage between hand calculations and SAP2000, version 19.0.0 is tabulated in Table 3.1.


Figure 3.3 P-Delta Static Non-Linear Load Case

Figure 3.4 P-Delta Moment Results for Column AB Using SAP2000

Table 3.1 Difference between Hand Calculations and SAP2000 Results

| Second-Order Moment <br> Using Moment <br> Amplification Method <br> (kN.m) | Second-Order Moment <br> Using SAP2000 (kN.m) | \% Difference |
| :---: | :---: | :---: |
| 1360 | 1285 | $5.35 \%$ |

Based on the results in Table 3.1, the percent of difference between hand calculations and SAP2000 is almost five percent in computing P-Delta effect. These results give us reliability in SAP2000 software in static analysis.

### 3.3 Verification on dynamic results

### 3.3.1 Verification of the Fundamental Period

The portal frame which is under investigation is shown in Fig. 3.5, it is one-story portal frame with a story height equals three meters and one bay with a span length equals six meters. Beam's section is $0.6 \mathrm{~m} \times 0.3 \mathrm{~m}$ in depth and width respectively. Column has 0.3 m square section. The structural period of this frame is calculated by both; hand calculations and SAP2000 software. Reinforced concrete is used to build this portal frame with the following properties; concrete elasticity modulus equals 23 GPa and Poisson's ratio equals 0.2 . Weight of 670 kN is applied on the portal
frame as a line mass (i.e. mass $=670 / 9.81$ ton). It is worth mentioning that, the source of mass in this study is the 670 kN , which means that, the structure's own weight is zeroed by entering the weight per unit volume of concrete as zero.


Figure 3.5 Portal Frame Dimensions (All dimensions are in meter)

### 3.3.1.1 Hand Calculations

Column's section moment of inertia $(\mathrm{I})=\frac{0.3^{4}}{12}=6.75 \times 10^{-4} \mathrm{~m}^{4}$
$\mathrm{EI}=23 \times 10^{6} \times 6.75 \times 10^{-4}=15500 \mathrm{kN} . \mathrm{m}^{2}$
Stiffness $(\mathrm{K})=\frac{2 \times 12 \times E I}{h^{3}}=\frac{2 \times 12 \times 15500}{3^{3}}=13800 \mathrm{kN} / \mathrm{m}$
Mass $(\mathrm{m})=\frac{\text { weight }}{9.81}=\frac{670}{9.81}=68.3 \mathrm{ton}$
The structural fundamental period can be found from the following equation:
$\mathrm{T}=2 \pi \sqrt{\frac{m}{K}}=2 \pi \sqrt{\frac{68.3}{13800}}=0.44$ second

### 3.3.1.2 Structural period using SAP2000 19

The mass is assigned on portal frame as line mass $=\frac{\mathrm{m}}{\text { length }}=\frac{68.3}{6}=$ 11.38ton/m.

The beam's moment of inertia modifier about 3 -axis is modified to 10 , in order to make the beam as a rigid element.

Structural period using SAP2000 19 software $=0.45$ second (see Fig. 3.6).

## Weformed Shape (MODAL) - Mode $1 ; T=0.452 ; \quad f=2.21237$



Figure 3.6 Structural Period Using SAP2000

The results are tabulated in Table 3.2 and the percent difference between hand calculations and SAP2000 results is calculated. It can be concluded that SAP2000 gives reasonable results with regarded to dynamic analysis because the percent difference is very small (i.e. less than five percent).

## Table 3.2 Difference between Hand Calculations and SAP2000 Results

| Structural Period Using <br> Hand Calculations <br> (seconds) | Structural Period Using <br> SAP2000 (seconds) | \% Difference |
| :---: | :---: | :---: |
| 0.44 | 0.45 | $2 \%$ |

### 3.3.2 Verification of Mode Shapes and the Fundamental Period for MDOF Structure

The model under investigation is the same as model in section 3.3.1, nevertheless, it consists of four storeys with the same mass on each storey which equals 11.38 ton $/ \mathrm{m}$ except the fourth floor has half mass of the previous value as shown in Fig. 3.7. The beam's moment of inertia modifier about 3 -axis is modified to 10 , in order to make the beam as a rigid element. To calculate the modal shapes and the natural frequencies; an eigenvalue problem must be solved (Eq. 3.1).

$$
\begin{equation*}
A x=\lambda x \tag{3.1}
\end{equation*}
$$

Where,
$\mathrm{x}_{\mathrm{i}}=$ eigenvectors
$\lambda_{i}=$ eigenvalues
$\mathrm{A}=\mathrm{M}^{-1} \mathrm{~K}$
$\mathrm{M}=$ mass matrix
$\mathrm{K}=$ stiffness matrix
The stiffness matrix can be constructed as follows, Stiffness for Each Storey (Ks) $=\frac{2 \times 12 \times E I}{h^{3}}=\frac{2 \times 12 \times 15500}{3^{3}}=13800 \mathrm{kN} / \mathrm{m}$

Stifness Matrix $(K)=\left[\begin{array}{cccc}2 \mathrm{Ks} & -\mathrm{Ks} & 0 & 0 \\ -\mathrm{Ks} & 2 \mathrm{Ks} & -\mathrm{Ks} & 0 \\ 0 & -\mathrm{Ks} & 2 \mathrm{Ks} & -\mathrm{Ks} \\ 0 & 0 & -\mathrm{Ks} & \mathrm{Ks}\end{array}\right]$
Stifness Matrix (K) $\left[\begin{array}{cccc}2 \cdot 13800 & -13800 & 0 & 0 \\ -13800 & 2 \cdot 13800 & -13800 & 0 \\ 0 & -13800 & 2 \cdot 13800 & -13800 \\ 0 & 0 & -13800 & 13800\end{array}\right]$
$\operatorname{Mass}$ Matrix $(\mathrm{m})=\left[\begin{array}{cccc}68.3 & 0 & 0 & 0 \\ 0 & 68.3 & 0 & 0 \\ 0 & 0 & 68.3 & 0 \\ 0 & 0 & 0 & 34.15\end{array}\right]$


Figure 3.7 Model for Verification of Mode Shapes

Maple software can be used to get the eigenvalues and eigenvectors for this eigenvalue problem as follows,

Eigenvalues $=\left[\begin{array}{c}30.76 \\ 249.47 \\ 558.76 \\ 777.47\end{array}\right]$

Eigenvectors $=\left[\begin{array}{cccc}0.38 & 0.92 & 0.92 & 0.38 \\ 0.71 & 0.71 & -0.71 & -0.71 \\ 0.92 & -0.38 & -0.38 & 0.92 \\ 1.00 & -1.00 & 1.00 & -1.00\end{array}\right]$

Eigenvalues represent the squared natural frequencies $\left(w^{2}\right)$ for every mode. Applying the following equation to get the natural period,

$$
\begin{equation*}
\text { Natural Period }(T)=\frac{2 \pi}{w} \tag{3.2}
\end{equation*}
$$

Where, w is the natural frequency.
Eigenvectors represent the mode shapes for every mode. Tables 3.3 and 3.4 summarize the natural period and mode shapes for every mode respectively from both hand calculations and SAP software.

Table 3.3 Natural Periods

| Mode Number | T(Eq. 3.2) | T (SAP2000) | Percent Error |
| :---: | :---: | :---: | :---: |
| 1 | 1.13 | 1.17 | $3.4 \%$ |
| 2 | 0.40 | 0.41 | $2.4 \%$ |
| 3 | 0.27 | 0.27 | $0.0 \%$ |
| 4 | 0.23 | 0.23 | $0.0 \%$ |

Table 3.4 Mode Shapes from Solving Eigenvalue Problem versus SAP

## Software

| Mode1 |  | Mode2 |  | Mode3 |  | Mode4 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Matrix | SAP2000 | Matrix | SAP2000 | Matrix | SAP2000 | Matrix | SAP2000 |
| 0.38 | 0.37 | 0.92 | 0.90 | 0.92 | 0.93 | 0.38 | 0.41 |
| 0.71 | 0.69 | 0.71 | 0.71 | -0.71 | -0.67 | -0.71 | -0.76 |
| 0.92 | 0.91 | -0.38 | -0.35 | -0.38 | -0.41 | 0.92 | 0.96 |
| 1.00 | 1.00 | -1.00 | -1.00 | 1.00 | 1.00 | -1.00 | -1.00 |

From previous tables, it is obvious the matching of results.

## Chapter Four

## Multi-Storey Effect on P-Delta Analysis

### 4.1 Introduction

Three fundamental parameters affect P-Delta results, which are material of construction, loads and structure's geometry. In the entire course of this study the material is retained to be reinforced concrete. Number of stories is updated to investigate its effect on second-order analysis results. Section 4.2 deals with changing the number of storeys by taking six different models (i.e. 5 -storey, 10 -storey, 15 -storey, 20 -storey, 25 -storey and 30storey) with fundamental periods abide by the proposed equation (i.e. section 12.8.2, ASCE 7) and studying its effect on P-Delta results. Section 4.3 deals with changing the number of storeys by taking three different models (i.e. 5-storey, 10-storey, 15-storey) with fundamental periods greater than the proposed equation (i.e. section 12.8.2, ASCE 7) and studying its effect on P-Delta results. SAP2000, version 19.0.0 is used in this study with hand calculations when it is needed. All models are located in Nablus city. The seismic zone factor 0.2 is considered which falls under zone 2 B . The soil which the structures are built on is considered as a rock. Wind loads or seismic loads act in the lateral direction. In this study, seismic loads are taken as the lateral applied loads. For purpose of secondorder analysis a reasonable estimate of flexural rigidity may be made if $\mathrm{I}=$ $0.4 \mathrm{I}_{\mathrm{g}}$ for beams and $0.8 \mathrm{I}_{\mathrm{g}}$ for columns [40]. Thus, previous values are
adopted for beams and columns in addition to coefficient equals 0.3 for slabs.

### 4.2 Multi-Storey Effect on P-Delta Analysis with Fundamental Period Abides by the Proposed Equation

On account of growing number of population with the shortage in suitable lands for construction especially in metropolises; people resort to build high-rise buildings to accommodate these formidable numbers of people who need dwellings. On the other hand, some countries compete with each other to have the tallest buildings in the world. Wherefore, the tendency of people to build high buildings is not neoteric tendency, it has been since the existence of human on this planet. The pyramids that have been built by pharaohs which are the only wonder of the seven wonders of the ancient world that has survived to the present day have been the tallest man-made structures on earth for millenniums of years until the competition of Lincoln Cathedral. Building high has difficulties and hardships which stimulate us into creative ways to be overcome. For instance, when people have built high-rise buildings, the need arose to the invention of elevators to carry people all the way up or down. Furthermore, constructing skyscrapers leads to thinking in ways to evacuate all these people in emergency cases from all the floors at the same time. Previous two examples of the difficulties and hardships that were created by high-rise buildings are also applied to structural engineering especially in P-Delta effect. P-Delta effect exists in low-rise as well as high-rise structures.

However, its amount in low-rise buildings can be neglected in contrast to high-rise buildings in which P-Delta effect must be included in design. The hardships accompanied with high-rise buildings must be overcome by designing for it. But the million dollar question is after how many regular storeys the P-Delta effect must be included in design. This question will be answered at the end of this chapter for reinforced concrete moment resisting frame structures.

P-Delta effect becomes a tremendous amount while increasing the building's height. In this section, the increase in difference between firstorder analysis and second-order analysis is investigated due to increasing the building's height. Six models are used; model number one (M5) is a five-storey reinforced concrete portal frame and similarly M10, M15, M20, M25 and M30. The six models are assumed as standard occupancy structures. In the following sections, every model has been studied separately; and in the comments and conclusion section; the results from each model have been compared with each other. All models have the same plan, material properties, section properties, live loads and superimposed dead loads except number of stories and columns' sections.

### 4.2.1 Five-Storey model (M5)

M5 is a five-storey reinforced concrete portal frame with fixed supports. It consists from three spans in $x$-direction with span length equals 6 m and three spans in y-direction with span length equals 6 m as shown in Figs. 4.1 and 4.2. The beams' dimensions are $0.70 \mathrm{~m} \times 0.75 \mathrm{~m}$ in depth and width
respectively. The columns are $0.75 \mathrm{~m} \times 0.75 \mathrm{~m}$ square section. The slab is solid with thickness equals 0.15 m . All of the floors have a constant height which equals 3.40 m with a total height equals 17 m . Sections properties for frame's sections and area's section are summarized in Table 4.1. First-order moment and second-order moment are determined on all columns and the column which gives the maximum percent between first and second-order has been adopted.


Figure 4.1 M5, M10, M15, M20, M25 and M30 Plan (All dimensions are in meter)


Figure 4.2 M5 Model in 3D view

## Table 4.1 Sections Properties

| Beam |  | Column |  | Slab |
| :---: | :---: | :---: | :---: | :---: |
| Depth(m) | Width(m) | Depth(m) | Width(m) | Thickness(m) |
| 0.70 | 0.75 | 0.75 | 0.75 | 0.15 |

### 4.2.1.1 Material Properties

In the past, the construction materials consisted from thatch, adobe and wood. In the present era, there is two main materials can be used in
construction, which are reinforced concrete and/or steel. The process which the material has been chosen for this research was based on the local practice. The construction material that is used in most projects in Palestine and in the contiguous countries is reinforced concrete material. Reinforced concrete predominates on other materials in construction in these countries because it does not need a high skilled workmanship like other materials (for example steel), also it does not need continuous maintenance (like steel). "Cast in place reinforced concrete offers outstanding resistance to explosion or impact. Moreover, it can endure very high temperatures from fire for long time without loss in structural integrity" (Alfred G. Gerosa, president of Concrete Alliance Inc.). In respect to wind loads, concrete has very high mass compared to steel, so it can resist wind loads more effectively. At the same time, steel is more ductile than reinforced concrete, which makes it a reasonable choice in seismic zones. "Steel framing does very well under high loads because it is ductile, which means it has the ability to bend without breaking and can absorb that kind of energy" (Larry Williams, president of the Washington, D.C.-based Steel Framing Alliance of cold-formed steel). In this study, the construction material was chosen to be reinforced concrete; its properties are tabulated in the following table, Table 4.2. For all structural elements, the yielding stress of steel $\left(\mathrm{F}_{\mathrm{y}}\right)$ equals 420 MPa .

Table 4.2 Concrete Properties

| Compressive <br> Strength $^{1}$ | Compressive <br> Strength $^{2}$ | Elasticity <br> Modulus $^{1}$ | Elasticity <br> Modulus |  |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| ${ }^{2}$ | Poisson's | Weight <br> Ratio | Per <br> Unit <br> Volume |  |  |  |
| Value | 24 MPa | 32 MPa | 23 GPa | 26.6 GPa | 0.2 | $25 \mathrm{kN} / \mathrm{m}^{3}$ |

1 Concrete used for slabs and beams.
2 Concrete used for columns.

### 4.2.1.2 Vertical loads

The vertical loads are divided into two main categories: live loads and dead loads. Live loads refer to the loads that can change over time such as people, furniture, vehicles, and the like. It is measured as kilo newton per square meter ( $\mathrm{kN} / \mathrm{m}^{2}$ ). Building codes prescribe minimum live loads requirements for various buildings and structures based on their type of usage. For example, the live load for a classroom differs from live load for a single family residence. Dead loads refer to the loads that do not change over time and it is divided into two types: structure's own weight, which refers to the weight of structural elements, and the superimposed dead load, which refers to the non-structural elements' weights such as, partitions, plumbing and HVAC (HVAC stands for Heating, Ventilation, and Air Conditioning). Table 4.3 summarizes live loads and dead loads that have been used in this study.

## Table 4.3 Vertical Load Values

|  | Live <br> Load | Dead Load | Superimposed <br> Dead Load (SD) |
| :---: | :---: | :---: | :---: |
| Value | $3 \mathrm{kN} / \mathrm{m}^{2}$ | Structure's own weight | $4.5 \mathrm{kN} / \mathrm{m}^{2}$ |

### 4.2.1.3 Seismic loads

Palestine is located in the Middle East region to the west of the Dead Sea basin. Series of major earthquakes occurred in the eastern Mediterranean region and in the Middle East. Seismicity in Palestine comes from the movement along the Dead Sea Transform-DST. The DST also known as the Dead Sea Fault System, is a north-south striking left-lateral shear zone extending from the incipient oceanic ridge (Red Sea) in the south to the Taurus plate collision in the north (Turkey). About 105-110 km of leftlateral displacement between the African and Arabian tectonic plates took place along this fault system during the last 15 million years. The average rate of motion during the last 5 million years is 5 millimeters per year (United States Geological Survey) as shown in Fig. 4.3. Fig. 4.4 shows the pressure ridges in the Araba Valley region in the south of Dead Sea basin.


Figure 4.3 Tectonic Setting of the Dead Sea Transform (United States Geological Survey)

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Figure 4.4 Pressure Ridges in the Araba Valley (Review of Geophysics, 2009)
Seismic forces threaten any structure on earth, not only in the highly seismic regions but everywhere on the planet can be affected by an earthquake. So every structure is designed to withstand these forces by following local or international building codes and regulations. Earthquake comes with uncertainty in magnitude, location and time. Many building codes tried to compute the maximum base shear that can apply on the structure based on numerous parameters such as seismic coefficients, zone factor, soil type, structure's weight and natural period of the structure. After computing the base shear on the structure, its value can be distributed on the floors based on their weights and cumulative heights. In this section and the following sections, the base shear is computed using modal response spectrum analysis compatible with International Building Code 2015 (IBC 2015) and ASCE 7 code. As stated previously, the six models are located in Nablus city on a rock soil. The direction at which the
seismic forces are applied is x -direction. However, the plan is identical in both directions and the columns' sections are square sections, consequently, both directions will give the same results. Every structure, and portion thereof, including nonstructural components that are permanently attached to structures and their supports and attachments, shall be designed and constructed to resist the effects of earthquake motions in accordance with ASCE 7 (IBC 2015, section 1613.1). The exempted structures from previous phrase are very limited number of one and two-family dwellings which are located in very low seismic zones or for some structures that do not pose a risk on humans if they have been demolished during an earthquake (for example, agricultural storage structures intended only for incidental human occupancy).

The determination of base shear and its distribution on floors using IBC 2015 with accordance with ASCE 7 equations follows the steps bellow:

## STEP 1 (Seismic Zone Factor)

The seismic zone factor $(\mathrm{Z})$ is defined as the expected peak acceleration on the surface of a certain rock's type as a percent of gravity acceleration. Palestine is separated into four districts based on seismic zone factor as shown in Fig. 4.5 with $10 \%$ probability of exceedance in 50 years. Nablus is located in zone 2 B with seismic zone factor $(\mathrm{Z})$ equals 0.20 . As can be seen from Fig. 4.5 the higher seismic zone factor is located in areas which are close to the Dead Sea Transform (for example, Jericho city), and its value diminishes with keeping away from Dead Sea Transform. It is worth
mentioning that most West Bank districts are located in seismic zones 2 A and 2B with seismic zone factor ranges from 0.15 to 0.2 .

STEP 2 (Site Classification)
IBC 2015 states that based on the site soil properties, the site shall be classified in accordance with Chapter 20 of ASCE 7 as shown in Table 4.4. If the soil properties are not known, site class D shall be used unless there is information state that soil classes E or F are present. All models are located in Nablus city on a rock soil. According to Table 4.4, the site class is $B$.

## Seismic Zone Factor,Z



Figure 4.5 Palestine Seismic Map (Boore et al., 1997)

Table 4.4 Site Classification (Table20.3-1, ASCE 7)

| Site Class | $\bar{v}_{s}$ | $\bar{N}$ or $\bar{N}_{c h}$ | $\bar{s}_{u}$ |
| :---: | :---: | :---: | :---: |
| A. Hard rock | $>5,000 \mathrm{ft} / \mathrm{s}$ | NA | NA |
| B. Rock | 2,500 to $5,000 \mathrm{ft} / \mathrm{s}$ | NA | NA |
| C. Very dense soil and soft rock | 1,200 to $2,500 \mathrm{ft} / \mathrm{s}$ | $>50$ | >2,000 psf |
| D. Stiff soil | 600 to $1,200 \mathrm{ft} / \mathrm{s}$ | 15 to 50 | 1,000 to $2,000 \mathrm{psf}$ |
| E. Soft clay soil | <600 ft/s | $<15$ | <1,000 psf |
|  | Any profile with more than 10 ft of soil having the following characteristics: <br> -Plasticity index $P I>20$, <br> -Moisture content $w \geq 40 \%$, <br> -Undrained shear strength $\bar{s}_{u}<500 \mathrm{psf}$ |  |  |
| F. Soils requiring site response analysis in accordance with Section 21.1 | See Section 20.3.1 |  |  |

## STEP 3 (Design Basis Earthquake, DBE)

The maximum considered earthquake (MCE) is divided into two types: $\mathrm{S}_{\mathrm{s}}$, which refers to MCE for short period ( 0.2 second), $S_{1}$, which refers to MCE for long period ( 1 second) for a $2 \%$ probability of exceedance in 50years. Design Basis Earthquake (DBE) defines the peak horizontal acceleration with $10 \%$ probability of exceedance in $50 y e a r s$. For a $10 \%$ probability of exceedance; $\mathrm{S}_{\mathrm{s}^{*}}$ can be taken as 2.5 Z , and $\mathrm{S}_{1^{*}}$ can be taken as 1.25 Z (The National Steering Committee for Earthquake Preparedness, 2009).

$$
\begin{gather*}
\mathrm{S}_{\mathrm{s}^{*}}=2.5 \mathrm{Z}  \tag{4.1}\\
\mathrm{~S}_{1^{*}}=1.25 \mathrm{Z} \tag{4.2}
\end{gather*}
$$

Where, $S_{s *}$ refers to an equivalent 0.67 MCE for 0.2 -second period, $\mathrm{S}_{1^{*}}$ refers to an equivalent 0.67 MCE for 1 -second period and Z is the seismic zone factor.

By applying previous equations, we get:
$\mathrm{S}_{\mathrm{s}^{*}}=2.5 \mathrm{Z}=2.5 \times 0.2=0.5$
$S_{1^{*}}=1.25 Z=1.25 \times 0.2=0.25$

Concerning the $2 / 3$ reduction value, UBC97 was based on $10 \%$ probability of exceedance in 50 years. IBC adopts the same response spectrum but changes to $2 \%$ of exceedance in 50 years. The $2 / 3$ term reduces the value to the same $10 \%$ of exceedance value. Whereas this works well in California where the $2 \%$ and $10 \%$ varies over period uniformly and where usually earthquake codes are developed, Eastern United States has no uniform variation between $2 \%$ and $10 \%$. Thus they did not approve the $2 / 3$ rule. Israeli code adopts IBC and $10 \%$ of exceedance and did not use the 2/3 reduction.

## STEP 4 (Values of Site Coefficients)

It is substantial to consider the influence of soil conditions on the spectral design accelerations which can be achieved by site coefficient $F_{a}$ at short period ( 0.2 second) and site coefficient $F_{v}$ at long period (1second) as shown in Tables 4.5 and 4.6 respectively. It can be concluded from these tables that rock and hard rock do not amplify spectral design accelerations for both short period and long period, but on the contrary, hard rock minifies spectral design accelerations. On the other hand, soft clay, stiff soil or soft rock amplify the spectral design accelerations which can reach to tremendous amount in the case of soft clay soil.

For M5 model, with referring to step 2, the site class is B (thus values for $F_{a}$ and $F_{v}$ are equal to 1 for any value of $S_{s}$ and $S_{1}$ ). Consequently, $F_{a}$ equals 1 and $F_{v}$ equals 1 (i.e. no change in spectral design accelerations).

Table 4.5 Values of Site Coefficient Fa (Table 1613.3.3(1), IBC 2015)

| STEECLASs | MAPPED SPECTRAL RESPONSE ACCELERATION AT SHORT PERIOD |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | $s_{s} \leq 0.25$ | $s_{s}=0.50$ | $s_{s}=0.75$ | $s_{s}=1.00$ | $s_{s} \geq 1.25$ |
| A | 0.8 | 0.8 | 0.8 | 0.8 | 0.8 |
| B | 1.0 | 1.0 | 1.0 | 1.0 | 1.0 |
| C | 1.2 | 1.2 | 1.1 | 1.0 | 1.0 |
| D | 1.6 | 1.4 | 1.2 | 1.1 | 1.0 |
| E | 2.5 | 1.7 | 1.2 | 0.9 | 0.9 |
| F | Note b | Note b | Note b | Note b | Note b |

a. Use straight-inine interpolation for intermediate values of mapped spectral fesponse acceleation at shoot period, $S_{\mathrm{S}}$,
b. Values slall be detemined in accordance with Section 11.4 .7 of ASCE 7 .

Table 4.6 Values of Site Coefficient Fv (Table 1613.3.3(2), IBC 2015)

| SITE CLASS | MAPPED SPECTRAL RESPONSE ACCELERATION AT 1-SECOND PERIOD |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\varsigma_{1} \leq 0.1$ | $\varsigma_{1}=0.2$ | $\varsigma_{1}=0.3$ | $\varsigma_{1}=0.4$ | $\varsigma_{1} \geq 0.5$ |
| A | 0.8 | 0.8 | 0.8 | 0.8 | 0.8 |
| B | 1.0 | 1.0 | 1.0 | 1.0 | 1.0 |
| C | 1.7 | 1.6 | 1.5 | 1.4 | 1.3 |
| D | 2.4 | 2.0 | 1.8 | 1.6 | 1.5 |
| E | 3.5 | 3.2 | 2.8 | 2.4 | 2.4 |
| F | Note b | Note b | Note b | Note b | Note b |

a. Use straight-line interpolation for intermediate values of mapped spectral response acceleation at 1 -second period, $S_{1}$.
b. Values shall be determined in accordance with Section 11.4 .7 of ASCE 7 .

## STEP 5 (Considering Soil Conditions)

The spectral design acceleration considering the influence of the soil conditions on-site in short periods ( $\mathrm{S}_{\mathrm{DS}}$ ), and the spectral design acceleration considering the influence of the soil conditions on-site in period of one second ( $\mathrm{S}_{\mathrm{D} 1}$ ) can be determined from the following equations:

$$
\begin{gather*}
\mathrm{S}_{\mathrm{DS}}=\mathrm{F}_{\mathrm{a}} \mathrm{~S}_{\mathrm{s}^{*}}  \tag{4.3}\\
\mathrm{~S}_{\mathrm{D} 1}=\mathrm{F}_{\mathrm{v}} \mathrm{~S}_{\mathrm{l}^{*}} \tag{4.4}
\end{gather*}
$$

Where: $S_{D S}, F_{a}, F_{v}, S_{s^{*}}$ and $S_{1^{*}}$ are as defined before.
By applying previous equations, we get:
$S_{D S}=1 \times 0.50=0.50$
$S_{D 1}=1 \times 0.25=0.25$

STEP 6 (Risk Category of the Structure)
IBC 2015 assigns the risk category of a certain structure based on the intended use of the structure. It is described by roman numerals from I to IV. Where, I risk category represents structures with low hazard to human lives in the event of failure (for example, agricultural facilities), and IV risk category represents essential facilities such as hospitals or fire stations. For M5 model, the risk category is II as shown in Table 4.7.

## Table 4.7 Risk Category of Buildings and Other Structures (Table

## 1604.5, IBC 2015)

| FISK CATEOORY | NATURE OF OCCUPANCY |
| :---: | :---: |
| I | Buildings and other structures that represent a low hazard to human life in the event of faihre, including but not limited to: <br> - Agricultural facilities. <br> - Certain temporary facilities. <br> - Minor storage facilities. |
| II | Buildings and other structures except those listed in Risk Categories I, III and IV. |
| III | Buildings and other structures that represent a substartial hazard to human life in the event of failure, including but not limited to: <br> - Buildings and other structures whose primary occupancy is public assembly with an occupant load greater than 300. <br> - Buildings and other structures containing Group E occupancies with an occupant load greater than 250 . <br> - Buildings and other structures containing educational occupancies for students above the 12 th grade with an occupant load greater than 500 . <br> - Group I-2 occupancies with an occupant load of 50 or more resident care recipients but not having surgery or emsergency treatment facilities. <br> - Group I-3 occupancies. <br> - Any other occupancy with an occupant load greater than 5,000.* <br> - Power-generating stations, water treatment facilities for potable water, wastewater treatment facilities and other public utility facilities not included in Risk Category IV. <br> - Buildings and other structures not included in Risk Category IV containing quantities of toxic or explosive materials that: <br> Exceed maximum allowable quantities per control area as given in Table 307.1(1) or 307.1(2) or per outdoor control area in accordance with the International Five Code; and <br> Are sufficient to pose a threat to the public if released. ${ }^{\text {b }}$ |
| IV | Buildings and other structures designated as essential facilities, including but not limited to: <br> - Group I-2 occupancies having surgery or emergency treatment facilities. <br> - Fire, rescue, ambulance and police stations and emergency vehicle garages. <br> - Designated earthquake, hurricane or other emergency shelters. <br> - Designated emergency preparedness, communications and operations centers and other facilities required for emsergency response. <br> - Power-generating stations and other public utility facilities required as emergency backup facilities for Risk Category IV structures. <br> - Buildings and other structures containing quantities of highly toxic materials that: <br> Exceed maximum allowable quantities per control area as given in Table 307.1(2) or per outdoor control area in accordance with the International Fire Code, and <br> Are sufficient to pose a threat to the public if released ${ }^{\text {b }}$ <br> - Aviation control towers, air traffic control centers and emergency aircraft hangars. <br> - Buildings and other structures having critical national defense functions. <br> - Water storage facilities and pump structures required to maintain water pressure for fire suppression. |

2. For purpones of occupant load calculation, occupancies required by Table 1004.12 to use grons floor area calculations thall be parmittod to use net floor areas
3. to detormine the total occupunt load.
4. Where approved by the building officinl, the clasification of buildings mad othar structuren as Risk Caragory III or IV based on their quantites of toxic,
 Section 1.5 .3 of ASCE 7 that a reloaso of the toxic, highly toxic or explowive materials is not menficient to pose a threst to the public.

## STEP 7 (Seismic Design Category)

The International Building Code 2015 (IBC 2015) determines the suitable seismic design category based on the spectral design accelerations after considering the influence of the soil conditions for short- period and 1second period as shown in Tables 4.8 and 4.9 respectively. For M5 model, seismic design category is $D$.

Table 4.8 Seismic Design Category Based on Short-Period (0.2 second) Response Acceleration (Table 1613.3.5(1), IBC 2015)

| VALUE OF $S_{\text {DS }}$ | RISK CATEGORY |  |  |
| :---: | :---: | :---: | :---: |
|  | Ior II | IIII | IV |
| $S_{D S}<0.167 \mathrm{~g}$ | A | A | A |
| $0.167 \mathrm{~g} \leq S_{D S}<0.33 \mathrm{~g}$ | B | B | C |
| $0.33 \mathrm{~g} \leq S_{D S}<0.50 \mathrm{~g}$ | C | C | D |
| $0.50 \mathrm{~g} \leq S_{D S}$ | D | D | D |

Table 4.9 Seismic Design Category Based on 1-Second Period Response Acceleration (Table 1613.3.5(2), IBC 2015)

| VALUE OF $\mathrm{S}_{01}$ | RISK CATEGORY |  |  |
| :---: | :---: | :---: | :---: |
|  | Ior II | III | IV |
| $S_{D 1}<0.067 \mathrm{~g}$ | A | A | A |
| $0.067 \mathrm{~g} \leq S_{D 1}<0.133 \mathrm{~g}$ | B | B | C |
| $0.133 \mathrm{~g} \leq S_{D 1}<0.20 \mathrm{~g}$ | C | C | D |
| $0.20 \mathrm{~g} \leq S_{D 1}$ | D | D | D |

## STEP 8 (Seismic Importance Factor)

Based on the suitable risk category of the structure in IBC 2015, the seismic importance factor can be determined from ASCE 7. The importance factor is multiplier that reflects the significance of the structure which increases or decreases the base shear value. The risk category and importance factor are intended to protect the public's safety. Consequently, risk category and importance factor are not intended to prevent the aesthetics or functionality aspects during and after strong earthquakes for low risk category structures. For this reason, ASCE 7 increases the importance factors for facilities which are intended to maintain its functionality after severe earthquakes such as hospitals. For M5 model,
after referring to step 6 and Table 4.10, the seismic importance factor equals 1.

Table 4.10 Importance Factors by Risk Category of Buildings and Other Structures (Table 1.5-2, ASCE 7)
$\left.\begin{array}{ccccc}\hline \hline \begin{array}{c}\text { Risk Category } \\ \text { from }\end{array} & \begin{array}{c}\text { Snow Importance } \\ \text { Factor, }\end{array} & \begin{array}{c}\text { Ice Importance } \\ \text { Factor-Thickness, } \\ \text { Table 1.5-1 }\end{array} & I_{s} & \begin{array}{c}\text { Ice Importance } \\ \text { Factor-Wind, } \\ I_{w}\end{array}\end{array} \begin{array}{c}\text { Seismic Importance } \\ \text { Factor, }\end{array}\right]$
${ }^{a}$ The component importance factor, $I_{p}$, applicable to earthquake loads, is not included in this table because it is dependent on the importance of the individual component rather than that of the building as a whole, or its occupancy. Refer to Section 13.1.3.

## STEP 9 (Permitted analytical procedure)

ASCE 7 permits three analytical procedures to determine the forces on the structure based on the seismic design category, irregularity, structural system and structure's height. The procedures are as follows; Equivalent Lateral Force Analysis, Modal Response Spectrum Analysis and Seismic Response History Procedure as shown in Table 4.11.

For M5 model; the seismic design category is D, and the structures does not include any irregularity condition and not exceeding 48.8m in structural height. As a result, both Modal Response Spectrum and Equivalent Lateral Force Analyses are permitted. Modal Response Spectrum Analysis is adopted in the entire course of this study, because it gives more reasonable shear force distribution. Likewise, it is permitted for all structures' heights.

Table 4.11 Permitted Analytical Procedures (Table 12.6-1, ASCE 7)

| Seismic <br> Design <br> Category | Structural Characteristics | Equivalent Lateral Force Analysis, Section $12.8^{\circ}$ | Modal Response Spectrum Analysis, Section $12.9^{\text {a }}$ | Seismic Response History Procedures, Chapter $16^{a}$ |
| :---: | :---: | :---: | :---: | :---: |
| B, C | All structures | P | P | P |
| D, E, F | Risk Category I or II buildings not exceeding 2 stories above the base | P | P | P |
|  | Structures of light frame construction | P | P | P |
|  | Structures with no structural irregularities and not exceeding 160 ft in structural height | P | P | P |
|  | Structures exceeding 160 ft in structural height with no structural irregularities and with $T<3.5 T$, | P | P | p |
|  | Structures not exceeding 160 ft in structural height and having only horizontal irregularities of Type 2, 3, 4, or 5 in Table 12.3-1 or vertical irregularities of Type 4, 5a, or 5b in Table 12.3-2 | P | P | P |
|  | All other structures | NP | P | P |

${ }^{\text {ap: }}$ : Permitted; NP: Not Permitted; $T_{s}=S_{D 1} / S_{D s}$

## STEP 10 (Design coefficients)

Design coefficients can be determined based on the seismic-force resisting system. As shown in Table 4.12. In M5 model case, the seismic-force resisting system is special reinforced concrete moment frames. The design coefficients can be summarized in Table 4.13.

Table 4.12 Design Coefficients and Factors for Seismic Force-Resisting Systems (Table 12.2-1, ASCE 7)

| Seismic Force-Resisting System | ASCE 7 <br> Section <br> Where <br> Detailing <br> Requirements <br> Are Specified | Response Modification Coefficient, $\mathrm{R}^{a}$ | Overstrength Factor, $\Omega_{0}{ }^{g}$ | Deflection Amplification Factor, $\mathrm{C}_{\mathrm{d}}{ }^{b}$ | Structural System Limitations Including Structural Height, $h_{n}$ (ft) Limits ${ }^{c}$ |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  | Seismic Design Category |  |  |  |  |
|  |  |  |  |  | B | C | $\mathrm{D}^{d}$ | $\mathrm{E}^{d}$ | $\mathrm{F}^{e}$ |
| C. MOMENT-RESISTING FRAME SYSTEMS |  |  |  |  |  |  |  |  |  |
| 1. Steel special moment frames | $\begin{aligned} & 14.1 \text { and } \\ & 12.2 .5 .5 \end{aligned}$ | 8 | 3 | 51/2 | NL | NL | NL | NL | NL |
| 2. Steel special truss moment frames | 14.1 | 7 | 3 | 51/2 | NL | NL | 160 | 100 | NP |
| 3. Steel intermediate moment frames | $\begin{aligned} & 12.2 .5 .7 \text { and } \\ & 14.1 \end{aligned}$ | $41 / 2$ | 3 | 4 | NL | NL | $35^{h}$ | $\mathrm{NP}^{\text {h }}$ | $\mathrm{NP}^{h}$ |
| 4. Steel ordinary moment frames | 12.2.5.6 and $14.1$ | $31 / 2$ | 3 | 3 | NL | NL | $\mathrm{NP}^{i}$ | NP ${ }^{\text {i }}$ | $\mathrm{NP}^{i}$ |
| 5. Special reinforced concrete moment frames ${ }^{n}$ | $\begin{aligned} & 12.2 .5 .5 \text { and } \\ & 14.2 \end{aligned}$ | 8 | 3 | 51/2 | NL | NL | NL | NL | NL |
| 6. Intermediate reinforced concrete moment frames | 14.2 | 5 | 3 | $41 / 2$ | NL | NL | NP | NP | NP |

Table 4.13 M5 Design Coefficients and Factors

| Response Modification <br> Coefficient, $\mathbf{R}$ | Over Strength <br> Factor, $\boldsymbol{\Omega}_{\mathbf{0}}$ | Deflection Amplification <br> Factor, $\mathbf{C}_{\mathbf{d}}$ |
| :---: | :---: | :---: |
| 8 | 3 | 5.5 |

## STEP 11 (Effective Seismic Weight)

In order to determine the base shear that applies on the structure, the total seismic dead load must be computed which includes the structure's own weight in addition to super imposed dead load. Beams' weights, columns' weights, slab's weight and super imposed dead loads in one floor are calculated and tabulated in Tables 4.14-4.17.

## Table 4.14 Beams Weight in One Storey

|  | Number <br> of Beams | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Beams <br> Weight |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| Calculations | 24 | $0.55 \mathrm{~m}^{*}$ | 0.75 m | 6 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1485 kN |

* $0.7-0.15=0.55 \mathrm{~m}$


## Table 4.15 Columns Weight in One Storey

|  | Number <br> of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| Calculations | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

Table 4.16 Slab Weight in One Storey

|  | Slab Area | Slab <br> Thickness | Concrete Weight <br> Per Unit Volume | Slab <br> Weight |
| :--- | :---: | :---: | :---: | :---: |
| Calculations | $324 \mathrm{~m}^{2}$ | 0.15 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1215 kN |

## Table 4.17 Super Imposed Dead Load on One Storey

|  | Slab Area | SD Load Per <br> Unit Area | SD Load |
| :---: | :---: | :---: | :---: |
| Calculations | $324 \mathrm{~m}^{2}$ | $4.5 \mathrm{kN} / \mathrm{m}^{2}$ | 1458 kN |

The weight of each storey can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last storey is exempted from half columns' weights as tabulated in Table 4.18. The verification on an example of SAP2000 weight calculations is tabulated in Table 4.19 and Fig. 4.6.

## Table 4.18 M5 Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 4923 |
| 2 | 4923 |
| 3 | 4923 |
| 4 | 4923 |
| 5 | 4540 |
| Total | 24232 |

Table 4.19 M5-Super Imposed Dead Load

| Super Imposed Dead Load Using <br> Hand Calculations (kN) | Super Imposed Dead Load <br> Using SAP2000 (kN) | \%Error |
| :---: | :---: | :---: |
| 7290 | 7290 | $0 \%$ |



Figure 4.6 M5-Super Imposed Dead Load Using SAP2000

## STEP 12 (Fundamental Period of the Structure)

The natural period of the structure is defined as the time needed to complete one cycle of vibration, which is the inverse of structure's natural frequency. The structure's fundamental period is parameter which is of tremendous importance to earthquake engineering, because when the forcing period nears the natural period of the structure, it will endure the massive vibration and experience the largest loss. This phenomenon is called Resonance, which plays an essential part in survival or failure to survival of structure during and after earthquake. In this study, the
fundamental period of the structure (T) is determined from SAP2000 software analysis. For M5 model, the fundamental period equals 0.69 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental periods for multi-degree structures using the following equation:

$$
\begin{equation*}
\mathrm{T}=2 \pi \sqrt{\frac{\sum_{\mathrm{i}=1}^{\mathrm{n}} \mathrm{~W}_{\mathrm{i}} \delta_{\mathrm{i}}^{2}}{\mathrm{~g} \sum_{\mathrm{i}=1}^{\mathrm{n}} \mathrm{f}_{\mathrm{i}} \delta_{\mathrm{i}}}} \tag{4.5}
\end{equation*}
$$

Where, $\delta_{\mathrm{i}}$ is lateral displacement of storey $\mathrm{i}, \mathrm{f}_{\mathrm{i}}$ is the horizontal load applied at storey $\mathrm{i}, \mathrm{W}_{\mathrm{i}}$ is the weight of storey $\mathrm{i}, \mathrm{n}$ is number of storeys in the structure. The values of $f_{i}$ represent any lateral force distribution in accordance with the principles of structural mechanics. The elastic deflections, $\delta_{\mathrm{i}}$, shall be calculated using the applied lateral forces as shown in Table 4.20.

Table 4.20 M5-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N} \mathbf{N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})^{\mathbf{b}}$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta} \mathbf{i}$ | $\mathbf{W}_{\mathbf{i}}{\boldsymbol{\delta} \mathbf{i}^{\mathbf{2}}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 4923 | 1000 | 0.006 | $3.12 \mathrm{E}-05$ | 5.59 | 0.15 |
| 2 | 4923 | 1000 | 0.015 | 0.00021 | 14.54 | 1.04 |
| 3 | 4923 | 1000 | 0.022 | 0.00050 | 22.45 | 2.48 |
| 4 | 4923 | 1000 | 0.028 | 0.00079 | 28.17 | 3.91 |
| 5 | 4540 | 1000 | 0.032 | 0.00100 | 31.67 | 4.55 |
|  |  |  |  |  | 102.42 | 12.14 |

a Assumed forces
b Deflections corresponding to the assumed forces

By applying Eq. 4.5 with Table 4.20, we get:
$\mathrm{T}=0.69$ second
$\%$ error $=\frac{\text { SAP Period }- \text { Approximated Period }}{\text { SAP Period }} \times 100 \%$
$=\frac{0.69-0.69}{0.69} \times 100 \%=0.0 \%$
STEP 13 (Maximum Period Limit)
ASCE 7 in section 12.8 .2 states that, the fundamental period, T , shall not exceed the product of the coefficient for upper limit on calculated period $\left(\mathrm{C}_{\mathrm{u}}\right)$ from Table 4.21 and the approximate fundamental period, $\mathrm{T}_{\mathrm{a}}$.

$$
\begin{equation*}
\mathrm{T}_{\mathrm{a}}=\mathrm{C}_{\mathrm{t}} \mathrm{~h}_{\mathrm{n}}{ }^{\mathrm{x}} \tag{4.6}
\end{equation*}
$$

Where, $\mathrm{h}_{\mathrm{n}}$ is the structural height and the coefficients $\mathrm{C}_{\mathrm{t}}$ and x are determined from Table 4.22.

Table 4.21 Coefficient for Upper Limit on Calculated Period (Table

## 12.8-1, ASCE 7)

Design Spectral Response Acceleration
Parameter at $1 \mathrm{~s}, S_{D 1}$
Coefficient $C_{u}$
$\geq 0.4 \quad 1.4$
0.3 1.4
$0.2 \quad 1.5$
0.15 1.6
$\begin{array}{ll}\leq 0.1 & 1.7\end{array}$

## Table 4.22 Values of Approximate Period Parameters Ct and x (Table

## 12.8-2, ASCE 7)

| Structure Type | $C_{t}$ | $x$ |
| :--- | :---: | :---: |
| Moment-resisting frame systems in which the frames resist $100 \%$ of the required seismic force <br> and are not enclosed or adjoined by components that are more rigid and will prevent the frames <br> from deflecting where subjected to seismic forces: |  |  |
| Steel moment-resisting frames | $0.028(0.0724)^{a}$ | 0.8 |
| Concrete moment-resisting frames | $0.016(0.0466)^{a}$ | 0.9 |
| Steel eccentrically braced frames in accordance with Table 12.2-1 lines B1 or D1 | $0.03(0.0731)^{a}$ | 0.75 |
| Steel buckling-restrained braced frames | $0.03(0.0731)^{a}$ | 0.75 |
| All other structural systems | $0.02(0.0488)^{a}$ | 0.75 |

${ }^{a}$ Metric equivalents are shown in parentheses.

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=5 \times 3.40=17 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 17^{0.9}=0.597$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 0.597=0.87$ second $>0.69$ second $\quad O K$

## STEP 14 (Response Spectrum Graph)

In earthquake engineering, there is many tools that can be used in designing of structures to support seismic loads, one of these tools is the response
spectrum which is a graph relates the response parameter (i.e. acceleration, displacement or velocity) with natural period or natural frequency of the structure for single degree of freedom oscillator as shown in Figs. 4.7 and 4.8. The parameters in the response spectrum can be determined from the following equations:

$$
\begin{gather*}
\mathrm{T}_{0}=0.2 \frac{\mathrm{~S}_{\mathrm{D} 1}}{\mathrm{~S}_{\mathrm{DS}}}  \tag{4.7}\\
\mathrm{~T}_{\mathrm{S}}=\frac{\mathrm{S}_{\mathrm{D} 1}}{\mathrm{~S}_{\mathrm{DS}}} \tag{4.8}
\end{gather*}
$$

Where, $\mathrm{S}_{\mathrm{DS}}$ and $\mathrm{S}_{\mathrm{D} 1}$ are as defined before, long-period transition period $\left(\mathrm{T}_{\mathrm{L}}\right)$ for Palestine equals 4 second.


Figure 4.7 Generic Elastic Response Spectrum


Figure 4.8 M5-Elastic Response Spectrum

In determining the base shear value and for design purposes; the inelastic response spectrum is used by dividing the elastic response accelerations by the response modification coefficient and multiplying it with the seismic importance factor as shown in Figs. 4.9 and 4.10.


Figure 4.9 Generic Inelastic Response Spectrum


Figure 4.10 M5-Inelastic Response Spectrum
STEP 15 (Determination of Seismic Base Shear using Equivalent Lateral Force Analysis)

The seismic base shear (V) in a given direction is determined in accordance with the following equation:

$$
\begin{equation*}
\mathrm{V}=\mathrm{C}_{\mathrm{s}} \mathrm{~W} \tag{4.9}
\end{equation*}
$$

Where:
W is the effective seismic weight
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS}} \times \mathrm{I}}{\mathrm{R}} \\ \frac{\mathrm{S}_{\mathrm{D} 1 \times I}}{\mathrm{R} \times \mathrm{T}}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 0.69}\end{array}=\left\{\begin{array}{l}0.063 \\ 0.045\end{array}=0.045\right.\right.\right.$
$\mathrm{S}_{\mathrm{DS}}, \mathrm{S}_{\mathrm{D} 1}, \mathrm{I}, \mathrm{R}$ and T are as defined before.
By applying Eq. 4.9, we get:
$\mathrm{V}=0.045 \times 24232=1090 \mathrm{kN}$

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

As stated earlier, Modal Response Spectrum Analysis was used using SAP2000 analysis software after defining the response spectrum function with the appropriate parameters as shown in Fig. 4.11. After applying the modal response spectrum analysis on the structure as a load case, the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,


Figure 4.11 Response Spectrum Function Using SAP2000

ASCE 7 section 12.9.4.1 states that, the base shear (V) using the modal response spectrum procedure shall not be less than 85 percent of the
calculated base shear ( V ) using the equivalent lateral force procedure as shown in Table 4.23.

## Table 4.23 M5-Base Shear Selection

| Base shear using MRSA <br> $(\mathbf{1}$ <br> $(\mathbf{k N})$ | Base shear using ELF <br> 2 <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5 \times \text { ELF } ^ { \mathbf { 2 } }}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 936 | 1090 | 926 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, base shear equals 936 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.24.

## Table 4.24 M5-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 936 | 57 |
| 2 | 6.80 | 879 | 131 |
| 3 | 10.2 | 748 | 188 |
| 4 | 13.6 | 560 | 254 |
| 5 | 17.0 | 306 | 306 |
|  |  |  |  |

## Verification of SAP2000 response shear results:

To verify response shear results which are created by SAP2000 software, new expressions shall be defined as follows:

$$
\begin{equation*}
\beta_{n}=\frac{\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{~W}_{i} \emptyset_{\text {in }}}{\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{~W}_{i} \emptyset_{i n}{ }^{2}} \tag{4.10}
\end{equation*}
$$

Where
$\beta_{\mathrm{n}}=$ participation factor for mode n
$\mathrm{i}=$ level number
$\mathrm{n}=$ mode number
$\mathrm{N}=$ total number of storeys
$\mathrm{W}_{\mathrm{i}}=$ weight at level i
$\Phi_{\text {in }}=$ mode shape value at level i in mode n

$$
\begin{equation*}
\mathrm{W}_{\mathrm{en}}=\frac{\left(\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{~W}_{i} \phi_{\mathrm{in}}\right)^{2}}{\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{~W}_{i} \phi_{i n}{ }^{2}} \tag{4.11}
\end{equation*}
$$

Where
$\mathrm{W}_{\mathrm{en}}=$ effective weight in mode n
$\mathrm{i}, \mathrm{n}, \mathrm{N}, \mathrm{W}_{\mathrm{i}}$ and $\Phi_{\mathrm{in}}$ are as defined before

$$
\begin{equation*}
\mathrm{F}_{\mathrm{in}}=\mathrm{aWi} \beta_{\mathrm{n}} \Phi_{\mathrm{in}} \tag{4.12}
\end{equation*}
$$

Where
$\mathrm{F}_{\text {in }}=$ force at each level i for mode n
$\mathrm{a}=$ spectral acceleration of SDOF system with a period corresponding to mode n
$\beta_{\mathrm{n}} \Phi_{\mathrm{in}}=$ the product of participation factor and mode shape, it is called participation function

Note: For reader's convenience previous symbols correspond with the book of Dynamics of Structures "Theory and Applications to Earthquake Engineering" by Anil K. Chopra as follows:
$\beta_{\mathrm{n}}=\Gamma_{\mathrm{n}}$
$\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{W}_{i} \emptyset_{\text {in }}=\mathrm{L}_{\mathrm{n}}$
$\sum_{\mathrm{i}=1}^{\mathrm{N}} \mathrm{W}_{i} \emptyset_{\text {in }}{ }^{2}=\mathrm{W}_{\mathrm{n}}$
$\mathrm{W}_{\mathrm{en}}=\mathrm{W}_{\mathrm{n}} *$
The applying of previous equations (Eqs.4.10, 4.11 and 4.12) is summarized in Tables 4.25-4.28.

Table 4.25 M5-Mode 1 Modal Mass Participation Ratio

| Mode 1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathrm{W}_{\mathrm{i}}(\mathrm{kN})$ | $\Phi_{\mathrm{i}}$ | $\Phi_{\mathrm{i}}{ }^{2}$ | $\mathrm{W}_{\mathrm{i}} \Phi_{\mathrm{i}}$ | $\mathrm{W}_{\mathrm{i}} \Phi_{\mathrm{i}}{ }^{2}$ |
| 1 | 4923 | 0.15 | 0.02 | 744.2 | 112.5 |
| 2 | 4923 | 0.42 | 0.17 | 2050.5 | 854.1 |
| 3 | 4923 | 0.67 | 0.45 | 3317.6 | 2235.8 |
| 4 | 4923 | 0.87 | 0.76 | 4300.1 | 3755.9 |
| 5 | 4540 | 1.00 | 1.00 | 4540.0 | 4540.0 |
|  |  |  | $\Sigma$ | 14952.4 | 11498.3 |
| $\beta_{1}=\frac{14952.4}{11498.3}=1.3$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 1}=\frac{14952 . \mathrm{4}^{2}}{11498.3}=19444 \mathrm{kN}$ |  |  |  |  |  |

Modal mass participation ratio of Mode $1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \%$

$$
=\frac{19444}{24232} \times 100 \%=80.2 \%
$$

Table 4.26 M5-Mode 1 Shear Distribution

| Mode 1 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\begin{gathered} \mathrm{T} \\ \text { (second) } \end{gathered}$ | $\mathrm{a}^{*}$ | W (kN) | $\beta \Phi_{\mathrm{i}}$ | Force (kN) | Shear (kN) |
| 1 | 0.69 | 0.045 | 4923 | 0.197 | 43.5 | 874.7 |
| 2 | 0.69 | 0.045 | 4923 | 0.541 | 120.0 | 831.2 |
| 3 | 0.69 | 0.045 | 4923 | 0.876 | 194.1 | 711.2 |
| 4 | 0.69 | 0.045 | 4923 | 1.136 | 251.6 | 517.1 |
| 5 | 0.69 | 0.045 | 4540 | 1.300 | 265.6 | 265.6 |

* response acceleration corresponding to mode period

Table 4.27 M5-Mode 2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathrm{W}_{\mathrm{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\Phi_{\mathrm{i}}{ }^{2}$ | $\mathrm{W}_{\mathrm{i}} \Phi_{\mathrm{i}}$ | $\mathrm{W}_{\mathrm{i}} \Phi_{\mathrm{i}}{ }^{2}$ |
| 1 | 4923 | 0.513 | 0.263 | 2525.2 | 1295.2 |
| 2 | 4923 | 1.000 | 1.000 | 4923.0 | 4923.0 |
| 3 | 4923 | 0.760 | 0.578 | 3741.6 | 2843.7 |
| 4 | 4923 | -0.098 | 0.010 | -481.9 | 47.2 |
| 5 | 4540 | -0.998 | 0.995 | -4529.1 | 4518.2 |
|  |  |  | $\Sigma$ | 6178.8 | 13627.4 |
| $\beta_{2}=\frac{6178.8}{13627.4}=0.453$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{6178.8^{2}}{13627.4}=2802 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 2=\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \% \\ =\frac{2802}{24232} \times 100 \%=11.6 \% \end{gathered}$ |  |  |  |  |  |

## Table 4.28 M5-Mode 2 Shear Distribution

| Mode 2 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | T <br> (second) | $\mathrm{a}^{*}$ | $\mathrm{~W}(\mathrm{kN})$ | $\beta \Phi_{\mathrm{i}}$ | Force (kN) | Shear (kN) |  |
| 1 | 0.21 | 0.063 | 4923 | 0.232 | 72.1 | 176.3 |  |
| 2 | 0.21 | 0.063 | 4923 | 0.453 | 140.5 | 104.3 |  |
| 3 | 0.21 | 0.063 | 4923 | 0.344 | 106.8 | -36.2 |  |
| 4 | 0.21 | 0.063 | 4923 | -0.044 | -13.8 | -143.0 |  |
| 5 | 0.21 | 0.063 | 4540 | -0.452 | -129.3 | -129.3 |  |

* response acceleration from response spectrum corresponding to mode period

Analysis shall include a sufficient number of modes to obtain a combined modal participation of at least $90 \%$ of the actual mass in each orthogonal
level (ASCE 7, section 12.9.1). For M5 case, mode1 and mode 2 give $91.8 \%$ of the actual mass. Thus, mode1 and mode 2 are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table 4.29. There are different ways in which results from the modes are combined to arrive at the final response, such as: CQC, SRSS or Absolute. CQC method is used to account for some interaction of modes when the modes are closely spaced. While the square root of sum of squares (SRSS) is used when the modes spaced far apart. Absolute is rarely used because it is quite unconservative. It was noted that the modes are spaced far apart so SRSS is adopted in this research.

## Table 4.29 M5-Shear Force Determination Using Both Hand

 Calculations and SAP2000| Level | Shear Using Hand Calculations <br> $(\mathrm{kN})$ | Shear Using <br> SAP2000 $(\mathrm{kN})$ | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 893 | 936 | $4.5 \%$ |
| 2 | 838 | 879 | $4.6 \%$ |

* acceptance limit is $10 \%$


### 4.2.1.4 P-Delta Effect

Two load cases on SAP2000 software are created. Number one: a linear load case which includes dead loads, live loads and seismic forces; Number two: a nonlinear load case with geometric nonlinearity which includes dead loads, live loads and seismic forces. The bending moment is determined on all columns in all floors and the column which gives the maximum percent difference between first-order analysis and second-order analysis is
adopted as tabulated in Table 4.30. Fig. 4.12 shows the numbering sequence for all columns in all models.


Figure 4.12 Models Numbering Sequence

Table 4.30 M5 Percent Difference between First-Order and SecondOrder

| M5 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | MSD+D+L | Mpositive <br> EQ. | MNEGATIVE <br> EQ. | First- <br> Order <br> Moment | Second- <br> Order <br> Moment | Percent |
| 1 | 26.7 | 164.1 | -164.1 | 190.8 | 192.6 | 0.96 |
| 2 | 0.2 | 179.7 | -179.7 | 179.9 | 181.9 | 1.07 |
| 3 | -0.2 | 179.7 | -179.7 | -179.9 | -181.9 | 1.07 |
| 4 | -26.7 | 164.1 | -164.1 | -190.8 | -192.6 | 0.96 |
| 5 | 44.1 | 165.0 | -165.0 | 209.1 | 210.9 | 0.83 |
| 6 | 0.1 | 180.7 | -180.7 | 180.8 | 182.6 | 0.96 |
| 7 | -0.1 | 180.7 | -180.7 | -180.8 | -182.6 | 0.96 |

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| 8 | -44.1 | 165.0 | -165.0 | -209.1 | -210.9 | 0.83 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 9 | 44.1 | 165.0 | -165.0 | 209.1 | 210.9 | 0.83 |
| 10 | 0.1 | 180.7 | -180.7 | 180.8 | 182.6 | 0.96 |
| 11 | -0.1 | 180.7 | -180.7 | -180.8 | -182.6 | 0.96 |
| 12 | -44.1 | 165.0 | -165.0 | -209.1 | -210.9 | 0.83 |
| 13 | 26.7 | 164.1 | -164.1 | 190.8 | 192.6 | 0.96 |
| 14 | 0.2 | 179.7 | -179.7 | 179.9 | 181.9 | 1.07 |
| 15 | -0.2 | 179.7 | -179.7 | -179.9 | -181.9 | 1.07 |
| 16 | -26.7 | 164.1 | -164.1 | -190.8 | -192.6 | 0.96 |
| 17 | 72.4 | 80.2 | -80.2 | 152.5 | 153.8 | 0.81 |
| 18 | -0.4 | 127.0 | -127.0 | -127.4 | -129.3 | 1.46 |
| 19 | 0.4 | 127.0 | -127.0 | 127.4 | 129.3 | $\underline{1.46}$ |
| 20 | -72.4 | 80.2 | -80.2 | -152.5 | -153.8 | 0.81 |
| 21 | 123.0 | 81.3 | -81.3 | 204.3 | 205.5 | 0.61 |
| 22 | -0.2 | 129.3 | -129.3 | -129.5 | -131.4 | 1.43 |
| 23 | 0.2 | 129.3 | -129.3 | 129.5 | 131.4 | 1.43 |
| 24 | -123.0 | 81.3 | -81.3 | -204.3 | -205.5 | 0.61 |
| 25 | 123.0 | 81.3 | -81.3 | 204.3 | 205.5 | 0.61 |
| 26 | -0.2 | 129.3 | -129.3 | -129.5 | -131.4 | 1.43 |
| 27 | 0.2 | 129.3 | -129.3 | 129.5 | 131.4 | 1.43 |
| 28 | -123.0 | 81.3 | -81.3 | -204.3 | -205.5 | 0.61 |
| 29 | 72.4 | 80.2 | -80.2 | 152.5 | 153.8 | 0.81 |
| 30 | -0.4 | 127.0 | -127.0 | -127.4 | -129.3 | $\underline{1.46}$ |
| 31 | 0.4 | 127.0 | -127.0 | 127.4 | 129.3 | $\underline{1.46}$ |
| 32 | -72.4 | 80.2 | -80.2 | -152.5 | -153.8 | 0.81 |
| 33 | 66.4 | 44.3 | -44.3 | 110.8 | 111.2 | 0.40 |
| 34 | 2.6 | 90.3 | -90.3 | 92.9 | 93.9 | 1.13 |
| 35 | -2.6 | 90.3 | -90.3 | -92.9 | -93.9 | 1.13 |
| 36 | -66.4 | 44.3 | -44.3 | -110.8 | -111.2 | 0.40 |
| 37 | 112.8 | 45.6 | -45.6 | 158.4 | 158.8 | 0.29 |
| 38 | 4.5 | 92.5 | -92.5 | 97.0 | 98.1 | 1.11 |
| 39 | -4.5 | 92.5 | -92.5 | -97.0 | -98.1 | 1.11 |
| 40 | -112.8 | 45.6 | -45.6 | -158.4 | -158.8 | 0.29 |
| 41 | 112.8 | 45.6 | -45.6 | 158.4 | 158.8 | 0.29 |
| 42 | 4.5 | 92.5 | -92.5 | 97.0 | 98.1 | 1.11 |
| 43 | -4.5 | 92.5 | -92.5 | -97.0 | -98.1 | 1.11 |
| 44 | -112.8 | 45.6 | -45.6 | -158.4 | -158.8 | 0.29 |
| 45 | 66.4 | 44.3 | -44.3 | 110.8 | 111.2 | 0.40 |
| 46 | 2.6 | 90.3 | -90.3 | 92.9 | 93.9 | 1.13 |
| 47 | -2.6 | 90.3 | -90.3 | -92.9 | -93.9 | 1.13 |
| 48 | -66.4 | 44.3 | -44.3 | -110.8 | -111.2 | 0.40 |
| 49 | 67.2 | 19.6 | -19.6 | 86.8 | 86.8 | 0.00 |
| 50 | 4.2 | 56.5 | -56.5 | 60.7 | 61.1 | 0.62 |
| 51 | -4.2 | 56.5 | -56.5 | -60.7 | -61.1 | 0.62 |
| 52 | -67.2 | 19.6 | -19.6 | -86.8 | -86.8 | 0.00 |
| 53 | 113.7 | 20.7 | -20.7 | 134.4 | 134.4 | 0.00 |
| 54 | 7.0 | 58.4 | -58.4 | 65.3 | 65.8 | 0.64 |

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| 55 | -7.0 | 58.4 | -58.4 | -65.3 | -65.8 | 0.64 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 56 | -113.7 | 20.7 | -20.7 | -134.4 | -134.4 | 0.00 |
| 57 | 113.7 | 20.7 | -20.7 | 134.4 | 134.4 | 0.00 |
| 58 | 7.0 | 58.4 | -58.4 | 65.3 | 65.8 | 0.64 |
| 59 | -7.0 | 58.4 | -58.4 | -65.3 | -65.8 | 0.64 |
| 60 | -113.7 | 20.7 | -20.7 | -134.4 | -134.4 | 0.00 |
| 61 | 67.2 | 19.6 | -19.6 | 86.8 | 86.8 | 0.00 |
| 62 | 4.2 | 56.5 | -56.5 | 60.7 | 61.1 | 0.62 |
| 63 | -4.2 | 56.5 | -56.5 | -60.7 | -61.1 | 0.62 |
| 64 | -67.2 | 19.6 | -19.6 | -86.8 | -86.8 | 0.00 |
| 65 | 80.5 | -2.5 | 2.5 | 77.9 | 77.9 | 0.00 |
| 66 | 2.9 | 25.2 | -25.2 | 28.1 | 28.2 | 0.20 |
| 67 | -2.9 | 25.2 | -25.2 | -28.1 | -28.2 | 0.20 |
| 68 | -80.5 | -2.5 | 2.5 | -83.0 | -83.2 | 0.22 |
| 69 | 137.3 | -1.9 | 1.9 | 139.2 | 139.4 | 0.13 |
| 70 | 6.1 | 26.5 | -26.5 | 32.7 | 32.8 | 0.26 |
| 71 | -6.1 | 26.5 | -26.5 | -32.7 | -32.8 | 0.26 |
| 72 | -137.3 | -1.9 | 1.9 | -139.2 | -139.4 | 0.13 |
| 73 | 137.3 | -1.9 | 1.9 | 139.2 | 139.4 | 0.13 |
| 74 | 6.1 | 26.5 | -26.5 | 32.7 | 32.8 | 0.26 |
| 75 | -6.1 | 26.5 | -26.5 | -32.7 | -32.8 | 0.26 |
| 76 | -137.3 | -1.9 | 1.9 | -139.2 | -139.4 | 0.13 |
| 77 | 80.5 | -2.5 | 2.5 | 83.0 | 83.2 | 0.22 |
| 78 | 2.9 | 25.2 | -25.2 | 28.1 | 28.2 | 0.20 |
| 79 | -2.9 | 25.2 | -25.2 | -28.1 | -28.2 | 0.20 |
| 80 | -80.5 | -2.5 | 2.5 | -83.0 | -83.2 | 0.22 |

* All calculations are on the bottom of columns


## Verification of Second-Order Analysis

To verify second-order analysis results, it is required to determine both structure P-Delta effect and member P-Delta effect. Table 4.31 summarizes the first-order bending moments from all load cases on the column number four.

## Table 4.31 M5-First-Order Moment Analysis on Column Number

Four

|  | Moment on Base <br> $(\mathbf{k N . m})$ | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -5.48 | 11.5 |
| Super Imposed Load | -8.23 | 17.2 |
| Dead Load | -12.9 | 27.1 |
| Earthquake Load | -164 | 9.3 |

$\sum \mathrm{P}=$ summation of axial forces $=29475 \mathrm{kN}$
$\mathrm{I}_{\text {Column,Crack }}=0.8 \times \mathrm{I}_{\text {Column }}=0.8 \times \frac{0.75^{4}}{12}=0.0211 \mathrm{~m}^{4}$
$\mathrm{I}_{\text {Beam,Crack }}=0.4 \times \mathrm{I}_{\text {Beam }}=0.4 \times \frac{0.75 \times 0.7^{3}}{12}=8.58 \times 10^{-3} \mathrm{~m}^{4}$
$\mathrm{E}_{\text {Column }}=26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$
$E_{\text {Beam }}=23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$

## P- $\Delta$ effect (structure P-Delta effect)

## For middle:

After applying Eq.2.13, we get:
$\mathrm{G}_{\mathrm{A}}=\frac{26.6 \times 10^{6} \times 0.0211 \times 2 / 3.4}{23 \times 10^{6} \times 8.58 \times 10^{-3} \times 2 / 6}=5.02$
$\mathrm{G}_{\mathrm{B}}=0$ (Fixed Support)
The effective length factor $\mathrm{K}=1.5$ (From Fig. 2.16)

$$
\mathrm{P}_{\text {ek middle }}=\frac{\pi^{2} E I}{(K L)^{2}}=\frac{\pi^{2} \times 26.6 \times 10^{6} \times 0.0211}{(1.5 \times 3.4)^{2}}=212970 \mathrm{kN}
$$

## For Edge:

After applying Eq.2.13, we get:
$\mathrm{G}_{\mathrm{A}}=\frac{26.6 \times 10^{6} \times 0.0211 \times 2 / 3.4}{23 \times 10^{6} \times 8.58 \times 10^{-3} / 6}=10.03$
$G_{B}=0($ Fixed Support $)$

The effective length factor $\mathrm{K}=1.68$ (From Fig. 2.16)
$\mathrm{P}_{\text {ek edge }}=\frac{\pi^{2} E I}{(K L)^{2}}=\frac{\pi^{2} \times 26.6 \times 10^{6} \times 0.0211}{(1.68 \times 3.4)^{2}}=169780 \mathrm{kN}$
$\sum \mathrm{P}_{\text {ek }}=8 \times \mathrm{P}_{\text {ek middle }}+8 \times \mathrm{P}_{\text {ek edge }}=3062000 \mathrm{kN}$
After applying Eq.2.11, we get:
$\mathrm{B}_{2}=\left(\frac{1}{1-\sum \mathrm{P} / \sum \mathrm{P}_{\mathrm{ek}}}\right)=\frac{1}{1-29475 / 3062000}=1.01$

## P- $\boldsymbol{\delta}$ Effect (member P-Delta effect)

By applying Eq. 2.8, we get
$C_{m}=0.6-0.4\left(\frac{M_{A}}{M_{B}}\right)$
Where, $M_{B}$ is the larger of the two end moments.
$\mathrm{M}_{\mathrm{A}}=\mathrm{M}_{\text {Live }}+\mathrm{M}_{\mathrm{SD}}+\mathrm{M}_{\text {Dead }}=-26.6 \mathrm{kN} . \mathrm{m}$
$\mathrm{M}_{\mathrm{B}}=\mathrm{M}_{\mathrm{Live}}+\mathrm{M}_{\mathrm{SD}}+\mathrm{M}_{\text {Dead }}=55.8 \mathrm{kN} . \mathrm{m}$
$C_{m}=0.6-0.4\left(\frac{26.6}{55.8}\right)=0.41$
$\mathrm{P}=$ axial load on prescribed column $=1057 \mathrm{kN}($ SAP2000 $)$
$\mathrm{G}_{\mathrm{A}}=\frac{26.6 \times 10^{6} \times 0.0211 \times 2 / 3.4}{23 \times 10^{6} \times 6.25 \times 10^{-3} / 6}=10.03$
$G_{B}=0($ Fixed Support $)$

The effective length factor $K=0.7$ (From Fig. 2.14)
$P_{e}=\frac{\pi^{2} \mathrm{EI}}{(\mathrm{KL})^{2}}=\frac{\pi^{2} \times 26.6 \times 10^{6} \times 0.0211}{(0.7 \times 3.4)^{2}}=977935 \mathrm{kN}$

By applying Eq. 2.6, we get

$$
\mathrm{B}_{1}=\frac{\mathrm{C}_{\mathrm{m}}}{1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{e}}}}=\frac{0.41}{1-\frac{1057}{977935}}=0.41\left(\text { Take } \mathrm{B}_{1}=1\right)
$$

$\mathrm{M}_{\mathrm{u}}=\mathrm{B}_{2} \mathrm{M}_{1 \text { sway }}+\mathrm{B}_{1} \mathrm{M}_{1 \text { non-sway }}=1.01 \times-164-5.48-8.23-12.9=-192.3 \mathrm{kN} . \mathrm{m}$
The second-order moment results for prescribed column from both hand calculations and SAP2000 are tabulated in Table 3.32.

Table 4.32 M5-Difference between Hand Calculations and SAP2000 Results on Column Number Four

|  | Second-Order Analysis <br> Using Moment <br> Amplification Method | Second-Order <br> Analysis Using <br> SAP2000 | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending <br> Moment <br> (kN.m) $\operatorname{-192.3}$ | -192.6 | $0.1 \%$ |  |

### 4.2.1.5 Storey Drift Determination

The storey drift shall be computed as the difference of the deflections at the center of mass at the top and bottom of the storey under consideration (ASCE 7, section 12.8.6). Storey drift is an important indicator of structural behavior in performance-based seismic analysis especially for high-rise structures. During an earthquake, large lateral forces can be imposed on structures which require the designer to assess the effects of this deformation on both structural and non-structural elements. Storey drift has three primary effects on a structure; the movement can affect the structural elements (such as beams and columns); the movements can affect
non-structural elements (such as the windows and cladding); and the movements can affect adjacent structures.

Design provisions for moment frame and eccentric braced frame structures have requirements to ensure the ability of the structure to sustain inelastic rotations resulting from deformation and drift. Without proper consideration of the expected movement of the structure, the lateral force resisting system might experience premature failure and a corresponding loss of strength. In addition, if the lateral deflections of any structure become too large, $\mathrm{P}-\Delta$ effects can cause instability of the structure and potentially result in collapse. Thus, it is vital to determine the storey drift for every structure and making sure it is under the allowable limits prescribed in ASCE 7 as follows:

$$
\begin{align*}
& \Delta=\delta_{\mathrm{x}}-\delta_{\mathrm{x}-1}  \tag{4.13}\\
& \delta_{\mathrm{x}}=\frac{\mathrm{C}_{\mathrm{d}} \delta_{\mathrm{xe}}}{\mathrm{I}_{\mathrm{e}}} \tag{4.14}
\end{align*}
$$

Where
$\mathrm{C}_{\mathrm{d}}=$ the deflection amplification factor in Table 12.2, ASCE 7
$\delta_{\mathrm{xe}}=$ Elastic displacement
$\mathrm{I}_{\mathrm{e}}=$ importance factor
The storey drifts are determined and checked for allowable limit and tabulated in Table 4.33.

## Table 4.33 M5 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})^{\mathbf{b}}$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})^{\mathbf{c}}$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 1.13 | 6.21 | 6.21 | 52 | OK |
| 2 | 3.10 | 17.06 | 10.85 | 52 | OK |
| 3 | 5.02 | 27.60 | 10.54 | 52 | OK |
| 4 | 6.53 | 35.91 | 8.30 | 52 | OK |
| 5 | 7.51 | 41.31 | 5.40 | 52 | OK |

a Elastic displacement from seismic load case
b Amplified displacement (ASCE 7, Eq. 12.8-15)
c $\Delta_{\text {Allowable }}=0.0153 h_{\text {sx }}($ ASCE 7, Table 12.12-1 $)$

### 4.2.1.6 Stability Coefficient

The stability coefficient is an indicator to indicate the need for considering P-Delta effect in the analysis. ASCE 7, section 12.8 .7 states that P-Delta effects on story shears and moments, the resulting member forces and moments, and the story drifts induced by these effects are not required to be considered where stability coefficient $(\theta)$ as determined by the following equation is equal to or less than 0.10 :

$$
\begin{equation*}
\theta=\frac{\mathrm{P}_{\mathrm{x}} \Delta \mathrm{I}_{\mathrm{e}}}{\mathrm{~V}_{\mathrm{x}} \mathrm{~h}_{\mathrm{sx}} \mathrm{C}_{\mathrm{d}}} \tag{4.15}
\end{equation*}
$$

Where
$P_{x}=$ the total vertical design load at and above level $x$ in $k N$; where computing $\mathrm{P}_{\mathrm{x}}$, no individual load factor need exceed 1.0
$\Delta=$ the design story drift as defined in equation 4.13
$\mathrm{I}_{\mathrm{e}}=$ importance factor
$\mathrm{V}_{\mathrm{x}}=$ the seismic shear force acting between levels x and $\mathrm{x}-1$
$\mathrm{h}_{\mathrm{sx}}=$ the story height below level x
$\mathrm{C}_{\mathrm{d}}=$ the deflection amplification factor in Table 12.2-1, ASCE 7
The Storeys' coefficients are determined and tabulated in Table4.34.

Table 4.34 M5 Stability Coefficient

| M5 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |  |
| 1 | 29475 | 6.21 | 936 | 3400 | 5.5 | 0.010 |  |
| 2 | 23580 | 10.85 | 879 | 3400 | 5.5 | $\mathbf{0 . 0 1 6}$ |  |
| 3 | 17685 | 10.54 | 748 | 3400 | 5.5 | 0.013 |  |
| 4 | 11790 | 8.30 | 560 | 3400 | 5.5 | 0.009 |  |
| 5 | 5130 | 5.40 | 306 | 3400 | 5.5 | 0.005 |  |

The maximum stability coefficient for model M5 is less than 0.10. Thus, PDelta effect is not required to be considered. It is worth mentioning that the maximum difference between first-order moment and second-order moment on the critical column is very close to the stability coefficient's value. Furthermore, the stability coefficient gives more conservative value in this case.

### 4.2.2 Ten-Storey model (M10)

M10 is a ten-storey reinforced concrete portal frame with fixed supports with a total height equals 34 m . M10 has the same plan, material properties, section properties, storey's height, live loads, and superimposed dead loads as M5 except number of stories and columns' sections. The columns' sections are as shown in Table 4.35. Elevation of M10 is illustrated in Fig. 4.13.

Table 4.35 M10-Columns' Sections

|  | Column |  |
| :---: | :---: | :---: |
| Storey | Depth (m) | Width (m) |
| $1-5$ | 0.90 | 0.90 |
| $6-10$ | 0.75 | 0.75 |
|  |  |  |



Figure 4.13 M10 Model in 3-D view

Parameters that can be used to determine the base shear using International Building Code 2015 (IBC 2015) in co-operation with ASCE 7 can be determined in the same way as M5 model. M10 has the same values as M5 from step 1 until step 10 as summarized in Table 4.36. The Calculations for the rest values are resumed from step 11.

Table 4.36 M10 Model Summary from Step One to Ten

| Step | Parameter | Value |
| :---: | :---: | :---: |
| 1 | Seismic Zone Factor (Z) | 0.20 |
| 2 | Site Classification | B |
| 3 | Ss | 0.50 |
|  | $\mathrm{S}_{1}$ | 0.25 |
| 4 | $\mathrm{F}_{\mathrm{a}}$ | 1 |
|  | $\mathrm{F}_{\mathrm{v}}$ | 1 |
| 5 | SDS | 0.50 |
|  | $\mathrm{S}_{\text {D1 }}$ | 0.25 |
| 6 | Risk Category of the Structure | II |
| 7 | Seismic Design Category | D |
| 8 | Seismic Importance Factor | 1 |
| 9 | Permitted Analytical Procedure | MRSA* is used |
| 10 | Response Modification Coefficient, R | 8 |
|  | Over Strength Factor, $\Omega_{0}$ | 3 |
|  | Deflection Amplification Factor, $\mathrm{C}_{\mathrm{d}}$ | 5.5 |

* Modal Response Spectrum Analysis


## STEP 11(Effective Seismic Weight)

Each floor weight is determined in the same way as calculating M5 weights except for M10 model, the columns' sections are changing throughout the elevation. M10 columns' weights in each storey are determined and tabulated in Table 4.37. The weight of each storey can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last storey is exempted from half columns' weights as tabulated in Table 4.38.

Table 4.37 M10-Columns Weight in each Storey

| Storey | Number of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1-5$ | 16 | 0.9 m | 0.9 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1102 kN |
| $6-10$ | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

Table 4.38 M10-Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 5260 |
| 2 | 5260 |
| 3 | 5260 |
| 4 | 5260 |
| 5 | 5260 |
| 6 | 4923 |
| 7 | 4923 |
| 8 | 4923 |
| 9 | 4923 |
| 10 | 4540 |
| Total | 50532 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure (T) is determined from SAP2000, version 19.0.0 software analysis. For M10 model, the fundamental period equals 1.38 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental periods using the Eq. 4.5 as shown in Table 4.39.

By applying Eq. 4.5 with Table 4.39
$\mathrm{T}=1.37$ second
$\%$ error $=\frac{\text { SAP Period }- \text { Approximated Period }}{\text { SAP Period }} \times 100 \%$

$$
=\frac{1.38-1.37}{1.38} \times 100 \%=0.72 \%<10 \% \text { Acceptable }
$$

Table 4.39 M10-Verification of SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N} \mathbf{N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})^{\mathbf{b}}$ | $\boldsymbol{\delta}_{\mathbf{i}}{ }^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 5260 | 1000 | 0.008 | $7.03 \mathrm{E}-05$ | 8.4 | 0.37 |
| 2 | 5260 | 1000 | 0.025 | 0.0006 | 24.9 | 3.25 |
| 3 | 5260 | 1000 | 0.043 | 0.0019 | 43.1 | 9.78 |
| 4 | 5260 | 1000 | 0.061 | 0.0037 | 60.7 | 19.37 |
| 5 | 5260 | 1000 | 0.077 | 0.0059 | 76.7 | 30.93 |
| 6 | 4923 | 1000 | 0.091 | 0.0084 | 91.4 | 41.16 |
| 7 | 4923 | 1000 | 0.103 | 0.0107 | 103.3 | 52.53 |
| 8 | 4923 | 1000 | 0.112 | 0.0126 | 112.4 | 62.15 |
| 9 | 4923 | 1000 | 0.119 | 0.0141 | 118.7 | 69.40 |
| 10 | 4540 | 1000 | 0.123 | 0.0151 | 122.8 | 68.41 |
|  |  |  |  | $\sum$ | 762.3 | 357.4 |

a Assumed forces
b The displacement corresponding to the assumed forces

## STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=10 \times 3.40=34 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 34^{0.9}=1.11$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 1.11=1.61$ second $>1.38$ second $\quad O K$

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for M10 model are created in the same way as creating the response spectrums for M5 model as shown in Figs. 4.14 and 4.15 respectively.

STEP 15 (Determination of Seismic Base Shear using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS}} \times \mathrm{I}}{R} \\ \frac{\mathrm{~S}_{\mathrm{D} 1 \times \mathrm{I}}}{\mathrm{R} \times \mathrm{T}}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 1.38}\end{array}=\left\{\begin{array}{l}0.063 \\ 0.022\end{array}=0.022\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.022 \times 50532=1112 \mathrm{kN}$


Figure 4.14 M10-Elastic Response Spectrum


Figure 4.15 M10-Inelastic Response Spectrum

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum was defined as for M5 model and the base shear $(\mathrm{V})$ has been computed by using section cut at columns of ground floor for the response spectrum analysis case and the result is as follows, $\mathrm{V}=960 \mathrm{kN}$

Table 4.40 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

## Table 4.40 M10-Base Shear Selection

| Base shear using MRSA <br> ( <br> $(\mathbf{k N})$ | Base shear using ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 960 | 1112 | 945 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base-shear using equivalent lateral force procedure. Thus, base shear equals 960 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.41.

## Table 4.41 M10-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})^{*}$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 960 | 26 |
| 2 | 6.80 | 934 | 59 |
| 3 | 10.2 | 875 | 70 |
| 4 | 13.6 | 805 | 71 |
| 5 | 17.0 | 734 | 68 |
| 6 | 20.4 | 666 | 69 |
| 7 | 23.8 | 597 | 88 |
| 8 | 27.2 | 509 | 119 |
| 9 | 30.6 | 390 | 169 |
| 10 | 34.0 | 221 | 221 |
|  |  |  |  |

* Verification is in appendix A

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B and verified in appendix C. The critical column is tabulated in Table 4.42 .

Table 4.42 M10-Bending Moment Results on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 168.7 | 173.8 | $3.05 \%$ |

Storey drifts and stability coefficients for model M10 are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.43 and 4.44 respectively.

Table 4.43 M10 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.85 | 4.69 | 4.69 | 52 | OK |
| 2 | 2.60 | 14.28 | 9.58 | 52 | OK |
| 3 | 4.61 | 25.35 | 11.08 | 52 | OK |
| 4 | 6.64 | 36.51 | 11.15 | 52 | OK |
| 5 | 8.59 | 47.22 | 10.71 | 52 | OK |
| 6 | 10.52 | 57.86 | 10.64 | 52 | OK |
| 7 | 12.24 | 67.33 | 9.48 | 52 | OK |
| 8 | 13.71 | 75.38 | 8.04 | 52 | OK |
| 9 | 14.84 | 81.61 | 6.23 | 52 | OK |
| 10 | 15.60 | 85.80 | 4.19 | 52 | OK |

Table 4.44 M10 Stability Coefficient

|  |  |  | $\mathbf{M 1 0}$ |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta \mathbf{( m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 60633 | 4.69 | 960 | 3400 | 5.5 | 0.016 |
| 2 | 54401 | 9.58 | 934 | 3400 | 5.5 | 0.030 |
| 3 | 48170 | 11.08 | 875 | 3400 | 5.5 | $\underline{\mathbf{0 . 0 3 3}}$ |
| 4 | 41938 | 11.15 | 805 | 3400 | 5.5 | 0.031 |
| 5 | 35707 | 10.71 | 734 | 3400 | 5.5 | 0.028 |
| 6 | 29475 | 10.64 | 666 | 3400 | 5.5 | 0.025 |
| 7 | 23580 | 9.48 | 597 | 3400 | 5.5 | 0.020 |
| 8 | 17685 | 8.04 | 509 | 3400 | 5.5 | 0.015 |
| 9 | 11790 | 6.23 | 390 | 3400 | 5.5 | 0.010 |
| 10 | 5130 | 4.19 | 221 | 3400 | 5.5 | 0.005 |

The P-Delta analysis is not required based on the stability coefficient which is less than 0.1 . The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.42) on the critical column. However, the stability coefficient gives more conservative result compared to the percent difference on the critical column.

### 4.2.3 Fifteen-Storey model (M15)

M15 is a fifteen-storey reinforced concrete portal frame with fixed supports with a total height equals 51 m . M15 has the same plan, material properties, section properties, story's height, live loads, and superimposed dead loads as M5 except number of stories and columns' sections. The columns' sections are as shown in Table 4.45.

## Table 4.45 M15-Columns' Sections

|  | Column |  |
| :---: | :---: | :---: |
| Storey | Depth (m) | Width (m) |
| $1-5$ | 1.00 | 1.00 |
| $6-10$ | 0.90 | 0.90 |
| $11-15$ | 0.75 | 0.75 |
|  |  |  |

Elevation of M15 is illustrated in Fig. 4.16. Parameters that can be used to determine the base shear using International Building Code 2015 (IBC 2015) in co-operation with ASCE 7 can be determined in the same way as M5 model. M15 has the same values as M5 from step 1 until step 10 as summarized in Table 4.36. The Calculations for the rest values are resumed from step 11.


Figure 4.16 M15 Model in 3-D view

## STEP 11 (Effective Seismic Weight)

The columns' sections change throughout the elevation. M15 columns' weights in each storey are determined and tabulated in Table 4.46. The weight of each storey can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last storey is exempted from half columns' weights as tabulated in Table 4.47.

Table 4.46 M15-Columns Weight in each Storey

| Storey | Number of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1-5$ | 16 | 1.0 m | 1.0 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1360 kN |
| $6-10$ | 16 | 0.9 m | 0.9 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1102 kN |
| $11-15$ | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

Table 4.47 M15-Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 5518 |
| 2 | 5518 |
| 3 | 5518 |
| 4 | 5518 |
| 5 | 5518 |
| 6 | 5260 |
| 7 | 5260 |
| 8 | 5260 |
| 9 | 5260 |
| 10 | 5260 |
| 11 | 4923 |
| 12 | 4923 |
| 13 | 4923 |
| 14 | 4923 |
| 15 | 4540 |
| Total | 78122 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure (T) is determined from SAP2000, version 19.0.0 software analysis. For M15 model, the fundamental period equals 2.09 second.

## Verification of SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.48.

By applying Eq. 4.5 with Table 4.48
Approximated period $=2.08$ second
$\%$ error $=\frac{\text { Approximated Period }- \text { SAP Period }}{\text { SAP Period }} \times 100 \%=0.48 \%<10 \%$
Acceptable

Table 4.48 M15-Verification of SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N} \mathbf{N}$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})^{\mathbf{b}}$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}} \mathbf{2}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 5518 | 1000 | 0.011 | 0.0001 | 10.7 | 0.6 |
| 2 | 5518 | 1000 | 0.033 | 0.0011 | 33.4 | 6.2 |
| 3 | 5518 | 1000 | 0.061 | 0.0037 | 60.7 | 20.3 |
| 4 | 5518 | 1000 | 0.089 | 0.0079 | 88.9 | 43.6 |
| 5 | 5518 | 1000 | 0.116 | 0.0136 | 116.5 | 74.9 |
| 6 | 5260 | 1000 | 0.143 | 0.0205 | 143.1 | 107.7 |
| 7 | 5260 | 1000 | 0.167 | 0.0280 | 167.5 | 147.5 |
| 8 | 5260 | 1000 | 0.189 | 0.0359 | 189.5 | 188.8 |
| 9 | 5260 | 1000 | 0.209 | 0.0437 | 209.1 | 230.0 |
| 10 | 5260 | 1000 | 0.227 | 0.0513 | 226.6 | 270.0 |
| 11 | 4923 | 1000 | 0.242 | 0.0588 | 242.4 | 289.3 |
| 12 | 4923 | 1000 | 0.255 | 0.0652 | 255.3 | 320.8 |
| 13 | 4923 | 1000 | 0.265 | 0.0704 | 265.3 | 346.5 |
| 14 | 4923 | 1000 | 0.273 | 0.0743 | 272.6 | 365.9 |
| 15 | 4540 | 1000 | 0.278 | 0.0771 | 277.6 | 349.8 |
|  |  |  |  | $\sum$ | 2559 | 2762 |

a Assumed forces
b The displacement corresponding to the assumed forces

## STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=15 \times 3.40=51 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 51^{0.9}=1.60$ seconds
After referring to Table 4.21
$\mathrm{S}_{\mathrm{D} 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 1.60=2.32$ second $>2.09$ second $\quad O K$

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for M15 model are created in the same way as previously stated and the resulted graphs are as shown in Figs. 4.17 and 4.18 respectively.


Figure 4.17 M15-Elastic Response Spectrum


Figure 4.18 M15-Inelastic Response Spectrum
STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear (V) in a given direction is determined by applying Eq. 4.9 as follows:
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS}} \times \mathrm{I}}{} \\ \frac{\mathrm{S}_{\mathrm{D} 1} \times \mathrm{I}}{} \\ \mathrm{R} \times \mathrm{T}\end{array}=\left\{\begin{array}{c}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 2.09}\end{array}=\left\{\begin{array}{c}0.063 \\ 0.0149\end{array}=0.0149\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0149 \times 78122=1164 \mathrm{kN}$

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum was defined as for M5 model and the base shear $(\mathrm{V})$ has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=1007 \mathrm{kN}$
Table 4.49 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

Table 4.49 M15-Base Shear Selection

| Base shear using MRSA $^{\mathbf{1}}$ <br> $(\mathbf{k N})$ | Base shear using ELF $^{2}$ <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 1007 | 1164 | 989 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, base shear equals 1007 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.50.

Table 4.50 M15-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})^{*}$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 1007 | 14 |
| 2 | 6.80 | 993 | 39 |
| 3 | 10.2 | 954 | 53 |
| 4 | 13.6 | 901 | 52 |
| 5 | 17.0 | 849 | 46 |
| 6 | 20.4 | 803 | 44 |
| 7 | 23.8 | 759 | 45 |
| 8 | 27.2 | 714 | 45 |
| 9 | 30.6 | 669 | 46 |
| 10 | 34.0 | 623 | 51 |
| 11 | 37.4 | 572 | 61 |
| 12 | 40.8 | 511 | 75 |
| 13 | 44.2 | 436 | 105 |
| 14 | 47.6 | 331 | 151 |
| 15 | 51.0 | 180 | 180 |

* Verification is in appendix A

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B and verified in appendix C. The critical column is tabulated in Table 4.51.

## Table 4.51 Bending Moment Results for M15 Model on Critical

 Column|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 167.1 | 175.5 | $4.97 \%$ |

Storey drifts and stability coefficients for model M15 are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.52 and 4.53 respectively.

## Table 4.52 M15 Drift Check

| Level | $\boldsymbol{\delta}_{\text {xe }}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta \mathbf{( m m})$ | $\Delta_{\text {Allowable }}(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.75 | 4.14 | 4.14 | 52 | OK |
| 2 | 2.40 | 13.19 | 9.04 | 52 | OK |
| 3 | 4.42 | 24.29 | 11.10 | 52 | OK |
| 4 | 6.56 | 36.06 | 11.77 | 52 | OK |
| 5 | 8.70 | 47.87 | 11.81 | 52 | OK |
| 6 | 10.85 | 59.69 | 11.82 | 52 | OK |
| 7 | 12.92 | 71.04 | 11.35 | 52 | OK |
| 8 | 14.88 | 81.86 | 10.82 | 52 | OK |
| 9 | 16.75 | 92.11 | 10.25 | 52 | OK |
| 10 | 18.51 | 101.82 | 9.71 | 52 | OK |
| 11 | 20.26 | 111.43 | 9.61 | 52 | OK |
| 12 | 21.82 | 120.02 | 8.59 | 52 | OK |
| 13 | 23.16 | 127.36 | 7.34 | 52 | OK |
| 14 | 24.20 | 133.12 | 5.76 | 52 | OK |
| 15 | 24.93 | 137.10 | 3.98 | 52 | OK |

Table 4.53 M15 Stability Coefficient

| $\mathbf{M 1 5}$ |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 93083 | 4.14 | 1007 | 3400 | 5.5 | 0.020 |
| 2 | 86593 | 9.04 | 993 | 3400 | 5.5 | 0.042 |
| 3 | 80103 | 11.1 | 954 | 3400 | 5.5 | 0.050 |
| 4 | 73613 | 11.77 | 901 | 3400 | 5.5 | $\mathbf{0 . 0 5 1}$ |
| 5 | 67123 | 11.81 | 849 | 3400 | 5.5 | 0.050 |
| 6 | 60633 | 11.82 | 803 | 3400 | 5.5 | 0.048 |
| 7 | 54401 | 11.35 | 759 | 3400 | 5.5 | 0.044 |
| 8 | 48170 | 10.82 | 714 | 3400 | 5.5 | 0.039 |
| 9 | 41938 | 10.25 | 669 | 3400 | 5.5 | 0.034 |
| 10 | 35707 | 9.71 | 623 | 3400 | 5.5 | 0.030 |
| 11 | 29475 | 9.61 | 572 | 3400 | 5.5 | 0.026 |
| 12 | 23580 | 8.59 | 511 | 3400 | 5.5 | 0.021 |
| 13 | 17685 | 7.34 | 436 | 3400 | 5.5 | 0.016 |
| 14 | 11790 | 5.76 | 331 | 3400 | 5.5 | 0.011 |
| 15 | 5130 | 3.98 | 180 | 3400 | 5.5 | 0.006 |

The P-Delta analysis is not required based on the stability coefficient which is less than 0.1. The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.51) on the critical column. However, the stability coefficient gives more
conservative result compared to the percent difference on the critical column.

### 4.2.4 Twenty-Storey model (M20)

M20 is a twenty-story reinforced concrete portal frame with fixed supports with a total height equals 68m. M20 has the same plan, material properties, section properties, story's height, live loads, and superimposed dead loads as M5 except number of stories and columns' sections. The columns' sections are shrunken to keep economical design as shown in Table 4.54. Elevation of M20 is illustrated in Fig. 4.19.

## Table 4.54 M20-Columns' Sections

|  | Column |  |
| :---: | :---: | :---: |
| Storey | Depth (m) | Width (m) |
| $1-5$ | 1.10 | 1.10 |
| $6-10$ | 1.00 | 1.00 |
| $11-15$ | 0.90 | 0.90 |
| $16-20$ | 0.75 | 0.75 |
|  |  |  |

Parameters that can be used to determine the base shear using International Building Code 2015 (IBC 2015) in co-operation with ASCE 7 can be determined in the same way as M5 model. M20 has the same values as M5 from step 1 until step 10 as summarized in Table 4.36. The Calculations for the rest values are resumed from step 11.

## STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M5 weights except for M20 model, the columns' sections are changing
throughout the elevation. M20 columns' weights in each storey are determined and tabulated in Table 4.55. The weight of each storey can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last storey is exempted from half columns' weights as tabulated in Table 4.56.


Figure 4.19 M20 Model in 3-D view
Table 4.55 M20-Columns Weight in each Storey

| Storey | Number of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1-5$ | 16 | 1.1 m | 1.1 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1645 kN |
| $6-10$ | 16 | 1.0 m | 1.0 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1360 kN |
| $11-15$ | 16 | 0.9 m | 0.9 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1102 kN |
| $16-20$ | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

Table 4.56 M20-Storeys' Weights

| Storey | Weight (kN) |
| :---: | :---: |
| 1 | 5803 |
| 2 | 5803 |
| 3 | 5803 |
| 4 | 5803 |
| 5 | 5803 |
| 6 | 5518 |
| 7 | 5518 |
| 8 | 5518 |
| 9 | 5518 |
| 10 | 5518 |
| 11 | 5260 |
| 12 | 5260 |
| 13 | 5260 |
| 14 | 5260 |
| 15 | 5260 |
| 16 | 4923 |
| 17 | 4923 |
| 18 | 4923 |
| 19 | 4923 |
| 20 | 4540 |
| Total (kN) | 107137 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure ( T ) is determined from SAP2000, version 19.0.0 software analysis. For M20 model, the fundamental period equals 2.82 second.

## Verification on SAP2000 fundamental period

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.57.

Table $4.57 \mathrm{M} 20-V e r i f i c a t i o n ~ o n ~ S A P 2000 ~ F u n d a m e n t a l ~ P e r i o d ~$

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N} \mathbf{N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k} \mathbf{N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})^{\mathbf{b}}$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \mathbf{\delta i}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 5803 | 1000 | 0.012 | 0.0001 | 12.2 | 0.9 |
| 2 | 5803 | 1000 | 0.040 | 0.0016 | 39.6 | 9.1 |
| 3 | 5803 | 1000 | 0.074 | 0.0055 | 74.2 | 32.0 |
| 4 | 5803 | 1000 | 0.112 | 0.0125 | 111.8 | 72.5 |
| 5 | 5803 | 1000 | 0.150 | 0.0225 | 149.9 | 130.4 |
| 6 | 5518 | 1000 | 0.188 | 0.0353 | 188.0 | 194.9 |
| 7 | 5518 | 1000 | 0.224 | 0.0504 | 224.5 | 278.0 |
| 8 | 5518 | 1000 | 0.259 | 0.0671 | 259.0 | 370.2 |
| 9 | 5518 | 1000 | 0.291 | 0.0850 | 291.5 | 468.9 |
| 10 | 5518 | 1000 | 0.322 | 0.1036 | 321.9 | 571.8 |
| 11 | 5260 | 1000 | 0.351 | 0.1229 | 350.6 | 646.7 |
| 12 | 5260 | 1000 | 0.377 | 0.1419 | 376.7 | 746.5 |
| 13 | 5260 | 1000 | 0.400 | 0.1602 | 400.3 | 842.8 |
| 14 | 5260 | 1000 | 0.421 | 0.1776 | 421.4 | 934.2 |
| 15 | 5260 | 1000 | 0.440 | 0.1939 | 440.3 | 1019.8 |
| 16 | 4923 | 1000 | 0.458 | 0.2095 | 457.7 | 1031.1 |
| 17 | 4923 | 1000 | 0.472 | 0.2228 | 472.0 | 1096.6 |
| 18 | 4923 | 1000 | 0.483 | 0.2337 | 483.4 | 1150.6 |
| 19 | 4923 | 1000 | 0.492 | 0.2423 | 492.2 | 1192.7 |
| 20 | 4540 | 1000 | 0.499 | 0.2486 | 498.6 | 1128.7 |
|  |  |  |  | $\sum$ | 6066 | 11918 |

a Assumed forces
b The displacement corresponding to the assumed forces

By applying Eq. 4.5 with Table 4.57
$\mathrm{T}=2.81$ second
$\%$ error $=\frac{\text { Approximated Period }- \text { SAP Period }}{\text { SAP Period }} \times 100 \%=0.35 \%<10 \%$
Acceptable

## STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=20 \times 3.40=68 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 68^{0.9}=2.07$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 2.07=3.0$ second $>2.82$ second $\quad O K$

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for M20 model are created in the same way as creating the response spectrums for previous models as shown in Figs. 4.20 and 4.21 respectively.


Figure 4.20 M20-Elastic Response Spectrum


Figure 4.21 M20-Inelastic Response Spectrum

STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:
$C_{s}$ is the minimum of $\left\{\begin{array}{l}\frac{S_{D S} \times I}{R} \\ \frac{S_{D 1} \times I}{R \times T}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 2.82}\end{array}=\left\{\begin{array}{l}0.063 \\ 0.011\end{array}=0.011\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.011 \times 107137=1178 \mathrm{kN}$

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum was defined as for M5 model and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=1030 \mathrm{kN}$
Table 4.58 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

Table 4.58 M20-Base Shear Selection

| Base shear using MRSA <br> $(\mathbf{1}$ <br> $(\mathbf{k N})$ | Base shear using ELF <br> ( <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 1030 | 1178 | 1001 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base-shear using equivalent lateral force procedure. Thus, the base shear equals 1030 kN .

STEP 17 (Vertical Distribution of Seismic Base Shear)
Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.59.

Table 4.59 M20-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})^{*}$ | Force on each <br> Floor (kN) |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 1030 | 10 |
| 2 | 6.80 | 1020 | 27 |
| 3 | 10.2 | 993 | 41 |
| 4 | 13.6 | 952 | 45 |
| 5 | 17.0 | 907 | 38 |
| 6 | 20.4 | 869 | 31 |
| 7 | 23.8 | 838 | 28 |
| 8 | 27.2 | 810 | 31 |
| 9 | 30.6 | 779 | 36 |
| 10 | 34.0 | 743 | 38 |
| 11 | 37.4 | 705 | 38 |
| 12 | 40.8 | 667 | 39 |
| 13 | 44.2 | 628 | 43 |
| 14 | 47.6 | 585 | 45 |
| 15 | 51.0 | 540 | 43 |
| 16 | 54.4 | 497 | 46 |
| 17 | 57.8 | 451 | 66 |
| 18 | 61.2 | 385 | 98 |
| 19 | 64.6 | 287 | 135 |
| 20 | 68.0 | 152 | 152 |
|  |  |  |  |

*Verification is in appendix A

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B and verified in appendix C. The critical column is tabulated in Table 4.60.

Table 4.60 Bending Moment Results for M20 Model on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 165.9 | 178.1 | $7.35 \%$ |

Storey drifts and stability coefficients for model M20 are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.61 and 4.62 respectively.

Table 4.61 M20 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.66 | 3.61 | 3.61 | 52 | OK |
| 2 | 2.16 | 11.87 | 8.27 | 52 | OK |
| 3 | 4.09 | 22.48 | 10.61 | 52 | OK |
| 4 | 6.21 | 34.16 | 11.67 | 52 | OK |
| 5 | 8.40 | 46.22 | 12.06 | 52 | OK |
| 6 | 10.64 | 58.52 | 12.30 | 52 | OK |
| 7 | 12.85 | 70.69 | 12.17 | 52 | OK |
| 8 | 15.02 | 82.60 | 11.92 | 52 | OK |
| 9 | 17.13 | 94.21 | 11.61 | 52 | OK |
| 10 | 19.18 | 105.48 | 11.26 | 52 | OK |
| 11 | 21.19 | 116.56 | 11.09 | 52 | OK |
| 12 | 23.12 | 127.15 | 10.58 | 52 | OK |
| 13 | 24.95 | 137.21 | 10.07 | 52 | OK |
| 14 | 26.68 | 146.74 | 9.53 | 52 | OK |
| 15 | 28.32 | 155.76 | 9.02 | 52 | OK |
| 16 | 29.95 | 164.70 | 8.94 | 52 | OK |
| 17 | 31.42 | 172.79 | 8.09 | 52 | OK |
| 18 | 32.69 | 179.78 | 6.99 | 52 | OK |
| 19 | 33.70 | 185.35 | 5.56 | 52 | OK |
| 20 | 34.42 | 189.33 | 3.99 | 52 | OK |

Table 4.62 M20 Stability Coefficient

| $\mathbf{M 2 0}$ |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 126961 | 3.61 | 1030 | 3400 | 5.5 | 0.024 |
| 2 | 120185 | 8.27 | 1020 | 3400 | 5.5 | 0.052 |
| 3 | 113410 | 10.61 | 993 | 3400 | 5.5 | 0.065 |
| 4 | 106634 | 11.67 | 952 | 3400 | 5.5 | 0.070 |
| 5 | 99858 | 12.06 | 907 | 3400 | 5.5 | $\mathbf{0 . 0 7 1}$ |
| 6 | 93083 | 12.3 | 869 | 3400 | 5.5 | 0.070 |
| 7 | 86593 | 12.17 | 838 | 3400 | 5.5 | 0.067 |
| 8 | 80103 | 11.92 | 810 | 3400 | 5.5 | 0.063 |
| 9 | 73613 | 11.61 | 779 | 3400 | 5.5 | 0.059 |
| 10 | 67123 | 11.26 | 743 | 3400 | 5.5 | 0.054 |
| 11 | 60633 | 11.09 | 705 | 3400 | 5.5 | 0.051 |
| 12 | 54401 | 10.58 | 667 | 3400 | 5.5 | 0.046 |
| 13 | 48170 | 10.07 | 628 | 3400 | 5.5 | 0.041 |
| 14 | 41938 | 9.53 | 585 | 3400 | 5.5 | 0.037 |
| 15 | 35707 | 9.02 | 540 | 3400 | 5.5 | 0.032 |
| 16 | 29475 | 8.94 | 497 | 3400 | 5.5 | 0.028 |
| 17 | 23580 | 8.09 | 451 | 3400 | 5.5 | 0.023 |
| 18 | 17685 | 6.99 | 385 | 3400 | 5.5 | 0.017 |
| 19 | 11790 | 5.56 | 287 | 3400 | 5.5 | 0.012 |
| 20 | 5130 | 3.99 | 152 | 3400 | 5.5 | 0.007 |

The P-Delta analysis is not required based on the stability coefficient which is less than 0.1 . The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.60) on the critical column. However, the stability coefficient starts to give unconservative result compared to the percent difference on the critical column.

### 4.2.5 Twenty Five-Storey model (M25)

M25 is a twenty five-storey reinforced concrete portal frame with fixed supports with a total height equals 85 m . M25 has the same plan, material properties, section properties, story's height, live loads, and superimposed dead loads as M5 except number of stories and columns' sections. The
columns' sections are shrunken to keep economical design as shown in Table 4.63. Elevation of M25 is illustrated in Fig. 4.22.

## Table 4.63 M25-Columns' Sections

|  | Column |  |
| :---: | :---: | :---: |
| Storey | Depth (m) | Width (m) |
| $1-5$ | 1.20 | 1.20 |
| $6-10$ | 1.10 | 1.10 |
| $11-15$ | 1.00 | 1.00 |
| $16-20$ | 0.90 | 0.90 |
| $21-25$ | 0.75 | 0.75 |
|  |  |  |

Parameters that can be used to determine the base shear using International Building Code 2015 (IBC 2015) in co-operation with ASCE7-10 can be determined in the same way as M5 model. M25 has the same values as M5 from step 1 until step 10 as summarized in Table 4.36. The Calculations for the remaining values are resumed from step 11.

## STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M5 weights except for M25 model, the columns' sections are changing throughout the elevation. M25 columns' weights in each storey are determined and tabulated in Table 4.64. The weight of each story can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last storey is exempted from half columns' weights as tabulated in Table 4.65.


Figure 4.22 M25 Model in 3-D view
Table 4.64 M25-Columns Weight in each Storey

| Storey | Number of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1-5$ | 16 | 1.2 m | 1.2 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1958 kN |
| $6-10$ | 16 | 1.1 m | 1.1 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1645 kN |
| $11-15$ | 16 | 1.0 m | 1.0 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1360 kN |
| $16-20$ | 16 | 0.9 m | 0.9 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1102 kN |
| $21-25$ | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

## Table 4.65 M25-Storeys' Weights

| Storey | Weight (kN) |
| :---: | :---: |
| 1 | 6116 |
| 2 | 6116 |
| 3 | 6116 |
| 4 | 6116 |
| 5 | 6116 |
| 6 | 5803 |
| 7 | 5803 |
| 8 | 5803 |
| 9 | 5803 |
| 10 | 5803 |
| 11 | 5518 |
| 12 | 5518 |
| 13 | 5518 |
| 14 | 5518 |
| 15 | 5518 |
| 16 | 5260 |
| 17 | 5260 |
| 18 | 5260 |
| 19 | 5260 |
| 20 | 5260 |
| 21 | 4923 |
| 22 | 4923 |
| 23 | 4923 |
| 24 | 4993 |
| 25 | 137717 |
| Total (kN) |  |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure ( T ) is determined from SAP2000, version 19.0.0 software analysis. For M25 model, the fundamental period equals 3.57 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.66.

Table 4.66 M25-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N} \mathbf{N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k} \mathbf{N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})^{\mathbf{b}}$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 6116 | 1000 | 0.013 | 0.0002 | 13.1 | 1.1 |
| 2 | 6116 | 1000 | 0.044 | 0.0019 | 44.0 | 11.8 |
| 3 | 6116 | 1000 | 0.085 | 0.0071 | 84.5 | 43.7 |
| 4 | 6116 | 1000 | 0.130 | 0.0169 | 129.9 | 103.2 |
| 5 | 6116 | 1000 | 0.177 | 0.0315 | 177.4 | 192.5 |
| 6 | 5803 | 1000 | 0.226 | 0.0510 | 225.9 | 296.1 |
| 7 | 5803 | 1000 | 0.274 | 0.0749 | 273.7 | 434.8 |
| 8 | 5803 | 1000 | 0.320 | 0.1026 | 320.3 | 595.3 |
| 9 | 5803 | 1000 | 0.365 | 0.1334 | 365.2 | 773.9 |
| 10 | 5803 | 1000 | 0.408 | 0.1667 | 408.3 | 967.5 |
| 11 | 5518 | 1000 | 0.450 | 0.2024 | 449.9 | 1117.1 |
| 12 | 5518 | 1000 | 0.489 | 0.2393 | 489.2 | 1320.4 |
| 13 | 5518 | 1000 | 0.526 | 0.2768 | 526.1 | 1527.2 |
| 14 | 5518 | 1000 | 0.561 | 0.3144 | 560.7 | 1734.8 |
| 15 | 5518 | 1000 | 0.593 | 0.3519 | 593.2 | 1941.5 |
| 16 | 5260 | 1000 | 0.624 | 0.3892 | 623.9 | 2047.4 |
| 17 | 5260 | 1000 | 0.652 | 0.4251 | 652.0 | 2235.8 |
| 18 | 5260 | 1000 | 0.677 | 0.4590 | 677.5 | 2414.4 |
| 19 | 5260 | 1000 | 0.701 | 0.4908 | 700.6 | 2581.7 |
| 20 | 5260 | 1000 | 0.721 | 0.5205 | 721.4 | 2737.8 |
| 21 | 4923 | 1000 | 0.741 | 0.5487 | 740.7 | 2701.2 |
| 22 | 4923 | 1000 | 0.757 | 0.5730 | 757.0 | 2821.1 |
| 23 | 4923 | 1000 | 0.770 | 0.5935 | 770.4 | 2922.0 |
| 24 | 4923 | 1000 | 0.781 | 0.6102 | 781.1 | 3003.8 |
| 25 | 4540 | 1000 | 0.790 | 0.6233 | 789.5 | 2829.9 |
|  | 137717 |  |  | $\sum$ | 11876 | 37356 |

a Assumed forces
b The displacement corresponding to the assumed forces

By applying Eq. 4.5 with Table 4.58
$\mathrm{T}=3.56$ second
$\%$ error $=\frac{\text { Approximated Period }- \text { SAP Period }}{\text { SAP Period }} \times 100 \%=0.28 \%<10 \%$
Acceptable

STEP 13 (Maximum Period Limit)
After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=25 \times 3.40=85 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 85^{0.9}=2.54 \mathrm{~seconds}$
After referring to Table 4.21
$\mathrm{S}_{\mathrm{D} 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 2.54=3.68$ second $>3.57$ second $\quad O K$
STEP 14 (Response Spectrum Graph)
The elastic and inelastic response spectrums for M25 model are created in the same way as creating the response spectrums for previous models as shown in Figs. 4.23 and 4.24 respectively.


Figure 4.23 M25-Elastic Response Spectrum


Figure 4.24 M25-Inelastic Response Spectrum
STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear ( V ) in a given direction is determined by applying Eq. 4.9 as follows:
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS} \times \mathrm{I}}}{R} \\ \frac{\mathrm{~S}_{\mathrm{D} 1 \times \mathrm{I}}}{\mathrm{R} \times \mathrm{T}}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 3.57}\end{array}=\left\{\begin{array}{c}0.063 \\ 0.0087\end{array}=0.0087\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0087 \times 137717=1198 \mathrm{kN}$

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum function was defined as for M5 model and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=1044 \mathrm{kN}$

Table 4.67 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

## Table 4.67 M25-Base Shear Selection

| Base shear using MRSA $^{\mathbf{1}}$ <br> $(\mathbf{k N})$ | Base shear using ELF <br> 2 <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 1044 | 1198 | 1018 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base-shear using equivalent lateral force procedure. Thus, the base shear equals 1044 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.68.

Table 4.68 M25-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})^{*}$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 1044 | 6 |
| 2 | 6.80 | 1038 | 19 |
| 3 | 10.2 | 1019 | 32 |
| 4 | 13.6 | 987 | 39 |
| 5 | 17.0 | 948 | 37 |
| 6 | 20.4 | 911 | 31 |
| 7 | 23.8 | 880 | 23 |
| 8 | 27.2 | 857 | 22 |
| 9 | 30.6 | 835 | 23 |
| 10 | 34.0 | 812 | 28 |
| 11 | 37.4 | 784 | 31 |
| 12 | 40.8 | 753 | 31 |
| 13 | 44.2 | 722 | 29 |
| 14 | 47.6 | 693 | 28 |
| 15 | 51.0 | 665 | 30 |
| 16 | 54.4 | 635 | 34 |
| 17 | 57.8 | 601 | 38 |
| 18 | 61.2 | 563 | 39 |
| 19 | 64.6 | 524 | 36 |
| 20 | 68.0 | 488 | 34 |
| 21 | 71.4 | 454 | 44 |
| 22 | 74.8 | 410 | 65 |
| 23 | 78.2 | 345 | 94 |
| 24 | 81.6 | 251 | 121 |
| 25 | 85.0 | 130 | 130 |

*Verification is in appendix A
First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B and verified in appendix C. The critical column is tabulated in Table 4.69.

Table 4.69 Bending Moment Results for M25 Model on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 195.8 | 215.0 | $9.82 \%$ |

Storey drifts and stability coefficients for model M25 are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.70 and 4.71 respectively.

Table 4.70 M25 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.57 | 3.15 | 3.15 | 52 | OK |
| 2 | 1.94 | 10.66 | 7.50 | 52 | OK |
| 3 | 3.75 | 20.63 | 9.97 | 52 | OK |
| 4 | 5.80 | 31.93 | 11.30 | 52 | OK |
| 5 | 7.98 | 43.88 | 11.95 | 52 | OK |
| 6 | 10.23 | 56.24 | 12.36 | 52 | OK |
| 7 | 12.49 | 68.68 | 12.44 | 52 | OK |
| 8 | 14.74 | 81.06 | 12.38 | 52 | OK |
| 9 | 16.96 | 93.31 | 12.24 | 52 | OK |
| 10 | 19.16 | 105.38 | 12.07 | 52 | OK |
| 11 | 21.34 | 117.36 | 11.99 | 52 | OK |
| 12 | 23.46 | 129.04 | 11.68 | 52 | OK |
| 13 | 25.52 | 140.39 | 11.35 | 52 | OK |
| 14 | 27.53 | 151.40 | 11.02 | 52 | OK |
| 15 | 29.47 | 162.11 | 10.71 | 52 | OK |
| 16 | 31.40 | 172.67 | 10.56 | 52 | OK |
| 17 | 33.23 | 182.79 | 10.11 | 52 | OK |
| 18 | 34.99 | 192.43 | 9.64 | 52 | OK |
| 19 | 36.65 | 2201.59 | 9.17 | 52 | OK |
| 20 | 38.24 | 210.33 | 8.73 | 52 | OK |
| 21 | 39.82 | 219.02 | 8.69 | 52 | OK |
| 22 | 41.26 | 226.93 | 7.91 | 52 | OK |
| 23 | 42.51 | 233.79 | 6.86 | 52 | OK |
| 24 | 43.51 | 239.31 | 5.52 | 52 | OK |
| 25 | 44.26 | 243.42 | 4.10 | 52 | OK |

Table 4.71 M25 Stability Coefficient

| $\mathbf{M 2 5}$ |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 162403 | 3.15 | 1044 | 3400 | 5.5 | 0.026 |
| 2 | 155315 | 7.5 | 1038 | 3400 | 5.5 | 0.060 |
| 3 | 148226 | 9.97 | 1019 | 3400 | 5.5 | 0.078 |
| 4 | 141138 | 11.3 | 987 | 3400 | 5.5 | 0.086 |
| 5 | 134049 | 11.95 | 948 | 3400 | 5.5 | 0.090 |
| 6 | 126961 | 12.36 | 911 | 3400 | 5.5 | $\mathbf{0 . 0 9 2}$ |
| 7 | 120185 | 12.44 | 880 | 3400 | 5.5 | 0.091 |
| 8 | 113410 | 12.38 | 857 | 3400 | 5.5 | 0.088 |
| 9 | 106634 | 12.24 | 835 | 3400 | 5.5 | 0.084 |
| 10 | 99858 | 12.07 | 812 | 3400 | 5.5 | 0.079 |
| 11 | 93083 | 11.99 | 784 | 3400 | 5.5 | 0.076 |
| 12 | 86593 | 11.68 | 753 | 3400 | 5.5 | 0.072 |
| 13 | 80103 | 11.35 | 722 | 3400 | 5.5 | 0.067 |
| 14 | 73613 | 11.02 | 693 | 3400 | 5.5 | 0.063 |
| 15 | 67123 | 10.71 | 665 | 3400 | 5.5 | 0.058 |
| 16 | 60633 | 10.56 | 635 | 3400 | 5.5 | 0.054 |
| 17 | 54401 | 10.11 | 601 | 3400 | 5.5 | 0.049 |
| 18 | 48170 | 9.64 | 563 | 3400 | 5.5 | 0.044 |
| 19 | 41938 | 9.17 | 524 | 3400 | 5.5 | 0.039 |
| 20 | 35707 | 8.73 | 488 | 3400 | 5.5 | 0.034 |
| 21 | 29475 | 8.69 | 454 | 3400 | 5.5 | 0.030 |
| 22 | 23580 | 7.91 | 410 | 3400 | 5.5 | 0.024 |
| 23 | 17685 | 6.86 | 345 | 3400 | 5.5 | 0.019 |
| 24 | 11790 | 5.52 | 251 | 3400 | 5.5 | 0.014 |
| 25 | 5130 | 4.1 | 130 | 3400 | 5.5 | 0.009 |

The P-Delta analysis is not required based on the stability coefficient which is less than 0.1 . The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.69) on the critical column. However, the stability coefficient gives slightly unconservative result compared to the percent difference on the critical column.

### 4.2.6 Thirty-Storey model (M30)

M30 is a thirty-story reinforced concrete portal frame with fixed supports with a total height equals 102 m . M30 has the same plan, material
properties, section properties, story's height, live loads, and superimposed dead loads as M5 except number of stories and columns' sections. The columns' sections are shrunken to keep economical design as shown in Table 4.72. Elevation of M30 is illustrated in Fig. 4.25.

Table 4.72 M30-Columns' Sections

|  | Column |  |
| :---: | :---: | :---: |
| Storey | Depth (m) | Width (m) |
| $1-5$ | 1.3 | 1.3 |
| $6-10$ | 1.2 | 1.2 |
| $11-15$ | 1.1 | 1.1 |
| $16-20$ | 1.0 | 1.0 |
| $21-25$ | 0.9 | 0.9 |
| $26-30$ | 0.75 | 0.75 |
|  |  |  |



Figure 4.25 M30 Model in 3-D view

Parameters that can be used to determine the base shear using International Building Code 2015 (IBC 2015) in co-operation with ASCE7-10 can be determined in the same way as M5 model. M30 has the same values as M5 from step 1 until step 10 as summarized in Table 4.36. The Calculations for the rest values are resumed from step 11.

## STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M5 weights except for M30 model, the columns' sections are changing throughout the elevation. M30 columns' weights in each storey are determined and tabulated in Table 4.73. The weight of each storey can be determined as the sum of beams' weights, columns' weights, slab's weight, and superimposed dead loads. The last story is exempted from half columns' weights as tabulated in Table 4.74.

## Table 4.73 M30-Columns Weight in each Storey

| Storey | Number of <br> Columns | Depth | Width | Length | Concrete <br> Weight Per <br> Unit Volume | Columns <br> Weight |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1-5$ | 16 | 1.3 m | 1.3 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 2298 kN |
| $6-10$ | 16 | 1.2 m | 1.2 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1958 kN |
| $11-15$ | 16 | 1.1 m | 1.1 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1645 kN |
| $16-20$ | 16 | 1.0 m | 1.0 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1360 kN |
| $21-25$ | 16 | 0.9 m | 0.9 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 1102 kN |
| $26-30$ | 16 | 0.75 m | 0.75 m | 3.40 m | $25 \mathrm{kN} / \mathrm{m}^{3}$ | 765 kN |

## Table 4.74 M30-Storeys' Weights

| Storey | Weight (kN) |
| :---: | :---: |
| 1 | 6456 |
| 2 | 6456 |
| 3 | 6456 |
| 4 | 6456 |
| 5 | 6456 |
| 6 | 6116 |
| 7 | 6116 |
| 8 | 6116 |
| 9 | 6116 |
| 10 | 6116 |
| 11 | 5803 |
| 12 | 5803 |
| 13 | 5803 |
| 14 | 5803 |
| 15 | 5803 |
| 16 | 5518 |
| 17 | 5518 |
| 18 | 5518 |
| 19 | 5518 |
| 20 | 5518 |
| 21 | 5260 |
| 22 | 5260 |
| 23 | 5260 |
| 24 | 5260 |
| 25 | 5260 |
| 26 | 4923 |
| 27 | 4923 |
| 28 | 4923 |
| 29 | 4923 |
| 30 | 4540 |
| Total (kN) | 169997 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure (T) is determined from SAP2000, version 19.0.0 software analysis. For M30 model, the fundamental period equals 4.33 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.75.

Table 4.75 M30-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N})$ | $\mathrm{f}_{\mathrm{i}}(\mathrm{kN})^{\text {a }}$ | $\delta_{i}(\mathrm{~m})^{\text {b }}$ | $\delta_{i}{ }^{2}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\delta} \mathbf{i}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 6456 | 1000 | 0.014 | 0.0002 | 13.7 | 1.2 |
| 2 | 6456 | 1000 | 0.047 | 0.0022 | 47.0 | 14.3 |
| 3 | 6456 | 1000 | 0.092 | 0.0085 | 92.1 | 54.8 |
| 4 | 6456 | 1000 | 0.144 | 0.0207 | 144.0 | 133.9 |
| 5 | 6456 | 1000 | 0.200 | 0.0398 | 199.6 | 257.1 |
| 6 | 6116 | 1000 | 0.257 | 0.0662 | 257.4 | 405.1 |
| 7 | 6116 | 1000 | 0.316 | 0.0997 | 315.7 | 609.5 |
| 8 | 6116 | 1000 | 0.374 | 0.1395 | 373.5 | 853.2 |
| 9 | 6116 | 1000 | 0.430 | 0.1851 | 430.3 | 1132.2 |
| 10 | 6116 | 1000 | 0.486 | 0.2359 | 485.7 | 1442.6 |
| 11 | 5803 | 1000 | 0.540 | 0.2914 | 539.8 | 1691.0 |
| 12 | 5803 | 1000 | 0.592 | 0.3503 | 591.9 | 2032.9 |
| 13 | 5803 | 1000 | 0.642 | 0.4120 | 641.8 | 2390.6 |
| 14 | 5803 | 1000 | 0.690 | 0.4757 | 689.7 | 2760.5 |
| 15 | 5803 | 1000 | 0.736 | 0.5411 | 735.6 | 3140.0 |
| 16 | 5518 | 1000 | 0.780 | 0.6081 | 779.8 | 3355.8 |
| 17 | 5518 | 1000 | 0.822 | 0.6750 | 821.6 | 3724.9 |
| 18 | 5518 | 1000 | 0.861 | 0.7413 | 861.0 | 4090.7 |
| 19 | 5518 | 1000 | 0.898 | 0.8066 | 898.1 | 4450.7 |
| 20 | 5518 | 1000 | 0.933 | 0.8705 | 933.0 | 4803.6 |
| 21 | 5260 | 1000 | 0.966 | 0.9335 | 966.2 | 4910.4 |
| 22 | 5260 | 1000 | 0.997 | 0.9934 | 996.7 | 5225.5 |
| 23 | 5260 | 1000 | 1.025 | 1.0500 | 1024.7 | 5522.9 |
| 24 | 5260 | 1000 | 1.050 | 1.1030 | 1050.2 | 5801.5 |
| 25 | 5260 | 1000 | 1.074 | 1.1524 | 1073.5 | 6061.7 |
| 26 | 4923 | 1000 | 1.095 | 1.1995 | 1095.2 | 5905.3 |
| 27 | 4923 | 1000 | 1.114 | 1.2408 | 1113.9 | 6108.7 |
| 28 | 4923 | 1000 | 1.130 | 1.2764 | 1129.8 | 6283.7 |
| 29 | 4923 | 1000 | 1.143 | 1.3063 | 1142.9 | 6430.8 |
| 30 | 4540 | 1000 | 1.154 | 1.3311 | 1153.7 | 6043.2 |
|  |  |  |  | $\sum$ | 20598 | 95638 |

a Assumed forces
b The displacement corresponding to the assumed forces

By applying Eq. 4.5 with Table 4.75
$\mathrm{T}=4.32$ second
$\%$ error $=\frac{\text { Approximated Period-SAP Period }}{\text { SAP Period }} \times 100 \%=0.23 \%<10 \%$ OK
STEP 13 (Maximum Period Limit)
After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=30 \times 3.40=102 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 102^{0.9}=3.0$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 3.0=4.35$ second $>4.33$ second $\quad O K$

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for M30 model are created in the same way as creating the response spectrums for previous models as shown in Figs. 4.26 and 4.27 respectively.

STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:

$$
\mathrm{C}_{\mathrm{s}}=0.0067
$$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0067 \times 169997=1140 \mathrm{kN}$


Figure 4.26 M30-Elastic Response Spectrum


Figure 4.27 M30-Inelastic Response Spectrum

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum function was defined as for M5 model and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=1005 \mathrm{kN}$

Table 4.76 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

Table 4.76 M30-Base Shear Selection

| Base shear using MRSA <br> $(\mathbf{1}$ <br> $(\mathbf{k N})$ | Base shear using ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 1005 | 1140 | 969 |

[^0]2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is larger than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, the base shear equals 1005 kN .

STEP 17 (Vertical Distribution of Seismic Base Shear)
Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.77.

Table 4.77 M30-Base Shear Distribution on Floors

| * | Floor | h(m) | Shear Force $(\mathbf{k N})^{*}$ | Force on each Floor (kN) |
| :---: | :---: | :---: | :---: | :---: |
|  | 1 | 3.40 | 1005 | 5 |
|  | 2 | 6.80 | 1000 | 14 |
|  | 3 | 10.2 | 986 | 25 |
|  | 4 | 13.6 | 961 | 32 |
|  | 5 | 17.0 | 929 | 35 |
|  | 6 | 20.4 | 894 | 32 |
|  | 7 | 23.8 | 862 | 26 |
|  | 8 | 27.2 | 836 | 20 |
|  | 9 | 30.6 | 816 | 17 |
|  | 10 | 34.0 | 799 | 16 |
|  | 11 | 37.4 | 783 | 21 |
|  | 12 | 40.8 | 762 | 24 |
|  | 13 | 44.2 | 738 | 26 |
|  | 14 | 47.6 | 712 | 25 |
|  | 15 | 51.0 | 687 | 21 |
|  | 16 | 54.4 | 666 | 17 |
|  | 17 | 57.8 | 649 | 16 |
|  | 18 | 61.2 | 633 | 20 |
|  | 19 | 64.6 | 613 | 25 |
|  | 20 | 68.0 | 588 | 30 |
|  | 21 | 71.4 | 558 | 32 |
|  | 22 | 74.8 | 526 | 31 |
|  | 23 | 78.2 | 495 | 27 |
|  | 24 | 81.6 | 468 | 26 |
|  | 25 | 85.0 | 442 | 30 |
|  | 26 | 88.4 | 412 | 44 |
|  | 27 | 91.8 | 368 | 64 |
|  | 28 | 95.2 | 304 | 87 |
|  | 29 | 98.6 | 217 | 106 |
|  | 30 | 102 | 111 | 111 |
|  |  |  |  |  |

Verification is in appendix A

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B and verified in appendix C. The critical column is tabulated in Table 4.78.

Table 4.78 Bending Moment Results for M30 Model on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 183.5 | 206.3 | $12.40 \%$ |

Storey drifts and stability coefficients for model M30 are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.79 and 4.80 respectively.

Table 4.79 M30 Drift Check

| Level | $\delta_{\text {xe }}(\mathrm{mm})$ | $\delta_{\mathbf{x}}(\mathrm{mm})$ | $\Delta$ (mm) | $\Delta_{\text {Allowable }}(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.48 | 2.63 | 2.63 | 52 | OK |
| 2 | 1.65 | 9.09 | 6.45 | 52 | OK |
| 3 | 3.26 | 17.91 | 8.82 | 52 | OK |
| 4 | 5.12 | 28.14 | 10.23 | 52 | OK |
| 5 | 7.12 | 39.16 | 11.03 | 52 | OK |
| 6 | 9.22 | 50.72 | 11.56 | 52 | OK |
| 7 | 11.37 | 62.52 | 11.79 | 52 | OK |
| 8 | 13.52 | 74.37 | 11.85 | 52 | OK |
| 9 | 15.67 | 86.20 | 11.83 | 52 | OK |
| 10 | 17.81 | 97.96 | 11.76 | 52 | OK |
| 11 | 19.95 | 109.71 | 11.75 | 52 | OK |
| 12 | 22.05 | 121.30 | 11.59 | 52 | OK |
| 13 | 24.12 | 132.69 | 11.39 | 52 | OK |
| 14 | 26.16 | 143.86 | 11.18 | 52 | OK |
| 15 | 28.15 | 154.84 | 10.98 | 52 | OK |
| 16 | 30.14 | 165.75 | 10.91 | 52 | OK |
| 17 | 32.08 | 176.45 | 10.70 | 52 | OK |
| 18 | 33.99 | 186.94 | 10.49 | 52 | OK |
| 19 | 35.85 | 197.19 | 10.25 | 52 | OK |
| 20 | 37.67 | 207.19 | 10.00 | 52 | OK |
| 21 | 39.47 | 217.06 | 9.87 | 52 | OK |
| 22 | 41.19 | 226.54 | 9.48 | 52 | OK |
| 23 | 42.84 | 235.64 | 9.10 | 52 | OK |
| 24 | 44.43 | 244.37 | 8.73 | 52 | OK |
| 25 | 45.96 | 252.77 | 8.40 | 52 | OK |
| 26 | 47.48 | 261.15 | 8.38 | 52 | OK |
| 27 | 48.87 | 268.78 | 7.64 | 52 | OK |
| 28 | 50.08 | 275.43 | 6.65 | 52 | OK |
| 29 | 51.06 | 280.85 | 5.42 | 52 | OK |
| 30 | 51.82 | 285.00 | 4.16 | 52 | OK |

Table 4.80 M30 Stability Coefficient

| $\mathbf{M 3 0}$ |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta \mathbf{( m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k} \mathbf{N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 199545 | 2.63 | 1005 | 3400 | 5.5 | 0.028 |
| 2 | 192117 | 6.45 | 1000 | 3400 | 5.5 | 0.066 |
| 3 | 184688 | 8.82 | 986 | 3400 | 5.5 | 0.088 |
| 4 | 177260 | 10.23 | 961 | 3400 | 5.5 | 0.101 |
| 5 | 169831 | 11.03 | 929 | 3400 | 5.5 | 0.108 |
| 6 | 162403 | 11.56 | 894 | 3400 | 5.5 | 0.112 |
| 7 | 155315 | 11.79 | 862 | 3400 | 5.5 | $\mathbf{0 . 1 1 4}$ |
| 8 | 148226 | 11.85 | 836 | 3400 | 5.5 | 0.112 |
| 9 | 141138 | 11.83 | 816 | 3400 | 5.5 | 0.109 |
| 10 | 134049 | 11.76 | 799 | 3400 | 5.5 | 0.106 |
| 11 | 126961 | 11.75 | 783 | 3400 | 5.5 | 0.102 |
| 12 | 120185 | 11.59 | 762 | 3400 | 5.5 | 0.098 |
| 13 | 113410 | 11.39 | 738 | 3400 | 5.5 | 0.094 |
| 14 | 106634 | 11.18 | 712 | 3400 | 5.5 | 0.090 |
| 15 | 99858 | 10.98 | 687 | 3400 | 5.5 | 0.085 |
| 16 | 93083 | 10.91 | 666 | 3400 | 5.5 | 0.082 |
| 17 | 86593 | 10.7 | 649 | 3400 | 5.5 | 0.076 |
| 18 | 80103 | 10.49 | 633 | 3400 | 5.5 | 0.071 |
| 19 | 73613 | 10.25 | 613 | 3400 | 5.5 | 0.066 |
| 20 | 67123 | 10 | 588 | 3400 | 5.5 | 0.061 |
| 21 | 60633 | 9.87 | 558 | 3400 | 5.5 | 0.057 |
| 22 | 54401 | 9.48 | 526 | 3400 | 5.5 | 0.052 |
| 23 | 48170 | 9.1 | 495 | 3400 | 5.5 | 0.047 |
| 24 | 41938 | 8.73 | 468 | 3400 | 5.5 | 0.042 |
| 25 | 35707 | 8.4 | 442 | 3400 | 5.5 | 0.036 |
| 26 | 29475 | 8.38 | 412 | 3400 | 5.5 | 0.032 |
| 27 | 23580 | 7.64 | 368 | 3400 | 5.5 | 0.026 |
| 28 | 17685 | 6.65 | 304 | 3400 | 5.5 | 0.021 |
| 29 | 11790 | 5.42 | 217 | 3400 | 5.5 | 0.016 |
| 30 | 5130 | 4.16 | 111 | 3400 | 5.5 | 0.010 |

The P-Delta analysis is required based on the stability coefficient which is larger than 0.1 which is assured by the percent difference between firstorder and second order analyses on the critical column also is larger than 0.1 .The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.78) on the critical column. However, the stability coefficient gives unconservative result compared to the percent difference on the critical column.

### 4.3 Multi-Storey Effect on P-Delta Analysis with Fundamental Period does not Abide by the Proposed Equation

In previous section, P -Delta effect on six models with different number of storeys was investigated taking into account that theirs fundamental periods are less than the proposed equation limit (Eq. 4.6). In this section, P-Delta effect on the same first three models is investigated taking into account that theirs fundamental periods are greater than the maximum proposed limit. To achieve this condition; the previous three models are investigated with the beam's section is switched to a smaller section which is 0.4 m and 0.60 m in depth and width respectively. Furthermore, columns have gotten smaller which is addressed in every model. The models are TM5, TM10 and TM15. For instance, TM5 refers to five-storey model with fundamental period greater the maximum proposed limit. It is worth mentioning that all other parameters are kept the same as previous models except beam's section and columns' sections are changed (for example TM5 is the same as M5 except the beam's section and columns' sections have gotten smaller). The bending moment from first-order analysis and second-order analysis are determined on all columns in all floors and the critical column is adopted.

### 4.3.1 Five-Storey model (TM5)

TM5's columns are 0.6 m square section. In order to avoid repeating the same description of the model, refer to section 4.2.1 for more information about this model. It has the same results until step10; the calculations are continued from step 11 as follows:

## STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M5 weights and tabulated in Table 4.81.

Table 4.81 TM5-Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 3704 |
| 2 | 3704 |
| 3 | 3704 |
| 4 | 3704 |
| 5 | 3459 |
| Total | 18275 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period was computed using the analysis software SAP2000 which equals 1.23 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.82.

## Table 4.82 TM5-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{*}$ | $\boldsymbol{\delta}_{\mathbf{i}}(\mathbf{m})$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 3704 | 1000 | 0.019 | 0.0004 | 19.3 | 1.4 |
| 2 | 3704 | 1000 | 0.055 | 0.0030 | 54.6 | 11.0 |
| 3 | 3704 | 1000 | 0.089 | 0.0079 | 88.9 | 29.3 |
| 4 | 3704 | 1000 | 0.116 | 0.0134 | 116.0 | 49.8 |
| 5 | 3459 | 1000 | 0.135 | 0.0182 | 134.9 | 63.0 |
|  |  |  |  | $\sum$ | 414 | 155 |

[^1]By applying Eq. 4.5 with Table 4.82
$\mathrm{T}=1.23$ second
$\%$ error $=\frac{\text { Approximated Period }- \text { SAP Period }}{\text { SAP Period }} \times 100 \%=0 \%$

STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$x=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=5 \times 3.40=17 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 17^{0.9}=0.59$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 0.59=0.86$ second $<1.23$ second (Use $\mathrm{T}=0.86$ second in base shear calculations).

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for TM5 model are created in the same way as creating the response spectrums for previous models as shown in Figs. 4.28 and 4.29 respectively.

STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS}} \times \mathrm{I}}{R} \\ \frac{\mathrm{~S}_{\mathrm{D} 1 \times I}}{\mathrm{R} \times \mathrm{T}}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 0.86}\end{array}=\left\{\begin{array}{c}0.063 \\ 0.0363\end{array}=0.0363\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0363 \times 18275=663 \mathrm{kN}$


Figure 4.28 TM5-Elastic Response Spectrum


Figure 4.29 TM5-Inelastic Response Spectrum

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum function was defined as previously mentioned and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=389 \mathrm{kN}$
Table 4.83 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

Table 4.83 TM5-Base Shear Selection

| Base shear using MRSA <br> $(\mathbf{1}$ <br> $(\mathbf{k N})$ | Base shear using ELF <br> (kN) | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 389 | 663 | 564 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is less than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, the base shear has to scale to $0.85 \times$ ELF which equals 564 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.84 and scale the value to $0.85 \times$ ELF.

## Table 4.84 TM5-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 564 | 45 |
| 2 | 6.80 | 519 | 86 |
| 3 | 10.2 | 434 | 81 |
| 4 | 13.6 | 352 | 128 |
| 5 | 17.0 | 225 | 225 |
|  |  |  |  |

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B . The critical column is tabulated in Table 4.85 .

## Table 4.85 Bending Moment Results for TM5 Model n Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | -84.0 | -88.2 | $4.95 \%$ |

Storey drifts and stability coefficients for TM5 model are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.86 and 4.87 respectively.

Table 4.86 TM5 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{x}}$ <br> $(\mathbf{m m})^{\mathbf{b}}$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})^{\mathbf{c}}$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 2.40 | 13.20 | 13.20 | 52 | OK |
| 2 | 7.12 | 39.14 | 25.94 | 52 | OK |
| 3 | 12.11 | 66.62 | 27.48 | 52 | OK |
| 4 | 16.49 | 90.67 | 24.05 | 52 | OK |
| 5 | 19.85 | 109.18 | 18.51 | 52 | OK |

Table 4.87 TM5 Stability Coefficient

| TM5 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 23373 | 13.20 | 564 | 3400 | 5.5 | 0.029 |
| 2 | 18698 | 25.94 | 519 | 3400 | 5.5 | $\mathbf{0 . 0 5 0}$ |
| 3 | 14024 | 27.48 | 434 | 3400 | 5.5 | 0.047 |
| 4 | 9349 | 24.05 | 352 | 3400 | 5.5 | 0.034 |
| 5 | 4185 | 18.51 | 225 | 3400 | 5.5 | 0.018 |

The P-Delta analysis is not required based on the stability coefficient which is less than 0.1 which is assured by the percent difference between firstorder and second order analyses on the critical column which is also less than 0.1 .The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.85) on the critical column. However, the stability coefficient gives conservative result compared to the percent difference on the critical column.

### 4.3.2 Ten-Storey model (TM10)

TM10's columns' sections are square sections and are summarized in Table 4.88. TM10 has the same results as M10 until step 10; the calculations are resumed from step 11 as follows;

Table 4.88 TM10 columns' sections

| Level | Depth | Width |
| :---: | :---: | :---: |
| $1-5$ | 0.7 m | 0.7 m |
| $6-10$ | 0.6 m | 0.6 m |

STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M10 weights and tabulated in Table 4.89.

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure (T) is determined from SAP2000, version 19.0.0 software analysis. For TM10 model, the fundamental period equals 2.56 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.90.

Table 4.89 TM10-Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 3879 |
| 2 | 3879 |
| 3 | 3879 |
| 4 | 3879 |
| 5 | 3879 |
| 6 | 3704 |
| 7 | 3704 |
| 8 | 3704 |
| 9 | 3704 |
| 10 | 3459 |
| Total | 37670 |

Table 4.90 TM10-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}}(\mathbf{k N})$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}(\mathbf{m})^{\mathbf{b}}}$ | $\boldsymbol{\delta}_{\mathbf{i}}^{\mathbf{2}\left(\mathbf{1 0}^{-\mathbf{3}}\right)}$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 3879 | 1000 | 0.032 | 0.001 | 31.7 | 3.9 |
| 2 | 3879 | 1000 | 0.100 | 0.010 | 100.0 | 38.8 |
| 3 | 3879 | 1000 | 0.181 | 0.033 | 181.3 | 127.5 |
| 4 | 3879 | 1000 | 0.263 | 0.069 | 263.4 | 269.1 |
| 5 | 3879 | 1000 | 0.341 | 0.116 | 340.7 | 450.2 |
| 6 | 3704 | 1000 | 0.411 | 0.169 | 410.5 | 624.2 |
| 7 | 3704 | 1000 | 0.467 | 0.218 | 467.3 | 808.8 |
| 8 | 3704 | 1000 | 0.511 | 0.261 | 511.0 | 967.2 |
| 9 | 3704 | 1000 | 0.542 | 0.294 | 542.5 | 1090.0 |
| 10 | 3459 | 1000 | 0.564 | 0.318 | 563.9 | 1100.0 |
|  |  |  |  | $\sum$ | 3412 | 5480 |

a Assumed forces
b The displacement corresponding to the assumed forces
By applying Eq. 4.5 with Table 4.90
$\mathrm{T}=2.54$ second
$\%$ error $=\frac{\text { Approximated Period }- \text { SAP Period }}{\text { SAP Period }} \times 100 \%=0.8 \%<10 \%$
Acceptable

## STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=10 \times 3.40=34 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 34^{0.9}=1.11$ seconds
After referring to Table 4.21
$\mathrm{S}_{\mathrm{D} 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 1.11=1.61$ second $<2.56$ second (Use $\mathrm{T}=1.61$ second for base shear calculations).

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for TM10 model are created in the same way as creating the response spectrums for M10 model as shown in Figs. 4.30 and 4.31 respectively.

STEP 15 (Determination of Seismic Base Shear using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:
$\mathrm{C}_{\mathrm{s}}$ is the minimum of $\left\{\begin{array}{l}\frac{\mathrm{S}_{\mathrm{DS} \times \mathrm{I}}}{R} \\ \frac{\mathrm{~S}_{\mathrm{D} 1 \times \mathrm{I}}}{\mathrm{R} \times \mathrm{T}}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 1.61}\end{array}=\left\{\begin{array}{c}0.063 \\ 0.0194\end{array}=0.0194\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0194 \times 37670=731 \mathrm{kN}$


Figure 4.30 TM10-Elastic Response Spectrum


Figure 4.31 TM10-Inelastic Response Spectrum

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum was defined as for M5 model and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=411 \mathrm{kN}$

Table 4.91 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1

## Table 4.91 TM10-Base Shear Selection

| Base shear using MRSA <br> $(\mathbf{1}$ <br> $(\mathbf{k N})$ | Base shear using ELF $^{2}$ <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 411 | 731 | 621 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is less than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, the base shear has to scale to $0.85 \times$ ELF which equals 621 kN .

STEP 17 (Vertical Distribution of Seismic Base Shear)
Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.92 and scale the value to $0.85 \times$ ELF.

Table 4.92 TM10-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 621 | 26 |
| 2 | 6.80 | 595 | 53 |
| 3 | 10.2 | 542 | 44 |
| 4 | 13.6 | 499 | 35 |
| 5 | 17.0 | 464 | 39 |
| 6 | 20.4 | 425 | 39 |
| 7 | 23.8 | 385 | 48 |
| 8 | 27.2 | 337 | 53 |
| 9 | 30.6 | 284 | 97 |
| 10 | 34.0 | 187 | 187 |
|  |  |  |  |

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B . The critical column is tabulated in Table 4.93.

Table 4.93 TM10-Bending Moment Results on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 90.7 | 102.5 | $12.91 \%$ |

Storey drifts and stability coefficients for TM10 model are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.94 and 4.95 respectively.

Table 4.94 TM10 Drift Check

| Level | $\boldsymbol{\delta}_{\mathbf{x e}}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 2.09 | 11.49 | 6.46 | 52 | OK |
| 2 | 6.73 | 37.00 | 25.50 | 52 | OK |
| 3 | 12.41 | 68.28 | 31.28 | 52 | OK |
| 4 | 18.40 | 101.21 | 32.94 | 52 | OK |
| 5 | 24.34 | 133.88 | 32.67 | 52 | OK |
| 6 | 30.15 | 165.81 | 31.94 | 52 | OK |
| 7 | 35.42 | 194.83 | 29.01 | 52 | OK |
| 8 | 40.04 | 220.20 | 25.37 | 52 | OK |
| 9 | 43.85 | 241.19 | 20.99 | 52 | OK |
| 10 | 46.76 | 257.20 | 16.01 | 52 | OK |

Table 4.95 TM10 Stability Coefficient

| TM10 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 47630 | 6.46 | 621 | 3400 | 5.5 | 0.026 |
| 2 | 42779 | 25.50 | 595 | 3400 | 5.5 | 0.098 |
| 3 | 37927 | 31.28 | 542 | 3400 | 5.5 | $\mathbf{0 . 1 1 7}$ |
| 4 | 33076 | 32.94 | 499 | 3400 | 5.5 | $\mathbf{0 . 1 1 7}$ |
| 5 | 28224 | 32.67 | 464 | 3400 | 5.5 | 0.106 |
| 6 | 23373 | 31.94 | 425 | 3400 | 5.5 | 0.094 |
| 7 | 18698 | 29.01 | 385 | 3400 | 5.5 | 0.075 |
| 8 | 14024 | 25.37 | 337 | 3400 | 5.5 | 0.056 |
| 9 | 9349 | 20.99 | 284 | 3400 | 5.5 | 0.037 |
| 10 | 4185 | 16.01 | 187 | 3400 | 5.5 | 0.019 |

The P-Delta analysis is required based on the stability coefficient which is larger than 0.1 which is assured by the percent difference between firstorder and second order analyses on the critical column which is also larger than 0.1.The maximum stability coefficient's value is very close to the percent difference between first and second-order analyses (Table 4.93) on the critical column. However, the stability coefficient gives slightly unconservative result compared to the percent difference on the critical column

### 4.2.3 Fifteen-Storey model (TM15)

TM15's columns' sections are tabulated in Table 4.96. TM15 has the same results as M15 until step 10; the calculations are resumed from step 11 as follows;

Table 4.96 TM15 columns' sections

| Level | Depth | Width |
| :---: | :---: | :---: |
| $1-5$ | 0.9 m | 0.9 m |
| $6-10$ | 0.7 m | 0.7 m |
| $11-15$ | 0.6 m | 0.6 m |

## STEP 11 (Effective Seismic Weight)

Each floor weight has determined in the same way as calculating M15 weights and tabulated in Table 4.97.

Table 4.97 TM15-Floors' Weights

| Floor | Weight (kN) |
| :---: | :---: |
| 1 | 4314 |
| 2 | 4314 |
| 3 | 4314 |
| 4 | 4314 |
| 5 | 4314 |
| 6 | 3879 |
| 7 | 3879 |
| 8 | 3879 |
| 9 | 3879 |
| 10 | 3879 |
| 11 | 3704 |
| 12 | 3704 |
| 13 | 3704 |
| 14 | 3704 |
| 15 | 3459 |
| Total | 59240 |

STEP 12 (Fundamental Period of the Structure)
The fundamental period of the structure (T) is determined from SAP2000, version 19.0.0 software analysis. For TM15 model, the fundamental period equals 3.73 second.

## Verification on SAP2000 fundamental period:

Rayleigh-Ritz method can be used for calculating an approximation to the fundamental period using Eq. 4.5 as shown in Table 4.98.

Table 4.98 TM15-Verification on SAP2000 Fundamental Period

| Storey | $\mathbf{W}_{\mathbf{i}} \mathbf{( k N )}$ | $\mathbf{f}_{\mathbf{i}}(\mathbf{k N})^{\mathbf{a}}$ | $\boldsymbol{\delta}_{\mathbf{i}(\mathbf{m})^{\mathbf{b}}}$ | $\left.\boldsymbol{\delta}_{\mathbf{i}} \mathbf{2}^{\mathbf{( 1 0}} \mathbf{- 3}^{\mathbf{3}}\right)$ | $\mathbf{f}_{\mathbf{i}} \boldsymbol{\delta}_{\mathbf{i}}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\delta}^{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 4314 | 1000 | 0.030 | 0.0009 | 30.1 | 3.9 |
| 2 | 4314 | 1000 | 0.103 | 0.0107 | 103.4 | 46.1 |
| 3 | 4314 | 1000 | 0.202 | 0.0407 | 201.6 | 175.4 |
| 4 | 4314 | 1000 | 0.313 | 0.0979 | 312.9 | 422.5 |
| 5 | 4314 | 1000 | 0.430 | 0.1846 | 429.7 | 796.4 |
| 6 | 3879 | 1000 | 0.550 | 0.3020 | 549.6 | 1171.5 |
| 7 | 3879 | 1000 | 0.664 | 0.4413 | 664.3 | 1711.8 |
| 8 | 3879 | 1000 | 0.770 | 0.5933 | 770.3 | 2301.5 |
| 9 | 3879 | 1000 | 0.866 | 0.7498 | 865.9 | 2908.5 |
| 10 | 3879 | 1000 | 0.951 | 0.9046 | 951.1 | 3508.9 |
| 11 | 3704 | 1000 | 1.026 | 1.0517 | 1025.5 | 3895.5 |
| 12 | 3704 | 1000 | 1.085 | 1.1769 | 1084.9 | 4359.4 |
| 13 | 3704 | 1000 | 1.130 | 1.2776 | 1130.3 | 4732.4 |
| 14 | 3704 | 1000 | 1.163 | 1.3530 | 1163.2 | 5011.6 |
| 15 | 3459 | 1000 | 1.186 | 1.4064 | 1185.9 | 4864.8 |
|  |  |  |  | $\sum$ | 10469 | 35910 |

a Assumed forces
b The displacement corresponding to the assumed forces

By applying Eq. 4.5 with Table 4.98
$\mathrm{T}=3.72$ second
$\%$ error $=\frac{\text { Approximated Period-SAP Period }}{\text { SAP Period }} \times 100 \%=0.3 \%<10 \%$
Acceptable

## STEP 13 (Maximum Period Limit)

After referring to Table 4.22
$\mathrm{C}_{\mathrm{t}}=0.0466$ (Concrete moment-resisting frame)
$\mathrm{x}=0.9$
$\mathrm{h}_{\mathrm{n}}=$ number of storeys $\times$ storey's height $=15 \times 3.40=51 \mathrm{~m}$
Apply previous values in Eq. 4.6, we get
$\mathrm{T}_{\mathrm{a}}=0.0466 \times 51^{0.9}=1.60$ seconds
After referring to Table 4.21
$S_{D 1}=0.25$
Interpolation between $S_{D 1}$ equals 0.20 and $S_{D 1}$ equals 0.30 , we get
$\mathrm{C}_{\mathrm{u}}=1.45$
$\mathrm{C}_{\mathrm{u}} \times \mathrm{T}_{\mathrm{a}}=1.45 \times 1.60=2.32$ second $<3.73$ second (Use $\mathrm{T}=2.32$ second for base shear calculations).

## STEP 14 (Response Spectrum Graph)

The elastic and inelastic response spectrums for TM15 model are created as previously mentioned as shown in Figs. 4.32 and 4.33 respectively.

STEP 15 (Determination of Seismic Base Shear Using Equivalent Lateral Force Analysis)

The seismic base shear $(\mathrm{V})$ in a given direction is determined by applying Eq. 4.9 as follows:
$C_{s}$ is the minimum of $\left\{\begin{array}{l}\frac{S_{D S} \times I}{R} \\ \frac{S_{D 1} \times I}{R \times T}\end{array}=\left\{\begin{array}{l}\frac{0.50 \times 1}{8} \\ \frac{0.25 \times 1}{8 \times 2.32}\end{array}=\left\{\begin{array}{c}0.063 \\ 0.0135\end{array}=0.0135\right.\right.\right.$

By applying Eq. 4.9, we get:
$\mathrm{V}=0.0135 \times 59240=800 \mathrm{kN}$


Figure 4.32 TM15-Elastic Response Spectrum


Figure 4.33 TM15-Inelastic Response Spectrum

STEP 16 (Determination of Seismic Base Shear using Modal Response Spectrum Analysis)

The response spectrum function was defined as previously mentioned and the base shear (V) has been computed by using section cut at columns of ground floor and the result is as follows,
$\mathrm{V}=439 \mathrm{kN}$
Table 4.99 summarizes the selection process for the appropriate base shear value based on ASCE 7 section 12.9.4.1.

## Table 4.99 TM15-Base Shear Selection

| Base shear using MRSA <br> (kN) | Base shear using ELF <br> ( <br> $(\mathbf{k N})$ | $\mathbf{0 . 8 5} \times$ ELF $^{\mathbf{2}}$ <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: |
| 439 | 800 | 680 |

1 Modal Response Spectrum Analysis
2 Equivalent Lateral Force

The base shear using modal response spectrum analysis is less than 85 percent of the calculated base shear using equivalent lateral force procedure. Thus, the base shear has to scale to $0.85 \times$ ELF which equals 680 kN .

## STEP 17 (Vertical Distribution of Seismic Base Shear)

Seismic base shear distribution can be computed using SAP2000 by making section cut and reading the shear force from response analysis case at each level as shown in Table 4.100 and scale the value to $0.85 \times$ ELF.

## Table 4.100 TM15-Base Shear Distribution on Floors

| Floor | $\mathbf{h ( m )}$ | Shear Force <br> $(\mathbf{k N})$ | Force on each <br> Floor $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: |
| 1 | 3.40 | 680 | 14 |
| 2 | 6.80 | 666 | 36 |
| 3 | 10.2 | 630 | 46 |
| 4 | 13.6 | 584 | 39 |
| 5 | 17.0 | 545 | 20 |
| 6 | 20.4 | 525 | 19 |
| 7 | 23.8 | 507 | 31 |
| 8 | 27.2 | 476 | 34 |
| 9 | 30.6 | 441 | 28 |
| 10 | 34.0 | 414 | 33 |
| 11 | 37.4 | 381 | 39 |
| 12 | 40.8 | 342 | 33 |
| 13 | 44.2 | 310 | 43 |
| 14 | 47.6 | 266 | 102 |
| 15 | 51.0 | 164 | 164 |

First-order moment and second-order moment are determined on all columns in all floors and they are annexed in appendix B . The critical column is tabulated in Table 4.101.

## Table 4.101 Bending Moment Results for TM15 Model on Critical Column

|  | First-Order <br> Analysis | Second-Order <br> Analysis | Percent <br> Difference |
| :---: | :---: | :---: | :---: |
| Bending Moment <br> (kN.m) | 126.5 | 152.7 | $20.72 \%$ |

Storey drifts and stability coefficients for TM15 model are calculated in the same way as were calculated for model M5 and tabulated in Tables 4.102 and 4.103 respectively.

Table 4.102 TM15 Drift Check

| Level | $\boldsymbol{\delta}_{\text {xe }}(\mathbf{m m})$ | $\boldsymbol{\delta}_{\mathbf{x}}(\mathbf{m m})$ | $\Delta(\mathbf{m m})$ | $\Delta_{\text {Allowable }}$ <br> $(\mathbf{m m})$ | Check |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 1.45 | 7.98 | 4.55 | 52 | OK |
| 2 | 5.03 | 27.66 | 19.68 | 52 | OK |
| 3 | 9.89 | 54.41 | 26.75 | 52 | OK |
| 4 | 15.48 | 85.14 | 30.72 | 52 | OK |
| 5 | 21.44 | 117.91 | 32.78 | 52 | OK |
| 6 | 27.79 | 152.83 | 34.91 | 52 | OK |
| 7 | 34.20 | 188.12 | 35.29 | 52 | OK |
| 8 | 40.47 | 222.57 | 34.45 | 52 | OK |
| 9 | 46.47 | 255.56 | 33.00 | 52 | OK |
| 10 | 52.16 | 286.91 | 31.34 | 52 | OK |
| 11 | 57.60 | 316.81 | 29.90 | 52 | OK |
| 12 | 62.48 | 343.66 | 26.85 | 52 | OK |
| 13 | 66.77 | 367.23 | 23.57 | 52 | OK |
| 14 | 70.33 | 386.83 | 19.60 | 52 | OK |
| 15 | 73.04 | 401.73 | 14.90 | 52 | OK |

Table 4.103 TM15 Stability Coefficient

| TM15 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{P}_{\mathbf{x}}(\mathbf{k N})$ | $\Delta(\mathbf{m m})$ | $\mathbf{V}_{\mathbf{x}}(\mathbf{k N})$ | $\mathbf{h}_{\mathbf{s x}}(\mathbf{m m})$ | $\mathbf{C}_{\mathbf{d}}$ | $\boldsymbol{\theta}$ |
| 1 | 74063 | 4.55 | 680 | 3400 | 5.5 | 0.03 |
| 2 | 68776 | 19.68 | 666 | 3400 | 5.5 | 0.11 |
| 3 | 63490 | 26.75 | 630 | 3400 | 5.5 | 0.14 |
| 4 | 58203 | 30.72 | 584 | 3400 | 5.5 | 0.16 |
| 5 | 52917 | 32.78 | 545 | 3400 | 5.5 | $\mathbf{0 . 1 7}$ |
| 6 | 47630 | 34.91 | 525 | 3400 | 5.5 | $\mathbf{0 . 1 7}$ |
| 7 | 42779 | 35.29 | 507 | 3400 | 5.5 | 0.16 |
| 8 | 37927 | 34.45 | 476 | 3400 | 5.5 | 0.15 |
| 9 | 33076 | 33.00 | 441 | 3400 | 5.5 | 0.13 |
| 10 | 28224 | 31.34 | 414 | 3400 | 5.5 | 0.11 |
| 11 | 23373 | 29.90 | 381 | 3400 | 5.5 | 0.10 |
| 12 | 18698 | 26.85 | 342 | 3400 | 5.5 | 0.08 |
| 13 | 14024 | 23.57 | 310 | 3400 | 5.5 | 0.06 |
| 14 | 9349 | 19.60 | 266 | 3400 | 5.5 | 0.04 |
| 15 | 4185 | 14.90 | 164 | 3400 | 5.5 | 0.02 |

The P-Delta analysis is required based on the stability coefficient which is larger than 0.1 which is assured by the percent difference between firstorder and second order analyses on the critical column which is also larger than 0.1 . The maximum stability coefficient's value is to some extent close to the percent difference between first and second-order analyses (Table 4.101) on the critical column. However, the stability coefficient does not describe precisely the percent difference between first-order analysis and second order-analysis and gives lesser value and its value just indicates including or not including P-Delta analysis and this is noted for stability coefficient exceeds 0.1 .

### 4.4 Seismic Design of Reinforced Concrete Special Moment Frames

Special Moment Resisting Frames (SMRFs) are used as a part of the seismic resisting system. SMRF concept was introduced in the United States discharge about 1960. Their use was very limited until 1973 where
the Uniform Building Code first required use of the special frames in the highest seismicity regions. SMRF has more stringent requirements compared to Ordinary or Intermediate Moment Frames. The requirements in designing SMRF are addressed in ACI 318. Three types of elements are detailed to resist seismic loads without tremendous loss in stiffness or strength; these are beams, columns and beam-column joints [41]. They are detailed to resist internal forces (flexural, shear and axial forces) that result from imposed displacement as a result of earthquake ground shaking and to ensure that reinforced concrete special moment resisting frame maintains a high level of ductility and energy dissipation mechanism [43]. In this section, some special moment resisting frames main design regulations are explained and presented.

The good performance of SMRFs is greatly guaranteed if the following stringent design provisions are applied [42],

- The failure has to be a ductile failure; which means the designer shall avoid brittle failure such as shear failure.
- The flexural failure has to precede shear failure.
- Beam's distortion has to precede column's distortion.
- Joints are stronger than members.
- Use the concept of strong-column/weak-beam frame.


### 4.4.1 SMRFs Layout and Proportioning

Special Frame Beam Layout Requirements in ACI 318

- Comment 1: Beam clear span shall not be less than four times its effective depth.
- Comment 2: Width of beam web shall be larger than or equal the minimum of 0.3 of its depth, or 250 mm .

Special Frame Column Layout Requirements in ACI 318

- Comment 1: The shortest cross-sectional dimension of the column shall be 300 mm at least.
- Comment 2: The shortest cross-sectional dimension of the column shall be at least 0.4 of the other perpendicular dimension within the section.

Tables 4.104 and 4.105 summarize checking process on beams' and columns' dimensions respectively.

Table 4.104 Special Frame Beams Dimensions Checks

| Model | Beam |  | Comment 2 | Effective <br> Depth | Clear <br> Span | Comment 1 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Depth | Width |  |  |  |  |
| M Models | 0.7 m | 0.75 m | $>0.21 \mathrm{~m}$ | 0.64 m | 4.7 m | $>2.56 \mathrm{~m}$ |
| TM Models | 0.4 m | 0.6 m | $>0.12 \mathrm{~m}$ | 0.34 m | 4.8 m | $>1.36 \mathrm{~m}$ |

Table 4.105 Special Frame Columns Dimensions Checks

| Model | Base Column |  | Comment 1 | Comment 2 |
| :---: | :---: | :---: | :---: | :---: |
|  | Depth(m) | Width(m) |  | $>0.3 \mathrm{~m}$ |
| $>0.4$ of other <br> dimension |  |  |  |
| M5 | 0.75 | 0.75 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| M10 | 0.9 | 0.9 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| M20 | 1.0 | 1.0 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| M25 | 1.2 | 1.1 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| M30 | 1.3 | 1.3 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| TM5 | 0.6 | 0.6 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| TM10 | 0.7 | 0.7 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |
| TM15 | 0.9 | 0.9 | $>0.3 \mathrm{~m}$ | $>0.4$ of other <br> dimension |

### 4.4.2 SMRFs Material Properties

- ACI 318 states that "the specified concrete compressive strength shall be at least 21 MPa ". For all models in this research, a 24 MPa concrete compressive strength is used for beams and slabs, and 32MPa compressive strength is used for columns which are both larger than 21 MPa .
- ACI 318 states that "steel grades higher than 420 MPa are not permitted". Yielding stress of 420MPa is utilized for all steel bars.


### 4.4.3 Longitudinal and Transverse Beam Reinforcement

In this section, a beam which is part of the special moment resisting frame in M5 model as shown in Fig. 4.34 is designed and the ACI 318 provisions are checked.

### 4.4.3.1 Load Combinations

Structures, components, and foundations shall be designed so that their design strength equals or exceeds the effects of the factored loads in the following combinations (ASCE 7, section 2.3.2);

$$
\begin{align*}
& 1.4 \mathrm{D}  \tag{4.16}\\
& 1.2 \mathrm{D}+1.6 \mathrm{~L}  \tag{4.17}\\
& 1.2 \mathrm{D}+1.0 \mathrm{~L}+1.0 \mathrm{E} \tag{4.18}
\end{align*}
$$



Figure 4.34 Location of Beam which is verified (All dimensions are in meter)

$$
\begin{equation*}
0.9 \mathrm{D}+1.0 \mathrm{E} \tag{4.19}
\end{equation*}
$$

Where,
D = dead load
$\mathrm{L}=$ live load
$\mathrm{E}=$ for use in Eq. 4.18 and Eq. 4.19, shall be determined in accordance with Eq. 4.20 and Eq. 4.21 respectively (ASCE 7, section 12.4.2) as follows:

$$
\begin{align*}
& \mathrm{E}=\mathrm{E}_{\mathrm{h}}+\mathrm{E}_{\mathrm{v}}  \tag{4.20}\\
& \mathrm{E}=\mathrm{E}_{\mathrm{h}}-\mathrm{E}_{\mathrm{v}} \tag{4.21}
\end{align*}
$$

Where,
$E_{h}=$ effect of horizontal seismic forces $=\rho Q_{E}$
$\mathrm{E}_{\mathrm{v}}=$ effect of vertical seismic forces $=0.2 \mathrm{~S}_{\mathrm{DS}} \mathrm{D}$
Where,
$\rho=$ redundancy factor $=1.3($ ASCE 7, section 12.3.4 $)$
$\mathrm{Q}_{\mathrm{E}}=$ seismic effect of orthogonal loading
$\mathrm{S}_{\mathrm{DS}}$ and D are as defined before
Applying each quantity value in previous load combinations to get the final factored load pattern as follows:

$$
\begin{align*}
& 1.4 \mathrm{D}  \tag{4.22}\\
& 1.2 \mathrm{D}+1.6 \mathrm{~L}  \tag{4.23}\\
& 1.3 \mathrm{D}+1.0 \mathrm{~L}+1.3 \mathrm{QE}_{\mathrm{E}}  \tag{4.24}\\
& 0.8 \mathrm{D}+1.3 \mathrm{Q}_{\mathrm{E}} \tag{4.25}
\end{align*}
$$

During careful examination; Eq. 4.24 was giving the more critical values for the negative moment and Eq. 4.23 was giving the more critical values for the positive moment.

### 4.4.3.2 Beam Moments

The negative moment $\left(\mathrm{M}_{-\mathrm{ve}}\right)=-300 \mathrm{kN} . \mathrm{m}$
The positive moment $\left(\mathrm{M}_{+\mathrm{ve}}\right)=124 \mathrm{kN} . \mathrm{m}$
Based on ACI 318, positive moment strength is not to be less than $1 / 2$ of the negative moment strength provided at the face of the joint. Furthermore, the negative or positive moment at any section along the member is not to be less than $1 / 4$ of the maximum moment strength at face of either joint.

$$
\mathrm{M}_{+\mathrm{ve}}=0.5 * 300=150 \mathrm{kN} . \mathrm{m}>124 \mathrm{kN} . \mathrm{m} \quad\left(\text { Take } \mathrm{M}_{+\mathrm{ve}}=150 \mathrm{kN} . \mathrm{m}\right)
$$

### 4.4.3.3 Factored Axial Force

Based on ACI 318, section 21.5.1; the factored axial compressive force is considered in the beam if $\mathrm{P}_{\mathrm{u}}>0.1 \mathrm{f}_{\mathrm{c}} \mathrm{A}_{\mathrm{g}}$
$\mathrm{P}_{\mathrm{u}}=12.2 \mathrm{kN}<1260 \mathrm{kN} \quad$ (i.e. No need to consider axial force)

### 4.4.3.4 Longitudinal Reinforcement

$$
\begin{equation*}
\mathrm{d}=\mathrm{h}-\mathrm{C}_{\mathrm{c}}-\mathrm{d}_{\mathrm{h}}-0.5 \mathrm{~d}_{\mathrm{b}} \tag{4.26}
\end{equation*}
$$

Where,
$d=$ member effective depth
$\mathrm{h}=$ member depth $=700 \mathrm{~mm}$
$\mathrm{C}_{\mathrm{c}}=$ concrete cover $=40 \mathrm{~mm}$
$\mathrm{d}_{\mathrm{h}}=$ hoop diameter $\approx 10 \mathrm{~mm}$ (assumed)
$\mathrm{d}_{\mathrm{b}}=$ bar diameter $\approx 20 \mathrm{~mm}$ (assumed)
Applying previous values to Eq. 4.26 as follows:
$\mathrm{d}=700-40-10-10=640 \mathrm{~mm}$
The area of steel and minimum area of steel can be obtained from the following equations for $\mathrm{M}_{\mathrm{u}}$ equals $300 \mathrm{kN} . \mathrm{m}$ :

$$
A_{s}=\frac{0.85 f_{c}^{\prime} b_{w} d}{f y}\left[1-\sqrt{1-\frac{2.61 M_{u}}{f_{c}^{\prime} b_{w} d^{2}}}\right]
$$

$\mathrm{A}_{\mathrm{s}}=1280 \mathrm{~mm}^{2}$
$A_{s, \min }=\frac{0.25 \sqrt{f_{c}^{\prime}}}{f y} b_{w} d$

$$
\mathrm{A}_{\mathrm{s}, \min }=1400 \mathrm{~mm}^{2}
$$

$$
A_{s, \min }=\frac{1.4}{f y} b_{w} d
$$

$A_{s, \text { min }}=1600 \mathrm{~mm}^{2}$

Take $\mathrm{A}_{\mathrm{s}}=1600 \mathrm{~mm}^{2}$ (8Ф16)

Based on ACI 318, section 21.5.2.3; lap splices of flexural reinforcement shall be permitted only if hoop or spiral reinforcement is provided over the lap length. Spacing of the transverse reinforcement enclosing the lapspliced bars shall not exceed the smaller of $\mathrm{d} / 4$ or 100 mm . Lap splices shall not be used:
a. Within the joints;
b. Within the distance of twice member depth from the face of the joint.
c. At locations where analysis indicates flexural yielding caused by inelastic lateral displacements of the frame.

In the beam under investigation; lap splices are used with implementing the previous requirements with spacing of the transverse reinforcement equals 100 mm .

### 4.4.3.5 Shear Strength Requirements

The design shear force $\mathrm{V}_{\mathrm{e}}$ is to be determined from consideration of the static forces on the portion of the member between faces of the joint. It is assumed that moments of opposite sign corresponding to probable flexural moment strength $\mathrm{M}_{\mathrm{pr}}$ act at the joint faces and that the member is loaded with the factored tributary gravity load along its span (ACI 318, section
21.5.4). For calculations of $\mathrm{M}_{\mathrm{pr}}$ it is assumed that tensile strength in the longitudinal bars is $1.25 \mathrm{~F}_{\mathrm{y}}$. The calculations are as follows:
$\mathrm{V}_{\mathrm{g}}($ shear force from factored gravity load $)=168 \mathrm{kN}$

$$
\begin{equation*}
\mathrm{a}=\frac{\mathrm{A}_{\mathrm{s}} 1.25 \mathrm{~F}_{\mathrm{y}}}{0.85 \mathrm{f}_{\mathrm{c}} \mathrm{~b}_{\mathrm{w}}} \tag{4.27}
\end{equation*}
$$

Where,
$\mathrm{a}=$ depth of equivalent rectangular compressive block
$A_{s}, F_{y}, f_{c}$ and $b_{w}$ are as defined before
$\mathrm{A}_{\mathrm{s}}=1600 \mathrm{~mm}^{2}, \mathrm{~F}_{\mathrm{y}}=420 \mathrm{MPa}, \mathrm{f}_{\mathrm{c}}=24 \mathrm{MPa}, \mathrm{bw}=750 \mathrm{~mm}$, after applying these values in Eq. 12; we get:
$a=55 \mathrm{~mm}$
$\mathrm{M}_{\mathrm{pr}}(+\mathrm{ve})=\mathrm{M}_{\mathrm{pr}}(-\mathrm{ve})=\mathrm{A}_{\mathrm{s}} \times 1.25 \mathrm{~F}_{\mathrm{y}} \times(\mathrm{d}-\mathrm{a} / 2)$
$=515 \mathrm{kN} . \mathrm{m}$

$$
\begin{equation*}
\mathrm{V}_{\mathrm{d}}=\frac{\mathrm{M}_{\mathrm{pr}(+\mathrm{ve})}+\mathrm{M}_{\mathrm{pr}(-\mathrm{ve})}}{\mathrm{L}_{\mathrm{n}}} \tag{4.28}
\end{equation*}
$$

Where,
$\mathrm{V}_{\mathrm{d}}=$ design shear force
$\mathrm{L}_{\mathrm{n}}=$ clear span length $=5.25 \mathrm{~m}$
$\mathrm{V}_{\mathrm{d}}=196 \mathrm{kN}$

$$
\begin{equation*}
\mathrm{V}_{\mathrm{e}, \max }=\mathrm{V}_{\mathrm{g}}+\mathrm{V}_{\mathrm{d}} \tag{4.29}
\end{equation*}
$$

Where
$\mathrm{V}_{\mathrm{e}, \max }=$ maximum required shear strength
$\mathrm{V}_{\mathrm{g}}$ and $\mathrm{V}_{\mathrm{d}}$ are as defined before
$\mathrm{V}_{\mathrm{e}, \max }=365 \mathrm{kN}$

Based on ACI 318, section 21.5.4.2; transverse reinforcement over lengths identified in 21.5.3.1 shall be proportioned to resist shear assuming $\mathrm{V}_{\mathrm{c}}=0$ when the following conditions occur,

- The design shear force represents $1 / 2$ or more of the maximum required shear strength within those lengths;
- The factored axial compressive force including earthquake effect is less than $0.05 \mathrm{Ag}_{\mathrm{g}}$.

Since both conditions met, the concrete nominal shear strength assumed to equal zero.
$\mathrm{V}_{\mathrm{s}}=\frac{V_{u}}{\emptyset}=\frac{365}{0.75}=487 \mathrm{kN}$

To get the spacing, Eq. 4.30 can be used as follows:

$$
\begin{equation*}
\mathrm{V}_{\mathrm{s}}=\frac{\mathrm{A}_{\mathrm{vf}} \mathrm{~d}}{} \tag{4.30}
\end{equation*}
$$

Where,
$\mathrm{A}_{\mathrm{v}}=$ stirrup cross sectional area
For two-legged 10 mm transverse reinforcement,
$\mathrm{s}=87 \mathrm{~mm}$
Based on ACI 318, section 21.5.3.2, the first hoop is to be located at a distance not more than 50 mm from the face of the supporting member. Maximum spacing of such reinforcement is not to exceed the smallest of: $d / 4,6 d_{b}$ where $d_{b}$ is the diameter of the smallest longitudinal bars, and 150 mm .

The smallest value of previous limits is $6 \mathrm{~d}_{\mathrm{b}}=96 \mathrm{~mm}$
Use two-legged 10mm stirrup @ 80mm

For stirrups at other locations;
$\mathrm{V}_{\mathrm{c}}=0.17 \sqrt{\mathrm{f}_{\mathrm{c}}} \mathrm{b}_{\mathrm{w}} \mathrm{d}=400 \mathrm{kN}$
$\mathrm{V}_{\mathrm{u}}=260 \mathrm{kN}$
Since $\frac{\mathrm{V}_{\mathrm{u}}}{\emptyset}<\mathrm{V}_{\mathrm{c}} ; A_{\mathrm{v}, \text { min }} / \mathrm{s}$ shall be provided
$\mathrm{A}_{\mathrm{v}, \text { min }} / \mathrm{s}$ shall be greater of :
a. $\frac{0.062 \sqrt{f_{c}} b_{w}}{f_{y}}=0.54 \mathrm{~mm}^{2} / \mathrm{mm}$
b. $\frac{0.35 b_{w}}{f_{y}} \quad=0.625 \mathrm{~mm}^{2} / \mathrm{mm}$

Use $A_{v} / \mathrm{s}=0.625 \mathrm{~mm}^{2} / \mathrm{mm}$
For two-legged 10 mm transverse reinforcement; $\mathrm{s}=250 \mathrm{~mm}$
Based on ACI 318, section 21.5.3.4, where hoops are not required, stirrups with seismic hooks at both ends are to be spaced at a distance not more than $\mathrm{d} / 2$ throughout the length of the member.
$\mathrm{d} / 2=320 \mathrm{~mm}>200 \mathrm{~mm}$
Use two-legged 10mm stirrup @ 200mm
Previous results for the beam under consideration are detailed in Fig. 4.35.

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Figure 4.35 Longitudinal and Transverse Reinforcement for the Special Beam (All dimensions are in meter)



### 4.4.4 Design of Special Column

The columns which the beam in previous section is supported on; are designed, detailed and checked for ACI 318 special frame provisions.
$\mathrm{P}_{\mathrm{u}}=3588 \mathrm{kN}$ (from the critical load combination)
$\mathrm{M}_{\mathrm{u}}=235 \mathrm{kN} . \mathrm{m}$ (from the critical load combination)
$\mathrm{A}_{\mathrm{s}}(\mathrm{SAP} 2000)=5625 \mathrm{~mm}^{2} \quad(20 \emptyset 20 \mathrm{~mm})$
$0.1 \mathrm{f}_{\mathrm{c}} \mathrm{A}_{\mathrm{g}}=1800 \mathrm{kN}<3588 \mathrm{kN}$
Thus, requirements of section 21.6.2 in ACI 318 must be satisfied.
Based on ACI318, section 21.6.2.2, the flexural strengths of the columns shall satisfy the following equation:

$$
\begin{equation*}
\sum \mathrm{M}_{\mathrm{nc}} \geq 1.2 \sum \mathrm{M}_{\mathrm{nb}} \tag{4.31}
\end{equation*}
$$

Where,
$\sum \mathrm{M}_{\mathrm{nc}}=$ sum of nominal flexural strengths of columns framing into the joint, evaluated at the face of the joint
$\sum \mathrm{M}_{\mathrm{nb}}=$ sum of nominal flexural strengths of the beams framing into the joint, evaluated at the face of the joint

Considering the columns on both sides of the joint are of equal flexural strengths, the flexural strength of each column is determined using strength interaction diagrams using $\rho$ equals $1 \%, \gamma$ equals 0.85 , $\varnothing$ equals $0.65, \mathrm{P}_{\mathrm{u}}$ equals 3588 kN .
$M_{u}=1270 \mathrm{kN} . \mathrm{m}$ (from strength interaction diagram)
$\sum \mathrm{M}_{\mathrm{nc}}=1270+1270=2540 \mathrm{kN} . \mathrm{m}$
$\mathrm{a}=\frac{\mathrm{A}_{\mathrm{s}} \mathrm{F}_{\mathrm{y}}}{0.85 \mathrm{f}_{\mathrm{c}} \mathrm{b}_{\mathrm{w}}}=44 \mathrm{~mm} \quad$ (for previous beam)
$M_{b}(+v e)=M_{b}(-v e)=A_{s} \times F_{y} \times(d-a / 2)=415 k N . m$
$\sum \mathrm{M}_{\mathrm{nb}}=415 \times 4=1660$
$\frac{\sum \mathrm{M}_{\mathrm{nc}}}{\sum \mathrm{M}_{\mathrm{nb}}}=\frac{2540}{1660}=1.5>1.2$ (OK)

Based on ACI 318, section 21.6.3.1, the reinforcement ratio shall not be less than 0.01 and shall not exceed 0.06 .
$\rho=0.01$
Based on ACI 318, section 21.6.3.3, lap splices are only permitted within the center half of the member length and shall be designed as tension lap splices enclosed within transverse reinforcement.

For class B lap splice, $\mathrm{L}_{\mathrm{sp}}=1.3 \mathrm{~L}_{\mathrm{d}}=1.3 \times \frac{0.59 \times f_{y} \times d_{b}}{\sqrt{f_{c}}} \geq 300 \mathrm{~mm}$ $=1.15 \mathrm{~m}$ (Use 1.2m)

Based on ACI 318, section 21.6.4.2, transverse reinforcement shall be spaced at a distance not exceeding (a) $1 / 4$ of the minimum member dimension (b) six times the diameter of the longitudinal reinforcement and
(c) $\mathrm{S}_{\mathrm{x}}=220-0.085 \mathrm{~h}_{\mathrm{x}}$ where $\mathrm{h}_{\mathrm{x}}$ is the largest value of the distance from the centerline to centerline of tie legs.
(a) $0.25 \times 750=187 \mathrm{~mm}$
(b) $6 \times 20=120 \mathrm{~mm}$
(c) $\mathrm{h}_{\mathrm{x}}=126 \mathrm{~mm}, \mathrm{~S}_{\mathrm{x}}=209 \mathrm{~mm}$

Use spacing equals 100 mm between transverse reinforcement.
Based on ACI 318, section 21.6.4.4, the total cross section area of rectangular hoop reinforcement, $\mathrm{A}_{\text {sh }}$, shall not be less than that required by Eqs. 4.32 and 4.33.

$$
\begin{gather*}
\mathrm{A}_{\mathrm{sh}, 1}=0.3 \frac{\mathrm{sb}_{\mathrm{c}} \mathrm{f}_{\mathrm{c}}}{\mathrm{f}_{\mathrm{y}}}\left[\left(\frac{\mathrm{~A}_{\mathrm{g}}}{\mathrm{~A}_{\mathrm{ch}}}\right)-1\right]  \tag{4.32}\\
\mathrm{A}_{\mathrm{sh}, 2}=0.09 \frac{\mathrm{sb}_{\mathrm{c}} \mathrm{f}_{\mathrm{c}}}{\mathrm{fy}} \tag{4.33}
\end{gather*}
$$

Where,
$\mathrm{s}=$ tie spacing
$\mathrm{A}_{\mathrm{ch}}=$ core area
$\mathrm{b}_{\mathrm{c}}=$ the core dimension perpendicular to the tie legs that constitute $\mathrm{A}_{\mathrm{ch}}$
$\mathrm{s}=100 \mathrm{~mm}, \mathrm{~b}_{\mathrm{c}}=670 \mathrm{~mm}, \mathrm{~A}_{\mathrm{ch}}=670 \mathrm{~mm} \times 670 \mathrm{~mm}, A_{\mathrm{g}}=750 \mathrm{~mm} \times 50 \mathrm{~mm}, \mathrm{f}_{\mathrm{c}}=$ 32 MPa
$\mathrm{A}_{\text {sh }, 1}=388 \mathrm{~mm}^{2}$
$\mathrm{A}_{\mathrm{sh}, 2}=460 \mathrm{~mm}^{2}$
Use $A_{\text {sh }}=460 \mathrm{~mm}^{2}$
Use $\emptyset 12 \mathrm{~mm}$ tie plus four $\emptyset 12 \mathrm{~mm}$ cross tie @ 100 mm

$$
\begin{equation*}
A_{s h}=6 \times 113=670 \mathrm{~mm}^{2}>460 \mathrm{~mm}^{2} \tag{OK}
\end{equation*}
$$

Based on ACI 318, section 21.6.4.1, transverse reinforcement in amount specified before shall be provided over a length $L_{0}$ from each joint face and on both sides of any section where flexural yielding is likely to occur as a result of inelastic lateral displacements of the frame. The length $L_{0}$ shall not be less than the largest of the following:
a. the depth of the member at the joint face $=750 \mathrm{~mm}$
b. $1 / 6$ of the clear span of the member $=\frac{1}{6} \times 3250=542 \mathrm{~mm}$
c. 450 mm
i.e. 750 mm

Based on ACI 318, section 21.6.4.5, beyond $\mathrm{L}_{0}$, the column shall contain spiral or hoop reinforcement with center to center spacing not exceeding the smaller of six times the diameter of the smallest longitudinal column bars and 150 mm .
$S_{\text {max }, 1}=6 \times 20=120 \mathrm{~mm}$
$\mathrm{S}_{\mathrm{max}, 2}=150 \mathrm{~mm}$
i.e. $S_{\max }=150 \mathrm{~mm}$

The design shear force, $\mathrm{V}_{\mathrm{e}}$, shall be determined from consideration of the maximum forces that can be generated at the faces of the joint at each end of the member. These forces shall be determined using the maximum probable moment strengths, $\mathrm{M}_{\mathrm{pr}}$, at each end of the member associated with the range of axial loads on the member (ACI 318, section 21.6.5.1). $\mathrm{P}_{\mathrm{u}}=1300 \mathrm{kN}$ (factored axial load)

$$
\begin{equation*}
\gamma=\frac{\mathrm{h}-2 * \mathrm{c}}{\mathrm{~h}} \tag{4.34}
\end{equation*}
$$

Where,
$\mathrm{h}=$ column dimension perpendicular to the axis of rotation
$\mathrm{c}=$ concrete cover
From Eq. 4.34,
$\gamma=\frac{\mathrm{h}-2 * \mathrm{c}}{\mathrm{h}}=\frac{750-2 \times 40}{750}=0.9$
$\frac{\mathrm{P}_{\mathrm{u}}}{\not \mathrm{A}_{\mathrm{g}} \mathrm{f}_{\mathrm{c}}}=\frac{1300 \times 1000}{0.65 \times 750 \times 750 \times 32}=0.11$

$$
\begin{equation*}
\rho=\frac{\mathrm{A}_{\mathrm{s}}}{\mathrm{~b} \times \mathrm{d}} \tag{4.35}
\end{equation*}
$$

Where,
$\mathrm{A}_{\mathrm{s}}=$ area of steel
$\mathrm{b}=$ column width
$\mathrm{d}=$ column depth $=750-40-(20 / 2)-12=688 \mathrm{~mm}$
From Eq. 4.35,
$\rho=\frac{\mathrm{A}_{\mathrm{s}}}{\mathrm{b} \times \mathrm{d}}=\frac{6283}{750 \times 688}=0.012$
Based on the previous values and the proper column interaction diagram with the indispensable interpolation; we can get the following values:
$\frac{\mathrm{M}_{\mathrm{u}}}{\emptyset_{\mathrm{g}} \mathrm{f}_{\mathrm{c}} \mathrm{h}}=0.142$
$\frac{\mathrm{M}_{\mathrm{u}}}{\varnothing \mathrm{Ag}_{\mathrm{c}} \mathrm{h}}=\frac{\mathrm{M}_{\mathrm{u}}}{0.65 \times 750 \times 750 \times 32 \times 750}=0.09 \rightarrow \mathrm{M}_{\mathrm{u}}=790 \mathrm{kN} . \mathrm{m}$
$\mathrm{V}_{\mathrm{e}}=\frac{790 \times 4 \times 0.5}{3.25}=486 \mathrm{kN}$
$\mathrm{V}_{\mathrm{c}}=0.17 \sqrt{\mathrm{f}_{\mathrm{c}}} \mathrm{b}_{\mathrm{w}} \mathrm{d}=498 \mathrm{kN}$
$\mathrm{V}_{\mathrm{e}} / \emptyset>\mathrm{V}_{\mathrm{c}}$
$\mathrm{V}_{\mathrm{s}}=\mathrm{V}_{\mathrm{e}} / \emptyset-\mathrm{V}_{\mathrm{c}}=150 \mathrm{kN}$
$\frac{\mathrm{A}_{\mathrm{v}}}{\mathrm{s}}=\frac{\mathrm{V}_{\mathrm{S}}}{\mathrm{F}_{\mathrm{y}} \mathrm{d}}=0.52 \mathrm{~mm}^{2} / \mathrm{mm}$
For spacing $100 \mathrm{~mm} \rightarrow \mathrm{~A}_{\mathrm{v}}=52 \mathrm{~mm}^{2}$
$\mathrm{A}_{\mathrm{v}}\left(\right.$ available within the length $\left.\mathrm{L}_{0}\right)=670 \mathrm{~mm}^{2}>52 \mathrm{~mm}^{2} \quad(\mathrm{OK})$
Previous results for the special column are detailed in Figs. 4.36 and 4.37.


Figure 4.36 Special Column Detailing (All dimensions are in meter)


Figure 4.37 Special Column Cross Section (All dimensions are in meter)

### 4.4.5 Strong-Column/Weak-Beam Frame Concept

This concept means that columns are stiffer than the beams frame into them. The purpose of this concept in earthquake engineering is that when a building sways during an earthquake; if it has weak columns, drifts tend to concentrate in one or few stories as shown in Fig. 4.38-a, and may override the column drift limit. On the other hand, if the building has strong columns, the drifts will be more uniformly distributed as shown in Fig. 4.38-c. Furthermore, columns support the weight of the entire building, whereas beams support only the gravity loads of the floor which they form apart; thus, column's failure has a greater effect than failure of beam [41].

Studies (e.g. Kuntz and Browning 2003) have shown that the full beam mechanism can only be achieved if the column-to-beam strength ratio on the order of four. As this ratio is impractical in most cases; ACI 318 adopted this concept by specifying column-to-beam strength ratio of 1.2. Table 4.106 summarizes column-to-beam strength for previous example in model M5.

(a) Story mechanism

(b) Intermediate mechanism

(c) Beam mechanism

Figure 4.38 Strong-Column/Weak-Beam Concept Illustration

## Table 4.106 Strong-Column/Weak-Beam Checks

| Model | flexural capacity of <br> the column (kN.m) | flexural capacity of <br> the beam (kN.m) | column- <br> to-beam <br> strength | Comment |
| :---: | :---: | :---: | :---: | :---: |
| M5 | 790 | 515 | 1.53 | $>1.2$ |

### 4.5 Results and discussion

For the models which abide by the proposed equation period limit; the relationship between the fundamental period and the maximum stability coefficient is drawn in Fig. 4.39. The relationship between the fundamental period and the maximum percent difference between firstorder and second-order analyses (i.e. on the critical column) is drawn in Fig 4.40. Table 4.107 summarizes the maximum stability coefficient and the maximum percent difference. Table 4.108 tabulates the location of critical columns. For the models which do not abide by the proposed equation period limit; the relationship between the fundamental period and the maximum stability coefficient is drawn in Fig. 4.41. The relationship between the fundamental period and the maximum percent difference between first-order and second-order analyses (i.e. on the critical column) is drawn in Fig 4.42. Table 4.109 summarizes the maximum stability coefficient and the maximum percent difference. Table 4.110 tabulates the location of critical columns.


Figure 4.39 Maximum Stability Coefficient versus Fundamental Period for the Abiding Models by the Proposed Equation


Figure 4.40 Maximum Percent Difference between First and Second-order Analyses on the Critical Column Versus the Fundamental Period for the Abiding Models by the Proposed Equation

Table 4.107 Maximum Stability Coefficient versus Maximum Percent Difference between First and Second-Order Analyses for the Abiding Models by the Proposed Equation

| Model | Maximum Stability <br> Coefficient ( $\boldsymbol{\theta})$ | Maximum Percent Difference |
| :---: | :---: | :---: |
| M5 | $1.60 \%$ | $1.46 \%$ |
| M10 | $3.30 \%$ | $3.05 \%$ |
| M15 | $5.10 \%$ | $4.97 \%$ |
| M20 | $7.10 \%$ | $7.35 \%$ |
| M25 | $9.20 \%$ | $9.82 \%$ |
| M30 | $11.40 \%$ | $12.40 \%$ |

Table 4.108 Location of critical Columns for the Abiding Models by the Proposed Equation

| Model | Location of Critical Columns |
| :---: | :---: |
| M5 | Second Storey |
| M10 | Second Storey |
| M15 | Third Storey |
| M20 | Fourth Storey |
| M25 | Fourth Storey |
| M30 | Fifth Storey |

200


Figure 4.41 Maximum Stability Coefficient versus Fundamental Period for the nonAbiding Models by the Proposed Equation


Figure 4.42 Maximum Percent Difference between First and Second-order Analyses on the Critical Column Versus the Fundamental Period for the non- Abiding Models by the Proposed Equation

Table 4.109 Maximum Stability Coefficient versus Maximum Percent Difference between First and Second-Order Analyses for the nonAbiding Models by the Proposed Equation

| Model | Maximum Stability <br> Coefficient ( $\boldsymbol{\theta})$ | Maximum Percent <br> Difference |
| :---: | :---: | :---: |
| TM5 | $5 \%$ | $4.95 \%$ |
| TM10 | $11.7 \%$ | $12.91 \%$ |
| TM15 | $17 \%$ | $20.72 \%$ |

Table 4.110 Location of Critical Columns for the non-Aboding Models by the Proposed Equation

| Model | Location of Critical Columns |
| :---: | :---: |
| TM5 | Second Storey |
| TM10 | Third Storey |
| TM15 | Fourth Storey |

It can be noted from Figs. 4.39 and 4.41 that the period limits after which the P-Delta analysis must be included in the analysis (i.e. stability coefficient exceeds $10 \%$ ) are 3.8 second and 2.2 second for the abiding and non-abiding models by the proposed equation respectively. It can be noted from Figs. 4.40 and 4.42 that the period limits after which the percent difference between first-order and second-order analyses exceeds $10 \%$ are 3.6 second and 2.1 second for the abiding and non-abiding models by the proposed equation respectively. It can be noticed from Tables 4.107 and 4.109 that for both the abiding and the non-abiding models to the proposed equation, the stability coefficient gives to some extent conservative results until $10 \%$, and then, the percent difference on the critical column gives
larger percentage (i.e. more conservative). Thus, it is concluded that when the stability coefficient exceeds $10 \%$, its value does not describe precisely the percent difference between first-order and second-order analyses rather than its value indicates the urgent need to include the P-Delta effect in the analysis. Furthermore, it can be elicited from Tables 4.108 and 4.110 that in the non-abiding models to the proposed equation, the location of the critical column has the trend to go upward compared to the same number of storeys model nevertheless it abides by the proposed equation, concurrently; it remains in the same location in the plan. From Tables 4.107 and 4.109 , it is noted that most likely engineers will not run into problems of P-Delta if they build reinforced concrete portal frames until 25 storeys and 8 storeys for abiding and non-abiding models by the proposed equation respectively.

## Chapter Five

## Summary, Conclusions, and Research Needs

### 5.1 Conclusions

### 5.1.1 General Conclusions

The research led to the following general conclusions:

1. When the number of storeys increases, the percent difference between first-order analysis and second-order analysis increases as well.

- The percent difference between first-order analysis and secondorder analysis is larger for the non-abiding models by the proposed equation (i.e. models that do not abide by Eq. 4.6).

2. When the number of storeys increases, the column which has the maximum percent difference between first-order and second-order moments (i.e. the critical column) goes upward.

- The location of the critical column in the plan remains the same in the all models (i.e. in the edge). However, its level increases upward with increasing number of storeys.
- The critical column has a larger willingness to go upward in the case of the non-abiding models by the proposed equation compared to the abiding models.

3. The drift is checked for all models and it is under the allowable limit, although there are problems in P-Delta above a certain number of
storeys which urges engineers not to neglect P-Delta if the drift is below the allowable limits.
4. The stability coefficient gives to some extent conservative results until $10 \%$, and then, the percent difference on the critical column gives larger percentage (i.e. more conservative). Thus, it is concluded that when the stability coefficient exceeds $10 \%$, its value does not describe precisely the percent difference between first-order and second-order analyses rather than its value indicates the urgent need to include the P Delta effect in the analysis.
5. The ASCE 7 limits using Eq. 4.6 in calculations of base shear, but it was believed that this equation has implicitly another objective which is if the fundamental period of the structure abides by this equation; the structure will not run into problems of P-Delta which is proved in this thesis.
6. Abiding by Eq. 4.6 as a structure's fundamental period limit allows building approximately another 17 storeys without considering P-Delta effect.

### 5.1.2 Specific Conclusions

The research led to the following specific conclusions:

1. Most probably, engineers will not run problems with P-Delta if they build reinforced concrete portal frames until 25 storeys for the models which abide by the proposed equation and 8storeys for the models which do not abide by the proposed equation for 10 percent of increase.
2. Most buildings in Palestine do not reach 15-storey which makes previous conclusion so valuable for structural engineers.
3. The period limits after which the P-Delta analysis must be included in the analysis are 3.6 second and 2.1 second for the abiding and nonabiding models by the proposed equation respectively.

### 5.2 Recommendations

In this research, it was decided to focus on well-defined problems for which accurate information could be obtained for deriving of the essential relationships. Research on the following topics is needed in order to generalize the results obtained in this research:

1. Variation of structural material from reinforced concrete to steel.
2. Examine other seismic force-resisting systems such as dual, shear wall systems and bracing.
3. Variation of plan and elevation configurations.
4. Study the interaction between torsion and P-delta effect.
5. Effect of inelasticity.
6. The ASCE 7-16 states that, the base shear using the modal response spectrum procedure shall not be less than 100 percent of the calculated base shear using the equivalent lateral force procedure. Thus, it is recommended to take this note into account in the future studies.

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## Appendix A

## Verification of Response Shear Results

## A. Verification of SAP2000 Response Shear Results

## A. 1 M10

Table A. 1 M10-Mode 1 Modal Mass Participation Ratio

| Mode1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathbf{i}}$ | $\boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W i g}_{\mathbf{i}} \mathrm{i}^{\mathbf{2}}$ |
| 1 | 5260 | -0.056 | 0.003 | -294.7 | 16.5 |
| 2 | 5260 | -0.172 | 0.030 | -904.0 | 155.4 |
| 3 | 5260 | -0.308 | 0.095 | -1619.5 | 498.6 |
| 4 | 5260 | -0.446 | 0.199 | -2347.6 | 1047.7 |
| 5 | 5260 | -0.579 | 0.335 | -3045.3 | 1763.1 |
| 6 | 4923 | -0.707 | 0.500 | -3482.8 | 2463.9 |
| 7 | 4923 | -0.816 | 0.665 | -4015.7 | 3275.6 |
| 8 | 4923 | -0.901 | 0.812 | -4435.6 | 3996.4 |
| 9 | 4923 | -0.962 | 0.925 | -4734.9 | 4554.0 |
| 10 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  |  |  | $\sum$ | -29420 | 22311 |
| $\beta_{1}=\frac{-29420}{22311}=-1.32$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 1}=\frac{-29420^{2}}{22311}=38794 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \% \\ =\frac{38794}{50532} \times 100 \%=77 \% \end{gathered}$ |  |  |  |  |  |

Table A. 2 M10-Mode1 Shear Distribution

| Mode1 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 1.38 | 0.022 | 5260 | 0.074 | 8.5 | 853.5 |  |
| 2 | 1.38 | 0.022 | 5260 | 0.227 | 26.2 | 844.9 |  |
| 3 | 1.38 | 0.022 | 5260 | 0.406 | 47.0 | 818.7 |  |
| 4 | 1.38 | 0.022 | 5260 | 0.589 | 68.1 | 771.7 |  |
| 5 | 1.38 | 0.022 | 5260 | 0.763 | 88.3 | 703.6 |  |
| 6 | 1.38 | 0.022 | 4923 | 0.933 | 101.0 | 615.3 |  |
| 7 | 1.38 | 0.022 | 4923 | 1.076 | 116.5 | 514.2 |  |
| 8 | 1.38 | 0.022 | 4923 | 1.188 | 128.7 | 397.7 |  |
| 9 | 1.38 | 0.022 | 4923 | 1.268 | 137.4 | 269.1 |  |
| 10 | 1.38 | 0.022 | 4540 | 1.319 | 131.7 | 131.7 |  |

Table A. 3 M10-Mode 2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathrm{i}}$ | $\boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W i g}_{\mathbf{i}} \mathrm{\Phi}^{\mathbf{i}}$ |
| 1 | 5260 | 0.173 | 0.030 | 909.7 | 157.3 |
| 2 | 5260 | 0.486 | 0.236 | 2555.1 | 1241.2 |
| 3 | 5260 | 0.758 | 0.575 | 3988.0 | 3023.7 |
| 4 | 5260 | 0.894 | 0.799 | 4701.7 | 4202.6 |
| 5 | 5260 | 0.859 | 0.737 | 4516.1 | 3877.3 |
| 6 | 4923 | 0.608 | 0.369 | 2991.1 | 1817.4 |
| 7 | 4923 | 0.185 | 0.034 | 910.7 | 168.5 |
| 8 | 4923 | -0.292 | 0.085 | -1438.1 | 420.1 |
| 9 | 4923 | -0.711 | 0.506 | -3501.1 | 2490.0 |
| 10 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  | 50532 |  | $\Sigma$ | 11093 | 21938 |
| $\beta_{2}=\frac{11093}{21938}=0.51$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{11093^{2}}{21938}=5609 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 2=\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \% \\ =\frac{5609}{50532}=11.1 \% \end{gathered}$ |  |  |  |  |  |

Table A. 4 M10-Mode2 Shear Distribution

| Mode 2 |  |  |  |  |  |  |  | Force | Shear <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \boldsymbol{\Phi}_{\mathbf{i}}$ | $(\mathbf{k N})$ | (sN |  |  |  |
| 2 | 0.44 | 0.063 | 5260 | 0.087 | 29.0 | 353.4 |  |  |  |
| 3 | 0.44 | 0.063 | 5260 | 0.246 | 81.4 | 324.4 |  |  |  |
| 4 | 0.44 | 0.063 | 5260 | 0.383 | 127.0 | 243.0 |  |  |  |
| 5 | 0.44 | 0.063 | 5260 | 0.452 | 149.8 | 116.0 |  |  |  |
| 6 | 0.44 | 0.063 | 5260 | 0.434 | 143.9 | -33.8 |  |  |  |
| 7 | 0.44 | 0.063 | 4923 | 0.094 | 29.0 | -273.0 |  |  |  |
| 8 | 0.44 | 0.063 | 4923 | -0.148 | -45.8 | -302.0 |  |  |  |
| 9 | 0.44 | 0.063 | 4923 | -0.360 | -111.5 | -256.2 |  |  |  |
| 10 | 0.44 | 0.063 | 4540 | -0.506 | -144.6 | -144.6 |  |  |  |

For M10 case, mode1 and mode 2 give almost $90 \%$ of the actual mass. Thus, mode1 and mode2 are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table A.5.

Table A. 5 M10-Shear Force Determination Using Both Hand Calculations and SAP2000

| Level | Shear Using Hand <br> Calculations (kN) | Shear Using <br> SAP2000 (kN) | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 923.8 | 960 | $3.7 \%$ |
| 2 | 905.0 | 934 | $3.1 \%$ |

[^2]
## A. 2 M15

Table A. 6 M15-Mode 1 Modal Mass Participation Ratio

| Mode1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathbf{i}}$ | $\Phi_{i}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathrm{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ |
| 1 | 5518 | -0.031 | 0.001 | -172.1 | 5.4 |
| 2 | 5518 | -0.100 | 0.010 | -551.9 | 55.2 |
| 3 | 5518 | -0.186 | 0.035 | -1025.3 | 190.5 |
| 4 | 5518 | -0.278 | 0.077 | -1536.1 | 427.6 |
| 5 | 5518 | -0.372 | 0.139 | -2055.0 | 765.3 |
| 6 | 5260 | -0.467 | 0.218 | -2455.8 | 1146.5 |
| 7 | 5260 | -0.557 | 0.310 | -2930.1 | 1632.2 |
| 8 | 5260 | -0.641 | 0.411 | -3372.8 | 2162.7 |
| 9 | 5260 | -0.718 | 0.516 | -3778.8 | 2714.7 |
| 10 | 5260 | -0.788 | 0.622 | -4146.9 | 3269.4 |
| 11 | 4923 | -0.853 | 0.728 | -4201.0 | 3584.9 |
| 12 | 4923 | -0.907 | 0.822 | -4464.6 | 4049.0 |
| 13 | 4923 | -0.949 | 0.900 | -4671.5 | 4432.8 |
| 14 | 4923 | -0.980 | 0.960 | -4822.5 | 4724.0 |
| 15 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  | 78122 |  | $\Sigma$ | -44724 | 33700 |
| $\beta_{1}=\frac{-44724}{33700}=-1.33$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{el}}=\frac{-44724^{2}}{33700}=59355 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \%$ $=\frac{59355}{78122}=76 \%$ |  |  |  |  |  |

Table A.7M15-Mode1 Shear Distribution

| Mode1 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 2.09 | 0.0149 | 5518 | 0.041 | 3.4 | 884.4 |  |
| 2 | 2.09 | 0.0149 | 5518 | 0.133 | 10.9 | 881.0 |  |
| 3 | 2.09 | 0.0149 | 5518 | 0.247 | 20.3 | 870.1 |  |
| 4 | 2.09 | 0.0149 | 5518 | 0.369 | 30.4 | 849.8 |  |
| 5 | 2.09 | 0.0149 | 5518 | 0.494 | 40.6 | 819.4 |  |
| 6 | 2.09 | 0.0149 | 5260 | 0.620 | 48.6 | 778.8 |  |
| 7 | 2.09 | 0.0149 | 5260 | 0.739 | 57.9 | 730.2 |  |
| 8 | 2.09 | 0.0149 | 5260 | 0.851 | 66.7 | 672.3 |  |
| 9 | 2.09 | 0.0149 | 5260 | 0.953 | 74.7 | 605.6 |  |
| 10 | 2.09 | 0.0149 | 5260 | 1.046 | 82.0 | 530.9 |  |
| 11 | 2.09 | 0.0149 | 4923 | 1.132 | 83.1 | 448.9 |  |
| 12 | 2.09 | 0.0149 | 4923 | 1.204 | 88.3 | 365.8 |  |
| 13 | 2.09 | 0.0149 | 4923 | 1.259 | 92.4 | 277.5 |  |
| 14 | 2.09 | 0.0149 | 4923 | 1.300 | 95.4 | 185.1 |  |
| 15 | 2.09 | 0.0149 | 4540 | 1.327 | 89.8 | 89.8 |  |

Table A. 8 M15-Mode 2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\Phi_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathrm{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ |
| 1 | 5518 | 0.093 | 0.009 | 513.5 | 47.8 |
| 2 | 5518 | 0.286 | 0.082 | 1575.8 | 450.0 |
| 3 | 5518 | 0.499 | 0.249 | 2751.7 | 1372.2 |
| 4 | 5518 | 0.686 | 0.471 | 3786.2 | 2597.9 |
| 5 | 5518 | 0.821 | 0.675 | 4532.5 | 3722.9 |
| 6 | 5260 | 0.881 | 0.776 | 4632.5 | 4079.9 |
| 7 | 5260 | 0.845 | 0.714 | 4444.8 | 3756.0 |
| 8 | 5260 | 0.720 | 0.519 | 3788.2 | 2728.2 |
| 9 | 5260 | 0.520 | 0.271 | 2736.0 | 1423.2 |
| 10 | 5260 | 0.266 | 0.071 | 1397.7 | 371.4 |
| 11 | 4923 | -0.046 | 0.002 | -224.6 | 10.2 |
| 12 | 4923 | -0.361 | 0.131 | -1778.6 | 642.6 |
| 13 | 4923 | -0.642 | 0.412 | -3158.9 | 2026.9 |
| 14 | 4923 | -0.858 | 0.736 | -4224.1 | 3624.4 |
| 15 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  | 78122 |  | $\Sigma$ | 16233 | 31394 |
| $\beta_{2}=\frac{16233}{31394}=0.52$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{16233^{2}}{31394}=8394 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 2=\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \% \\ =\frac{8394}{78122}=10.7 \% \end{gathered}$ |  |  |  |  |  |

Table A. 9 M15-Mode2 Shear Distribution

| Mode 2 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \boldsymbol{\Phi}_{\mathbf{i}}$ | Force (kN) | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 0.69 | 0.045 | 5518 | 0.048 | 11.9 | 377.7 |  |
| 2 | 0.69 | 0.045 | 5518 | 0.148 | 36.7 | 365.8 |  |
| 3 | 0.69 | 0.045 | 5518 | 0.258 | 64.0 | 329.1 |  |
| 4 | 0.69 | 0.045 | 5518 | 0.355 | 88.1 | 265.1 |  |
| 5 | 0.69 | 0.045 | 5518 | 0.425 | 105.5 | 177.0 |  |
| 6 | 0.69 | 0.045 | 5260 | 0.455 | 107.8 | 71.5 |  |
| 7 | 0.69 | 0.045 | 5260 | 0.437 | 103.4 | -36.3 |  |
| 8 | 0.69 | 0.045 | 5260 | 0.372 | 88.1 | -139.7 |  |
| 9 | 0.69 | 0.045 | 5260 | 0.269 | 63.7 | -227.9 |  |
| 10 | 0.69 | 0.045 | 5260 | 0.137 | 32.5 | -291.5 |  |
| 11 | 0.69 | 0.045 | 4923 | -0.024 | -5.2 | -324.0 |  |
| 12 | 0.69 | 0.045 | 4923 | -0.187 | -41.4 | -318.8 |  |
| 13 | 0.69 | 0.045 | 4923 | -0.332 | -73.5 | -277.4 |  |
| 14 | 0.69 | 0.045 | 4923 | -0.444 | -98.3 | -203.9 |  |
| 15 | 0.69 | 0.045 | 4540 | -0.517 | -105.6 | -105.6 |  |

Table A. 10 M15-Mode 3 Modal Mass Participation Ratio

| Mode3 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathrm{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathbf{i}}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{\text {2 }}$ |
| 1 | 5518 | -0.165 | 0.027 | -908.8 | 149.7 |
| 2 | 5518 | -0.470 | 0.221 | -2591.4 | 1217.0 |
| 3 | 5518 | -0.730 | 0.533 | -4028.6 | 2941.2 |
| 4 | 5518 | -0.838 | 0.703 | -4626.4 | 3878.8 |
| 5 | 5518 | -0.755 | 0.570 | -4164.6 | 3143.2 |
| 6 | 5260 | -0.464 | 0.215 | -2439.9 | 1131.7 |
| 7 | 5260 | -0.030 | 0.001 | -156.5 | 4.7 |
| 8 | 5260 | 0.420 | 0.177 | 2210.1 | 928.6 |
| 9 | 5260 | 0.764 | 0.584 | 4020.0 | 3072.2 |
| 10 | 5260 | 0.922 | 0.851 | 4851.7 | 4475.1 |
| 11 | 4923 | 0.806 | 0.650 | 3969.6 | 3200.9 |
| 12 | 4923 | 0.412 | 0.169 | 2026.7 | 834.3 |
| 13 | 4923 | -0.122 | 0.015 | -601.1 | 73.4 |
| 14 | 4923 | -0.633 | 0.401 | -3116.7 | 1973.1 |
| 15 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  | 78122 |  | $\Sigma$ | -10096 | 31564 |
| $\beta_{3}=\frac{-10096}{31564}=-0.32$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 3}=\frac{-10096^{2}}{31564}=3229 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 3=\frac{\mathrm{W}_{\mathrm{e} 3}}{\text { total weight }} \times 100 \% \\ =\frac{3229}{78122}=4.1 \% \end{gathered}$ |  |  |  |  |  |

Table A. 11 M15-Mode3 Shear Distribution

| Mode3 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 0.38 | 0.063 | 5518 | 0.053 | 18.3 | 203.4 |  |
| 2 | 0.38 | 0.063 | 5518 | 0.150 | 52.2 | 185.1 |  |
| 3 | 0.38 | 0.063 | 5518 | 0.234 | 81.2 | 132.9 |  |
| 4 | 0.38 | 0.063 | 5518 | 0.268 | 93.2 | 51.7 |  |
| 5 | 0.38 | 0.063 | 5518 | 0.241 | 83.9 | -41.5 |  |
| 6 | 0.38 | 0.063 | 5260 | 0.148 | 49.2 | -125.4 |  |
| 7 | 0.38 | 0.063 | 5260 | 0.010 | 3.2 | -174.6 |  |
| 8 | 0.38 | 0.063 | 5260 | -0.134 | -44.5 | -177.7 |  |
| 9 | 0.38 | 0.063 | 5260 | -0.244 | -81.0 | -133.2 |  |
| 10 | 0.38 | 0.063 | 5260 | -0.295 | -97.8 | -52.2 |  |
| 11 | 0.38 | 0.063 | 4923 | -0.258 | -80.0 | 45.6 |  |
| 12 | 0.38 | 0.063 | 4923 | -0.132 | -40.8 | 125.6 |  |
| 13 | 0.38 | 0.063 | 4923 | 0.039 | 12.1 | 166.4 |  |
| 14 | 0.38 | 0.063 | 4923 | 0.202 | 62.8 | 154.3 |  |
| 15 | 0.38 | 0.063 | 4540 | 0.320 | 91.5 | 91.5 |  |

For M15 case, mode1, mode 2 and mode 3 give $91 \%$ of the actual mass. Thus, mode1, mode 2 and mode 3are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table A.12.

Table A.12 M15-Shear Force Determination Using Both Hand Calculations and SAP2000

| Level | Shear Using Hand <br> Calculations (kN) | Shear Using <br> SAP2000 (kN) | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 983 | 1007 | $2.4 \%$ |
| 2 | 972 | 993 | $2.1 \%$ |

[^3]A. 3 M20

Table A.13 M20-Mode 1 Modal Mass Participation Ratio

|  | Mode1 |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathrm{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathbf{W i m i}^{\mathbf{2}}$ |
| 1 | 5803 | 0.020 | 0.0004 | 114.3 | 2.3 |
| 2 | 5803 | 0.065 | 0.0043 | 378.5 | 24.7 |
| 3 | 5803 | 0.124 | 0.0155 | 722.1 | 89.9 |
| 4 | 5803 | 0.191 | 0.0363 | 1106.2 | 210.9 |
| 5 | 5803 | 0.260 | 0.0676 | 1508.7 | 392.2 |
| 6 | 5518 | 0.331 | 0.1099 | 1828.9 | 606.2 |
| 7 | 5518 | 0.402 | 0.1620 | 2220.9 | 893.9 |
| 8 | 5518 | 0.472 | 0.2226 | 2603.3 | 1228.2 |
| 9 | 5518 | 0.539 | 0.2902 | 2972.6 | 1601.3 |
| 10 | 5518 | 0.603 | 0.3634 | 3326.6 | 2005.4 |
| 11 | 5260 | 0.665 | 0.4419 | 3496.7 | 2324.5 |
| 12 | 5260 | 0.722 | 0.5216 | 3798.9 | 2743.6 |
| 13 | 5260 | 0.775 | 0.6005 | 4076.2 | 3158.8 |
| 14 | 5260 | 0.823 | 0.6771 | 4328.2 | 3561.4 |
| 15 | 5260 | 0.866 | 0.7500 | 4555.2 | 3944.9 |
| 16 | 4923 | 0.906 | 0.8206 | 4459.6 | 4039.8 |
| 17 | 4923 | 0.939 | 0.8817 | 4622.5 | 4340.4 |
| 18 | 4923 | 0.965 | 0.9322 | 4753.1 | 4589.1 |
| 19 | 4923 | 0.986 | 0.9714 | 4852.0 | 4782.1 |
| 20 | 4540 | 1.000 | 1.0000 | 4540.0 | 4540.0 |
|  | 107137 |  | $\Sigma$ | 60264 | 45080 |
| $\beta_{1}=\frac{60264}{45080}=1.34$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{el}}=\frac{60264^{2}}{45080}=80564 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{gathered} \text { Modal mass participation ratio of Mode } 1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \% \\ =\frac{80564}{107137}=75.2 \% \end{gathered}$ |  |  |  |  |  |

Table A. 14 M20-Mode1 Shear Distribution

| Mode1 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 2.82 | 0.011 | 5803 | 0.026 | 1.7 | 886.2 |  |
| 2 | 2.82 | 0.011 | 5803 | 0.087 | 5.6 | 884.5 |  |
| 3 | 2.82 | 0.011 | 5803 | 0.166 | 10.6 | 879.0 |  |
| 4 | 2.82 | 0.011 | 5803 | 0.255 | 16.3 | 868.3 |  |
| 5 | 2.82 | 0.011 | 5803 | 0.348 | 22.2 | 852.1 |  |
| 6 | 2.82 | 0.011 | 5518 | 0.443 | 26.9 | 829.9 |  |
| 7 | 2.82 | 0.011 | 5518 | 0.538 | 32.7 | 803.0 |  |
| 8 | 2.82 | 0.011 | 5518 | 0.631 | 38.3 | 770.3 |  |
| 9 | 2.82 | 0.011 | 5518 | 0.720 | 43.7 | 732.1 |  |
| 10 | 2.82 | 0.011 | 5518 | 0.806 | 48.9 | 688.3 |  |
| 11 | 2.82 | 0.011 | 5260 | 0.889 | 51.4 | 639.4 |  |
| 12 | 2.82 | 0.011 | 5260 | 0.966 | 55.9 | 588.0 |  |
| 13 | 2.82 | 0.011 | 5260 | 1.036 | 59.9 | 532.1 |  |
| 14 | 2.82 | 0.011 | 5260 | 1.100 | 63.6 | 472.2 |  |
| 15 | 2.82 | 0.011 | 5260 | 1.158 | 67.0 | 408.6 |  |
| 16 | 2.82 | 0.011 | 4923 | 1.211 | 65.6 | 341.6 |  |
| 17 | 2.82 | 0.011 | 4923 | 1.255 | 68.0 | 276.0 |  |
| 18 | 2.82 | 0.011 | 4923 | 1.291 | 69.9 | 208.0 |  |
| 19 | 2.82 | 0.011 | 4923 | 1.318 | 71.4 | 138.1 |  |
| 20 | 2.82 | 0.011 | 4540 | 1.337 | 66.8 | 66.8 |  |

Table A. 15 M20-Mode 2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathrm{i}}$ | $\boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W i}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ |
| 1 | 5803 | 0.059 | 0.003 | 341.4 | 20.1 |
| 2 | 5803 | 0.189 | 0.036 | 1098.7 | 208.0 |
| 3 | 5803 | 0.348 | 0.121 | 2019.8 | 703.0 |
| 4 | 5803 | 0.508 | 0.258 | 2948.7 | 1498.4 |
| 5 | 5803 | 0.653 | 0.426 | 3788.1 | 2472.8 |
| 6 | 5518 | 0.768 | 0.590 | 4239.5 | 3257.3 |
| 7 | 5518 | 0.840 | 0.705 | 4633.9 | 3891.5 |
| 8 | 5518 | 0.863 | 0.745 | 4761.4 | 4108.5 |
| 9 | 5518 | 0.837 | 0.701 | 4618.6 | 3865.7 |
| 10 | 5518 | 0.765 | 0.585 | 4222.0 | 3230.4 |
| 11 | 5260 | 0.644 | 0.415 | 3387.3 | 2181.3 |
| 12 | 5260 | 0.479 | 0.230 | 2520.3 | 1207.6 |
| 13 | 5260 | 0.283 | 0.080 | 1488.9 | 421.5 |
| 14 | 5260 | 0.068 | 0.005 | 357.2 | 24.3 |
| 15 | 5260 | -0.155 | 0.024 | -814.4 | 126.1 |
| 16 | 4923 | -0.388 | 0.151 | -1911.8 | 742.4 |
| 17 | 4923 | -0.601 | 0.361 | -2956.9 | 1776.0 |
| 18 | 4923 | -0.778 | 0.605 | -3830.0 | 2979.6 |
| 19 | 4923 | -0.911 | 0.830 | -4486.2 | 4088.1 |
| 20 | 4540 | -1.000 | 1.000 | -4540.0 | 4540.0 |
|  | 107137 |  | $\sum$ | 21887 | 41342 |
| $\beta_{2}=\frac{21887}{41342}=0.53$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{21887^{2}}{41342}=11587 \mathrm{kN}$ |  |  |  |  |  |
| $\begin{aligned} \text { Modal mass participation ratio of Mode } 2 & =\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \% \\ & =\frac{11587}{107137}=11 \% \end{aligned}$ |  |  |  |  |  |

Table A. 16 M20-Mode2 Shear Distribution

| Mode 2 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | $\mathbf{S h e a r}$ <br> $(\mathbf{k N})$ |  |
| 1 | 0.94 | 0.033 | 5803 | 0.031 | 6.0 | 382.4 |  |
| 2 | 0.94 | 0.033 | 5803 | 0.100 | 19.2 | 376.4 |  |
| 3 | 0.94 | 0.033 | 5803 | 0.184 | 35.3 | 357.2 |  |
| 4 | 0.94 | 0.033 | 5803 | 0.269 | 51.5 | 321.9 |  |
| 5 | 0.94 | 0.033 | 5803 | 0.346 | 66.2 | 270.4 |  |
| 6 | 0.94 | 0.033 | 5518 | 0.407 | 74.1 | 204.2 |  |
| 7 | 0.94 | 0.033 | 5518 | 0.445 | 81.0 | 130.2 |  |
| 8 | 0.94 | 0.033 | 5518 | 0.457 | 83.2 | 49.2 |  |
| 9 | 0.94 | 0.033 | 5518 | 0.443 | 80.7 | -34.0 |  |
| 10 | 0.94 | 0.033 | 5518 | 0.405 | 73.8 | -114.7 |  |
| 11 | 0.94 | 0.033 | 5260 | 0.341 | 59.2 | -188.4 |  |
| 12 | 0.94 | 0.033 | 5260 | 0.254 | 44.0 | -247.6 |  |
| 13 | 0.94 | 0.033 | 5260 | 0.150 | 26.0 | -291.6 |  |
| 14 | 0.94 | 0.033 | 5260 | 0.036 | 6.2 | -317.6 |  |
| 15 | 0.94 | 0.033 | 5260 | -0.082 | -14.2 | -323.9 |  |
| 16 | 0.94 | 0.033 | 4923 | -0.206 | -33.4 | -309.7 |  |
| 17 | 0.94 | 0.033 | 4923 | -0.318 | -51.7 | -276.3 |  |
| 18 | 0.94 | 0.033 | 4923 | -0.412 | -66.9 | -224.6 |  |
| 19 | 0.94 | 0.033 | 4923 | -0.482 | -78.4 | -157.7 |  |
| 20 | 0.94 | 0.033 | 4540 | -0.529 | -79.3 | -79.3 |  |

Table A. 17 M20-Mode3 Modal Mass Participation Ratio

| Mode3 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathrm{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathrm{i}}{ }^{2}$ |
| 1 | 5803 | 0.102 | 0.010 | 592.0 | 60.4 |
| 2 | 5803 | 0.314 | 0.099 | 1824.3 | 573.5 |
| 3 | 5803 | 0.543 | 0.294 | 3148.1 | 1707.9 |
| 4 | 5803 | 0.724 | 0.524 | 4201.3 | 3041.6 |
| 5 | 5803 | 0.822 | 0.676 | 4770.4 | 3921.6 |
| 6 | 5518 | 0.804 | 0.646 | 4434.1 | 3563.1 |
| 7 | 5518 | 0.655 | 0.429 | 3614.4 | 2367.5 |
| 8 | 5518 | 0.402 | 0.162 | 2217.6 | 891.2 |
| 9 | 5518 | 0.083 | 0.007 | 458.7 | 38.1 |
| 10 | 5518 | -0.255 | 0.065 | -1409.3 | 359.9 |
| 11 | 5260 | -0.569 | 0.323 | -2991.2 | 1701.0 |
| 12 | 5260 | -0.791 | 0.626 | -4161.4 | 3292.2 |
| 13 | 5260 | -0.886 | 0.785 | -4659.7 | 4127.8 |
| 14 | 5260 | -0.839 | 0.704 | -4412.5 | 3701.6 |
| 15 | 5260 | -0.662 | 0.439 | -3483.7 | 2307.3 |
| 16 | 4923 | -0.343 | 0.118 | -1687.5 | 578.5 |
| 17 | 4923 | 0.057 | 0.003 | 280.8 | 16.0 |
| 18 | 4923 | 0.455 | 0.207 | 2237.7 | 1017.1 |
| 19 | 4923 | 0.781 | 0.610 | 3845.3 | 3003.4 |
| 20 | 4540 | 1.000 | 1.000 | 4540.0 | 4540.0 |
|  | 107137 |  | $\Sigma$ | 13359 | 40810 |
| $\beta_{3}=\frac{13359}{40810}=0.33$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 3}=\frac{13359^{2}}{40810}=4373 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $3=\frac{\mathrm{W}_{\mathrm{e} 3}}{\text { total weight }} \times 100 \%=\frac{4373}{107137}=4.1 \%$ |  |  |  |  |  |

Table A. 18 M20-Mode3 Shear Distribution

| Mode3 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |  |
| 1 | 0.53 | 0.059 | 5803 | 0.033 | 11.4 | 258.0 |  |
| 2 | 0.53 | 0.059 | 5803 | 0.103 | 35.2 | 246.6 |  |
| 3 | 0.53 | 0.059 | 5803 | 0.178 | 60.8 | 211.4 |  |
| 4 | 0.53 | 0.059 | 5803 | 0.237 | 81.1 | 150.6 |  |
| 5 | 0.53 | 0.059 | 5803 | 0.269 | 92.1 | 69.4 |  |
| 6 | 0.53 | 0.059 | 5518 | 0.263 | 85.6 | -22.7 |  |
| 7 | 0.53 | 0.059 | 5518 | 0.214 | 69.8 | -108.4 |  |
| 8 | 0.53 | 0.059 | 5518 | 0.132 | 42.8 | -178.2 |  |
| 9 | 0.53 | 0.059 | 5518 | 0.027 | 8.9 | -221.0 |  |
| 10 | 0.53 | 0.059 | 5518 | -0.084 | -27.2 | -229.9 |  |
| 11 | 0.53 | 0.059 | 5260 | -0.186 | -57.8 | -202.6 |  |
| 12 | 0.53 | 0.059 | 5260 | -0.259 | -80.4 | -144.9 |  |
| 13 | 0.53 | 0.059 | 5260 | -0.290 | -90.0 | -64.5 |  |
| 14 | 0.53 | 0.059 | 5260 | -0.275 | -85.2 | 25.5 |  |
| 15 | 0.53 | 0.059 | 5260 | -0.217 | -67.3 | 110.7 |  |
| 16 | 0.53 | 0.059 | 4923 | -0.112 | -32.6 | 178.0 |  |
| 17 | 0.53 | 0.059 | 4923 | 0.019 | 5.4 | 210.6 |  |
| 18 | 0.53 | 0.059 | 4923 | 0.149 | 43.2 | 205.2 |  |
| 19 | 0.53 | 0.059 | 4923 | 0.256 | 74.3 | 162.0 |  |
| 20 | 0.53 | 0.059 | 4540 | 0.327 | 87.7 | 87.7 |  |

For M20 case, mode1, mode 2 and mode 3 give $90.3 \%$ of the actual mass. Thus, mode1, mode 2 and mode 3 are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table A.19.

Table A.19 M20-Shear Force Determination Using Both Hand Calculations and SAP2000

| Level | Shear Using Hand <br> Calculations (kN) | Shear Using <br> SAP2000 (kN) | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 999 | 1030 | $3.0 \%$ |
| 2 | 992 | 1020 | $2.7 \%$ |

[^4]
## A. 4 M25

Table A.20 M25-Mode 1 Modal Mass Participation Ratio

| Mode1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\Phi_{\text {i }}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathrm{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{\mathbf{2}}$ |
| 1 | 6116 | 0.013 | 0.0002 | 81.9 | 1.1 |
| 2 | 6116 | 0.045 | 0.0021 | 278.1 | 12.6 |
| 3 | 6116 | 0.089 | 0.0079 | 541.9 | 48.0 |
| 4 | 6116 | 0.138 | 0.0191 | 844.6 | 116.6 |
| 5 | 6116 | 0.191 | 0.0365 | 1169.0 | 223.4 |
| 6 | 5803 | 0.247 | 0.0609 | 1431.9 | 353.3 |
| 7 | 5803 | 0.303 | 0.0920 | 1760.3 | 534.0 |
| 8 | 5803 | 0.360 | 0.1294 | 2087.9 | 751.2 |
| 9 | 5803 | 0.416 | 0.1728 | 2411.9 | 1002.5 |
| 10 | 5803 | 0.470 | 0.2213 | 2729.8 | 1284.1 |
| 11 | 5518 | 0.525 | 0.2751 | 2894.3 | 1518.1 |
| 12 | 5518 | 0.577 | 0.3324 | 3181.5 | 1834.4 |
| 13 | 5518 | 0.626 | 0.3924 | 3456.7 | 2165.4 |
| 14 | 5518 | 0.674 | 0.4542 | 3718.9 | 2506.4 |
| 15 | 5518 | 0.719 | 0.5171 | 3968.2 | 2853.6 |
| 16 | 5260 | 0.762 | 0.5812 | 4009.9 | 3056.9 |
| 17 | 5260 | 0.802 | 0.6437 | 4220.0 | 3385.7 |
| 18 | 5260 | 0.839 | 0.7038 | 4412.7 | 3701.8 |
| 19 | 5260 | 0.872 | 0.7606 | 4587.4 | 4000.8 |
| 20 | 5260 | 0.902 | 0.8140 | 4745.8 | 4281.8 |
| 21 | 4923 | 0.930 | 0.8652 | 4579.2 | 4259.4 |
| 22 | 4923 | 0.954 | 0.9095 | 4695.0 | 4477.6 |
| 23 | 4923 | 0.973 | 0.9467 | 4790.0 | 4660.6 |
| 24 | 4923 | 0.988 | 0.9767 | 4865.3 | 4808.2 |
| 25 | 4540 | 1.000 | 1.0000 | 4540.0 | 4540.0 |
|  | 137717 |  | $\Sigma$ | 76002 | 56378 |
| $\beta_{1}=\frac{76002}{56378}=1.35$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 1}=\frac{76002^{2}}{56378}=102458 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \%=\frac{102458}{137717}=74.4 \%$ |  |  |  |  |  |

Table A.21 M25-Mode1 Shear Distribution

| Mode1 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $\mathbf{( k N )}$ |
| 1 | 3.57 | 0.0087 | 6116 | 0.018 | 1.0 | 891.4 |
| 2 | 3.57 | 0.0087 | 6116 | 0.061 | 3.3 | 890.4 |
| 3 | 3.57 | 0.0087 | 6116 | 0.119 | 6.4 | 887.2 |
| 4 | 3.57 | 0.0087 | 6116 | 0.186 | 9.9 | 880.8 |
| 5 | 3.57 | 0.0087 | 6116 | 0.258 | 13.7 | 870.9 |
| 6 | 3.57 | 0.0087 | 5803 | 0.333 | 16.8 | 857.2 |
| 7 | 3.57 | 0.0087 | 5803 | 0.409 | 20.6 | 840.4 |
| 8 | 3.57 | 0.0087 | 5803 | 0.485 | 24.5 | 819.7 |
| 9 | 3.57 | 0.0087 | 5803 | 0.560 | 28.3 | 795.3 |
| 10 | 3.57 | 0.0087 | 5803 | 0.634 | 32.0 | 767.0 |
| 11 | 3.57 | 0.0087 | 5518 | 0.707 | 33.9 | 735.0 |
| 12 | 3.57 | 0.0087 | 5518 | 0.777 | 37.3 | 701.0 |
| 13 | 3.57 | 0.0087 | 5518 | 0.844 | 40.5 | 663.7 |
| 14 | 3.57 | 0.0087 | 5518 | 0.909 | 43.6 | 623.2 |
| 15 | 3.57 | 0.0087 | 5518 | 0.969 | 46.5 | 579.5 |
| 16 | 3.57 | 0.0087 | 5260 | 1.028 | 47.0 | 533.0 |
| 17 | 3.57 | 0.0087 | 5260 | 1.082 | 49.5 | 486.0 |
| 18 | 3.57 | 0.0087 | 5260 | 1.131 | 51.8 | 436.5 |
| 19 | 3.57 | 0.0087 | 5260 | 1.176 | 53.8 | 384.7 |
| 20 | 3.57 | 0.0087 | 5260 | 1.216 | 55.7 | 330.9 |
| 21 | 3.57 | 0.0087 | 4923 | 1.254 | 53.7 | 275.3 |
| 22 | 3.57 | 0.0087 | 4923 | 1.286 | 55.1 | 221.6 |
| 23 | 3.57 | 0.0087 | 4923 | 1.312 | 56.2 | 166.5 |
| 24 | 3.57 | 0.0087 | 4923 | 1.332 | 57.1 | 110.3 |
| 25 | 3.57 | 0.0087 | 4540 | 1.348 | 53.2 | 53.2 |

Table A. 22 M25-Mode 2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W i}_{\mathbf{i}}(\mathbf{k N})$ | $\mathbf{\Phi}_{\mathbf{i}}$ | $\boldsymbol{\Phi}^{\mathbf{2}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W i}_{\mathbf{i}} \mathrm{I}^{\mathbf{i}}{ }^{\text {a }}$ |
| 1 | 6116 | -0.040 | 0.002 | -246.7 | 10.0 |
| 2 | 6116 | -0.134 | 0.018 | -820.7 | 110.1 |
| 3 | 6116 | -0.255 | 0.065 | -1558.1 | 396.9 |
| 4 | 6116 | -0.385 | 0.148 | -2352.7 | 905.0 |
| 5 | 6116 | -0.512 | 0.263 | -3134.0 | 1605.9 |
| 6 | 5803 | -0.629 | 0.396 | -3652.2 | 2298.6 |
| 7 | 5803 | -0.726 | 0.527 | -4210.7 | 3055.4 |
| 8 | 5803 | -0.797 | 0.635 | -4622.6 | 3682.3 |
| 9 | 5803 | -0.840 | 0.705 | -4871.8 | 4090.0 |
| 10 | 5803 | -0.854 | 0.729 | -4955.8 | 4232.3 |
| 11 | 5518 | -0.836 | 0.699 | -4612.5 | 3855.6 |
| 12 | 5518 | -0.783 | 0.613 | -4320.0 | 3382.1 |
| 13 | 5518 | -0.699 | 0.488 | -3856.1 | 2694.7 |
| 14 | 5518 | -0.588 | 0.345 | -3243.1 | 1906.1 |
| 15 | 5518 | -0.455 | 0.207 | -2508.3 | 1140.1 |
| 16 | 5260 | -0.299 | 0.089 | -1571.6 | 469.6 |
| 17 | 5260 | -0.128 | 0.016 | -675.0 | 86.6 |
| 18 | 5260 | 0.049 | 0.002 | 258.3 | 12.7 |
| 19 | 5260 | 0.227 | 0.051 | 1193.4 | 270.8 |
| 20 | 5260 | 0.400 | 0.160 | 2101.9 | 839.9 |
| 21 | 4923 | 0.571 | 0.326 | 2810.0 | 1603.9 |
| 22 | 4923 | 0.720 | 0.519 | 3546.9 | 2555.4 |
| 23 | 4923 | 0.843 | 0.711 | 4151.9 | 3501.5 |
| 24 | 4923 | 0.936 | 0.876 | 4608.2 | 4313.5 |
| 25 | 4540 | 1.000 | 1.000 | 4540.0 | 4540.0 |
|  | 137717 |  | $\Sigma$ | -28001 | 51559 |
| $\beta_{2}=\frac{-28001}{51559}=-0.54$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{-28001^{2}}{51559}=15207 \mathrm{kN}$ |  |  |  |  |  |
| $\text { Modal mass participation ratio of Mode } 2=\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \%=\frac{15207}{137717}=11 \%$ |  |  |  |  |  |

Table A. 23 M25-Mode2 Shear Distribution

| Mode2 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \boldsymbol{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |
| 1 | 1.19 | 0.026 | 6116 | 0.022 | 3.5 | 395.4 |
| 2 | 1.19 | 0.026 | 6116 | 0.073 | 11.6 | 391.9 |
| 3 | 1.19 | 0.026 | 6116 | 0.138 | 22.0 | 380.3 |
| 4 | 1.19 | 0.026 | 6116 | 0.209 | 33.2 | 358.3 |
| 5 | 1.19 | 0.026 | 6116 | 0.278 | 44.3 | 325.1 |
| 6 | 1.19 | 0.026 | 5803 | 0.342 | 51.6 | 280.8 |
| 7 | 1.19 | 0.026 | 5803 | 0.394 | 59.5 | 229.3 |
| 8 | 1.19 | 0.026 | 5803 | 0.433 | 65.3 | 169.8 |
| 9 | 1.19 | 0.026 | 5803 | 0.456 | 68.8 | 104.5 |
| 10 | 1.19 | 0.026 | 5803 | 0.464 | 70.0 | 35.8 |
| 11 | 1.19 | 0.026 | 5518 | 0.454 | 65.1 | -34.2 |
| 12 | 1.19 | 0.026 | 5518 | 0.425 | 61.0 | -99.4 |
| 13 | 1.19 | 0.026 | 5518 | 0.380 | 54.4 | -160.4 |
| 14 | 1.19 | 0.026 | 5518 | 0.319 | 45.8 | -214.8 |
| 15 | 1.19 | 0.026 | 5518 | 0.247 | 35.4 | -260.6 |
| 16 | 1.19 | 0.026 | 5260 | 0.162 | 22.2 | -296.0 |
| 17 | 1.19 | 0.026 | 5260 | 0.070 | 9.5 | -318.2 |
| 18 | 1.19 | 0.026 | 5260 | -0.027 | -3.6 | -327.7 |
| 19 | 1.19 | 0.026 | 5260 | -0.123 | -16.9 | -324.1 |
| 20 | 1.19 | 0.026 | 5260 | -0.217 | -29.7 | -307.2 |
| 21 | 1.19 | 0.026 | 4923 | -0.310 | -39.7 | -277.6 |
| 22 | 1.19 | 0.026 | 4923 | -0.391 | -50.1 | -237.9 |
| 23 | 1.19 | 0.026 | 4923 | -0.458 | -58.6 | -187.8 |
| 24 | 1.19 | 0.026 | 4923 | -0.508 | -65.1 | -129.2 |
| 25 | 1.19 | 0.026 | 4540 | -0.543 | -64.1 | -64.1 |

Table A. 24 M25-Mode 3 Modal Mass Participation Ratio

| Mode3 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W i}_{\mathbf{i}}(\mathbf{k N})$ | $\boldsymbol{\Phi}_{\mathbf{i}}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \mathrm{\Phi}_{\mathbf{i}}$ | $\mathbf{W} \mathbf{i} \Phi_{i}{ }^{2}$ |
| 1 | 6116 | 0.069 | 0.0048 | 423.0 | 29.3 |
| 2 | 6116 | 0.223 | 0.0498 | 1365.0 | 304.7 |
| 3 | 6116 | 0.407 | 0.1655 | 2488.4 | 1012.5 |
| 4 | 6116 | 0.581 | 0.3381 | 3556.3 | 2067.9 |
| 5 | 6116 | 0.721 | 0.5200 | 4410.1 | 3180.1 |
| 6 | 5803 | 0.801 | 0.6416 | 4648.4 | 3723.5 |
| 7 | 5803 | 0.802 | 0.6428 | 4652.5 | 3730.1 |
| 8 | 5803 | 0.723 | 0.5225 | 4194.8 | 3032.3 |
| 9 | 5803 | 0.572 | 0.3277 | 3321.7 | 1901.4 |
| 10 | 5803 | 0.365 | 0.1335 | 2120.6 | 775.0 |
| 11 | 5518 | 0.113 | 0.0128 | 624.9 | 70.8 |
| 12 | 5518 | -0.157 | 0.0247 | -867.3 | 136.3 |
| 13 | 5518 | -0.415 | 0.1722 | -2290.1 | 950.4 |
| 14 | 5518 | -0.634 | 0.4019 | -3498.4 | 2217.9 |
| 15 | 5518 | -0.795 | 0.6321 | -4387.0 | 3487.8 |
| 16 | 5260 | -0.877 | 0.7686 | -4611.4 | 4042.7 |
| 17 | 5260 | -0.858 | 0.7370 | -4515.6 | 3876.5 |
| 18 | 5260 | -0.746 | 0.5564 | -3923.7 | 2926.8 |
| 19 | 5260 | -0.553 | 0.3054 | -2906.8 | 1606.3 |
| 20 | 5260 | -0.300 | 0.0899 | -1576.8 | 472.7 |
| 21 | 4923 | 0.015 | 0.0002 | 74.9 | 1.1 |
| 22 | 4923 | 0.339 | 0.1146 | 1666.4 | 564.1 |
| 23 | 4923 | 0.628 | 0.3942 | 3090.9 | 1940.6 |
| 24 | 4923 | 0.852 | 0.7262 | 4195.3 | 3575.2 |
| 25 | 4540 | 1.000 | 1.0000 | 4540.0 | 4540.0 |
|  | 137717 |  | $\Sigma$ | 16796 | 50166 |
| $\beta_{3}=\frac{16796}{50166}=0.33$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 3}=\frac{16796^{2}}{50166}=5624 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $3=\frac{\mathrm{W}_{\mathrm{e} 3}}{\text { total weight }} \times 100 \%=\frac{5624}{137717}=4.1 \%$ |  |  |  |  |  |

Table A. 25 M25-Mode3 Shear Distribution

| Mode3 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \boldsymbol{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> (kN) |  |
| 1 | 0.67 | 0.047 | 6116 | 0.023 | 6.7 | 264.3 |  |
| 2 | 0.67 | 0.047 | 6116 | 0.075 | 21.5 | 257.7 |  |
| 3 | 0.67 | 0.047 | 6116 | 0.136 | 39.2 | 236.2 |  |
| 4 | 0.67 | 0.047 | 6116 | 0.195 | 56.0 | 197.0 |  |
| 5 | 0.67 | 0.047 | 6116 | 0.241 | 69.4 | 141.1 |  |
| 6 | 0.67 | 0.047 | 5803 | 0.268 | 73.1 | 71.7 |  |
| 7 | 0.67 | 0.047 | 5803 | 0.268 | 73.2 | -1.5 |  |
| 8 | 0.67 | 0.047 | 5803 | 0.242 | 66.0 | -74.7 |  |
| 9 | 0.67 | 0.047 | 5803 | 0.192 | 52.3 | -140.7 |  |
| 10 | 0.67 | 0.047 | 5803 | 0.122 | 33.4 | -193.0 |  |
| 11 | 0.67 | 0.047 | 5518 | 0.038 | 9.8 | -226.4 |  |
| 12 | 0.67 | 0.047 | 5518 | -0.053 | -13.6 | -236.2 |  |
| 13 | 0.67 | 0.047 | 5518 | -0.139 | -36.0 | -222.5 |  |
| 14 | 0.67 | 0.047 | 5518 | -0.212 | -55.1 | -186.5 |  |
| 15 | 0.67 | 0.047 | 5518 | -0.266 | -69.0 | -131.5 |  |
| 16 | 0.67 | 0.047 | 5260 | -0.294 | -72.6 | -62.4 |  |
| 17 | 0.67 | 0.047 | 5260 | -0.287 | -71.1 | 10.1 |  |
| 18 | 0.67 | 0.047 | 5260 | -0.250 | -61.7 | 81.2 |  |
| 19 | 0.67 | 0.047 | 5260 | -0.185 | -45.7 | 142.9 |  |
| 20 | 0.67 | 0.047 | 5260 | -0.100 | -24.8 | 188.7 |  |
| 21 | 0.67 | 0.047 | 4923 | 0.005 | 1.2 | 213.5 |  |
| 22 | 0.67 | 0.047 | 4923 | 0.113 | 26.2 | 212.3 |  |
| 23 | 0.67 | 0.047 | 4923 | 0.210 | 48.6 | 186.1 |  |
| 24 | 0.67 | 0.047 | 4923 | 0.285 | 66.0 | 137.5 |  |
| 25 | 0.67 | 0.047 | 4540 | 0.335 | 71.4 | 71.4 |  |

For M25 case, mode1, mode 2 and mode 3 give almost $90 \%$ of the actual mass. Thus, mode1 and mode2 and mode 3 are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table A. 26.

Table A.26 M25-Shear Force Determination Using Both Hand Calculations and SAP2000

| Level | Shear Using Hand <br> Calculations (kN) | Shear Using <br> SAP2000 (kN) | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 1010 | 1044 | $3.3 \%$ |
| 2 | 1006 | 1038 | $3.1 \%$ |

[^5]Table A.27 M30-Mode1 Modal Mass Participation Ratio

| Mode1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathrm{W}_{\mathrm{i}} \mathrm{\Phi}_{\mathrm{i}}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ |
| 1 | 6456 | -0.010 | 0.0001 | -61.5 | 0.6 |
| 2 | 6456 | -0.033 | 0.0011 | -214.0 | 7.1 |
| 3 | 6456 | -0.066 | 0.0043 | -423.7 | 27.8 |
| 4 | 6456 | -0.104 | 0.0108 | -670.3 | 69.6 |
| 5 | 6456 | -0.146 | 0.0212 | -939.4 | 136.7 |
| 6 | 6116 | -0.190 | 0.0361 | -1161.6 | 220.6 |
| 7 | 6116 | -0.236 | 0.0556 | -1442.3 | 340.1 |
| 8 | 6116 | -0.282 | 0.0797 | -1727.2 | 487.7 |
| 9 | 6116 | -0.329 | 0.1083 | -2012.5 | 662.2 |
| 10 | 6116 | -0.376 | 0.1410 | -2296.8 | 862.5 |
| 11 | 5803 | -0.422 | 0.1780 | -2448.5 | 1033.1 |
| 12 | 5803 | -0.467 | 0.2185 | -2712.5 | 1267.9 |
| 13 | 5803 | -0.512 | 0.2620 | -2970.3 | 1520.3 |
| 14 | 5803 | -0.555 | 0.3081 | -3221.3 | 1788.1 |
| 15 | 5803 | -0.597 | 0.3566 | -3465.5 | 2069.6 |
| 16 | 5518 | -0.638 | 0.4074 | -3522.1 | 2248.2 |
| 17 | 5518 | -0.678 | 0.4591 | -3738.9 | 2533.4 |
| 18 | 5518 | -0.715 | 0.5114 | -3946.0 | 2821.8 |
| 19 | 5518 | -0.751 | 0.5636 | -4142.6 | 3110.1 |
| 20 | 5518 | -0.784 | 0.6154 | -4328.7 | 3395.8 |
| 21 | 5260 | -0.817 | 0.6671 | -4296.3 | 3509.2 |
| 22 | 5260 | -0.847 | 0.7168 | -4453.2 | 3770.2 |
| 23 | 5260 | -0.874 | 0.7641 | -4597.9 | 4019.2 |
| 24 | 5260 | -0.899 | 0.8086 | -4730.0 | 4253.4 |
| 25 | 5260 | -0.922 | 0.8503 | -4850.3 | 4472.5 |
| 26 | 4923 | -0.943 | 0.8900 | -4644.4 | 4381.5 |
| 27 | 4923 | -0.962 | 0.9249 | -4734.5 | 4553.3 |
| 28 | 4923 | -0.977 | 0.9548 | -4810.4 | 4700.4 |
| 29 | 4923 | -0.990 | 0.9796 | -4872.4 | 4822.4 |
| 30 | 4540 | -1.000 | 1.0000 | -4540.0 | 4540.0 |
|  | 169997 |  | $\Sigma$ | -91975 | 67625 |
| $\beta_{1}=\frac{-91975}{67625}=-1.36$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{el}}=\frac{-91975^{2}}{6765}=125093 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $1=\frac{\mathrm{W}_{\mathrm{e} 1}}{\text { total weight }} \times 100 \%=\frac{125093}{169997}=74 \%$ |  |  |  |  |  |

Table A. 28 M30-Mode 1 Shear Distribution

|  | Mode1 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |
| 1 | 4.33 | 0.0067 | 6456 | 0.013 | 0.6 | 838.1 |
| 2 | 4.33 | 0.0067 | 6456 | 0.045 | 1.9 | 837.6 |
| 3 | 4.33 | 0.0067 | 6456 | 0.089 | 3.9 | 835.6 |
| 4 | 4.33 | 0.0067 | 6456 | 0.141 | 6.1 | 831.7 |
| 5 | 4.33 | 0.0067 | 6456 | 0.198 | 8.6 | 825.6 |
| 6 | 4.33 | 0.0067 | 6116 | 0.258 | 10.6 | 817.1 |
| 7 | 4.33 | 0.0067 | 6116 | 0.321 | 13.1 | 806.5 |
| 8 | 4.33 | 0.0067 | 6116 | 0.384 | 15.7 | 793.4 |
| 9 | 4.33 | 0.0067 | 6116 | 0.448 | 18.3 | 777.6 |
| 10 | 4.33 | 0.0067 | 6116 | 0.511 | 20.9 | 759.3 |
| 11 | 4.33 | 0.0067 | 5803 | 0.574 | 22.3 | 738.3 |
| 12 | 4.33 | 0.0067 | 5803 | 0.636 | 24.7 | 716.0 |
| 13 | 4.33 | 0.0067 | 5803 | 0.696 | 27.1 | 691.3 |
| 14 | 4.33 | 0.0067 | 5803 | 0.755 | 29.4 | 664.2 |
| 15 | 4.33 | 0.0067 | 5803 | 0.812 | 31.6 | 634.9 |
| 16 | 4.33 | 0.0067 | 5518 | 0.868 | 32.1 | 603.3 |
| 17 | 4.33 | 0.0067 | 5518 | 0.922 | 34.1 | 571.2 |
| 18 | 4.33 | 0.0067 | 5518 | 0.973 | 36.0 | 537.2 |
| 19 | 4.33 | 0.0067 | 5518 | 1.021 | 37.7 | 501.2 |
| 20 | 4.33 | 0.0067 | 5518 | 1.067 | 39.4 | 463.4 |
| 21 | 4.33 | 0.0067 | 5260 | 1.111 | 39.1 | 424.0 |
| 22 | 4.33 | 0.0067 | 5260 | 1.151 | 40.6 | 384.8 |
| 23 | 4.33 | 0.0067 | 5260 | 1.189 | 41.9 | 344.3 |
| 24 | 4.33 | 0.0067 | 5260 | 1.223 | 43.1 | 302.4 |
| 25 | 4.33 | 0.0067 | 5260 | 1.254 | 44.2 | 259.3 |
| 26 | 4.33 | 0.0067 | 4923 | 1.283 | 42.3 | 215.1 |
| 27 | 4.33 | 0.0067 | 4923 | 1.308 | 43.1 | 172.7 |
| 28 | 4.33 | 0.0067 | 4923 | 1.329 | 43.8 | 129.6 |
| 29 | 4.33 | 0.0067 | 4923 | 1.346 | 44.4 | 85.8 |
| 30 | 4.33 | 0.0067 | 4540 | 1.360 | 41.4 | 41.4 |

Table A. 29 M30-Mode2 Modal Mass Participation Ratio

| Mode 2 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathrm{W}_{\mathrm{i}}(\mathrm{kN})$ | $\Phi_{\text {i }}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W i}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}^{\mathbf{i}}{ }^{2}$ |
| 1 | 6456 | 0.029 | 0.0008 | 187.9 | 5.5 |
| 2 | 6456 | 0.099 | 0.0098 | 640.4 | 63.5 |
| 3 | 6456 | 0.193 | 0.0372 | 1244.5 | 239.9 |
| 4 | 6456 | 0.298 | 0.0887 | 1922.8 | 572.7 |
| 5 | 6456 | 0.406 | 0.1650 | 2622.1 | 1065.0 |
| 6 | 6116 | 0.512 | 0.2619 | 3130.0 | 1601.9 |
| 7 | 6116 | 0.608 | 0.3698 | 3719.3 | 2261.8 |
| 8 | 6116 | 0.691 | 0.4778 | 4227.5 | 2922.2 |
| 9 | 6116 | 0.758 | 0.5751 | 4638.3 | 3517.6 |
| 10 | 6116 | 0.808 | 0.6533 | 4943.3 | 3995.5 |
| 11 | 5803 | 0.838 | 0.7018 | 4861.5 | 4072.8 |
| 12 | 5803 | 0.843 | 0.7114 | 4894.7 | 4128.5 |
| 13 | 5803 | 0.826 | 0.6827 | 4794.8 | 3961.8 |
| 14 | 5803 | 0.787 | 0.6200 | 4569.3 | 3597.9 |
| 15 | 5803 | 0.729 | 0.5313 | 4229.7 | 3083.0 |
| 16 | 5518 | 0.649 | 0.4215 | 3582.3 | 2325.6 |
| 17 | 5518 | 0.550 | 0.3021 | 3033.1 | 1667.2 |
| 18 | 5518 | 0.434 | 0.1886 | 2396.5 | 1040.8 |
| 19 | 5518 | 0.307 | 0.0940 | 1691.6 | 518.6 |
| 20 | 5518 | 0.170 | 0.0289 | 937.7 | 159.3 |
| 21 | 5260 | 0.024 | 0.0006 | 126.4 | 3.0 |
| 22 | 5260 | -0.125 | 0.0155 | -655.3 | 81.6 |
| 23 | 5260 | -0.271 | 0.0737 | -1427.7 | 387.5 |
| 24 | 5260 | -0.413 | 0.1706 | -2172.7 | 897.4 |
| 25 | 5260 | -0.547 | 0.2992 | -2877.1 | 1573.7 |
| 26 | 4923 | -0.676 | 0.4575 | -3329.9 | 2252.3 |
| 27 | 4923 | -0.788 | 0.6209 | -3879.1 | 3056.6 |
| 28 | 4923 | -0.880 | 0.7736 | -4330.0 | 3808.5 |
| 29 | 4923 | -0.950 | 0.9020 | -4675.5 | 4440.4 |
| 30 | 4540 | -1.000 | 1.0000 | -4540.0 | 4540.0 |
|  | 169997 |  | $\Sigma$ | 34506 | 61842 |
| $\beta_{2}=\frac{34506}{61842}=0.56$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 2}=\frac{34506^{2}}{61842}=19254 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $2=\frac{\mathrm{W}_{\mathrm{e} 2}}{\text { total weight }} \times 100 \%=\frac{19254}{169997}=11.3 \%$ |  |  |  |  |  |

Table A. 30 M30-Mode2 Shear Distribution

| Mevel |  |  |  |  |  |  |  | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 1.46 | 0.021 | 6456 | 0.016 | 2.2 | 404.3 |  |  |  |  |  |  |  |
| 2 | 1.46 | 0.021 | 6456 | 0.055 | 7.5 | 402.1 |  |  |  |  |  |  |  |
| 3 | 1.46 | 0.021 | 6456 | 0.108 | 14.6 | 394.6 |  |  |  |  |  |  |  |
| 4 | 1.46 | 0.021 | 6456 | 0.166 | 22.5 | 380.0 |  |  |  |  |  |  |  |
| 5 | 1.46 | 0.021 | 6456 | 0.227 | 30.7 | 357.5 |  |  |  |  |  |  |  |
| 6 | 1.46 | 0.021 | 6116 | 0.286 | 36.7 | 326.8 |  |  |  |  |  |  |  |
| 7 | 1.46 | 0.021 | 6116 | 0.339 | 43.6 | 290.1 |  |  |  |  |  |  |  |
| 8 | 1.46 | 0.021 | 6116 | 0.386 | 49.5 | 246.5 |  |  |  |  |  |  |  |
| 9 | 1.46 | 0.021 | 6116 | 0.423 | 54.3 | 197.0 |  |  |  |  |  |  |  |
| 10 | 1.46 | 0.021 | 6116 | 0.451 | 57.9 | 142.6 |  |  |  |  |  |  |  |
| 11 | 1.46 | 0.021 | 5803 | 0.467 | 57.0 | 84.7 |  |  |  |  |  |  |  |
| 12 | 1.46 | 0.021 | 5803 | 0.471 | 57.4 | 27.8 |  |  |  |  |  |  |  |
| 13 | 1.46 | 0.021 | 5803 | 0.461 | 56.2 | -29.6 |  |  |  |  |  |  |  |
| 14 | 1.46 | 0.021 | 5803 | 0.439 | 53.5 | -85.8 |  |  |  |  |  |  |  |
| 15 | 1.46 | 0.021 | 5803 | 0.407 | 49.6 | -139.3 |  |  |  |  |  |  |  |
| 16 | 1.46 | 0.021 | 5518 | 0.362 | 42.0 | -188.9 |  |  |  |  |  |  |  |
| 17 | 1.46 | 0.021 | 5518 | 0.307 | 35.5 | -230.9 |  |  |  |  |  |  |  |
| 18 | 1.46 | 0.021 | 5518 | 0.242 | 28.1 | -266.4 |  |  |  |  |  |  |  |
| 19 | 1.46 | 0.021 | 5518 | 0.171 | 19.8 | -294.5 |  |  |  |  |  |  |  |
| 20 | 1.46 | 0.021 | 5518 | 0.095 | 11.0 | -314.3 |  |  |  |  |  |  |  |
| 21 | 1.46 | 0.021 | 5260 | 0.013 | 1.5 | -325.3 |  |  |  |  |  |  |  |
| 22 | 1.46 | 0.021 | 5260 | -0.070 | -7.7 | -326.8 |  |  |  |  |  |  |  |
| 23 | 1.46 | 0.021 | 5260 | -0.151 | -16.7 | -319.1 |  |  |  |  |  |  |  |
| 24 | 1.46 | 0.021 | 5260 | -0.230 | -25.5 | -302.4 |  |  |  |  |  |  |  |
| 25 | 1.46 | 0.021 | 5260 | -0.305 | -33.7 | -276.9 |  |  |  |  |  |  |  |
| 26 | 1.46 | 0.021 | 4923 | -0.377 | -39.0 | -243.2 |  |  |  |  |  |  |  |
| 27 | 1.46 | 0.021 | 4923 | -0.440 | -45.5 | -204.2 |  |  |  |  |  |  |  |
| 28 | 1.46 | 0.021 | 4923 | -0.491 | -50.7 | -158.7 |  |  |  |  |  |  |  |
| 29 | 1.46 | 0.021 | 4923 | -0.530 | -54.8 | -108.0 |  |  |  |  |  |  |  |
| 30 | 1.46 | 0.021 | 4540 | -0.558 | -53.2 | -53.2 |  |  |  |  |  |  |  |

Table A. 31 M30-Mode3 Modal Mass Participation Ratio

| Mode3 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{W}_{\mathbf{i}}(\mathrm{kN})$ | $\Phi_{i}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W}_{\mathrm{i}} \boldsymbol{\Phi}_{\mathrm{i}}$ | $\mathbf{W i}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}{ }^{\text {a }}$ |
| 1 | 6456 | 0.050 | 0.0025 | 321.2 | 16.0 |
| 2 | 6456 | 0.166 | 0.0275 | 1071.0 | 177.7 |
| 3 | 6456 | 0.313 | 0.0981 | 2021.6 | 633.0 |
| 4 | 6456 | 0.466 | 0.2171 | 3008.4 | 1401.9 |
| 5 | 6456 | 0.606 | 0.3670 | 3911.0 | 2369.3 |
| 6 | 6116 | 0.716 | 0.5121 | 4376.7 | 3132.0 |
| 7 | 6116 | 0.780 | 0.6079 | 4768.4 | 3717.7 |
| 8 | 6116 | 0.792 | 0.6274 | 4844.4 | 3837.1 |
| 9 | 6116 | 0.752 | 0.5649 | 4596.9 | 3455.1 |
| 10 | 6116 | 0.662 | 0.4381 | 4048.0 | 2679.3 |
| 11 | 5803 | 0.522 | 0.2726 | 3029.5 | 1581.6 |
| 12 | 5803 | 0.339 | 0.1150 | 1968.0 | 667.4 |
| 13 | 5803 | 0.129 | 0.0165 | 746.1 | 95.9 |
| 14 | 5803 | -0.094 | 0.0088 | -545.7 | 51.3 |
| 15 | 5803 | -0.314 | 0.0985 | -1821.6 | 571.8 |
| 16 | 5518 | -0.517 | 0.2676 | -2854.5 | 1476.6 |
| 17 | 5518 | -0.683 | 0.4667 | -3769.5 | 2575.0 |
| 18 | 5518 | -0.799 | 0.6391 | -4411.2 | 3526.4 |
| 19 | 5518 | -0.859 | 0.7378 | -4739.6 | 4071.1 |
| 20 | 5518 | -0.860 | 0.7398 | -4746.1 | 4082.2 |
| 21 | 5260 | -0.795 | 0.6322 | -4182.4 | 3325.5 |
| 22 | 5260 | -0.664 | 0.4407 | -3491.7 | 2317.9 |
| 23 | 5260 | -0.480 | 0.2301 | -2523.4 | 1210.5 |
| 24 | 5260 | -0.257 | 0.0662 | -1353.7 | 348.4 |
| 25 | 5260 | -0.013 | 0.0002 | -67.4 | 0.9 |
| 26 | 4923 | 0.257 | 0.0659 | 1263.5 | 324.3 |
| 27 | 4923 | 0.511 | 0.2607 | 2513.6 | 1283.4 |
| 28 | 4923 | 0.727 | 0.5289 | 3580.3 | 2603.7 |
| 29 | 4923 | 0.891 | 0.7947 | 4388.5 | 3912.1 |
| 30 | 4540 | 1.000 | 1.0000 | 4540.0 | 4540.0 |
|  | 169997 |  | $\Sigma$ | 20490 | 59985 |
| $\beta_{3}=\frac{20490}{59985}=0.34$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 3}=\frac{20490^{2}}{59985}=6999 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $3=\frac{\mathrm{W}_{\mathrm{e} 3}}{\text { total weight }} \times 100 \%=\frac{6999}{169997}=4.1 \%$ |  |  |  |  |  |

Table A. 32 M30-Mode3 Shear Distribution

| Mode3 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $\mathbf{( k N})$ |
| 1 | 0.82 | 0.038 | 6456 | 0.017 | 4.2 | 266.0 |
| 2 | 0.82 | 0.038 | 6456 | 0.057 | 13.9 | 261.8 |
| 3 | 0.82 | 0.038 | 6456 | 0.107 | 26.2 | 247.9 |
| 4 | 0.82 | 0.038 | 6456 | 0.159 | 39.1 | 221.7 |
| 5 | 0.82 | 0.038 | 6456 | 0.207 | 50.8 | 182.6 |
| 6 | 0.82 | 0.038 | 6116 | 0.244 | 56.8 | 131.8 |
| 7 | 0.82 | 0.038 | 6116 | 0.266 | 61.9 | 75.0 |
| 8 | 0.82 | 0.038 | 6116 | 0.271 | 62.9 | 13.1 |
| 9 | 0.82 | 0.038 | 6116 | 0.257 | 59.7 | -49.7 |
| 10 | 0.82 | 0.038 | 6116 | 0.226 | 52.5 | -109.4 |
| 11 | 0.82 | 0.038 | 5803 | 0.178 | 39.3 | -162.0 |
| 12 | 0.82 | 0.038 | 5803 | 0.116 | 25.5 | -201.3 |
| 13 | 0.82 | 0.038 | 5803 | 0.044 | 9.7 | -226.8 |
| 14 | 0.82 | 0.038 | 5803 | -0.032 | -7.1 | -236.5 |
| 15 | 0.82 | 0.038 | 5803 | -0.107 | -23.6 | -229.4 |
| 16 | 0.82 | 0.038 | 5518 | -0.177 | -37.1 | -205.8 |
| 17 | 0.82 | 0.038 | 5518 | -0.233 | -48.9 | -168.7 |
| 18 | 0.82 | 0.038 | 5518 | -0.273 | -57.3 | -119.8 |
| 19 | 0.82 | 0.038 | 5518 | -0.293 | -61.5 | -62.5 |
| 20 | 0.82 | 0.038 | 5518 | -0.294 | -61.6 | -1.0 |
| 21 | 0.82 | 0.038 | 5260 | -0.272 | -54.3 | 60.6 |
| 22 | 0.82 | 0.038 | 5260 | -0.227 | -45.3 | 114.9 |
| 23 | 0.82 | 0.038 | 5260 | -0.164 | -32.8 | 160.2 |
| 24 | 0.82 | 0.038 | 5260 | -0.088 | -17.6 | 193.0 |
| 25 | 0.82 | 0.038 | 5260 | -0.004 | -0.9 | 210.5 |
| 26 | 0.82 | 0.038 | 4923 | 0.088 | 16.4 | 211.4 |
| 27 | 0.82 | 0.038 | 4923 | 0.174 | 32.6 | 195.0 |
| 28 | 0.82 | 0.038 | 4923 | 0.248 | 46.5 | 162.4 |
| 29 | 0.82 | 0.038 | 4923 | 0.305 | 57.0 | 115.9 |
| 30 | 0.82 | 0.038 | 4540 | 0.342 | 58.9 | 58.9 |

Table A. 33 M30-Mode4 Modal Mass Participation Ratio

| Mode4 |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathrm{W}_{\mathrm{i}}(\mathrm{kN})$ | $\boldsymbol{\Phi}_{\mathbf{i}}$ | $\boldsymbol{\Phi}_{\mathbf{i}}{ }^{2}$ | $\mathbf{W}_{\mathbf{i}} \boldsymbol{\Phi}_{\mathbf{i}}$ | $\mathbf{W i m i}^{\mathbf{2}}$ |
| 1 | 6456 | 0.071 | 0.0051 | 460.1 | 32.8 |
| 2 | 6456 | 0.230 | 0.0530 | 1486.0 | 342.0 |
| 3 | 6456 | 0.417 | 0.1736 | 2689.8 | 1120.7 |
| 4 | 6456 | 0.586 | 0.3433 | 3782.6 | 2216.2 |
| 5 | 6456 | 0.707 | 0.4996 | 4563.5 | 3225.7 |
| 6 | 6116 | 0.750 | 0.5629 | 4588.5 | 3442.5 |
| 7 | 6116 | 0.697 | 0.4854 | 4261.1 | 2968.8 |
| 8 | 6116 | 0.552 | 0.3051 | 3378.1 | 1865.9 |
| 9 | 6116 | 0.335 | 0.1120 | 2047.0 | 685.1 |
| 10 | 6116 | 0.070 | 0.0049 | 429.8 | 30.2 |
| 11 | 5803 | -0.213 | 0.0456 | -1238.5 | 264.3 |
| 12 | 5803 | -0.475 | 0.2256 | -2756.4 | 1309.3 |
| 13 | 5803 | -0.677 | 0.4585 | -3929.4 | 2660.8 |
| 14 | 5803 | -0.793 | 0.6286 | -4601.0 | 3648.0 |
| 15 | 5803 | -0.810 | 0.6560 | -4700.0 | 3806.6 |
| 16 | 5518 | -0.712 | 0.5075 | -3931.0 | 2800.4 |
| 17 | 5518 | -0.504 | 0.2541 | -2781.7 | 1402.3 |
| 18 | 5518 | -0.219 | 0.0481 | -1210.7 | 265.7 |
| 19 | 5518 | 0.101 | 0.0102 | 557.3 | 56.3 |
| 20 | 5518 | 0.415 | 0.1723 | 2290.5 | 950.8 |
| 21 | 5260 | 0.682 | 0.4657 | 3589.6 | 2449.7 |
| 22 | 5260 | 0.848 | 0.7193 | 4461.0 | 3783.3 |
| 23 | 5260 | 0.888 | 0.7887 | 4671.4 | 4148.7 |
| 24 | 5260 | 0.798 | 0.6371 | 4198.4 | 3351.0 |
| 25 | 5260 | 0.595 | 0.3541 | 3129.9 | 1862.4 |
| 26 | 4923 | 0.270 | 0.0727 | 1327.8 | 358.1 |
| 27 | 4923 | -0.116 | 0.0135 | -571.7 | 66.4 |
| 28 | 4923 | -0.490 | 0.2405 | -2414.1 | 1183.8 |
| 29 | 4923 | -0.795 | 0.6318 | -3913.1 | 3110.4 |
| 30 | 4540 | -1.000 | 1.0000 | -4540.0 | 4540.0 |
|  | 169997 |  | $\Sigma$ | 15325 | 57948 |
| $\beta_{4}=\frac{15325}{57948}=0.26$ |  |  |  |  |  |
| $\mathrm{W}_{\mathrm{e} 4}=\frac{15325^{2}}{57948}=4053 \mathrm{kN}$ |  |  |  |  |  |
| Modal mass participation ratio of Mode $4=\frac{\mathrm{W}_{\mathrm{e} 3}}{\text { total weight }} \times 100 \%=\frac{4053}{169997}=2.4 \%$ |  |  |  |  |  |

Table A. 34 M30-Mode4 Shear Distribution

| Mode4 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Level | $\mathbf{T}$ <br> (second) | $\mathbf{a}$ | $\mathbf{W}(\mathbf{k N})$ | $\boldsymbol{\beta} \mathbf{\Phi}_{\mathbf{i}}$ | Force <br> $(\mathbf{k N})$ | Shear <br> $(\mathbf{k N})$ |
| 1 | 0.56 | 0.055 | 6456 | 0.019 | 6.7 | 222.9 |
| 2 | 0.56 | 0.055 | 6456 | 0.061 | 21.6 | 216.2 |
| 3 | 0.56 | 0.055 | 6456 | 0.110 | 39.1 | 194.6 |
| 4 | 0.56 | 0.055 | 6456 | 0.155 | 55.0 | 155.5 |
| 5 | 0.56 | 0.055 | 6456 | 0.187 | 66.4 | 100.5 |
| 6 | 0.56 | 0.055 | 6116 | 0.198 | 66.7 | 34.1 |
| 7 | 0.56 | 0.055 | 6116 | 0.184 | 62.0 | -32.7 |
| 8 | 0.56 | 0.055 | 6116 | 0.146 | 49.1 | -94.6 |
| 9 | 0.56 | 0.055 | 6116 | 0.089 | 29.8 | -143.8 |
| 10 | 0.56 | 0.055 | 6116 | 0.019 | 6.3 | -173.6 |
| 11 | 0.56 | 0.055 | 5803 | -0.056 | -18.0 | -179.8 |
| 12 | 0.56 | 0.055 | 5803 | -0.126 | -40.1 | -161.8 |
| 13 | 0.56 | 0.055 | 5803 | -0.179 | -57.2 | -121.7 |
| 14 | 0.56 | 0.055 | 5803 | -0.210 | -66.9 | -64.5 |
| 15 | 0.56 | 0.055 | 5803 | -0.214 | -68.4 | 2.4 |
| 16 | 0.56 | 0.055 | 5518 | -0.188 | -57.2 | 70.7 |
| 17 | 0.56 | 0.055 | 5518 | -0.133 | -40.5 | 127.9 |
| 18 | 0.56 | 0.055 | 5518 | -0.058 | -17.6 | 168.4 |
| 19 | 0.56 | 0.055 | 5518 | 0.027 | 8.1 | 186.0 |
| 20 | 0.56 | 0.055 | 5518 | 0.110 | 33.3 | 177.9 |
| 21 | 0.56 | 0.055 | 5260 | 0.180 | 52.2 | 144.6 |
| 22 | 0.56 | 0.055 | 5260 | 0.224 | 64.9 | 92.4 |
| 23 | 0.56 | 0.055 | 5260 | 0.235 | 67.9 | 27.5 |
| 24 | 0.56 | 0.055 | 5260 | 0.211 | 61.1 | -40.5 |
| 25 | 0.56 | 0.055 | 5260 | 0.157 | 45.5 | -101.5 |
| 26 | 0.56 | 0.055 | 4923 | 0.071 | 19.3 | -147.1 |
| 27 | 0.56 | 0.055 | 4923 | -0.031 | -8.3 | -166.4 |
| 28 | 0.56 | 0.055 | 4923 | -0.130 | -35.1 | -158.1 |
| 29 | 0.56 | 0.055 | 4923 | -0.210 | -56.9 | -123.0 |
| 30 | 0.56 | 0.055 | 4540 | -0.264 | -66.0 | -66.0 |

For M30 case, mode1, mode 2, mode 3 and mode 4 give $91.8 \%$ of the actual mass. Thus, mode1, mode2, mode 3 and mode 4 are sufficient for shear calculations. The difference between hand calculations and SAP2000 is carried out for the first and second storeys as shown in Table A. 35.

Table A. 35 M30-Shear Force Determination Using Both Hand Calculations and SAP2000

| Level | Shear Using Hand <br> Calculations (kN) | Shear Using <br> SAP2000 (kN) | Percent of <br> Error* |
| :---: | :---: | :---: | :---: |
| 1 | 993 | 1005 | $1.2 \%$ |
| 2 | 989 | 1000 | $1.1 \%$ |

[^6]
## Appendix B

## First and Second-Order Moments in All Columns

Table B. 1 M10 First and Second-Order Moments

| M10 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathbf{M}_{\text {SD }+\mathrm{D}+\mathrm{L}}$ | $\begin{gathered} \text { M }_{\text {Positive }} \\ \text { EQ. } \end{gathered}$ | $\begin{gathered} \mathbf{M}_{\text {Negative }} \\ \text { EQ. } \end{gathered}$ | First-Order Moment | SecondOrder <br> Moment | Percent |
| 1 | 28.5 | 227.2 | -227.2 | 255.6 | 260.3 | 1.85 |
| 2 | 1.5 | 241.2 | -241.2 | 242.7 | 247.6 | 2.03 |
| 3 | -1.5 | 241.2 | -241.2 | -242.7 | -247.6 | 2.03 |
| 4 | -28.5 | 227.2 | -227.2 | -255.6 | -260.3 | 1.85 |
| 5 | 47.4 | 228.1 | -228.1 | 275.5 | 280.0 | 1.65 |
| 6 | 1.9 | 242.2 | -242.2 | 244.1 | 248.7 | 1.89 |
| 7 | -1.9 | 242.2 | -242.2 | -244.1 | -248.7 | 1.89 |
| 8 | -47.4 | 228.1 | -228.1 | -275.5 | -280.0 | 1.65 |
| 9 | 47.4 | 228.1 | -228.1 | 275.5 | 280.0 | 1.65 |
| 10 | 1.9 | 242.2 | -242.2 | 244.1 | 248.7 | 1.89 |
| 11 | -1.9 | 242.2 | -242.2 | -244.1 | -248.7 | 1.89 |
| 12 | -47.4 | 228.1 | -228.1 | -275.5 | -280.0 | 1.65 |
| 13 | 28.5 | 227.2 | -227.2 | 255.6 | 260.3 | 1.85 |
| 14 | 1.5 | 241.2 | -241.2 | 242.7 | 247.6 | 2.03 |
| 15 | -1.5 | 241.2 | -241.2 | -242.7 | -247.6 | 2.03 |
| 16 | -28.5 | 227.2 | -227.2 | -255.6 | -260.3 | 1.85 |
| 17 | 73.9 | 123.5 | -123.5 | 197.4 | 201.4 | 2.03 |
| 18 | 1.5 | 167.1 | -167.1 | 168.7 | 173.8 | 3.05 |
| 19 | -1.5 | 167.1 | -167.1 | -168.7 | -173.8 | 3.05 |
| 20 | -73.9 | 123.5 | -123.5 | -197.4 | -201.4 | 2.03 |
| 21 | 128.4 | 124.4 | -124.4 | 252.8 | 256.8 | 1.59 |
| 22 | 3.0 | 169.5 | -169.5 | 172.5 | 177.6 | 2.97 |
| 23 | -3.0 | 169.5 | -169.5 | -172.5 | -177.6 | 2.97 |
| 24 | -128.4 | 124.4 | -124.4 | -252.8 | -256.8 | 1.59 |
| 25 | 128.4 | 124.4 | -124.4 | 252.8 | 256.8 | 1.59 |
| 26 | 3.0 | 169.5 | -169.5 | 172.5 | 177.6 | 2.97 |
| 27 | -3.0 | 169.5 | -169.5 | -172.5 | -177.6 | 2.97 |
| 28 | -128.4 | 124.4 | -124.4 | -252.8 | -256.8 | 1.59 |
| 29 | 73.9 | 123.5 | -123.5 | 197.4 | 201.4 | 2.03 |
| 30 | 1.5 | 167.1 | -167.1 | 168.7 | 173.8 | 3.05 |
| 31 | -1.5 | 167.1 | -167.1 | -168.7 | -173.8 | $\underline{3.05}$ |
| 32 | -73.9 | 123.5 | -123.5 | -197.4 | -201.4 | 2.03 |
| 33 | 70.7 | 76.2 | -76.2 | 146.8 | 149.4 | 1.73 |
| 34 | 6.0 | 128.4 | -128.4 | 134.4 | 138.5 | 3.04 |
| 35 | -6.0 | 128.4 | -128.4 | -134.4 | -138.5 | 3.04 |
| 36 | -70.7 | 76.2 | -76.2 | -146.8 | -149.4 | 1.73 |
| 37 | 121.0 | 77.6 | -77.6 | 198.6 | 201.1 | 1.28 |
| 38 | 9.8 | 131.0 | -131.0 | 140.9 | 145.0 | 2.93 |
| 39 | -9.8 | 131.0 | -131.0 | -140.9 | -145.0 | 2.93 |
| 40 | -121.0 | 77.6 | -77.6 | -198.6 | -201.1 | 1.28 |
| 41 | 121.0 | 77.6 | -77.6 | 198.6 | 201.1 | 1.28 |
| 42 | 9.8 | 131.0 | -131.0 | 140.9 | 145.0 | 2.93 |

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| 43 | -9.8 | 131.0 | -131.0 | -140.9 | -145.0 | 2.93 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -121.0 | 77.6 | -77.6 | -198.6 | -201.1 | 1.28 |
| 45 | 70.7 | 76.2 | -76.2 | 146.8 | 149.4 | 1.73 |
| 46 | 6.0 | 128.4 | -128.4 | 134.4 | 138.5 | 3.04 |
| 47 | -6.0 | 128.4 | -128.4 | -134.4 | -138.5 | 3.04 |
| 48 | -70.7 | 76.2 | -76.2 | -146.8 | -149.4 | 1.73 |
| 49 | 73.3 | 53.6 | -53.6 | 126.9 | 128.3 | 1.13 |
| 50 | 7.9 | 106.4 | -106.4 | 114.3 | 117.3 | 2.62 |
| 51 | -7.9 | 106.4 | -106.4 | -114.3 | -117.3 | 2.62 |
| 52 | -73.3 | 53.6 | -53.6 | -126.9 | -128.3 | 1.13 |
| 53 | 125.7 | 55.0 | -55.0 | 180.7 | 182.2 | 0.82 |
| 54 | 13.1 | 109.1 | -109.1 | 122.2 | 125.2 | 2.52 |
| 55 | -13.1 | 109.1 | -109.1 | -122.2 | -125.2 | 2.52 |
| 56 | -125.7 | 55.0 | -55.0 | -180.7 | -182.2 | 0.82 |
| 57 | 125.7 | 55.0 | -55.0 | 180.7 | 182.2 | 0.82 |
| 58 | 13.1 | 109.1 | -109.1 | 122.2 | 125.2 | 2.52 |
| 59 | -13.1 | 109.1 | -109.1 | -122.2 | -125.2 | 2.52 |
| 60 | -125.7 | 55.0 | -55.0 | -180.7 | -182.2 | 0.82 |
| 61 | 73.3 | 53.6 | -53.6 | 126.9 | 128.3 | 1.13 |
| 62 | 7.9 | 106.4 | -106.4 | 114.3 | 117.3 | 2.62 |
| 63 | -7.9 | 106.4 | -106.4 | -114.3 | -117.3 | 2.62 |
| 64 | -73.3 | 53.6 | -53.6 | -126.9 | -128.3 | 1.13 |
| 65 | 77.2 | 41.4 | -41.4 | 118.6 | 119.4 | 0.66 |
| 66 | 9.8 | 94.3 | -94.3 | 104.1 | 106.4 | 2.18 |
| 67 | -9.8 | 94.3 | -94.3 | -104.1 | -106.4 | 2.18 |
| 68 | -77.2 | 41.4 | -41.4 | -118.6 | -119.4 | 0.66 |
| 69 | 133.1 | 42.7 | -42.7 | 175.8 | 176.6 | 0.47 |
| 70 | 16.6 | 97.0 | -97.0 | 113.6 | 115.9 | 2.07 |
| 71 | -16.6 | 97.0 | -97.0 | -113.6 | -115.9 | 2.07 |
| 72 | -133.1 | 42.7 | -42.7 | -175.8 | -176.6 | 0.47 |
| 73 | 133.1 | 42.7 | -42.7 | 175.8 | 176.6 | 0.47 |
| 74 | 16.6 | 97.0 | -97.0 | 113.6 | 115.9 | 2.07 |
| 75 | -16.6 | 97.0 | -97.0 | -113.6 | -115.9 | 2.07 |
| 76 | -133.1 | 42.7 | -42.7 | -175.8 | -176.6 | 0.47 |
| 77 | 77.2 | 41.4 | -41.4 | 118.6 | 119.4 | 0.66 |
| 78 | 9.8 | 94.3 | -94.3 | 104.1 | 106.4 | 2.18 |
| 79 | -9.8 | 94.3 | -94.3 | -104.1 | -106.4 | 2.18 |
| 80 | -77.2 | 41.4 | -41.4 | -118.6 | -119.4 | 0.66 |
| 81 | 62.6 | 43.1 | -43.1 | 105.7 | 106.5 | 0.74 |
| 82 | 9.2 | 80.4 | -80.4 | 89.7 | 91.4 | 1.92 |
| 83 | -9.2 | 80.4 | -80.4 | -89.7 | -91.4 | 1.92 |
| 84 | -62.6 | 43.1 | -43.1 | -105.7 | -106.5 | 0.74 |
| 85 | 105.2 | 44.2 | -44.2 | 149.4 | 150.2 | 0.55 |
| 86 | 14.8 | 82.2 | -82.2 | 97.1 | 98.9 | 1.84 |
| 87 | -14.8 | 82.2 | -82.2 | -97.1 | -98.9 | 1.84 |
| 88 | -105.2 | 44.2 | -44.2 | -149.4 | -150.2 | 0.55 |
| 89 | 105.2 | 44.2 | -44.2 | 149.4 | 150.2 | 0.55 |

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| 90 | 14.8 | 82.2 | -82.2 | 97.1 | 98.9 | 1.84 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 91 | -14.8 | 82.2 | -82.2 | -97.1 | -98.9 | 1.84 |
| 92 | -105.2 | 44.2 | -44.2 | -149.4 | -150.2 | 0.55 |
| 93 | 62.6 | 43.1 | -43.1 | 105.7 | 106.5 | 0.74 |
| 94 | 9.2 | 80.4 | -80.4 | 89.7 | 91.4 | 1.92 |
| 95 | -9.2 | 80.4 | -80.4 | -89.7 | -91.4 | 1.92 |
| 96 | -62.6 | 43.1 | -43.1 | -105.7 | -106.5 | 0.74 |
| 97 | 79.5 | 31.8 | -31.8 | 111.3 | 111.6 | 0.26 |
| 98 | 11.9 | 74.3 | -74.3 | 86.2 | 87.4 | 1.40 |
| 99 | -11.9 | 74.3 | -74.3 | -86.2 | -87.4 | 1.40 |
| 100 | -79.5 | 31.8 | -31.8 | -111.3 | -111.6 | 0.26 |
| 101 | 133.8 | 32.8 | -32.8 | 166.6 | 166.9 | 0.19 |
| 102 | 19.6 | 76.2 | -76.2 | 95.8 | 97.1 | 1.32 |
| 103 | -19.6 | 76.2 | -76.2 | -95.8 | -97.1 | 1.32 |
| 104 | -133.8 | 32.8 | -32.8 | -166.6 | -166.9 | 0.19 |
| 105 | 133.8 | 32.8 | -32.8 | 166.6 | 166.9 | 0.19 |
| 106 | 19.6 | 76.2 | -76.2 | 95.8 | 97.1 | 1.32 |
| 107 | -19.6 | 76.2 | -76.2 | -95.8 | -97.1 | 1.32 |
| 108 | -133.8 | 32.8 | -32.8 | -166.6 | -166.9 | 0.19 |
| 109 | 79.5 | 31.8 | -31.8 | 111.3 | 111.6 | 0.26 |
| 110 | 11.9 | 74.3 | -74.3 | 86.2 | 87.4 | 1.40 |
| 111 | -11.9 | 74.3 | -74.3 | -86.2 | -87.4 | 1.40 |
| 112 | -79.5 | 31.8 | -31.8 | -111.3 | -111.6 | 0.26 |
| 113 | 78.9 | 23.8 | -23.8 | 102.7 | 102.8 | 0.06 |
| 114 | 13.5 | 58.8 | -58.8 | 72.3 | 73.0 | 0.91 |
| 115 | -13.5 | 58.8 | -58.8 | -72.3 | -73.0 | 0.91 |
| 116 | -78.9 | 23.8 | -23.8 | -102.7 | -102.8 | 0.06 |
| 117 | 132.6 | 24.7 | -24.7 | 157.3 | 157.4 | 0.05 |
| 118 | 22.1 | 60.4 | -60.4 | 82.5 | 83.2 | 0.84 |
| 119 | -22.1 | 60.4 | -60.4 | -82.5 | -83.2 | 0.84 |
| 120 | -132.6 | 24.7 | -24.7 | -157.3 | -157.4 | 0.05 |
| 121 | 132.6 | 24.7 | -24.7 | 157.3 | 157.4 | 0.05 |
| 122 | 22.1 | 60.4 | -60.4 | 82.5 | 83.2 | 0.84 |
| 123 | -22.1 | 60.4 | -60.4 | -82.5 | -83.2 | 0.84 |
| 124 | -132.6 | 24.7 | -24.7 | -157.3 | -157.4 | 0.05 |
| 125 | 78.9 | 23.8 | -23.8 | 102.7 | 102.8 | 0.06 |
| 126 | 13.5 | 58.8 | -58.8 | 72.3 | 73.0 | 0.91 |
| 127 | -13.5 | 58.8 | -58.8 | -72.3 | -73.0 | 0.91 |
| 128 | -78.9 | 23.8 | -23.8 | -102.7 | -102.8 | 0.06 |
| 129 | 78.7 | 12.1 | -12.1 | 90.8 | 90.7 | -0.13 |
| 130 | 15.3 | 39.7 | -39.7 | 54.9 | 55.2 | 0.47 |
| 131 | -15.3 | 39.7 | -39.7 | -54.9 | -55.2 | 0.47 |
| 132 | -78.7 | 12.1 | -12.1 | -90.8 | -90.7 | -0.13 |
| 133 | 131.5 | 12.9 | -12.9 | 144.4 | 144.3 | -0.08 |
| 134 | 24.5 | 41.0 | -41.0 | 65.5 | 65.7 | 0.43 |
| 135 | -24.5 | 41.0 | -41.0 | -65.5 | -65.7 | 0.43 |
| 136 | -131.5 | 12.9 | -12.9 | -144.4 | -144.3 | -0.08 |

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| 137 | 131.5 | 12.9 | -12.9 | 144.4 | 144.3 | -0.08 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 24.5 | 41.0 | -41.0 | 65.5 | 65.7 | 0.43 |
| 139 | -24.5 | 41.0 | -41.0 | -65.5 | -65.7 | 0.43 |
| 140 | -131.5 | 12.9 | -12.9 | -144.4 | -144.3 | -0.08 |
| 141 | 78.7 | 12.1 | -12.1 | 90.8 | 90.7 | -0.13 |
| 142 | 15.3 | 39.7 | -39.7 | 54.9 | 55.2 | 0.47 |
| 143 | -15.3 | 39.7 | -39.7 | -54.9 | -55.2 | 0.47 |
| 144 | -78.7 | 12.1 | -12.1 | -90.8 | -90.7 | -0.13 |
| 145 | 93.7 | -2.1 | 2.1 | 95.8 | 96.0 | 0.18 |
| 146 | 15.7 | 19.3 | -19.3 | 35.0 | 35.1 | 0.17 |
| 147 | -15.7 | 19.3 | -19.3 | -35.0 | -35.1 | 0.17 |
| 148 | -93.7 | -2.1 | 2.1 | -95.8 | -96.0 | 0.18 |
| 149 | 158.2 | -1.7 | 1.7 | 159.9 | 160.1 | 0.11 |
| 150 | 26.5 | 20.3 | -20.3 | 46.8 | 46.9 | 0.18 |
| 151 | -26.5 | 20.3 | -20.3 | -46.8 | -46.9 | 0.18 |
| 152 | -158.2 | -1.7 | 1.7 | -159.9 | -160.1 | 0.11 |
| 153 | 158.2 | -1.7 | 1.7 | 159.9 | 160.1 | 0.11 |
| 154 | 26.5 | 20.3 | -20.3 | 46.8 | 46.9 | 0.18 |
| 155 | -26.5 | 20.3 | -20.3 | -46.8 | -46.9 | 0.18 |
| 156 | -158.2 | -1.7 | 1.7 | -159.9 | -160.1 | 0.11 |
| 157 | 93.7 | -2.1 | 2.1 | 95.8 | 96.0 | 0.18 |
| 158 | 15.7 | 19.3 | -19.3 | 35.0 | 35.1 | 0.17 |
| 159 | -15.7 | 19.3 | -19.3 | -35.0 | -35.1 | 0.17 |
| 160 | -93.7 | -2.1 | 2.1 | -95.8 | -96.0 | 0.18 |

Table B. 2 M15 First and Second-Order Moments

| M15 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathrm{M}_{\text {SD+D+L }}$ | Mpositive EQ. | M $_{\text {negative }}$ <br> EQ. | FirstOrder Moment | Second-Order Moment | Percent |
| 1 | 29.5 | 288.8 | -288.8 | 318.3 | 327.4 | 2.85 |
| 2 | 2.2 | 302.1 | -302.1 | 304.4 | 313.7 | 3.07 |
| 3 | -2.2 | 302.1 | -302.1 | -304.4 | -313.7 | 3.07 |
| 4 | -29.5 | 288.8 | -288.8 | -318.3 | -327.4 | 2.85 |
| 5 | 49.0 | 289.8 | -289.8 | 338.7 | 347.5 | 2.59 |
| 6 | 2.8 | 303.0 | -303.0 | 305.9 | 314.8 | 2.91 |
| 7 | -2.8 | 303.0 | -303.0 | -305.9 | -314.8 | 2.91 |
| 8 | -49.0 | 289.8 | -289.8 | -338.7 | -347.5 | 2.59 |
| 9 | 49.0 | 289.8 | -289.8 | 338.7 | 347.5 | 2.59 |
| 10 | 2.8 | 303.0 | -303.0 | 305.9 | 314.8 | 2.91 |
| 11 | -2.8 | 303.0 | -303.0 | -305.9 | -314.8 | 2.91 |
| 12 | -49.0 | 289.8 | -289.8 | -338.7 | -347.5 | 2.59 |
| 13 | 29.5 | 288.8 | -288.8 | 318.3 | 327.4 | 2.85 |
| 14 | 2.2 | 302.1 | -302.1 | 304.4 | 313.7 | 3.07 |
| 15 | -2.2 | 302.1 | -302.1 | -304.4 | -313.7 | 3.07 |
| 16 | -29.5 | 288.8 | -288.8 | -318.3 | -327.4 | 2.85 |
| 17 | 73.9 | 168.2 | -168.2 | 242.1 | 250.3 | 3.35 |
| 18 | 2.3 | 209.6 | -209.6 | 211.9 | 221.6 | 4.60 |
| 19 | -2.3 | 209.6 | -209.6 | -211.9 | -221.6 | 4.60 |
| 20 | -73.9 | 168.2 | -168.2 | -242.1 | -250.3 | 3.35 |
| 21 | 129.4 | 169.1 | -169.1 | 298.5 | 306.6 | 2.73 |
| 22 | 4.2 | 211.9 | -211.9 | 216.1 | 225.9 | 4.51 |
| 23 | -4.2 | 211.9 | -211.9 | -216.1 | -225.9 | 4.51 |
| 24 | -129.4 | 169.1 | -169.1 | -298.5 | -306.6 | 2.73 |
| 25 | 129.4 | 169.1 | -169.1 | 298.5 | 306.6 | 2.73 |
| 26 | 4.2 | 211.9 | -211.9 | 216.1 | 225.9 | 4.51 |
| 27 | -4.2 | 211.9 | -211.9 | -216.1 | -225.9 | 4.51 |
| 28 | -129.4 | 169.1 | -169.1 | -298.5 | -306.6 | 2.73 |
| 29 | 73.9 | 168.2 | -168.2 | 242.1 | 250.3 | 3.35 |
| 30 | 2.3 | 209.6 | -209.6 | 211.9 | 221.6 | 4.60 |
| 31 | -2.3 | 209.6 | -209.6 | -211.9 | -221.6 | 4.60 |
| 32 | -73.9 | 168.2 | -168.2 | -242.1 | -250.3 | 3.35 |
| 33 | 72.9 | 105.8 | -105.8 | 178.7 | 184.6 | 3.29 |
| 34 | 7.7 | 159.5 | -159.5 | 167.1 | 175.5 | 4.97 |
| 35 | -7.7 | 159.5 | -159.5 | -167.1 | -175.5 | $\underline{4.97}$ |
| 36 | -72.9 | 105.8 | -105.8 | -178.7 | -184.6 | 3.29 |
| 37 | 124.9 | 107.2 | -107.2 | 232.2 | 238.0 | 2.54 |
| 38 | 12.1 | 162.3 | -162.3 | 174.4 | 182.8 | 4.79 |
| 39 | -12.1 | 162.3 | -162.3 | -174.4 | -182.8 | 4.79 |
| 40 | -124.9 | 107.2 | -107.2 | -232.2 | -238.0 | 2.54 |
| 41 | 124.9 | 107.2 | -107.2 | 232.2 | 238.0 | 2.54 |
| 42 | 12.1 | 162.3 | -162.3 | 174.4 | 182.8 | 4.79 |

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| 43 | -12.1 | 162.3 | -162.3 | -174.4 | -182.8 | 4.79 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -124.9 | 107.2 | -107.2 | -232.2 | -238.0 | 2.54 |
| 45 | 72.9 | 105.8 | -105.8 | 178.7 | 184.6 | 3.29 |
| 46 | 7.7 | 159.5 | -159.5 | 167.1 | 175.5 | $\underline{4.97}$ |
| 47 | -7.7 | 159.5 | -159.5 | -167.1 | -175.5 | $\underline{\mathbf{4 . 9 7}}$ |
| 48 | -72.9 | 105.8 | -105.8 | -178.7 | -184.6 | 3.29 |
| 49 | 75.7 | 73.4 | -73.4 | 149.1 | 153.0 | 2.62 |
| 50 | 10.0 | 130.8 | -130.8 | 140.8 | 147.4 | 4.72 |
| 51 | -10.0 | 130.8 | -130.8 | -140.8 | -147.4 | 4.72 |
| 52 | -75.7 | 73.4 | -73.4 | -149.1 | -153.0 | 2.62 |
| 53 | 129.8 | 74.9 | -74.9 | 204.7 | 208.7 | 1.94 |
| 54 | 16.1 | 133.7 | -133.7 | 149.8 | 156.6 | 4.51 |
| 55 | -16.1 | 133.7 | -133.7 | -149.8 | -156.6 | 4.51 |
| 56 | -129.8 | 74.9 | -74.9 | -204.7 | -208.7 | 1.94 |
| 57 | 129.8 | 74.9 | -74.9 | 204.7 | 208.7 | 1.94 |
| 58 | 16.1 | 133.7 | -133.7 | 149.8 | 156.6 | 4.51 |
| 59 | -16.1 | 133.7 | -133.7 | -149.8 | -156.6 | 4.51 |
| 60 | -129.8 | 74.9 | -74.9 | -204.7 | -208.7 | 1.94 |
| 61 | 75.7 | 73.4 | -73.4 | 149.1 | 153.0 | 2.62 |
| 62 | 10.0 | 130.8 | -130.8 | 140.8 | 147.4 | 4.72 |
| 63 | -10.0 | 130.8 | -130.8 | -140.8 | -147.4 | 4.72 |
| 64 | -75.7 | 73.4 | -73.4 | -149.1 | -153.0 | 2.62 |
| 65 | 79.0 | 56.4 | -56.4 | 135.5 | 138.0 | 1.88 |
| 66 | 12.6 | 115.3 | -115.3 | 127.9 | 133.2 | 4.19 |
| 67 | -12.6 | 115.3 | -115.3 | -127.9 | -133.2 | 4.19 |
| 68 | -79.0 | 56.4 | -56.4 | -135.5 | -138.0 | 1.88 |
| 69 | 135.9 | 57.8 | -57.8 | 193.7 | 196.3 | 1.35 |
| 70 | 20.5 | 118.4 | -118.4 | 138.8 | 144.4 | 3.96 |
| 71 | -20.5 | 118.4 | -118.4 | -138.8 | -144.4 | 3.96 |
| 72 | -135.9 | 57.8 | -57.8 | -193.7 | -196.3 | 1.35 |
| 73 | 135.9 | 57.8 | -57.8 | 193.7 | 196.3 | 1.35 |
| 74 | 20.5 | 118.4 | -118.4 | 138.8 | 144.4 | 3.96 |
| 75 | -20.5 | 118.4 | -118.4 | -138.8 | -144.4 | 3.96 |
| 76 | -135.9 | 57.8 | -57.8 | -193.7 | -196.3 | 1.35 |
| 77 | 79.0 | 56.4 | -56.4 | 135.5 | 138.0 | 1.88 |
| 78 | 12.6 | 115.3 | -115.3 | 127.9 | 133.2 | 4.19 |
| 79 | -12.6 | 115.3 | -115.3 | -127.9 | -133.2 | 4.19 |
| 80 | -79.0 | 56.4 | -56.4 | -135.5 | -138.0 | 1.88 |
| 81 | 72.8 | 52.5 | -52.5 | 125.3 | 127.3 | 1.61 |
| 82 | 13.8 | 102.5 | -102.5 | 116.3 | 120.6 | 3.71 |
| 83 | -13.8 | 102.5 | -102.5 | -116.3 | -120.6 | 3.71 |
| 84 | -72.8 | 52.5 | -52.5 | -125.3 | -127.3 | 1.61 |
| 85 | 123.1 | 54.0 | -54.0 | 177.0 | 179.1 | 1.18 |
| 86 | 21.8 | 105.0 | -105.0 | 126.8 | 131.3 | 3.51 |
| 87 | -21.8 | 105.0 | -105.0 | -126.8 | -131.3 | 3.51 |
| 88 | -123.1 | 54.0 | -54.0 | -177.0 | -179.1 | 1.18 |
| 89 | 123.1 | 54.0 | -54.0 | 177.0 | 179.1 | 1.18 |
|  |  |  |  |  |  |  |
| 7 |  |  |  |  |  |  |

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| 90 | 21.8 | 105.0 | -105.0 | 126.8 | 131.3 | 3.51 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 91 | -21.8 | 105.0 | -105.0 | -126.8 | -131.3 | 3.51 |
| 92 | -123.1 | 54.0 | -54.0 | -177.0 | -179.1 | 1.18 |
| 93 | 72.8 | 52.5 | -52.5 | 125.3 | 127.3 | 1.61 |
| 94 | 13.8 | 102.5 | -102.5 | 116.3 | 120.6 | 3.71 |
| 95 | -13.8 | 102.5 | -102.5 | -116.3 | -120.6 | 3.71 |
| 96 | -72.8 | 52.5 | -52.5 | -125.3 | -127.3 | 1.61 |
| 97 | 84.0 | 43.8 | -43.8 | 127.9 | 129.1 | 0.96 |
| 98 | 17.3 | 98.5 | -98.5 | 115.7 | 119.3 | 3.11 |
| 99 | -17.3 | 98.5 | -98.5 | -115.7 | -119.3 | 3.11 |
| 100 | -84.0 | 43.8 | -43.8 | -127.9 | -129.1 | 0.96 |
| 101 | 142.6 | 45.1 | -45.1 | 187.7 | 189.0 | 0.69 |
| 102 | 27.7 | 101.2 | -101.2 | 128.9 | 132.6 | 2.89 |
| 103 | -27.7 | 101.2 | -101.2 | -128.9 | -132.6 | 2.89 |
| 104 | -142.6 | 45.1 | -45.1 | -187.7 | -189.0 | 0.69 |
| 105 | 142.6 | 45.1 | -45.1 | 187.7 | 189.0 | 0.69 |
| 106 | 27.7 | 101.2 | -101.2 | 128.9 | 132.6 | 2.89 |
| 107 | -27.7 | 101.2 | -101.2 | -128.9 | -132.6 | 2.89 |
| 108 | -142.6 | 45.1 | -45.1 | -187.7 | -189.0 | 0.69 |
| 109 | 84.0 | 43.8 | -43.8 | 127.9 | 129.1 | 0.96 |
| 110 | 17.3 | 98.5 | -98.5 | 115.7 | 119.3 | 3.11 |
| 111 | -17.3 | 98.5 | -98.5 | -115.7 | -119.3 | 3.11 |
| 112 | -84.0 | 43.8 | -43.8 | -127.9 | -129.1 | 0.96 |
| 113 | 85.6 | 40.2 | -40.2 | 125.8 | 126.7 | 0.67 |
| 114 | 19.8 | 91.2 | -91.2 | 111.0 | 113.9 | 2.57 |
| 115 | -19.8 | 91.2 | -91.2 | -111.0 | -113.9 | 2.57 |
| 116 | -85.6 | 40.2 | -40.2 | -125.8 | -126.7 | 0.67 |
| 117 | 144.4 | 41.5 | -41.5 | 185.9 | 186.8 | 0.48 |
| 118 | 31.5 | 93.7 | -93.7 | 125.2 | 128.2 | 2.37 |
| 119 | -31.5 | 93.7 | -93.7 | -125.2 | -128.2 | 2.37 |
| 120 | -144.4 | 41.5 | -41.5 | -185.9 | -186.8 | 0.48 |
| 121 | 144.4 | 41.5 | -41.5 | 185.9 | 186.8 | 0.48 |
| 122 | 31.5 | 93.7 | -93.7 | 125.2 | 128.2 | 2.37 |
| 123 | -31.5 | 93.7 | -93.7 | -125.2 | -128.2 | 2.37 |
| 124 | -144.4 | 41.5 | -41.5 | -185.9 | -186.8 | 0.48 |
| 125 | 85.6 | 40.2 | -40.2 | 125.8 | 126.7 | 0.67 |
| 126 | 19.8 | 91.2 | -91.2 | 111.0 | 113.9 | 2.57 |
| 127 | -19.8 | 91.2 | -91.2 | -111.0 | -113.9 | 2.57 |
| 128 | -85.6 | 40.2 | -40.2 | -125.8 | -126.7 | 0.67 |
| 129 | 87.9 | 36.6 | -36.6 | 124.5 | 125.1 | 0.44 |
| 130 | 22.1 | 84.8 | -84.8 | 106.9 | 109.1 | 2.12 |
| 131 | -22.1 | 84.8 | -84.8 | -106.9 | -109.1 | 2.12 |
| 132 | -87.9 | 36.6 | -36.6 | -124.5 | -125.1 | 0.44 |
| 133 | 147.7 | 37.9 | -37.9 | 185.5 | 186.1 | 0.32 |
| 134 | 34.9 | 87.2 | -87.2 | 122.1 | 124.4 | 1.94 |
| 135 | -34.9 | 87.2 | -87.2 | -122.1 | -124.4 | 1.94 |
| 136 | -147.7 | 37.9 | -37.9 | -185.5 | -186.1 | 0.32 |

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| 137 | 147.7 | 37.9 | -37.9 | 185.5 | 186.1 | 0.32 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 34.9 | 87.2 | -87.2 | 122.1 | 124.4 | 1.94 |
| 139 | -34.9 | 87.2 | -87.2 | -122.1 | -124.4 | 1.94 |
| 140 | -147.7 | 37.9 | -37.9 | -185.5 | -186.1 | 0.32 |
| 141 | 87.9 | 36.6 | -36.6 | 124.5 | 125.1 | 0.44 |
| 142 | 22.1 | 84.8 | -84.8 | 106.9 | 109.1 | 2.12 |
| 143 | -22.1 | 84.8 | -84.8 | -106.9 | -109.1 | 2.12 |
| 144 | -87.9 | 36.6 | -36.6 | -124.5 | -125.1 | 0.44 |
| 145 | 91.7 | 32.7 | -32.7 | 124.4 | 124.7 | 0.30 |
| 146 | 24.0 | 80.0 | -80.0 | 104.0 | 105.9 | 1.81 |
| 147 | -24.0 | 80.0 | -80.0 | -104.0 | -105.9 | 1.81 |
| 148 | -91.7 | 32.7 | -32.7 | -124.4 | -124.7 | 0.30 |
| 149 | 155.1 | 33.7 | -33.7 | 188.8 | 189.2 | 0.22 |
| 150 | 38.5 | 82.3 | -82.3 | 120.8 | 122.8 | 1.63 |
| 151 | -38.5 | 82.3 | -82.3 | -120.8 | -122.8 | 1.63 |
| 152 | -155.1 | 33.7 | -33.7 | -188.8 | -189.2 | 0.22 |
| 153 | 155.1 | 33.7 | -33.7 | 188.8 | 189.2 | 0.22 |
| 154 | 38.5 | 82.3 | -82.3 | 120.8 | 122.8 | 1.63 |
| 155 | -38.5 | 82.3 | -82.3 | -120.8 | -122.8 | 1.63 |
| 156 | -155.1 | 33.7 | -33.7 | -188.8 | -189.2 | 0.22 |
| 157 | 91.7 | 32.7 | -32.7 | 124.4 | 124.7 | 0.30 |
| 158 | 24.0 | 80.0 | -80.0 | 104.0 | 105.9 | 1.81 |
| 159 | -24.0 | 80.0 | -80.0 | -104.0 | -105.9 | 1.81 |
| 160 | -91.7 | 32.7 | -32.7 | -124.4 | -124.7 | 0.30 |
| 161 | 73.9 | 36.2 | -36.2 | 110.1 | 110.7 | 0.54 |
| 162 | 19.9 | 69.6 | -69.6 | 89.5 | 91.0 | 1.71 |
| 163 | -19.9 | 69.6 | -69.6 | -89.5 | -91.0 | 1.71 |
| 164 | -73.9 | 36.2 | -36.2 | -110.1 | -110.7 | 0.54 |
| 165 | 121.8 | 37.1 | -37.1 | 158.9 | 159.5 | 0.40 |
| 166 | 30.8 | 71.1 | -71.1 | 101.9 | 103.5 | 1.55 |
| 167 | -30.8 | 71.1 | -71.1 | -101.9 | -103.5 | 1.55 |
| 168 | -121.8 | 37.1 | -37.1 | -158.9 | -159.5 | 0.40 |
| 169 | 121.8 | 37.1 | -37.1 | 158.9 | 159.5 | 0.40 |
| 170 | 30.8 | 71.1 | -71.1 | 101.9 | 103.5 | 1.55 |
| 171 | -30.8 | 71.1 | -71.1 | -101.9 | -103.5 | 1.55 |
| 172 | -121.8 | 37.1 | -37.1 | -158.9 | -159.5 | 0.40 |
| 173 | 73.9 | 36.2 | -36.2 | 110.1 | 110.7 | 0.54 |
| 174 | 19.9 | 69.6 | -69.6 | 89.5 | 91.0 | 1.71 |
| 175 | -19.9 | 69.6 | -69.6 | -89.5 | -91.0 | 1.71 |
| 176 | -73.9 | 36.2 | -36.2 | -110.1 | -110.7 | 0.54 |
| 177 | 93.0 | 26.3 | -26.3 | 119.3 | 119.6 | 0.18 |
| 178 | 24.9 | 64.2 | -64.2 | 89.1 | 90.2 | 1.25 |
| 179 | -24.9 | 64.2 | -64.2 | -89.1 | -90.2 | 1.25 |
| 180 | -93.0 | 26.3 | -26.3 | -119.3 | -119.6 | 0.18 |
| 181 | 153.6 | 27.1 | -27.1 | 180.7 | 180.9 | 0.13 |
| 182 | 39.0 | 65.9 | -65.9 | 104.8 | 106.0 | 1.11 |
| 183 | -39.0 | 65.9 | -65.9 | -104.8 | -106.0 | 1.11 |

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| 184 | -153.6 | 27.1 | -27.1 | -180.7 | -180.9 | 0.13 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 185 | 153.6 | 27.1 | -27.1 | 180.7 | 180.9 | 0.13 |
| 186 | 39.0 | 65.9 | -65.9 | 104.8 | 106.0 | 1.11 |
| 187 | -39.0 | 65.9 | -65.9 | -104.8 | -106.0 | 1.11 |
| 188 | -153.6 | 27.1 | -27.1 | -180.7 | -180.9 | 0.13 |
| 189 | 93.0 | 26.3 | -26.3 | 119.3 | 119.6 | 0.18 |
| 190 | 24.9 | 64.2 | -64.2 | 89.1 | 90.2 | 1.25 |
| 191 | -24.9 | 64.2 | -64.2 | -89.1 | -90.2 | 1.25 |
| 192 | -93.0 | 26.3 | -26.3 | -119.3 | -119.6 | 0.18 |
| 193 | 92.0 | 19.5 | -19.5 | 111.5 | 111.5 | 0.03 |
| 194 | 26.0 | 50.9 | -50.9 | 76.9 | 77.5 | 0.82 |
| 195 | -26.0 | 50.9 | -50.9 | -76.9 | -77.5 | 0.82 |
| 196 | -92.0 | 19.5 | -19.5 | -111.5 | -111.5 | 0.03 |
| 197 | 151.6 | 20.2 | -20.2 | 171.9 | 171.9 | 0.03 |
| 198 | 40.5 | 52.3 | -52.3 | 92.8 | 93.4 | 0.71 |
| 199 | -40.5 | 52.3 | -52.3 | -92.8 | -93.4 | 0.71 |
| 200 | -151.6 | 20.2 | -20.2 | -171.9 | -171.9 | 0.03 |
| 201 | 151.6 | 20.2 | -20.2 | 171.9 | 171.9 | 0.03 |
| 202 | 40.5 | 52.3 | -52.3 | 92.8 | 93.4 | 0.71 |
| 203 | -40.5 | 52.3 | -52.3 | -92.8 | -93.4 | 0.71 |
| 204 | -151.6 | 20.2 | -20.2 | -171.9 | -171.9 | 0.03 |
| 205 | 92.0 | 19.5 | -19.5 | 111.5 | 111.5 | 0.03 |
| 206 | 26.0 | 50.9 | -50.9 | 76.9 | 77.5 | 0.82 |
| 207 | -26.0 | 50.9 | -50.9 | -76.9 | -77.5 | 0.82 |
| 208 | -92.0 | 19.5 | -19.5 | -111.5 | -111.5 | 0.03 |
| 209 | 91.5 | 9.2 | -9.2 | 100.7 | 100.5 | -0.13 |
| 210 | 27.4 | 34.0 | -34.0 | 61.4 | 61.7 | 0.44 |
| 211 | -27.4 | 34.0 | -34.0 | -61.4 | -61.7 | 0.44 |
| 212 | -91.5 | 9.2 | -9.2 | -100.7 | -100.5 | -0.13 |
| 213 | 150.0 | 9.7 | -9.7 | 159.7 | 159.6 | -0.08 |
| 214 | 42.3 | 35.1 | -35.1 | 77.3 | 77.6 | 0.38 |
| 215 | -42.3 | 35.1 | -35.1 | -77.3 | -77.6 | 0.38 |
| 216 | -150.0 | 9.7 | -9.7 | -159.7 | -159.6 | -0.08 |
| 217 | 150.0 | 9.7 | -9.7 | 159.7 | 159.6 | -0.08 |
| 218 | 42.3 | 35.1 | -35.1 | 77.3 | 77.6 | 0.38 |
| 219 | -42.3 | 35.1 | -35.1 | -77.3 | -77.6 | 0.38 |
| 220 | -150.0 | 9.7 | -9.7 | -159.7 | -159.6 | -0.08 |
| 221 | 91.5 | 9.2 | -9.2 | 100.7 | 100.5 | -0.13 |
| 222 | 27.4 | 34.0 | -34.0 | 61.4 | 61.7 | 0.44 |
| 223 | -27.4 | 34.0 | -34.0 | -61.4 | -61.7 | 0.44 |
| 224 | -91.5 | 9.2 | -9.2 | -100.7 | -100.5 | -0.13 |
| 225 | 108.4 | -3.3 | 3.3 | 111.7 | 111.9 | 0.17 |
| 226 | 29.9 | 16.3 | -16.3 | 46.2 | 46.3 | 0.21 |
| 227 | -29.9 | 16.3 | -16.3 | -46.2 | -46.3 | 0.21 |
| 228 | -108.4 | -3.3 | 3.3 | -111.7 | -111.9 | 0.17 |
| 229 | 179.6 | -3.1 | 3.1 | 182.7 | 182.9 | 0.11 |
| 230 | 47.4 | 17.0 | -17.0 | 64.4 | 64.6 | 0.18 |
|  |  |  |  |  |  |  |
| 10 |  |  |  |  |  |  |

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| 231 | -47.4 | 17.0 | -17.0 | -64.4 | -64.6 | 0.18 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 232 | -179.6 | -3.1 | 3.1 | -182.7 | -182.9 | 0.11 |
| 233 | 179.6 | -3.1 | 3.1 | 182.7 | 182.9 | 0.11 |
| 234 | 47.4 | 17.0 | -17.0 | 64.4 | 64.6 | 0.18 |
| 235 | -47.4 | 17.0 | -17.0 | -64.4 | -64.6 | 0.18 |
| 236 | -179.6 | -3.1 | 3.1 | -182.7 | -182.9 | 0.11 |
| 237 | 108.4 | -3.3 | 3.3 | 111.7 | 111.9 | 0.17 |
| 238 | 29.9 | 16.3 | -16.3 | 46.2 | 46.3 | 0.21 |
| 239 | -29.9 | 16.3 | -16.3 | -46.2 | -46.3 | 0.21 |
| 240 | -108.4 | -3.3 | 3.3 | -111.7 | -111.9 | 0.17 |

Table B. 3 M20 First and Second-Order Moments

| M20 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathrm{M}_{\text {SD+D+L }}$ | Mpositive <br> ео. | $\underset{\text { EQ. }}{\mathbf{M}_{\text {NeGAtIVE }}}$ | First-Order Moment | SecondOrder <br> Moment | Percent |
| 1 | 30.5 | 353.0 | -353.0 | 383.4 | 399.2 | 4.10 |
| 2 | 2.8 | 365.4 | -365.4 | 368.1 | 384.3 | 4.38 |
| 3 | -2.8 | 365.4 | -365.4 | -368.1 | -384.3 | 4.38 |
| 4 | -30.5 | 353.0 | -353.0 | -383.4 | -399.2 | 4.10 |
| 5 | 50.2 | 354.0 | -354.0 | 404.3 | 419.6 | 3.79 |
| 6 | 3.5 | 366.2 | -366.2 | 369.7 | 385.3 | 4.21 |
| 7 | -3.5 | 366.2 | -366.2 | -369.7 | -385.3 | 4.21 |
| 8 | -50.2 | 354.0 | -354.0 | -404.3 | -419.6 | 3.79 |
| 9 | 50.2 | 354.0 | -354.0 | 404.3 | 419.6 | 3.79 |
| 10 | 3.5 | 366.2 | -366.2 | 369.7 | 385.3 | 4.21 |
| 11 | -3.5 | 366.2 | -366.2 | -369.7 | -385.3 | 4.21 |
| 12 | -50.2 | 354.0 | -354.0 | -404.3 | -419.6 | 3.79 |
| 13 | 30.5 | 353.0 | -353.0 | 383.4 | 399.2 | 4.10 |
| 14 | 2.8 | 365.4 | -365.4 | 368.1 | 384.3 | 4.38 |
| 15 | -2.8 | 365.4 | -365.4 | -368.1 | -384.3 | 4.38 |
| 16 | -30.5 | 353.0 | -353.0 | -383.4 | -399.2 | 4.10 |
| 17 | 73.1 | 218.6 | -218.6 | 291.7 | 306.3 | 5.00 |
| 18 | 2.4 | 256.1 | -256.1 | 258.5 | 275.2 | 6.44 |
| 19 | -2.4 | 256.1 | -256.1 | -258.5 | -275.2 | 6.44 |
| 20 | -73.1 | 218.6 | -218.6 | -291.7 | -306.3 | 5.00 |
| 21 | 128.7 | 219.4 | -219.4 | 348.0 | 362.7 | 4.21 |
| 22 | 4.4 | 258.3 | -258.3 | 262.7 | 279.4 | 6.36 |
| 23 | -4.4 | 258.3 | -258.3 | -262.7 | -279.4 | 6.36 |
| 24 | -128.7 | 219.4 | -219.4 | -348.0 | -362.7 | 4.21 |
| 25 | 128.7 | 219.4 | -219.4 | 348.0 | 362.7 | 4.21 |
| 26 | 4.4 | 258.3 | -258.3 | 262.7 | 279.4 | 6.36 |
| 27 | -4.4 | 258.3 | -258.3 | -262.7 | -279.4 | 6.36 |
| 28 | -128.7 | 219.4 | -219.4 | -348.0 | -362.7 | 4.21 |
| 29 | 73.1 | 218.6 | -218.6 | 291.7 | 306.3 | 5.00 |
| 30 | 2.4 | 256.1 | -256.1 | 258.5 | 275.2 | 6.44 |
| 31 | -2.4 | 256.1 | -256.1 | -258.5 | -275.2 | 6.44 |
| 32 | -73.1 | 218.6 | -218.6 | -291.7 | -306.3 | 5.00 |
| 33 | 74.1 | 141.3 | -141.3 | 215.4 | 226.8 | 5.30 |
| 34 | 8.3 | 193.2 | -193.2 | 201.5 | 216.2 | 7.30 |
| 35 | -8.3 | 193.2 | -193.2 | -201.5 | -216.2 | 7.30 |
| 36 | -74.1 | 141.3 | -141.3 | -215.4 | -226.8 | 5.30 |
| 37 | 127.0 | 142.7 | -142.7 | 269.7 | 281.1 | 4.24 |
| 38 | 12.9 | 196.0 | -196.0 | 208.8 | 223.6 | 7.08 |
| 39 | -12.9 | 196.0 | -196.0 | -208.8 | -223.6 | 7.08 |
| 40 | -127.0 | 142.7 | -142.7 | -269.7 | -281.1 | 4.24 |
| 41 | 127.0 | 142.7 | -142.7 | 269.7 | 281.1 | 4.24 |
| 42 | 12.9 | 196.0 | -196.0 | 208.8 | 223.6 | 7.08 |

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| 43 | -12.9 | 196.0 | -196.0 | -208.8 | -223.6 | 7.08 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -127.0 | 142.7 | -142.7 | -269.7 | -281.1 | 4.24 |
| 45 | 74.1 | 141.3 | -141.3 | 215.4 | 226.8 | 5.30 |
| 46 | 8.3 | 193.2 | -193.2 | 201.5 | 216.2 | 7.30 |
| 47 | -8.3 | 193.2 | -193.2 | -201.5 | -216.2 | 7.30 |
| 48 | -74.1 | 141.3 | -141.3 | -215.4 | -226.8 | 5.30 |
| 49 | 76.6 | 97.3 | -97.3 | 173.9 | 182.1 | 4.74 |
| 50 | 10.9 | 155.0 | -155.0 | 165.9 | 178.1 | 7.35 |
| 51 | -10.9 | 155.0 | -155.0 | -165.9 | -178.1 | 7.35 |
| 52 | -76.6 | 97.3 | -97.3 | -173.9 | -182.1 | 4.74 |
| 53 | 131.4 | 98.8 | -98.8 | 230.2 | 238.5 | 3.61 |
| 54 | 17.1 | 158.1 | -158.1 | 175.2 | 187.5 | 7.05 |
| 55 | -17.1 | 158.1 | -158.1 | -175.2 | -187.5 | 7.05 |
| 56 | -131.4 | 98.8 | -98.8 | -230.2 | -238.5 | 3.61 |
| 57 | 131.4 | 98.8 | -98.8 | 230.2 | 238.5 | 3.61 |
| 58 | 17.1 | 158.1 | -158.1 | 175.2 | 187.5 | 7.05 |
| 59 | -17.1 | 158.1 | -158.1 | -175.2 | -187.5 | 7.05 |
| 60 | -131.4 | 98.8 | -98.8 | -230.2 | -238.5 | 3.61 |
| 61 | 76.6 | 97.3 | -97.3 | 173.9 | 182.1 | 4.74 |
| 62 | 10.9 | 155.0 | -155.0 | 165.9 | 178.1 | 7.35 |
| 63 | -10.9 | 155.0 | -155.0 | -165.9 | -178.1 | 7.35 |
| 64 | -76.6 | 97.3 | -97.3 | -173.9 | -182.1 | 4.74 |
| 65 | 79.7 | 72.3 | -72.3 | 152.0 | 157.8 | 3.79 |
| 66 | 13.6 | 133.0 | -133.0 | 146.5 | 156.6 | 6.86 |
| 67 | -13.6 | 133.0 | -133.0 | -146.5 | -156.6 | 6.86 |
| 68 | -79.7 | 72.3 | -72.3 | -152.0 | -157.8 | 3.79 |
| 69 | 136.9 | 73.7 | -73.7 | 210.6 | 216.5 | 2.78 |
| 70 | 21.6 | 136.2 | -136.2 | 157.8 | 168.0 | 6.50 |
| 71 | -21.6 | 136.2 | -136.2 | -157.8 | -168.0 | 6.50 |
| 72 | -136.9 | 73.7 | -73.7 | -210.6 | -216.5 | 2.78 |
| 73 | 136.9 | 73.7 | -73.7 | 210.6 | 216.5 | 2.78 |
| 74 | 21.6 | 136.2 | -136.2 | 157.8 | 168.0 | 6.50 |
| 75 | -21.6 | 136.2 | -136.2 | -157.8 | -168.0 | 6.50 |
| 76 | -136.9 | 73.7 | -73.7 | -210.6 | -216.5 | 2.78 |
| 77 | 79.7 | 72.3 | -72.3 | 152.0 | 157.8 | 3.79 |
| 78 | 13.6 | 133.0 | -133.0 | 146.5 | 156.6 | 6.86 |
| 79 | -13.6 | 133.0 | -133.0 | -146.5 | -156.6 | 6.86 |
| 80 | -79.7 | 72.3 | -72.3 | -152.0 | -157.8 | 3.79 |
| 81 | 75.5 | 62.9 | -62.9 | 138.4 | 142.8 | 3.17 |
| 82 | 15.5 | 117.1 | -117.1 | 132.6 | 140.8 | 6.20 |
| 83 | -15.5 | 117.1 | -117.1 | -132.6 | -140.8 | 6.20 |
| 84 | -75.5 | 62.9 | -62.9 | -138.4 | -142.8 | 3.17 |
| 85 | 127.4 | 64.5 | -64.5 | 191.9 | 196.4 | 2.35 |
| 86 | 23.9 | 119.9 | -119.9 | 143.8 | 152.3 | 5.86 |
| 87 | -23.9 | 119.9 | -119.9 | -143.8 | -152.3 | 5.86 |
| 88 | -127.4 | 64.5 | -64.5 | -191.9 | -196.4 | 2.35 |
| 89 | 127.4 | 64.5 | -64.5 | 191.9 | 196.4 | 2.35 |

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| 90 | 23.9 | 119.9 | -119.9 | 143.8 | 152.3 | 5.86 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 91 | -23.9 | 119.9 | -119.9 | -143.8 | -152.3 | 5.86 |
| 92 | -127.4 | 64.5 | -64.5 | -191.9 | -196.4 | 2.35 |
| 93 | 75.5 | 62.9 | -62.9 | 138.4 | 142.8 | 3.17 |
| 94 | 15.5 | 117.1 | -117.1 | 132.6 | 140.8 | 6.20 |
| 95 | -15.5 | 117.1 | -117.1 | -132.6 | -140.8 | 6.20 |
| 96 | -75.5 | 62.9 | -62.9 | -138.4 | -142.8 | 3.17 |
| 97 | 85.8 | 52.8 | -52.8 | 138.6 | 141.5 | 2.12 |
| 98 | 19.3 | 112.7 | -112.7 | 132.0 | 139.0 | 5.36 |
| 99 | -19.3 | 112.7 | -112.7 | -132.0 | -139.0 | 5.36 |
| 100 | -85.8 | 52.8 | -52.8 | -138.6 | -141.5 | 2.12 |
| 101 | 145.4 | 54.2 | -54.2 | 199.6 | 202.7 | 1.52 |
| 102 | 30.3 | 115.8 | -115.8 | 146.0 | 153.3 | 5.00 |
| 103 | -30.3 | 115.8 | -115.8 | -146.0 | -153.3 | 5.00 |
| 104 | -145.4 | 54.2 | -54.2 | -199.6 | -202.7 | 1.52 |
| 105 | 145.4 | 54.2 | -54.2 | 199.6 | 202.7 | 1.52 |
| 106 | 30.3 | 115.8 | -115.8 | 146.0 | 153.3 | 5.00 |
| 107 | -30.3 | 115.8 | -115.8 | -146.0 | -153.3 | 5.00 |
| 108 | -145.4 | 54.2 | -54.2 | -199.6 | -202.7 | 1.52 |
| 109 | 85.8 | 52.8 | -52.8 | 138.6 | 141.5 | 2.12 |
| 110 | 19.3 | 112.7 | -112.7 | 132.0 | 139.0 | 5.36 |
| 111 | -19.3 | 112.7 | -112.7 | -132.0 | -139.0 | 5.36 |
| 112 | -85.8 | 52.8 | -52.8 | -138.6 | -141.5 | 2.12 |
| 113 | 88.3 | 48.8 | -48.8 | 137.1 | 139.3 | 1.57 |
| 114 | 22.4 | 106.5 | -106.5 | 128.9 | 134.9 | 4.63 |
| 115 | -22.4 | 106.5 | -106.5 | -128.9 | -134.9 | 4.63 |
| 116 | -88.3 | 48.8 | -48.8 | -137.1 | -139.3 | 1.57 |
| 117 | 148.6 | 50.3 | -50.3 | 198.9 | 201.1 | 1.14 |
| 118 | 34.8 | 109.5 | -109.5 | 144.3 | 150.5 | 4.27 |
| 119 | -34.8 | 109.5 | -109.5 | -144.3 | -150.5 | 4.27 |
| 120 | -148.6 | 50.3 | -50.3 | -198.9 | -201.1 | 1.14 |
| 121 | 148.6 | 50.3 | -50.3 | 198.9 | 201.1 | 1.14 |
| 122 | 34.8 | 109.5 | -109.5 | 144.3 | 150.5 | 4.27 |
| 123 | -34.8 | 109.5 | -109.5 | -144.3 | -150.5 | 4.27 |
| 124 | -148.6 | 50.3 | -50.3 | -198.9 | -201.1 | 1.14 |
| 125 | 88.3 | 48.8 | -48.8 | 137.1 | 139.3 | 1.57 |
| 126 | 22.4 | 106.5 | -106.5 | 128.9 | 134.9 | 4.63 |
| 127 | -22.4 | 106.5 | -106.5 | -128.9 | -134.9 | 4.63 |
| 128 | -88.3 | 48.8 | -48.8 | -137.1 | -139.3 | 1.57 |
| 129 | 91.1 | 44.9 | -44.9 | 136.0 | 137.6 | 1.18 |
| 130 | 25.0 | 101.0 | -101.0 | 126.1 | 131.2 | 4.04 |
| 131 | -25.0 | 101.0 | -101.0 | -126.1 | -131.2 | 4.04 |
| 132 | -91.1 | 44.9 | -44.9 | -136.0 | -137.6 | 1.18 |
| 133 | 152.7 | 46.4 | -46.4 | 199.1 | 200.8 | 0.86 |
| 134 | 38.8 | 103.9 | -103.9 | 142.7 | 148.0 | 3.70 |
| 135 | -38.8 | 103.9 | -103.9 | -142.7 | -148.0 | 3.70 |
| 136 | -152.7 | 46.4 | -46.4 | -199.1 | -200.8 | 0.86 |

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| 137 | 152.7 | 46.4 | -46.4 | 199.1 | 200.8 | 0.86 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 38.8 | 103.9 | -103.9 | 142.7 | 148.0 | 3.70 |
| 139 | -38.8 | 103.9 | -103.9 | -142.7 | -148.0 | 3.70 |
| 140 | -152.7 | 46.4 | -46.4 | -199.1 | -200.8 | 0.86 |
| 141 | 91.1 | 44.9 | -44.9 | 136.0 | 137.6 | 1.18 |
| 142 | 25.0 | 101.0 | -101.0 | 126.1 | 131.2 | 4.04 |
| 143 | -25.0 | 101.0 | -101.0 | -126.1 | -131.2 | 4.04 |
| 144 | -91.1 | 44.9 | -44.9 | -136.0 | -137.6 | 1.18 |
| 145 | 94.0 | 41.1 | -41.1 | 135.1 | 136.4 | 0.92 |
| 146 | 27.4 | 96.4 | -96.4 | 123.7 | 128.2 | 3.62 |
| 147 | -27.4 | 96.4 | -96.4 | -123.7 | -128.2 | 3.62 |
| 148 | -94.0 | 41.1 | -41.1 | -135.1 | -136.4 | 0.92 |
| 149 | 158.1 | 42.3 | -42.3 | 200.4 | 201.8 | 0.66 |
| 150 | 42.8 | 99.1 | -99.1 | 141.9 | 146.5 | 3.28 |
| 151 | -42.8 | 99.1 | -99.1 | -141.9 | -146.5 | 3.28 |
| 152 | -158.1 | 42.3 | -42.3 | -200.4 | -201.8 | 0.66 |
| 153 | 158.1 | 42.3 | -42.3 | 200.4 | 201.8 | 0.66 |
| 154 | 42.8 | 99.1 | -99.1 | 141.9 | 146.5 | 3.28 |
| 155 | -42.8 | 99.1 | -99.1 | -141.9 | -146.5 | 3.28 |
| 156 | -158.1 | 42.3 | -42.3 | -200.4 | -201.8 | 0.66 |
| 157 | 94.0 | 41.1 | -41.1 | 135.1 | 136.4 | 0.92 |
| 158 | 27.4 | 96.4 | -96.4 | 123.7 | 128.2 | 3.62 |
| 159 | -27.4 | 96.4 | -96.4 | -123.7 | -128.2 | 3.62 |
| 160 | -94.0 | 41.1 | -41.1 | -135.1 | -136.4 | 0.92 |
| 161 | 85.9 | 42.5 | -42.5 | 128.5 | 129.8 | 1.06 |
| 162 | 26.4 | 88.6 | -88.6 | 115.1 | 118.9 | 3.36 |
| 163 | -26.4 | 88.6 | -88.6 | -115.1 | -118.9 | 3.36 |
| 164 | -85.9 | 42.5 | -42.5 | -128.5 | -129.8 | 1.06 |
| 165 | 142.1 | 43.7 | -43.7 | 185.8 | 187.2 | 0.78 |
| 166 | 40.4 | 90.9 | -90.9 | 131.3 | 135.3 | 3.05 |
| 167 | -40.4 | 90.9 | -90.9 | -131.3 | -135.3 | 3.05 |
| 168 | -142.1 | 43.7 | -43.7 | -185.8 | -187.2 | 0.78 |
| 169 | 142.1 | 43.7 | -43.7 | 185.8 | 187.2 | 0.78 |
| 170 | 40.4 | 90.9 | -90.9 | 131.3 | 135.3 | 3.05 |
| 171 | -40.4 | 90.9 | -90.9 | -131.3 | -135.3 | 3.05 |
| 172 | -142.1 | 43.7 | -43.7 | -185.8 | -187.2 | 0.78 |
| 173 | 85.9 | 42.5 | -42.5 | 128.5 | 129.8 | 1.06 |
| 174 | 26.4 | 88.6 | -88.6 | 115.1 | 118.9 | 3.36 |
| 175 | -26.4 | 88.6 | -88.6 | -115.1 | -118.9 | 3.36 |
| 176 | -85.9 | 42.5 | -42.5 | -128.5 | -129.8 | 1.06 |
| 177 | 98.2 | 36.5 | -36.5 | 134.8 | 135.7 | 0.65 |
| 178 | 31.2 | 86.4 | -86.4 | 117.6 | 121.0 | 2.86 |
| 179 | -31.2 | 86.4 | -86.4 | -117.6 | -121.0 | 2.86 |
| 180 | -98.2 | 36.5 | -36.5 | -134.8 | -135.7 | 0.65 |
| 181 | 163.1 | 37.6 | -37.6 | 200.7 | 201.6 | 0.47 |
| 182 | 48.1 | 88.8 | -88.8 | 136.9 | 140.3 | 2.55 |
| 183 | -48.1 | 88.8 | -88.8 | -136.9 | -140.3 | 2.55 |

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| 184 | -163.1 | 37.6 | -37.6 | -200.7 | -201.6 | 0.47 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 185 | 163.1 | 37.6 | -37.6 | 200.7 | 201.6 | 0.47 |
| 186 | 48.1 | 88.8 | -88.8 | 136.9 | 140.3 | 2.55 |
| 187 | -48.1 | 88.8 | -88.8 | -136.9 | -140.3 | 2.55 |
| 188 | -163.1 | 37.6 | -37.6 | -200.7 | -201.6 | 0.47 |
| 189 | 98.2 | 36.5 | -36.5 | 134.8 | 135.7 | 0.65 |
| 190 | 31.2 | 86.4 | -86.4 | 117.6 | 121.0 | 2.86 |
| 191 | -31.2 | 86.4 | -86.4 | -117.6 | -121.0 | 2.86 |
| 192 | -98.2 | 36.5 | -36.5 | -134.8 | -135.7 | 0.65 |
| 193 | 99.4 | 33.9 | -33.9 | 133.3 | 134.0 | 0.48 |
| 194 | 33.2 | 80.3 | -80.3 | 113.5 | 116.2 | 2.39 |
| 195 | -33.2 | 80.3 | -80.3 | -113.5 | -116.2 | 2.39 |
| 196 | -99.4 | 33.9 | -33.9 | -133.3 | -134.0 | 0.48 |
| 197 | 164.1 | 35.0 | -35.0 | 199.1 | 199.8 | 0.35 |
| 198 | 51.0 | 82.5 | -82.5 | 133.4 | 136.3 | 2.12 |
| 199 | -51.0 | 82.5 | -82.5 | -133.4 | -136.3 | 2.12 |
| 200 | -164.1 | 35.0 | -35.0 | -199.1 | -199.8 | 0.35 |
| 201 | 164.1 | 35.0 | -35.0 | 199.1 | 199.8 | 0.35 |
| 202 | 51.0 | 82.5 | -82.5 | 133.4 | 136.3 | 2.12 |
| 203 | -51.0 | 82.5 | -82.5 | -133.4 | -136.3 | 2.12 |
| 204 | -164.1 | 35.0 | -35.0 | -199.1 | -199.8 | 0.35 |
| 205 | 99.4 | 33.9 | -33.9 | 133.3 | 134.0 | 0.48 |
| 206 | 33.2 | 80.3 | -80.3 | 113.5 | 116.2 | 2.39 |
| 207 | -33.2 | 80.3 | -80.3 | -113.5 | -116.2 | 2.39 |
| 208 | -99.4 | 33.9 | -33.9 | -133.3 | -134.0 | 0.48 |
| 209 | 101.5 | 30.6 | -30.6 | 132.1 | 132.5 | 0.32 |
| 210 | 35.2 | 74.3 | -74.3 | 109.5 | 111.7 | 2.00 |
| 211 | -35.2 | 74.3 | -74.3 | -109.5 | -111.7 | 2.00 |
| 212 | -101.5 | 30.6 | -30.6 | -132.1 | -132.5 | 0.32 |
| 213 | 166.8 | 31.6 | -31.6 | 198.5 | 198.9 | 0.24 |
| 214 | 53.8 | 76.3 | -76.3 | 130.1 | 132.3 | 1.75 |
| 215 | -53.8 | 76.3 | -76.3 | -130.1 | -132.3 | 1.75 |
| 216 | -166.8 | 31.6 | -31.6 | -198.5 | -198.9 | 0.24 |
| 217 | 166.8 | 31.6 | -31.6 | 198.5 | 198.9 | 0.24 |
| 218 | 53.8 | 76.3 | -76.3 | 130.1 | 132.3 | 1.75 |
| 219 | -53.8 | 76.3 | -76.3 | -130.1 | -132.3 | 1.75 |
| 220 | -166.8 | 31.6 | -31.6 | -198.5 | -198.9 | 0.24 |
| 221 | 101.5 | 30.6 | -30.6 | 132.1 | 132.5 | 0.32 |
| 222 | 35.2 | 74.3 | -74.3 | 109.5 | 111.7 | 2.00 |
| 223 | -35.2 | 74.3 | -74.3 | -109.5 | -111.7 | 2.00 |
| 224 | -101.5 | 30.6 | -30.6 | -132.1 | -132.5 | 0.32 |
| 225 | 105.2 | 27.2 | -27.2 | 132.4 | 132.7 | 0.22 |
| 226 | 37.4 | 69.9 | -69.9 | 107.3 | 109.1 | 1.71 |
| 227 | -37.4 | 69.9 | -69.9 | -107.3 | -109.1 | 1.71 |
| 228 | -105.2 | 27.2 | -27.2 | -132.4 | -132.7 | 0.22 |
| 229 | 174.3 | 27.9 | -27.9 | 202.2 | 202.6 | 0.16 |
| 230 | 57.7 | 71.8 | -71.8 | 129.6 | 131.5 | 1.48 |

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| 231 | -57.7 | 71.8 | -71.8 | -129.6 | -131.5 | 1.48 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 232 | -174.3 | 27.9 | -27.9 | -202.2 | -202.6 | 0.16 |
| 233 | 174.3 | 27.9 | -27.9 | 202.2 | 202.6 | 0.16 |
| 234 | 57.7 | 71.8 | -71.8 | 129.6 | 131.5 | 1.48 |
| 235 | -57.7 | 71.8 | -71.8 | -129.6 | -131.5 | 1.48 |
| 236 | -174.3 | 27.9 | -27.9 | -202.2 | -202.6 | 0.16 |
| 237 | 105.2 | 27.2 | -27.2 | 132.4 | 132.7 | 0.22 |
| 238 | 37.4 | 69.9 | -69.9 | 107.3 | 109.1 | 1.71 |
| 239 | -37.4 | 69.9 | -69.9 | -107.3 | -109.1 | 1.71 |
| 240 | -105.2 | 27.2 | -27.2 | -132.4 | -132.7 | 0.22 |
| 241 | 84.5 | 31.0 | -31.0 | 115.5 | 116.0 | 0.46 |
| 242 | 29.8 | 61.3 | -61.3 | 91.1 | 92.6 | 1.64 |
| 243 | -29.8 | 61.3 | -61.3 | -91.1 | -92.6 | 1.64 |
| 244 | -84.5 | 31.0 | -31.0 | -115.5 | -116.0 | 0.46 |
| 245 | 136.3 | 31.8 | -31.8 | 168.1 | 168.7 | 0.34 |
| 246 | 44.8 | 62.5 | -62.5 | 107.3 | 108.8 | 1.44 |
| 247 | -44.8 | 62.5 | -62.5 | -107.3 | -108.8 | 1.44 |
| 248 | -136.3 | 31.8 | -31.8 | -168.1 | -168.7 | 0.34 |
| 249 | 136.3 | 31.8 | -31.8 | 168.1 | 168.7 | 0.34 |
| 250 | 44.8 | 62.5 | -62.5 | 107.3 | 108.8 | 1.44 |
| 251 | -44.8 | 62.5 | -62.5 | -107.3 | -108.8 | 1.44 |
| 252 | -136.3 | 31.8 | -31.8 | -168.1 | -168.7 | 0.34 |
| 253 | 84.5 | 31.0 | -31.0 | 115.5 | 116.0 | 0.46 |
| 254 | 29.8 | 61.3 | -61.3 | 91.1 | 92.6 | 1.64 |
| 255 | -29.8 | 61.3 | -61.3 | -91.1 | -92.6 | 1.64 |
| 256 | -84.5 | 31.0 | -31.0 | -115.5 | -116.0 | 0.46 |
| 257 | 105.6 | 23.0 | -23.0 | 128.6 | 128.8 | 0.13 |
| 258 | 37.0 | 57.5 | -57.5 | 94.5 | 95.6 | 1.19 |
| 259 | -37.0 | 57.5 | -57.5 | -94.5 | -95.6 | 1.19 |
| 260 | -105.6 | 23.0 | -23.0 | -128.6 | -128.8 | 0.13 |
| 261 | 171.0 | 23.6 | -23.6 | 194.6 | 194.8 | 0.10 |
| 262 | 55.9 | 58.9 | -58.9 | 114.8 | 115.9 | 1.01 |
| 263 | -55.9 | 58.9 | -58.9 | -114.8 | -115.9 | 1.01 |
| 264 | -171.0 | 23.6 | -23.6 | -194.6 | -194.8 | 0.10 |
| 265 | 171.0 | 23.6 | -23.6 | 194.6 | 194.8 | 0.10 |
| 266 | 55.9 | 58.9 | -58.9 | 114.8 | 115.9 | 1.01 |
| 267 | -55.9 | 58.9 | -58.9 | -114.8 | -115.9 | 1.01 |
| 268 | -171.0 | 23.6 | -23.6 | -194.6 | -194.8 | 0.10 |
| 269 | 105.6 | 23.0 | -23.0 | 128.6 | 128.8 | 0.13 |
| 270 | 37.0 | 57.5 | -57.5 | 94.5 | 95.6 | 1.19 |
| 271 | -37.0 | 57.5 | -57.5 | -94.5 | -95.6 | 1.19 |
| 272 | -105.6 | 23.0 | -23.0 | -128.6 | -128.8 | 0.13 |
| 273 | 104.1 | 16.6 | -16.6 | 120.7 | 120.7 | 0.00 |
| 274 | 37.6 | 45.4 | -45.4 | 83.0 | 83.6 | 0.79 |
| 275 | -37.6 | 45.4 | -45.4 | -83.0 | -83.6 | 0.79 |
| 276 | -104.1 | 16.6 | -16.6 | -120.7 | -120.7 | 0.00 |
| 277 | 168.2 | 17.2 | -17.2 | 185.4 | 185.4 | 0.01 |

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| 278 | 56.6 | 46.5 | -46.5 | 103.2 | 103.9 | 0.66 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 279 | -56.6 | 46.5 | -46.5 | -103.2 | -103.9 | 0.66 |
| 280 | -168.2 | 17.2 | -17.2 | -185.4 | -185.4 | 0.01 |
| 281 | 168.2 | 17.2 | -17.2 | 185.4 | 185.4 | 0.01 |
| 282 | 56.6 | 46.5 | -46.5 | 103.2 | 103.9 | 0.66 |
| 283 | -56.6 | 46.5 | -46.5 | -103.2 | -103.9 | 0.66 |
| 284 | -168.2 | 17.2 | -17.2 | -185.4 | -185.4 | 0.01 |
| 285 | 104.1 | 16.6 | -16.6 | 120.7 | 120.7 | 0.00 |
| 286 | 37.6 | 45.4 | -45.4 | 83.0 | 83.6 | 0.79 |
| 287 | -37.6 | 45.4 | -45.4 | -83.0 | -83.6 | 0.79 |
| 288 | -104.1 | 16.6 | -16.6 | -120.7 | -120.7 | 0.00 |
| 289 | 103.4 | 6.9 | -6.9 | 110.3 | 110.2 | -0.14 |
| 290 | 38.7 | 29.7 | -29.7 | 68.5 | 68.8 | 0.44 |
| 291 | -38.7 | 29.7 | -29.7 | -68.5 | -68.8 | 0.44 |
| 292 | -103.4 | 6.9 | -6.9 | -110.3 | -110.2 | -0.14 |
| 293 | 166.1 | 7.3 | -7.3 | 173.4 | 173.3 | -0.09 |
| 294 | 57.9 | 30.6 | -30.6 | 88.5 | 88.8 | 0.37 |
| 295 | -57.9 | 30.6 | -30.6 | -88.5 | -88.8 | 0.37 |
| 296 | -166.1 | 7.3 | -7.3 | -173.4 | -173.3 | -0.09 |
| 297 | 166.1 | 7.3 | -7.3 | 173.4 | 173.3 | -0.09 |
| 298 | 57.9 | 30.6 | -30.6 | 88.5 | 88.8 | 0.37 |
| 299 | -57.9 | 30.6 | -30.6 | -88.5 | -88.8 | 0.37 |
| 300 | -166.1 | 7.3 | -7.3 | -173.4 | -173.3 | -0.09 |
| 301 | 103.4 | 6.9 | -6.9 | 110.3 | 110.2 | -0.14 |
| 302 | 38.7 | 29.7 | -29.7 | 68.5 | 68.8 | 0.44 |
| 303 | -38.7 | 29.7 | -29.7 | -68.5 | -68.8 | 0.44 |
| 304 | -103.4 | 6.9 | -6.9 | -110.3 | -110.2 | -0.14 |
| 305 | 122.1 | -4.0 | 4.0 | 126.1 | 126.3 | 0.18 |
| 306 | 43.1 | 14.3 | -14.3 | 57.4 | 57.6 | 0.26 |
| 307 | -43.1 | 14.3 | -14.3 | -57.4 | -57.6 | 0.26 |
| 308 | -122.1 | -4.0 | 4.0 | -126.1 | -126.3 | 0.18 |
| 309 | 198.3 | -3.9 | 3.9 | 202.2 | 202.4 | 0.12 |
| 310 | 65.7 | 14.9 | -14.9 | 80.6 | 80.8 | 0.21 |
| 311 | -65.7 | 14.9 | -14.9 | -80.6 | -80.8 | 0.21 |
| 312 | -198.3 | -3.9 | 3.9 | -202.2 | -202.4 | 0.12 |
| 313 | 198.3 | -3.9 | 3.9 | 202.2 | 202.4 | 0.12 |
| 314 | 65.7 | 14.9 | -14.9 | 80.6 | 80.8 | 0.21 |
| 315 | -65.7 | 14.9 | -14.9 | -80.6 | -80.8 | 0.21 |
| 316 | -198.3 | -3.9 | 3.9 | -202.2 | -202.4 | 0.12 |
| 317 | 122.1 | -4.0 | 4.0 | 126.1 | 126.3 | 0.18 |
| 318 | 43.1 | 14.3 | -14.3 | 57.4 | 57.6 | 0.26 |
| 319 | -43.1 | 14.3 | -14.3 | -57.4 | -57.6 | 0.26 |
| 320 | -122.1 | -4.0 | 4.0 | -126.1 | -126.3 | 0.18 |

Table B. 4 M25 First and Second-Order Moments

| M25 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathbf{M}_{\text {SD }+\mathrm{D}+\mathrm{L}}$ | Mpositive ео. | M $_{\text {negative }}$ <br> еQ. | First-Order Moment | Second- Order <br> Moment | Percent |
| 1 | 31.3 | 422.8 | -422.8 | 454.1 | 478.3 | 5.32 |
| 2 | 3.2 | 434.3 | -434.3 | 437.6 | 462.2 | 5.63 |
| 3 | -3.2 | 434.3 | -434.3 | -437.6 | -462.2 | 5.63 |
| 4 | -31.3 | 422.8 | -422.8 | -454.1 | -478.3 | 5.32 |
| 5 | 51.3 | 423.9 | -423.9 | 475.2 | 498.9 | 4.98 |
| 6 | 4.1 | 435.1 | -435.1 | 439.2 | 463.2 | 5.46 |
| 7 | -4.1 | 435.1 | -435.1 | -439.2 | -463.2 | 5.46 |
| 8 | -51.3 | 423.9 | -423.9 | -475.2 | -498.9 | 4.98 |
| 9 | 51.3 | 423.9 | -423.9 | 475.2 | 498.9 | 4.98 |
| 10 | 4.1 | 435.1 | -435.1 | 439.2 | 463.2 | 5.46 |
| 11 | -4.1 | 435.1 | -435.1 | -439.2 | -463.2 | 5.46 |
| 12 | -51.3 | 423.9 | -423.9 | -475.2 | -498.9 | 4.98 |
| 13 | 31.3 | 422.8 | -422.8 | 454.1 | 478.3 | 5.32 |
| 14 | 3.2 | 434.3 | -434.3 | 437.6 | 462.2 | 5.63 |
| 15 | -3.2 | 434.3 | -434.3 | -437.6 | -462.2 | 5.63 |
| 16 | -31.3 | 422.8 | -422.8 | -454.1 | -478.3 | 5.32 |
| 17 | 71.9 | 277.0 | -277.0 | 348.9 | 371.7 | 6.54 |
| 18 | 2.2 | 310.5 | -310.5 | 312.7 | 337.9 | 8.04 |
| 19 | -2.2 | 310.5 | -310.5 | -312.7 | -337.9 | 8.04 |
| 20 | -71.9 | 277.0 | -277.0 | -348.9 | -371.7 | 6.54 |
| 21 | 127.1 | 277.6 | -277.6 | 404.7 | 427.7 | 5.66 |
| 22 | 4.1 | 312.5 | -312.5 | 316.6 | 341.8 | 7.97 |
| 23 | -4.1 | 312.5 | -312.5 | -316.6 | -341.8 | 7.97 |
| 24 | -127.1 | 277.6 | -277.6 | -404.7 | -427.7 | 5.66 |
| 25 | 127.1 | 277.6 | -277.6 | 404.7 | 427.7 | 5.66 |
| 26 | 4.1 | 312.5 | -312.5 | 316.6 | 341.8 | 7.97 |
| 27 | -4.1 | 312.5 | -312.5 | -316.6 | -341.8 | 7.97 |
| 28 | -127.1 | 277.6 | -277.6 | -404.7 | -427.7 | 5.66 |
| 29 | 71.9 | 277.0 | -277.0 | 348.9 | 371.7 | 6.54 |
| 30 | 2.2 | 310.5 | -310.5 | 312.7 | 337.9 | 8.04 |
| 31 | -2.2 | 310.5 | -310.5 | -312.7 | -337.9 | 8.04 |
| 32 | -71.9 | 277.0 | -277.0 | -348.9 | -371.7 | 6.54 |
| 33 | 74.7 | 185.1 | -185.1 | 259.8 | 278.6 | 7.23 |
| 34 | 8.4 | 234.1 | -234.1 | 242.5 | 265.2 | 9.35 |
| 35 | -8.4 | 234.1 | -234.1 | -242.5 | -265.2 | 9.35 |
| 36 | -74.7 | 185.1 | -185.1 | -259.8 | -278.6 | 7.23 |
| 37 | 128.1 | 186.5 | -186.5 | 314.6 | 333.4 | 5.98 |
| 38 | 12.9 | 236.8 | -236.8 | 249.7 | 272.4 | 9.12 |
| 39 | -12.9 | 236.8 | -236.8 | -249.7 | -272.4 | 9.12 |
| 40 | -128.1 | 186.5 | -186.5 | -314.6 | -333.4 | 5.98 |
| 41 | 128.1 | 186.5 | -186.5 | 314.6 | 333.4 | 5.98 |
| 42 | 12.9 | 236.8 | -236.8 | 249.7 | 272.4 | 9.12 |

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| 43 | -12.9 | 236.8 | -236.8 | -249.7 | -272.4 | 9.12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -128.1 | 186.5 | -186.5 | -314.6 | -333.4 | 5.98 |
| 45 | 74.7 | 185.1 | -185.1 | 259.8 | 278.6 | 7.23 |
| 46 | 8.4 | 234.1 | -234.1 | 242.5 | 265.2 | 9.35 |
| 47 | -8.4 | 234.1 | -234.1 | -242.5 | -265.2 | 9.35 |
| 48 | -74.7 | 185.1 | -185.1 | -259.8 | -278.6 | 7.23 |
| 49 | 76.8 | 128.4 | -128.4 | 205.2 | 219.6 | 7.00 |
| 50 | 11.0 | 184.7 | -184.7 | 195.8 | 215.0 | 9.82 |
| 51 | -11.0 | 184.7 | -184.7 | -195.8 | -215.0 | 9.82 |
| 52 | -76.8 | 128.4 | -128.4 | -205.2 | -219.6 | 7.00 |
| 53 | 131.7 | 129.9 | -129.9 | 261.6 | 276.1 | 5.51 |
| 54 | 17.1 | 187.7 | -187.7 | 204.8 | 224.2 | 9.47 |
| 55 | -17.1 | 187.7 | -187.7 | -204.8 | -224.2 | 9.47 |
| 56 | -131.7 | 129.9 | -129.9 | -261.6 | -276.1 | 5.51 |
| 57 | 131.7 | 129.9 | -129.9 | 261.6 | 276.1 | 5.51 |
| 58 | 17.1 | 187.7 | -187.7 | 204.8 | 224.2 | 9.47 |
| 59 | -17.1 | 187.7 | -187.7 | -204.8 | -224.2 | 9.47 |
| 60 | -131.7 | 129.9 | -129.9 | -261.6 | -276.1 | 5.51 |
| 61 | 76.8 | 128.4 | -128.4 | 205.2 | 219.6 | 7.00 |
| 62 | 11.0 | 184.7 | -184.7 | 195.8 | 215.0 | 9.82 |
| 63 | -11.0 | 184.7 | -184.7 | -195.8 | -215.0 | $\underline{9.82}$ |
| 64 | -76.8 | 128.4 | -128.4 | -205.2 | -219.6 | 7.00 |
| 65 | 79.5 | 93.2 | -93.2 | 172.8 | 183.3 | 6.10 |
| 66 | 13.6 | 153.5 | -153.5 | 167.2 | 183.2 | 9.57 |
| 67 | -13.6 | 153.5 | -153.5 | -167.2 | -183.2 | 9.57 |
| 68 | -79.5 | 93.2 | -93.2 | -172.8 | -183.3 | 6.10 |
| 69 | 136.6 | 94.6 | -94.6 | 231.2 | 241.8 | 4.60 |
| 70 | 21.4 | 156.8 | -156.8 | 178.2 | 194.4 | 9.13 |
| 71 | -21.4 | 156.8 | -156.8 | -178.2 | -194.4 | 9.13 |
| 72 | -136.6 | 94.6 | -94.6 | -231.2 | -241.8 | 4.60 |
| 73 | 136.6 | 94.6 | -94.6 | 231.2 | 241.8 | 4.60 |
| 74 | 21.4 | 156.8 | -156.8 | 178.2 | 194.4 | 9.13 |
| 75 | -21.4 | 156.8 | -156.8 | -178.2 | -194.4 | 9.13 |
| 76 | -136.6 | 94.6 | -94.6 | -231.2 | -241.8 | 4.60 |
| 77 | 79.5 | 93.2 | -93.2 | 172.8 | 183.3 | 6.10 |
| 78 | 13.6 | 153.5 | -153.5 | 167.2 | 183.2 | 9.57 |
| 79 | -13.6 | 153.5 | -153.5 | -167.2 | -183.2 | 9.57 |
| 80 | -79.5 | 93.2 | -93.2 | -172.8 | -183.3 | 6.10 |
| 81 | 76.8 | 75.4 | -75.4 | 152.2 | 160.2 | 5.23 |
| 82 | 16.0 | 131.4 | -131.4 | 147.4 | 160.5 | 8.91 |
| 83 | -16.0 | 131.4 | -131.4 | -147.4 | -160.5 | 8.91 |
| 84 | -76.8 | 75.4 | -75.4 | -152.2 | -160.2 | 5.23 |
| 85 | 129.7 | 77.0 | -77.0 | 206.7 | 214.8 | 3.92 |
| 86 | 24.4 | 134.3 | -134.3 | 158.7 | 172.1 | 8.44 |
| 87 | -24.4 | 134.3 | -134.3 | -158.7 | -172.1 | 8.44 |
| 88 | -129.7 | 77.0 | -77.0 | -206.7 | -214.8 | 3.92 |
| 89 | 129.7 | 77.0 | -77.0 | 206.7 | 214.8 | 3.92 |

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| 90 | 24.4 | 134.3 | -134.3 | 158.7 | 172.1 | 8.44 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 91 | -24.4 | 134.3 | -134.3 | -158.7 | -172.1 | 8.44 |
| 92 | -129.7 | 77.0 | -77.0 | -206.7 | -214.8 | 3.92 |
| 93 | 76.8 | 75.4 | -75.4 | 152.2 | 160.2 | 5.23 |
| 94 | 16.0 | 131.4 | -131.4 | 147.4 | 160.5 | 8.91 |
| 95 | -16.0 | 131.4 | -131.4 | -147.4 | -160.5 | 8.91 |
| 96 | -76.8 | 75.4 | -75.4 | -152.2 | -160.2 | 5.23 |
| 97 | 86.0 | 61.3 | -61.3 | 147.2 | 152.8 | 3.77 |
| 98 | 19.7 | 123.3 | -123.3 | 143.0 | 154.3 | 7.91 |
| 99 | -19.7 | 123.3 | -123.3 | -143.0 | -154.3 | 7.91 |
| 100 | -86.0 | 61.3 | -61.3 | -147.2 | -152.8 | 3.77 |
| 101 | 145.7 | 62.7 | -62.7 | 208.4 | 214.1 | 2.73 |
| 102 | 30.5 | 126.5 | -126.5 | 157.1 | 168.7 | 7.40 |
| 103 | -30.5 | 126.5 | -126.5 | -157.1 | -168.7 | 7.40 |
| 104 | -145.7 | 62.7 | -62.7 | -208.4 | -214.1 | 2.73 |
| 105 | 145.7 | 62.7 | -62.7 | 208.4 | 214.1 | 2.73 |
| 106 | 30.5 | 126.5 | -126.5 | 157.1 | 168.7 | 7.40 |
| 107 | -30.5 | 126.5 | -126.5 | -157.1 | -168.7 | 7.40 |
| 108 | -145.7 | 62.7 | -62.7 | -208.4 | -214.1 | 2.73 |
| 109 | 86.0 | 61.3 | -61.3 | 147.2 | 152.8 | 3.77 |
| 110 | 19.7 | 123.3 | -123.3 | 143.0 | 154.3 | 7.91 |
| 111 | -19.7 | 123.3 | -123.3 | -143.0 | -154.3 | 7.91 |
| 112 | -86.0 | 61.3 | -61.3 | -147.2 | -152.8 | 3.77 |
| 113 | 89.1 | 55.5 | -55.5 | 144.5 | 148.6 | 2.85 |
| 114 | 23.0 | 116.4 | -116.4 | 139.5 | 149.2 | 6.95 |
| 115 | -23.0 | 116.4 | -116.4 | -139.5 | -149.2 | 6.95 |
| 116 | -89.1 | 55.5 | -55.5 | -144.5 | -148.6 | 2.85 |
| 117 | 149.7 | 57.1 | -57.1 | 206.8 | 211.1 | 2.07 |
| 118 | 35.4 | 119.6 | -119.6 | 155.0 | 165.0 | 6.44 |
| 119 | -35.4 | 119.6 | -119.6 | -155.0 | -165.0 | 6.44 |
| 120 | -149.7 | 57.1 | -57.1 | -206.8 | -211.1 | 2.07 |
| 121 | 149.7 | 57.1 | -57.1 | 206.8 | 211.1 | 2.07 |
| 122 | 35.4 | 119.6 | -119.6 | 155.0 | 165.0 | 6.44 |
| 123 | -35.4 | 119.6 | -119.6 | -155.0 | -165.0 | 6.44 |
| 124 | -149.7 | 57.1 | -57.1 | -206.8 | -211.1 | 2.07 |
| 125 | 89.1 | 55.5 | -55.5 | 144.5 | 148.6 | 2.85 |
| 126 | 23.0 | 116.4 | -116.4 | 139.5 | 149.2 | 6.95 |
| 127 | -23.0 | 116.4 | -116.4 | -139.5 | -149.2 | 6.95 |
| 128 | -89.1 | 55.5 | -55.5 | -144.5 | -148.6 | 2.85 |
| 129 | 92.0 | 51.3 | -51.3 | 143.4 | 146.5 | 2.19 |
| 130 | 25.9 | 111.5 | -111.5 | 137.4 | 145.9 | 6.17 |
| 131 | -25.9 | 111.5 | -111.5 | -137.4 | -145.9 | 6.17 |
| 132 | -92.0 | 51.3 | -51.3 | -143.4 | -146.5 | 2.19 |
| 133 | 154.1 | 52.9 | -52.9 | 207.0 | 210.3 | 1.59 |
| 134 | 39.6 | 114.6 | -114.6 | 154.2 | 163.0 | 5.68 |
| 135 | -39.6 | 114.6 | -114.6 | -154.2 | -163.0 | 5.68 |
| 136 | -154.1 | 52.9 | -52.9 | -207.0 | -210.3 | 1.59 |

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| 137 | 154.1 | 52.9 | -52.9 | 207.0 | 210.3 | 1.59 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 39.6 | 114.6 | -114.6 | 154.2 | 163.0 | 5.68 |
| 139 | -39.6 | 114.6 | -114.6 | -154.2 | -163.0 | 5.68 |
| 140 | -154.1 | 52.9 | -52.9 | -207.0 | -210.3 | 1.59 |
| 141 | 92.0 | 51.3 | -51.3 | 143.4 | 146.5 | 2.19 |
| 142 | 25.9 | 111.5 | -111.5 | 137.4 | 145.9 | 6.17 |
| 143 | -25.9 | 111.5 | -111.5 | -137.4 | -145.9 | 6.17 |
| 144 | -92.0 | 51.3 | -51.3 | -143.4 | -146.5 | 2.19 |
| 145 | 94.6 | 47.8 | -47.8 | 142.4 | 144.9 | 1.76 |
| 146 | 28.3 | 107.5 | -107.5 | 135.8 | 143.4 | 5.60 |
| 147 | -28.3 | 107.5 | -107.5 | -135.8 | -143.4 | 5.60 |
| 148 | -94.6 | 47.8 | -47.8 | -142.4 | -144.9 | 1.76 |
| 149 | 158.7 | 49.1 | -49.1 | 207.9 | 210.5 | 1.26 |
| 150 | 43.6 | 110.6 | -110.6 | 154.2 | 162.1 | 5.11 |
| 151 | -43.6 | 110.6 | -110.6 | -154.2 | -162.1 | 5.11 |
| 152 | -158.7 | 49.1 | -49.1 | -207.9 | -210.5 | 1.26 |
| 153 | 158.7 | 49.1 | -49.1 | 207.9 | 210.5 | 1.26 |
| 154 | 43.6 | 110.6 | -110.6 | 154.2 | 162.1 | 5.11 |
| 155 | -43.6 | 110.6 | -110.6 | -154.2 | -162.1 | 5.11 |
| 156 | -158.7 | 49.1 | -49.1 | -207.9 | -210.5 | 1.26 |
| 157 | 94.6 | 47.8 | -47.8 | 142.4 | 144.9 | 1.76 |
| 158 | 28.3 | 107.5 | -107.5 | 135.8 | 143.4 | 5.60 |
| 159 | -28.3 | 107.5 | -107.5 | -135.8 | -143.4 | 5.60 |
| 160 | -94.6 | 47.8 | -47.8 | -142.4 | -144.9 | 1.76 |
| 161 | 88.8 | 48.4 | -48.4 | 137.2 | 139.7 | 1.83 |
| 162 | 28.3 | 100.3 | -100.3 | 128.6 | 135.4 | 5.25 |
| 163 | -28.3 | 100.3 | -100.3 | -128.6 | -135.4 | 5.25 |
| 164 | -88.8 | 48.4 | -48.4 | -137.2 | -139.7 | 1.83 |
| 165 | 146.5 | 49.8 | -49.8 | 196.3 | 199.0 | 1.35 |
| 166 | 42.7 | 102.9 | -102.9 | 145.6 | 152.6 | 4.80 |
| 167 | -42.7 | 102.9 | -102.9 | -145.6 | -152.6 | 4.80 |
| 168 | -146.5 | 49.8 | -49.8 | -196.3 | -199.0 | 1.35 |
| 169 | 146.5 | 49.8 | -49.8 | 196.3 | 199.0 | 1.35 |
| 170 | 42.7 | 102.9 | -102.9 | 145.6 | 152.6 | 4.80 |
| 171 | -42.7 | 102.9 | -102.9 | -145.6 | -152.6 | 4.80 |
| 172 | -146.5 | 49.8 | -49.8 | -196.3 | -199.0 | 1.35 |
| 173 | 88.8 | 48.4 | -48.4 | 137.2 | 139.7 | 1.83 |
| 174 | 28.3 | 100.3 | -100.3 | 128.6 | 135.4 | 5.25 |
| 175 | -28.3 | 100.3 | -100.3 | -128.6 | -135.4 | 5.25 |
| 176 | -88.8 | 48.4 | -48.4 | -137.2 | -139.7 | 1.83 |
| 177 | 99.8 | 42.5 | -42.5 | 142.3 | 144.1 | 1.29 |
| 178 | 33.1 | 98.7 | -98.7 | 131.8 | 138.0 | 4.68 |
| 179 | -33.1 | 98.7 | -98.7 | -131.8 | -138.0 | 4.68 |
| 180 | -99.8 | 42.5 | -42.5 | -142.3 | -144.1 | 1.29 |
| 181 | 165.5 | 43.7 | -43.7 | 209.2 | 211.1 | 0.93 |
| 182 | 50.4 | 101.4 | -101.4 | 151.8 | 158.2 | 4.21 |
| 183 | -50.4 | 101.4 | -101.4 | -151.8 | -158.2 | 4.21 |

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| 184 | -165.5 | 43.7 | -43.7 | -209.2 | -211.1 | 0.93 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 185 | 165.5 | 43.7 | -43.7 | 209.2 | 211.1 | 0.93 |
| 186 | 50.4 | 101.4 | -101.4 | 151.8 | 158.2 | 4.21 |
| 187 | -50.4 | 101.4 | -101.4 | -151.8 | -158.2 | 4.21 |
| 188 | -165.5 | 43.7 | -43.7 | -209.2 | -211.1 | 0.93 |
| 189 | 99.8 | 42.5 | -42.5 | 142.3 | 144.1 | 1.29 |
| 190 | 33.1 | 98.7 | -98.7 | 131.8 | 138.0 | 4.68 |
| 191 | -33.1 | 98.7 | -98.7 | -131.8 | -138.0 | 4.68 |
| 192 | -99.8 | 42.5 | -42.5 | -142.3 | -144.1 | 1.29 |
| 193 | 102.0 | 40.4 | -40.4 | 142.4 | 143.9 | 1.06 |
| 194 | 35.7 | 94.0 | -94.0 | 129.7 | 135.1 | 4.16 |
| 195 | -35.7 | 94.0 | -94.0 | -129.7 | -135.1 | 4.16 |
| 196 | -102.0 | 40.4 | -40.4 | -142.4 | -143.9 | 1.06 |
| 197 | 167.9 | 41.7 | -41.7 | 209.6 | 211.2 | 0.77 |
| 198 | 54.0 | 96.6 | -96.6 | 150.6 | 156.2 | 3.71 |
| 199 | -54.0 | 96.6 | -96.6 | -150.6 | -156.2 | 3.71 |
| 200 | -167.9 | 41.7 | -41.7 | -209.6 | -211.2 | 0.77 |
| 201 | 167.9 | 41.7 | -41.7 | 209.6 | 211.2 | 0.77 |
| 202 | 54.0 | 96.6 | -96.6 | 150.6 | 156.2 | 3.71 |
| 203 | -54.0 | 96.6 | -96.6 | -150.6 | -156.2 | 3.71 |
| 204 | -167.9 | 41.7 | -41.7 | -209.6 | -211.2 | 0.77 |
| 205 | 102.0 | 40.4 | -40.4 | 142.4 | 143.9 | 1.06 |
| 206 | 35.7 | 94.0 | -94.0 | 129.7 | 135.1 | 4.16 |
| 207 | -35.7 | 94.0 | -94.0 | -129.7 | -135.1 | 4.16 |
| 208 | -102.0 | 40.4 | -40.4 | -142.4 | -143.9 | 1.06 |
| 209 | 104.5 | 38.4 | -38.4 | 142.9 | 144.1 | 0.87 |
| 210 | 38.1 | 90.2 | -90.2 | 128.3 | 133.0 | 3.70 |
| 211 | -38.1 | 90.2 | -90.2 | -128.3 | -133.0 | 3.70 |
| 212 | -104.5 | 38.4 | -38.4 | -142.9 | -144.1 | 0.87 |
| 213 | 171.4 | 39.6 | -39.6 | 211.0 | 212.3 | 0.63 |
| 214 | 57.4 | 92.7 | -92.7 | 150.1 | 155.0 | 3.28 |
| 215 | -57.4 | 92.7 | -92.7 | -150.1 | -155.0 | 3.28 |
| 216 | -171.4 | 39.6 | -39.6 | -211.0 | -212.3 | 0.63 |
| 217 | 17.4 | 39.6 | -39.6 | 211.0 | 212.3 | 0.63 |
| 218 | 57.4 | 92.7 | -92.7 | 150.1 | 155.0 | 3.28 |
| 219 | -57.4 | 92.7 | -92.7 | -150.1 | -155.0 | 3.28 |
| 220 | -171.4 | 39.6 | -39.6 | -211.0 | -212.3 | 0.63 |
| 221 | 104.5 | 38.4 | -38.4 | 142.9 | 144.1 | 0.87 |
| 222 | 38.1 | 90.2 | -90.2 | 128.3 | 133.0 | 3.70 |
| 223 | -38.1 | 90.2 | -90.2 | -128.3 | -133.0 | 3.70 |
| 224 | -104.5 | 38.4 | -38.4 | -142.9 | -144.1 | 0.87 |
| 225 | 107.1 | 36.2 | -36.2 | 143.3 | 144.4 | 0.72 |
| 226 | 40.3 | 87.2 | -87.2 | 127.5 | 131.7 | 3.34 |
| 227 | -40.3 | 87.2 | -87.2 | -127.5 | -131.7 | 3.34 |
| 228 | -107.1 | 36.2 | -36.2 | -143.3 | -144.4 | 0.72 |
| 229 | 176.4 | 37.2 | -37.2 | 213.6 | 214.7 | 0.52 |
| 230 | 61.1 | 89.7 | -89.7 | 150.7 | 155.2 | 2.94 |
|  |  |  |  |  |  |  |
| 10.7 |  |  |  |  |  |  |

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| 231 | -61.1 | 89.7 | -89.7 | -150.7 | -155.2 | 2.94 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 232 | -176.4 | 37.2 | -37.2 | -213.6 | -214.7 | 0.52 |
| 233 | 176.4 | 37.2 | -37.2 | 213.6 | 214.7 | 0.52 |
| 234 | 61.1 | 89.7 | -89.7 | 150.7 | 155.2 | 2.94 |
| 235 | -61.1 | 89.7 | -89.7 | -150.7 | -155.2 | 2.94 |
| 236 | -176.4 | 37.2 | -37.2 | -213.6 | -214.7 | 0.52 |
| 237 | 107.1 | 36.2 | -36.2 | 143.3 | 144.4 | 0.72 |
| 238 | 40.3 | 87.2 | -87.2 | 127.5 | 131.7 | 3.34 |
| 239 | -40.3 | 87.2 | -87.2 | -127.5 | -131.7 | 3.34 |
| 240 | -107.1 | 36.2 | -36.2 | -143.3 | -144.4 | 0.72 |
| 241 | 97.4 | 38.0 | -38.0 | 135.4 | 136.6 | 0.91 |
| 242 | 37.4 | 80.6 | -80.6 | 118.0 | 121.7 | 3.16 |
| 243 | -37.4 | 80.6 | -80.6 | -118.0 | -121.7 | 3.16 |
| 244 | -97.4 | 38.0 | -38.0 | -135.4 | -136.6 | 0.91 |
| 245 | 157.7 | 39.1 | -39.1 | 196.8 | 198.1 | 0.66 |
| 246 | 55.7 | 82.6 | -82.6 | 138.3 | 142.2 | 2.79 |
| 247 | -55.7 | 82.6 | -82.6 | -138.3 | -142.2 | 2.79 |
| 248 | -157.7 | 39.1 | -39.1 | -196.8 | -198.1 | 0.66 |
| 249 | 157.7 | 39.1 | -39.1 | 196.8 | 198.1 | 0.66 |
| 250 | 55.7 | 82.6 | -82.6 | 138.3 | 142.2 | 2.79 |
| 251 | -55.7 | 82.6 | -82.6 | -138.3 | -142.2 | 2.79 |
| 252 | -157.7 | 39.1 | -39.1 | -196.8 | -198.1 | 0.66 |
| 253 | 97.4 | 38.0 | -38.0 | 135.4 | 136.6 | 0.91 |
| 254 | 37.4 | 80.6 | -80.6 | 118.0 | 121.7 | 3.16 |
| 255 | -37.4 | 80.6 | -80.6 | -118.0 | -121.7 | 3.16 |
| 256 | -97.4 | 38.0 | -38.0 | -135.4 | -136.6 | 0.91 |
| 257 | 110.6 | 32.3 | -32.3 | 142.9 | 143.7 | 0.56 |
| 258 | 43.2 | 78.5 | -78.5 | 121.7 | 125.0 | 2.70 |
| 259 | -43.2 | 78.5 | -78.5 | -121.7 | -125.0 | 2.70 |
| 260 | -110.6 | 32.3 | -32.3 | -142.9 | -143.7 | 0.56 |
| 261 | 179.9 | 33.2 | -33.2 | 213.1 | 214.0 | 0.40 |
| 262 | 64.8 | 80.5 | -80.5 | 145.4 | 148.8 | 2.34 |
| 263 | -64.8 | 80.5 | -80.5 | -145.4 | -148.8 | 2.34 |
| 264 | -179.9 | 33.2 | -33.2 | -213.1 | -214.0 | 0.40 |
| 265 | 179.9 | 33.2 | -33.2 | 213.1 | 214.0 | 0.40 |
| 266 | 64.8 | 80.5 | -80.5 | 145.4 | 148.8 | 2.34 |
| 267 | -64.8 | 80.5 | -80.5 | -145.4 | -148.8 | 2.34 |
| 268 | -179.9 | 33.2 | -33.2 | -213.1 | -214.0 | 0.40 |
| 269 | 110.6 | 32.3 | -32.3 | 142.9 | 143.7 | 0.56 |
| 270 | 43.2 | 78.5 | -78.5 | 121.7 | 125.0 | 2.70 |
| 271 | -43.2 | 78.5 | -78.5 | -121.7 | -125.0 | 2.70 |
| 272 | -110.6 | 32.3 | -32.3 | -142.9 | -143.7 | 0.56 |
| 273 | 111.4 | 29.7 | -29.7 | 141.2 | 141.7 | 0.41 |
| 274 | 44.9 | 72.6 | -72.6 | 117.5 | 120.1 | 2.28 |
| 275 | -44.9 | 72.6 | -72.6 | -117.5 | -120.1 | 2.28 |
| 276 | -111.4 | 29.7 | -29.7 | -141.2 | -141.7 | 0.41 |
| 277 | 180.3 | 30.6 | -30.6 | 210.9 | 211.6 | 0.30 |

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| 278 | 67.0 | 74.5 | -74.5 | 141.5 | 144.2 | 1.96 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 279 | -67.0 | 74.5 | -74.5 | -141.5 | -144.2 | 1.96 |
| 280 | -180.3 | 30.6 | -30.6 | -210.9 | -211.6 | 0.30 |
| 281 | 180.3 | 30.6 | -30.6 | 210.9 | 211.6 | 0.30 |
| 282 | 67.0 | 74.5 | -74.5 | 141.5 | 144.2 | 1.96 |
| 283 | -67.0 | 74.5 | -74.5 | -141.5 | -144.2 | 1.96 |
| 284 | -180.3 | 30.6 | -30.6 | -210.9 | -211.6 | 0.30 |
| 285 | 111.4 | 29.7 | -29.7 | 141.2 | 141.7 | 0.41 |
| 286 | 44.9 | 72.6 | -72.6 | 117.5 | 120.1 | 2.28 |
| 287 | -44.9 | 72.6 | -72.6 | -117.5 | -120.1 | 2.28 |
| 288 | -111.4 | 29.7 | -29.7 | -141.2 | -141.7 | 0.41 |
| 289 | 113.3 | 27.1 | -27.1 | 140.3 | 140.7 | 0.28 |
| 290 | 46.6 | 67.5 | -67.5 | 114.1 | 116.3 | 1.91 |
| 291 | -46.6 | 67.5 | -67.5 | -114.1 | -116.3 | 1.91 |
| 292 | -113.3 | 27.1 | -27.1 | -140.3 | -140.7 | 0.28 |
| 293 | 182.6 | 27.9 | -27.9 | 210.5 | 211.0 | 0.20 |
| 294 | 69.3 | 69.2 | -69.2 | 138.6 | 140.8 | 1.63 |
| 295 | -69.3 | 69.2 | -69.2 | -138.6 | -140.8 | 1.63 |
| 296 | -182.6 | 27.9 | -27.9 | -210.5 | -211.0 | 0.20 |
| 297 | 182.6 | 27.9 | -27.9 | 210.5 | 211.0 | 0.20 |
| 298 | 69.3 | 69.2 | -69.2 | 138.6 | 140.8 | 1.63 |
| 299 | -69.3 | 69.2 | -69.2 | -138.6 | -140.8 | 1.63 |
| 300 | -182.6 | 27.9 | -27.9 | -210.5 | -211.0 | 0.20 |
| 301 | 113.3 | 27.1 | -27.1 | 140.3 | 140.7 | 0.28 |
| 302 | 46.6 | 67.5 | -67.5 | 114.1 | 116.3 | 1.91 |
| 303 | -46.6 | 67.5 | -67.5 | -114.1 | -116.3 | 1.91 |
| 304 | -113.3 | 27.1 | -27.1 | -140.3 | -140.7 | 0.28 |
| 305 | 117.0 | 24.5 | -24.5 | 141.6 | 141.8 | 0.18 |
| 306 | 49.0 | 64.2 | -64.2 | 113.2 | 115.1 | 1.63 |
| 307 | -49.0 | 64.2 | -64.2 | -113.2 | -115.1 | 1.63 |
| 308 | -117.0 | 24.5 | -24.5 | -141.6 | -141.8 | 0.18 |
| 309 | 190.1 | 25.2 | -25.2 | 215.3 | 215.6 | 0.14 |
| 310 | 73.5 | 66.0 | -66.0 | 139.5 | 141.4 | 1.38 |
| 311 | -73.5 | 66.0 | -66.0 | -139.5 | -141.4 | 1.38 |
| 312 | -190.1 | 25.2 | -25.2 | -215.3 | -215.6 | 0.14 |
| 313 | 190.1 | 25.2 | -25.2 | 215.3 | 215.6 | 0.14 |
| 314 | 73.5 | 66.0 | -66.0 | 139.5 | 141.4 | 1.38 |
| 315 | -73.5 | 66.0 | -66.0 | -139.5 | -141.4 | 1.38 |
| 316 | -190.1 | 25.2 | -25.2 | -215.3 | -215.6 | 0.14 |
| 317 | 117.0 | 24.5 | -24.5 | 141.6 | 141.8 | 0.18 |
| 318 | 49.0 | 64.2 | -64.2 | 113.2 | 115.1 | 1.63 |
| 319 | -49.0 | 64.2 | -64.2 | -113.2 | -115.1 | 1.63 |
| 320 | -117.0 | 24.5 | -24.5 | -141.6 | -141.8 | 0.18 |
| 321 | 93.6 | 28.2 | -28.2 | 121.9 | 122.4 | 0.41 |
| 322 | 38.5 | 56.5 | -56.5 | 94.9 | 96.4 | 1.59 |
| 323 | -38.5 | 56.5 | -56.5 | -94.9 | -96.4 | 1.59 |
| 324 | -93.6 | 28.2 | -28.2 | -121.9 | -122.4 | 0.41 |

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| 325 | 148.3 | 28.9 | -28.9 | 177.2 | 177.7 | 0.30 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 326 | 56.3 | 57.6 | -57.6 | 113.9 | 115.4 | 1.36 |
| 327 | -56.3 | 57.6 | -57.6 | -113.9 | -115.4 | 1.36 |
| 328 | -148.3 | 28.9 | -28.9 | -177.2 | -177.7 | 0.30 |
| 329 | 148.3 | 28.9 | -28.9 | 177.2 | 177.7 | 0.30 |
| 330 | 56.3 | 57.6 | -57.6 | 113.9 | 115.4 | 1.36 |
| 331 | -56.3 | 57.6 | -57.6 | -113.9 | -115.4 | 1.36 |
| 332 | -148.3 | 28.9 | -28.9 | -177.2 | -177.7 | 0.30 |
| 333 | 93.6 | 28.2 | -28.2 | 121.9 | 122.4 | 0.41 |
| 334 | 38.5 | 56.5 | -56.5 | 94.9 | 96.4 | 1.59 |
| 335 | -38.5 | 56.5 | -56.5 | -94.9 | -96.4 | 1.59 |
| 336 | -93.6 | 28.2 | -28.2 | -121.9 | -122.4 | 0.41 |
| 337 | 116.6 | 20.2 | -20.2 | 136.8 | 136.9 | 0.10 |
| 338 | 47.5 | 52.5 | -52.5 | 100.0 | 101.1 | 1.15 |
| 339 | -47.5 | 52.5 | -52.5 | -100.0 | -101.1 | 1.15 |
| 340 | -116.6 | 20.2 | -20.2 | -136.8 | -136.9 | 0.10 |
| 341 | 185.3 | 20.7 | -20.7 | 206.0 | 206.1 | 0.08 |
| 342 | 69.8 | 53.7 | -53.7 | 123.5 | 124.7 | 0.96 |
| 343 | -69.8 | 53.7 | -53.7 | -123.5 | -124.7 | 0.96 |
| 344 | -185.3 | 20.7 | -20.7 | -206.0 | -206.1 | 0.08 |
| 345 | 185.3 | 20.7 | -20.7 | 206.0 | 206.1 | 0.08 |
| 346 | 69.8 | 53.7 | -53.7 | 123.5 | 124.7 | 0.96 |
| 347 | -69.8 | 53.7 | -53.7 | -123.5 | -124.7 | 0.96 |
| 348 | -185.3 | 20.7 | -20.7 | -206.0 | -206.1 | 0.08 |
| 349 | 116.6 | 20.2 | -20.2 | 136.8 | 136.9 | 0.10 |
| 350 | 47.5 | 52.5 | -52.5 | 100.0 | 101.1 | 1.15 |
| 351 | -47.5 | 52.5 | -52.5 | -100.0 | -101.1 | 1.15 |
| 352 | -116.6 | 20.2 | -20.2 | -136.8 | -136.9 | 0.10 |
| 353 | 114.7 | 13.9 | -13.9 | 128.5 | 128.5 | -0.02 |
| 354 | 47.6 | 40.7 | -40.7 | 88.3 | 89.0 | 0.78 |
| 355 | -47.6 | 40.7 | -40.7 | -88.3 | -89.0 | 0.78 |
| 356 | -114.7 | 13.9 | -13.9 | -128.5 | -128.5 | -0.02 |
| 357 | 181.9 | 14.3 | -14.3 | 196.2 | 196.2 | -0.01 |
| 358 | 69.9 | 41.7 | -41.7 | 111.6 | 112.3 | 0.64 |
| 359 | -69.9 | 41.7 | -41.7 | -111.6 | -112.3 | 0.64 |
| 360 | -181.9 | 14.3 | -14.3 | -196.2 | -196.2 | -0.01 |
| 361 | 181.9 | 14.3 | -14.3 | 196.2 | 196.2 | -0.01 |
| 362 | 69.9 | 41.7 | -41.7 | 111.6 | 112.3 | 0.64 |
| 363 | -69.9 | 41.7 | -41.7 | -111.6 | -112.3 | 0.64 |
| 364 | -181.9 | 14.3 | -14.3 | -196.2 | -196.2 | -0.01 |
| 365 | 114.7 | 13.9 | -13.9 | 128.5 | 128.5 | -0.02 |
| 366 | 47.6 | 40.7 | -40.7 | 88.3 | 89.0 | 0.78 |
| 367 | -47.6 | 40.7 | -40.7 | -88.3 | -89.0 | 0.78 |
| 368 | -114.7 | 13.9 | -13.9 | -128.5 | -128.5 | -0.02 |
| 369 | 113.8 | 4.9 | -4.9 | 118.7 | 118.5 | -0.15 |
| 370 | 48.5 | 26.1 | -26.1 | 74.6 | 75.0 | 0.46 |
| 371 | -48.5 | 26.1 | -26.1 | -74.6 | -75.0 | 0.46 |

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| 372 | -113.8 | 4.9 | -4.9 | -118.7 | -118.5 | -0.15 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 373 | 179.4 | 5.3 | -5.3 | 184.6 | 184.5 | -0.10 |
| 374 | 70.7 | 26.8 | -26.8 | 97.5 | 97.9 | 0.37 |
| 375 | -70.7 | 26.8 | -26.8 | -97.5 | -97.9 | 0.37 |
| 376 | -179.4 | 5.3 | -5.3 | -184.6 | -184.5 | -0.10 |
| 377 | 179.4 | 5.3 | -5.3 | 184.6 | 184.5 | -0.10 |
| 378 | 70.7 | 26.8 | -26.8 | 97.5 | 97.9 | 0.37 |
| 379 | -70.7 | 26.8 | -26.8 | -97.5 | -97.9 | 0.37 |
| 380 | -179.4 | 5.3 | -5.3 | -184.6 | -184.5 | -0.10 |
| 381 | 113.8 | 4.9 | -4.9 | 118.7 | 118.5 | -0.15 |
| 382 | 48.5 | 26.1 | -26.1 | 74.6 | 75.0 | 0.46 |
| 383 | -48.5 | 26.1 | -26.1 | -74.6 | -75.0 | 0.46 |
| 384 | -113.8 | 4.9 | -4.9 | -118.7 | -118.5 | -0.15 |
| 385 | 133.9 | -4.5 | 4.5 | 138.4 | 138.7 | 0.19 |
| 386 | 54.5 | 12.7 | -12.7 | 67.2 | 67.4 | 0.30 |
| 387 | -54.5 | 12.7 | -12.7 | -67.2 | -67.4 | 0.30 |
| 388 | -133.9 | -4.5 | 4.5 | -138.4 | -138.7 | 0.19 |
| 389 | 213.7 | -4.5 | 4.5 | 218.2 | 218.5 | 0.13 |
| 390 | 80.8 | 13.1 | -13.1 | 93.9 | 94.2 | 0.24 |
| 391 | -80.8 | 13.1 | -13.1 | -93.9 | -94.2 | 0.24 |
| 392 | -213.7 | -4.5 | 4.5 | -218.2 | -218.5 | 0.13 |
| 393 | 213.7 | -4.5 | 4.5 | 218.2 | 218.5 | 0.13 |
| 394 | 80.8 | 13.1 | -13.1 | 93.9 | 94.2 | 0.24 |
| 395 | -80.8 | 13.1 | -13.1 | -93.9 | -94.2 | 0.24 |
| 396 | -213.7 | -4.5 | 4.5 | -218.2 | -218.5 | 0.13 |
| 397 | 133.9 | -4.5 | 4.5 | 138.4 | 138.7 | 0.19 |
| 398 | 54.5 | 12.7 | -12.7 | 67.2 | 67.4 | 0.30 |
| 399 | -54.5 | 12.7 | -12.7 | -67.2 | -67.4 | 0.30 |
| 400 | -133.9 | -4.5 | 4.5 | -138.4 | -138.7 | 0.19 |

Table B. 5 M30 First and Second-Order Moments

| M30 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathrm{M}_{\text {SD+D+L }}$ | $\begin{gathered} \text { M }_{\text {Positive }} \\ \text { EQ. } \end{gathered}$ | $\mathrm{M}_{\text {Negative }}$ е Q . | FirstOrder Moment | SecondOrder Moment | Percent |
| 1 | 32.0 | 474.5 | -474.5 | 506.4 | 539.8 | 6.59 |
| 2 | 3.6 | 484.6 | -484.6 | 488.3 | 522.2 | 6.94 |
| 3 | -3.6 | 484.6 | -484.6 | -488.3 | -522.2 | 6.94 |
| 4 | -32.0 | 474.5 | -474.5 | -506.4 | -539.8 | 6.59 |
| 5 | 52.2 | 475.6 | -475.6 | 527.8 | 560.6 | 6.22 |
| 6 | 4.6 | 485.3 | -485.3 | 489.9 | 523.1 | 6.77 |
| 7 | -4.6 | 485.3 | -485.3 | -489.9 | -523.1 | 6.77 |
| 8 | -52.2 | 475.6 | -475.6 | -527.8 | -560.6 | 6.22 |
| 9 | 52.2 | 475.6 | -475.6 | 527.8 | 560.6 | 6.22 |
| 10 | 4.6 | 485.3 | -485.3 | 489.9 | 523.1 | 6.77 |
| 11 | -4.6 | 485.3 | -485.3 | -489.9 | -523.1 | 6.77 |
| 12 | -52.2 | 475.6 | -475.6 | -527.8 | -560.6 | 6.22 |
| 13 | 32.0 | 474.5 | -474.5 | 506.4 | 539.8 | 6.59 |
| 14 | 3.6 | 484.6 | -484.6 | 488.3 | 522.2 | 6.94 |
| 15 | -3.6 | 484.6 | -484.6 | -488.3 | -522.2 | 6.94 |
| 16 | -32.0 | 474.5 | -474.5 | -506.4 | -539.8 | 6.59 |
| 17 | 70.5 | 325.8 | -325.8 | 396.4 | 428.3 | 8.05 |
| 18 | 1.9 | 354.1 | -354.1 | 356.0 | 390.3 | 9.63 |
| 19 | -1.9 | 354.1 | -354.1 | -356.0 | -390.3 | 9.63 |
| 20 | -70.5 | 325.8 | -325.8 | -396.4 | -428.3 | 8.05 |
| 21 | 125.1 | 326.4 | -326.4 | 451.5 | 483.5 | 7.09 |
| 22 | 3.7 | 355.8 | -355.8 | 359.4 | 393.8 | 9.57 |
| 23 | -3.7 | 355.8 | -355.8 | -359.4 | -393.8 | 9.57 |
| 24 | -125.1 | 326.4 | -326.4 | -451.5 | -483.5 | 7.09 |
| 25 | 125.1 | 326.4 | -326.4 | 451.5 | 483.5 | 7.09 |
| 26 | 3.7 | 355.8 | -355.8 | 359.4 | 393.8 | 9.57 |
| 27 | -3.7 | 355.8 | -355.8 | -359.4 | -393.8 | 9.57 |
| 28 | -125.1 | 326.4 | -326.4 | -451.5 | -483.5 | 7.09 |
| 29 | 70.5 | 325.8 | -325.8 | 396.4 | 428.3 | 8.05 |
| 30 | 1.9 | 354.1 | -354.1 | 356.0 | 390.3 | 9.63 |
| 31 | -1.9 | 354.1 | -354.1 | -356.0 | -390.3 | 9.63 |
| 32 | -70.5 | 325.8 | -325.8 | -396.4 | -428.3 | 8.05 |
| 33 | 74.9 | 225.4 | -225.4 | 300.3 | 327.5 | 9.08 |
| 34 | 8.2 | 268.6 | -268.6 | 276.8 | 308.1 | 11.33 |
| 35 | -8.2 | 268.6 | -268.6 | -276.8 | -308.1 | 11.33 |
| 36 | -74.9 | 225.4 | -225.4 | -300.3 | -327.5 | 9.08 |
| 37 | 128.7 | 226.5 | -226.5 | 355.2 | 382.5 | 7.68 |
| 38 | 12.5 | 271.0 | -271.0 | 283.5 | 314.9 | 11.10 |
| 39 | -12.5 | 271.0 | -271.0 | -283.5 | -314.9 | 11.10 |
| 40 | -128.7 | 226.5 | -226.5 | -355.2 | -382.5 | 7.68 |
| 41 | 128.7 | 226.5 | -226.5 | 355.2 | 382.5 | 7.68 |
| 42 | 12.5 | 271.0 | -271.0 | 283.5 | 314.9 | 11.10 |

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$\left.\begin{array}{|r|r|r|r|r|r|r|}\hline 43 & -12.5 & 271.0 & -271.0 & -283.5 & -314.9 & 11.10 \\ \hline 44 & -128.7 & 226.5 & -226.5 & -355.2 & -382.5 & 7.68 \\ \hline 45 & 74.9 & 225.4 & -225.4 & 300.3 & 327.5 & 9.08 \\ \hline 46 & 8.2 & 268.6 & -268.6 & 276.8 & 308.1 & 11.33 \\ \hline 47 & -8.2 & 268.6 & -268.6 & -276.8 & -308.1 & 11.33 \\ \hline 48 & -74.9 & 225.4 & -225.4 & -300.3 & -327.5 & 9.08 \\ \hline 49 & 76.7 & 158.8 & -158.8 & 235.5 & 257.3 & 9.23 \\ \hline 50 & 10.8 & 210.0 & -210.0 & 220.8 & 247.9 & 12.25 \\ \hline 51 & -10.8 & 210.0 & -210.0 & -220.8 & -247.9 & 12.25 \\ \hline 52 & -76.7 & 158.8 & -158.8 & -235.5 & -257.3 & 9.23 \\ \hline 53 & 131.6 & 160.2 & -160.2 & 291.8 & 313.6 & 7.48 \\ \hline 54 & 16.6 & 212.8 & -212.8 & 229.4 & 256.6 & 11.88 \\ \hline 55 & -16.6 & 212.8 & -212.8 & -229.4 & -256.6 & 11.88 \\ \hline 56 & -131.6 & 160.2 & -160.2 & -291.8 & -313.6 & 7.48 \\ \hline 57 & 131.6 & 160.2 & -160.2 & 291.8 & 313.6 & 7.48 \\ \hline 58 & 16.6 & 212.8 & -212.8 & 229.4 & 256.6 & 11.88 \\ \hline 59 & -16.6 & 212.8 & -212.8 & -229.4 & -256.6 & 11.88 \\ \hline 60 & -131.6 & 160.2 & -160.2 & -291.8 & -313.6 & 7.48 \\ \hline 61 & 76.7 & 158.8 & -158.8 & 235.5 & 257.3 & 9.23 \\ \hline 62 & 10.8 & 210.0 & -210.0 & 220.8 & 247.9 & 12.25 \\ \hline 63 & -10.8 & 210.0 & -210.0 & -220.8 & -247.9 & 12.25 \\ \hline 64 & -76.7 & 158.8 & -158.8 & -235.5 & -257.3 & 9.23 \\ \hline 65 & 79.0 & 114.6 & -114.6 & 193.6 & 210.2 & 8.59 \\ \hline 66 & 13.3 & 170.2 & -170.2 & 183.5 & 206.3 & \mathbf{1 2 . 4 0} \\ \hline 67 & -13.3 & 170.2 & -170.2 & -183.5 & -206.3 & \mathbf{1 2 . 4 0} \\ \hline 68 & -79.0 & 114.6 & -114.6 & -193.6 & -210.2 & 8.59 \\ \hline 69 & 135.6 & 115.9 & -115.9 & 251.5 & 268.3 & 6.65 \\ \hline 70 & 20.6 & 173.3 & -173.3 & 193.9 & 217.0 & 11.88 \\ \hline 71 & -20.6 & 173.3 & -173.3 & -193.9 & -217.0 & 11.88 \\ \hline 72 & -135.6 & 115.9 & -115.9 & -251.5 & -268.3 & 6.65 \\ \hline 73 & 135.6 & 115.9 & -115.9 & 251.5 & 268.3 & 6.65 \\ \hline 74 & 20.6 & 173.3 & -173.3 & 193.9 & 217.0 & 11.88 \\ \hline 75 & -20.6 & 173.3 & -173.3 & -193.9 & -217.0 & 11.88 \\ \hline 76 & -135.6 & 115.9 & -115.9 & -251.5 & -268.3 & 6.65 \\ \hline 77 & 79.0 & 114.6 & -114.6 & 193.6 & 210.2 & 8.59 \\ \hline 78 & 13.3 & 170.2 & -170.2 & 183.5 & 206.3 & \mathbf{1 2 . 4 0} \\ \hline 79 & -13.3 & 170.2 & -170.2 & -183.5 & -206.3 & \mathbf{1 2 . 4 0} \\ \hline 80 & -79.0 & 114.6 & -114.6 & -193.6 & -210.2 & 8.59 \\ \hline 87.3 & 87.9 & -87.9 & 165.3 & 178.0 & 7.67 \\ \hline 81 & 77.3 & 15.9 & 141.3 & -141.3 & 157.1 & 175.9\end{array}\right) 11.939$.

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| 90 | 24.1 | 144.0 | -144.0 | 168.1 | 187.2 | 11.33 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 91 | -24.1 | 144.0 | -144.0 | -168.1 | -187.2 | 11.33 |
| 92 | -130.7 | 89.5 | -89.5 | -220.2 | -233.0 | 5.84 |
| 93 | 77.3 | 87.9 | -87.9 | 165.3 | 178.0 | 7.67 |
| 94 | 15.9 | 141.3 | -141.3 | 157.1 | 175.9 | 11.93 |
| 95 | -15.9 | 141.3 | -141.3 | -157.1 | -175.9 | 11.93 |
| 96 | -77.3 | 87.9 | -87.9 | -165.3 | -178.0 | 7.67 |
| 97 | 85.4 | 68.4 | -68.4 | 153.8 | 162.9 | 5.94 |
| 98 | 19.4 | 127.5 | -127.5 | 146.9 | 162.9 | 10.91 |
| 99 | -19.4 | 127.5 | -127.5 | -146.9 | -162.9 | 10.91 |
| 100 | -85.4 | 68.4 | -68.4 | -153.8 | -162.9 | 5.94 |
| 101 | 144.8 | 69.8 | -69.8 | 214.6 | 223.9 | 4.33 |
| 102 | 29.8 | 130.6 | -130.6 | 160.4 | 176.8 | 10.22 |
| 103 | -29.8 | 130.6 | -130.6 | -160.4 | -176.8 | 10.22 |
| 104 | -144.8 | 69.8 | -69.8 | -214.6 | -223.9 | 4.33 |
| 105 | 144.8 | 69.8 | -69.8 | 214.6 | 223.9 | 4.33 |
| 106 | 29.8 | 130.6 | -130.6 | 160.4 | 176.8 | 10.22 |
| 107 | -29.8 | 130.6 | -130.6 | -160.4 | -176.8 | 10.22 |
| 108 | -144.8 | 69.8 | -69.8 | -214.6 | -223.9 | 4.33 |
| 109 | 85.4 | 68.4 | -68.4 | 153.8 | 162.9 | 5.94 |
| 110 | 19.4 | 127.5 | -127.5 | 146.9 | 162.9 | 10.91 |
| 111 | -19.4 | 127.5 | -127.5 | -146.9 | -162.9 | 10.91 |
| 112 | -85.4 | 68.4 | -68.4 | -153.8 | -162.9 | 5.94 |
| 113 | 88.8 | 58.9 | -58.9 | 147.7 | 154.5 | 4.60 |
| 114 | 22.8 | 117.8 | -117.8 | 140.6 | 154.3 | 9.73 |
| 115 | -22.8 | 117.8 | -117.8 | -140.6 | -154.3 | 9.73 |
| 116 | -88.8 | 58.9 | -58.9 | -147.7 | -154.5 | 4.60 |
| 117 | 149.4 | 60.4 | -60.4 | 209.8 | 216.8 | 3.32 |
| 118 | 34.7 | 120.9 | -120.9 | 155.6 | 169.6 | 9.03 |
| 119 | -34.7 | 120.9 | -120.9 | -155.6 | -169.6 | 9.03 |
| 120 | -149.4 | 60.4 | -60.4 | -209.8 | -216.8 | 3.32 |
| 121 | 149.4 | 60.4 | -60.4 | 209.8 | 216.8 | 3.32 |
| 122 | 34.7 | 120.9 | -120.9 | 155.6 | 169.6 | 9.03 |
| 123 | -34.7 | 120.9 | -120.9 | -155.6 | -169.6 | 9.03 |
| 124 | -149.4 | 60.4 | -60.4 | -209.8 | -216.8 | 3.32 |
| 125 | 88.8 | 58.9 | -58.9 | 147.7 | 154.5 | 4.60 |
| 126 | 22.8 | 117.8 | -117.8 | 140.6 | 154.3 | 9.73 |
| 127 | -22.8 | 117.8 | -117.8 | -140.6 | -154.3 | 9.73 |
| 128 | -88.8 | 58.9 | -58.9 | -147.7 | -154.5 | 4.60 |
| 129 | 91.8 | 53.5 | -53.5 | 145.3 | 150.5 | 3.54 |
| 130 | 25.7 | 112.2 | -112.2 | 137.8 | 149.8 | 8.66 |
| 131 | -25.7 | 112.2 | -112.2 | -137.8 | -149.8 | 8.66 |
| 132 | -91.8 | 53.5 | -53.5 | -145.3 | -150.5 | 3.54 |
| 133 | 153.8 | 55.0 | -55.0 | 208.8 | 214.1 | 2.55 |
| 134 | 39.0 | 115.2 | -115.2 | 154.2 | 166.5 | 7.98 |
| 135 | -39.0 | 115.2 | -115.2 | -154.2 | -166.5 | 7.98 |
| 136 | -153.8 | 55.0 | -55.0 | -208.8 | -214.1 | 2.55 |
|  |  |  |  |  |  |  |
| 10 |  |  |  |  |  |  |

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| 137 | 153.8 | 55.0 | -55.0 | 208.8 | 214.1 | 2.55 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 39.0 | 115.2 | -115.2 | 154.2 | 166.5 | 7.98 |
| 139 | -39.0 | 115.2 | -115.2 | -154.2 | -166.5 | 7.98 |
| 140 | -153.8 | 55.0 | -55.0 | -208.8 | -214.1 | 2.55 |
| 141 | 91.8 | 53.5 | -53.5 | 145.3 | 150.5 | 3.54 |
| 142 | 25.7 | 112.2 | -112.2 | 137.8 | 149.8 | 8.66 |
| 143 | -25.7 | 112.2 | -112.2 | -137.8 | -149.8 | 8.66 |
| 144 | -91.8 | 53.5 | -53.5 | -145.3 | -150.5 | 3.54 |
| 145 | 94.0 | 50.0 | -50.0 | 144.0 | 148.1 | 2.81 |
| 146 | 28.0 | 108.5 | -108.5 | 136.5 | 147.1 | 7.83 |
| 147 | -28.0 | 108.5 | -108.5 | -136.5 | -147.1 | 7.83 |
| 148 | -94.0 | 50.0 | -50.0 | -144.0 | -148.1 | 2.81 |
| 149 | 157.6 | 51.4 | -51.4 | 209.0 | 213.2 | 2.01 |
| 150 | 42.7 | 111.5 | -111.5 | 154.2 | 165.3 | 7.16 |
| 151 | -42.7 | 111.5 | -111.5 | -154.2 | -165.3 | 7.16 |
| 152 | -157.6 | 51.4 | -51.4 | -209.0 | -213.2 | 2.01 |
| 153 | 157.6 | 51.4 | -51.4 | 209.0 | 213.2 | 2.01 |
| 154 | 42.7 | 111.5 | -111.5 | 154.2 | 165.3 | 7.16 |
| 155 | -42.7 | 111.5 | -111.5 | -154.2 | -165.3 | 7.16 |
| 156 | -157.6 | 51.4 | -51.4 | -209.0 | -213.2 | 2.01 |
| 157 | 94.0 | 50.0 | -50.0 | 144.0 | 148.1 | 2.81 |
| 158 | 28.0 | 108.5 | -108.5 | 136.5 | 147.1 | 7.83 |
| 159 | -28.0 | 108.5 | -108.5 | -136.5 | -147.1 | 7.83 |
| 160 | -94.0 | 50.0 | -50.0 | -144.0 | -148.1 | 2.81 |
| 161 | 90.0 | 50.3 | -50.3 | 140.2 | 144.0 | 2.68 |
| 162 | 28.7 | 102.5 | -102.5 | 131.2 | 140.8 | 7.26 |
| 163 | -28.7 | 102.5 | -102.5 | -131.2 | -140.8 | 7.26 |
| 164 | -90.0 | 50.3 | -50.3 | -140.2 | -144.0 | 2.68 |
| 165 | 148.4 | 51.7 | -51.7 | 200.2 | 204.1 | 1.96 |
| 166 | 43.0 | 105.2 | -105.2 | 148.2 | 158.0 | 6.63 |
| 167 | -43.0 | 105.2 | -105.2 | -148.2 | -158.0 | 6.63 |
| 168 | -148.4 | 51.7 | -51.7 | -200.2 | -204.1 | 1.96 |
| 169 | 148.4 | 51.7 | -51.7 | 200.2 | 204.1 | 1.96 |
| 170 | 43.0 | 105.2 | -105.2 | 148.2 | 158.0 | 6.63 |
| 171 | -43.0 | 105.2 | -105.2 | -148.2 | -158.0 | 6.63 |
| 172 | -148.4 | 51.7 | -51.7 | -200.2 | -204.1 | 1.96 |
| 173 | 90.0 | 50.3 | -50.3 | 140.2 | 144.0 | 2.68 |
| 174 | 28.7 | 102.5 | -102.5 | 131.2 | 140.8 | 7.26 |
| 175 | -28.7 | 102.5 | -102.5 | -131.2 | -140.8 | 7.26 |
| 176 | -90.0 | 50.3 | -50.3 | -140.2 | -144.0 | 2.68 |
| 177 | 99.6 | 44.6 | -44.6 | 144.2 | 147.0 | 1.96 |
| 178 | 33.2 | 101.1 | -101.1 | 134.2 | 143.1 | 6.57 |
| 179 | -33.2 | 101.1 | -101.1 | -134.2 | -143.1 | 6.57 |
| 180 | -99.6 | 44.6 | -44.6 | -144.2 | -147.0 | 1.96 |
| 181 | 165.2 | 45.9 | -45.9 | 211.0 | 214.0 | 1.41 |
| 182 | 50.1 | 103.9 | -103.9 | 154.0 | 163.2 | 5.93 |
| 183 | -50.1 | 103.9 | -103.9 | -154.0 | -163.2 | 5.93 |

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| 184 | -165.2 | 45.9 | -45.9 | -211.0 | -214.0 | 1.41 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 185 | 165.2 | 45.9 | -45.9 | 211.0 | 214.0 | 1.4 |
| 186 | 50.1 | 103.9 | -103.9 | 154.0 | 163.2 | 5.93 |
| 187 | -50.1 | 103.9 | -103.9 | -154.0 | -163.2 | 5.93 |
| 188 | -165.2 | 45.9 | -45.9 | -211.0 | -214.0 | 1.41 |
| 189 | 99.6 | 44.6 | -44.6 | 144.2 | 147.0 | 1.96 |
| 190 | 33.2 | 101.1 | -101.1 | 134.2 | 143.1 | 6.57 |
| 191 | -33.2 | 101.1 | -101.1 | -134.2 | -143.1 | 6.57 |
| 192 | -99.6 | 44.6 | -44.6 | -144.2 | -147.0 | 1.96 |
| 193 | 102.4 | 42.1 | -42.1 | 144.5 | 146.9 | 1.64 |
| 194 | 36.1 | 96.7 | -96.7 | 132.8 | 140.7 | 5.98 |
| 195 | -36.1 | 96.7 | -96.7 | -132.8 | -140.7 | 5.98 |
| 196 | -102.4 | 42.1 | -42.1 | -144.5 | -146.9 | 1.64 |
| 197 | 168.5 | 43.4 | -43.4 | 211.9 | 214.4 | 1.19 |
| 198 | 54.1 | 99.4 | -99.4 | 153.5 | 161.7 | 5.36 |
| 199 | -54.1 | 99.4 | -99.4 | -153.5 | -161.7 | 5.36 |
| 200 | -168.5 | 43.4 | -43.4 | -211.9 | -214.4 | 1.19 |
| 201 | 168.5 | 43.4 | -43.4 | 211.9 | 214.4 | 1.19 |
| 202 | 54.1 | 99.4 | -99.4 | 153.5 | 161.7 | 5.36 |
| 203 | -54.1 | 99.4 | -99.4 | -153.5 | -161.7 | 5.36 |
| 204 | -168.5 | 43.4 | -43.4 | -211.9 | -214.4 | 1.19 |
| 205 | 102.4 | 42.1 | -42.1 | 144.5 | 146.9 | 1.64 |
| 206 | 36.1 | 96.7 | -96.7 | 132.8 | 140.7 | 5.98 |
| 207 | -36.1 | 96.7 | -96.7 | -132.8 | -140.7 | 5.98 |
| 208 | -102.4 | 42.1 | -42.1 | -144.5 | -146.9 | 1.64 |
| 209 | 105.0 | 39.9 | -39.9 | 144.9 | 146.9 | 1.40 |
| 210 | 38.6 | 93.2 | -93.2 | 131.8 | 139.0 | 5.48 |
| 211 | -38.6 | 93.2 | -93.2 | -131.8 | -139.0 | 5.48 |
| 212 | -105.0 | 39.9 | -39.9 | -144.9 | -146.9 | 1.40 |
| 213 | 172.2 | 41.1 | -41.1 | 213.3 | 215.5 | 1.01 |
| 214 | 57.7 | 95.8 | -95.8 | 153.5 | 161.0 | 4.88 |
| 215 | -57.7 | 95.8 | -95.8 | -153.5 | -161.0 | 4.88 |
| 216 | -172.2 | 41.1 | -41.1 | -213.3 | -215.5 | 1.01 |
| 217 | 172.2 | 41.1 | -41.1 | 213.3 | 215.5 | 1.01 |
| 218 | 57.7 | 95.8 | -95.8 | 153.5 | 161.0 | 4.88 |
| 219 | -57.7 | 95.8 | -95.8 | -153.5 | -161.0 | 4.88 |
| 220 | -172.2 | 41.1 | -41.1 | -213.3 | -215.5 | 1.01 |
| 221 | 105.0 | 39.9 | -39.9 | 144.9 | 146.9 | 1.40 |
| 222 | 38.6 | 93.2 | -93.2 | 131.8 | 139.0 | 5.48 |
| 223 | -38.6 | 93.2 | -93.2 | -131.8 | -139.0 | 5.48 |
| 224 | -105.0 | 39.9 | -39.9 | -144.9 | -146.9 | 1.40 |
| 225 | 107.2 | 38.4 | -38.4 | 145.6 | 147.4 | 1.24 |
| 226 | 40.8 | 91.1 | -91.1 | 131.8 | 138.5 | 5.09 |
| 227 | -40.8 | 91.1 | -91.1 | -131.8 | -138.5 | 5.09 |
| 228 | -107.2 | 38.4 | -38.4 | -145.6 | -147.4 | 1.24 |
| 230 | 176.3 | 39.4 | -39.4 | 215.7 | 217.6 | 0.89 |
| 230 | 61.2 | 93.6 | -93.6 | 154.9 | 161.8 | 4.50 |
|  |  |  |  |  |  |  |

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| 231 | -61.2 | 93.6 | -93.6 | -154.9 | -161.8 | 4.50 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 232 | -176.3 | 39.4 | -39.4 | -215.7 | -217.6 | 0.89 |
| 233 | 176.3 | 39.4 | -39.4 | 215.7 | 217.6 | 0.89 |
| 234 | 61.2 | 93.6 | -93.6 | 154.9 | 161.8 | 4.50 |
| 235 | -61.2 | 93.6 | -93.6 | -154.9 | -161.8 | 4.50 |
| 236 | -176.3 | 39.4 | -39.4 | -215.7 | -217.6 | 0.89 |
| 237 | 107.2 | 38.4 | -38.4 | 145.6 | 147.4 | 1.24 |
| 238 | 40.8 | 91.1 | -91.1 | 131.8 | 138.5 | 5.09 |
| 239 | -40.8 | 91.1 | -91.1 | -131.8 | -138.5 | 5.09 |
| 240 | -107.2 | 38.4 | -38.4 | -145.6 | -147.4 | 1.24 |
| 241 | 100.1 | 41.1 | -41.1 | 141.2 | 143.3 | 1.48 |
| 242 | 39.1 | 86.8 | -86.8 | 126.0 | 132.1 | 4.87 |
| 243 | -39.1 | 86.8 | -86.8 | -126.0 | -132.1 | 4.87 |
| 244 | -100.1 | 41.1 | -41.1 | -141.2 | -143.3 | 1.48 |
| 245 | 161.9 | 42.3 | -42.3 | 204.2 | 206.4 | 1.08 |
| 246 | 57.8 | 89.0 | -89.0 | 146.8 | 153.2 | 4.32 |
| 247 | -57.8 | 89.0 | -89.0 | -146.8 | -153.2 | 4.32 |
| 248 | -161.9 | 42.3 | -42.3 | -204.2 | -206.4 | 1.08 |
| 249 | 161.9 | 42.3 | -42.3 | 204.2 | 206.4 | 1.08 |
| 250 | 57.8 | 89.0 | -89.0 | 146.8 | 153.2 | 4.32 |
| 251 | -57.8 | 89.0 | -89.0 | -146.8 | -153.2 | 4.32 |
| 252 | -161.9 | 42.3 | -42.3 | -204.2 | -206.4 | 1.08 |
| 253 | 100.1 | 41.1 | -41.1 | 141.2 | 143.3 | 1.48 |
| 254 | 39.1 | 86.8 | -86.8 | 126.0 | 132.1 | 4.87 |
| 255 | -39.1 | 86.8 | -86.8 | -126.0 | -132.1 | 4.87 |
| 256 | -100.1 | 41.1 | -41.1 | -141.2 | -143.3 | 1.48 |
| 257 | 111.7 | 37.8 | -37.8 | 149.5 | 151.1 | 1.08 |
| 258 | 44.8 | 87.6 | -87.6 | 132.3 | 138.1 | 4.35 |
| 259 | -44.8 | 87.6 | -87.6 | -132.3 | -138.1 | 4.35 |
| 260 | -111.7 | 37.8 | -37.8 | -149.5 | -151.1 | 1.08 |
| 261 | 181.7 | 38.8 | -38.8 | 220.6 | 222.3 | 0.77 |
| 262 | 66.6 | 89.9 | -89.9 | 156.5 | 162.5 | 3.81 |
| 263 | -66.6 | 89.9 | -89.9 | -156.5 | -162.5 | 3.81 |
| 264 | -181.7 | 38.8 | -38.8 | -220.6 | -222.3 | 0.77 |
| 265 | 181.7 | 38.8 | -38.8 | 220.6 | 222.3 | 0.77 |
| 266 | 66.6 | 89.9 | -89.9 | 156.5 | 162.5 | 3.81 |
| 267 | -66.6 | 89.9 | -89.9 | -156.5 | -162.5 | 3.81 |
| 268 | -181.7 | 38.8 | -38.8 | -220.6 | -222.3 | 0.77 |
| 269 | 111.7 | 37.8 | -37.8 | 149.5 | 151.1 | 1.08 |
| 270 | 44.8 | 87.6 | -87.6 | 132.3 | 138.1 | 4.35 |
| 271 | -44.8 | 87.6 | -87.6 | -132.3 | -138.1 | 4.35 |
| 272 | -111.7 | 37.8 | -37.8 | -149.5 | -151.1 | 1.08 |
| 273 | 113.5 | 36.9 | -36.9 | 150.4 | 151.8 | 0.93 |
| 274 | 47.0 | 84.7 | -84.7 | 131.6 | 136.7 | 3.89 |
| 275 | -47.0 | 84.7 | -84.7 | -131.6 | -136.7 | 3.89 |
| 276 | -113.5 | 36.9 | -36.9 | -150.4 | -151.8 | 0.93 |
| 277 | 183.5 | 38.0 | -38.0 | 221.4 | 222.9 | 0.67 |

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| 278 | 69.5 | 86.9 | -86.9 | 156.4 | 161.6 | 3.39 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 279 | -69.5 | 86.9 | -86.9 | -156.4 | -161.6 | 3.39 |
| 280 | -183.5 | 38.0 | -38.0 | -221.4 | -222.9 | 0.67 |
| 281 | 183.5 | 38.0 | -38.0 | 221.4 | 222.9 | 0.67 |
| 282 | 69.5 | 86.9 | -86.9 | 156.4 | 161.6 | 3.39 |
| 283 | -69.5 | 86.9 | -86.9 | -156.4 | -161.6 | 3.39 |
| 284 | -183.5 | 38.0 | -38.0 | -221.4 | -222.9 | 0.67 |
| 285 | 113.5 | 36.9 | -36.9 | 150.4 | 151.8 | 0.93 |
| 286 | 47.0 | 84.7 | -84.7 | 131.6 | 136.7 | 3.89 |
| 287 | -47.0 | 84.7 | -84.7 | -131.6 | -136.7 | 3.89 |
| 288 | -113.5 | 36.9 | -36.9 | -150.4 | -151.8 | 0.93 |
| 289 | 115.7 | 34.7 | -34.7 | 150.5 | 151.6 | 0.78 |
| 290 | 49.0 | 81.2 | -81.2 | 130.2 | 134.8 | 3.49 |
| 291 | -49.0 | 81.2 | -81.2 | -130.2 | -134.8 | 3.49 |
| 292 | -115.7 | 34.7 | -34.7 | -150.5 | -151.6 | 0.78 |
| 293 | 186.5 | 35.8 | -35.8 | 222.2 | 223.5 | 0.56 |
| 294 | 72.4 | 83.3 | -83.3 | 155.7 | 160.4 | 3.02 |
| 295 | -72.4 | 83.3 | -83.3 | -155.7 | -160.4 | 3.02 |
| 296 | -186.5 | 35.8 | -35.8 | -222.2 | -223.5 | 0.56 |
| 297 | 186.5 | 35.8 | -35.8 | 222.2 | 223.5 | 0.56 |
| 298 | 72.4 | 83.3 | -83.3 | 155.7 | 160.4 | 3.02 |
| 299 | -72.4 | 83.3 | -83.3 | -155.7 | -160.4 | 3.02 |
| 300 | -186.5 | 35.8 | -35.8 | -222.2 | -223.5 | 0.56 |
| 301 | 115.7 | 34.7 | -34.7 | 150.5 | 151.6 | 0.78 |
| 302 | 49.0 | 81.2 | -81.2 | 130.2 | 134.8 | 3.49 |
| 303 | -49.0 | 81.2 | -81.2 | -130.2 | -134.8 | 3.49 |
| 304 | -115.7 | 34.7 | -34.7 | -150.5 | -151.6 | 0.78 |
| 305 | 118.2 | 31.9 | -31.9 | 150.1 | 151.0 | 0.65 |
| 306 | 51.1 | 77.8 | -77.8 | 128.9 | 133.0 | 3.17 |
| 307 | -51.1 | 77.8 | -77.8 | -128.9 | -133.0 | 3.17 |
| 308 | -118.2 | 31.9 | -31.9 | -150.1 | -151.0 | 0.65 |
| 309 | 191.1 | 32.8 | -32.8 | 223.9 | 224.9 | 0.46 |
| 310 | 75.8 | 79.9 | -79.9 | 155.7 | 160.0 | 2.72 |
| 311 | -75.8 | 79.9 | -79.9 | -155.7 | -160.0 | 2.72 |
| 312 | -191.1 | 32.8 | -32.8 | -223.9 | -224.9 | 0.46 |
| 313 | 191.1 | 32.8 | -32.8 | 223.9 | 224.9 | 0.46 |
| 314 | 75.8 | 79.9 | -79.9 | 155.7 | 160.0 | 2.72 |
| 315 | -75.8 | 79.9 | -79.9 | -155.7 | -160.0 | 2.72 |
| 316 | -191.1 | 32.8 | -32.8 | -223.9 | -224.9 | 0.46 |
| 317 | 118.2 | 31.9 | -31.9 | 150.1 | 151.0 | 0.65 |
| 318 | 51.1 | 77.8 | -77.8 | 128.9 | 133.0 | 3.17 |
| 319 | -51.1 | 77.8 | -77.8 | -128.9 | -133.0 | 3.17 |
| 320 | -118.2 | 31.9 | -31.9 | -150.1 | -151.0 | 0.65 |
| 321 | 107.0 | 32.9 | -32.9 | 140.0 | 141.1 | 0.82 |
| 322 | 46.7 | 71.2 | -71.2 | 117.9 | 121.5 | 3.04 |
| 323 | -46.7 | 71.2 | -71.2 | -117.9 | -121.5 | 3.04 |
| 324 | -107.0 | 32.9 | -32.9 | -140.0 | -141.1 | 0.82 |

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| 325 | 170.3 | 33.8 | -33.8 | 204.1 | 205.3 | 0.60 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 326 | 68.1 | 72.9 | -72.9 | 141.0 | 144.7 | 2.62 |
| 327 | -68.1 | 72.9 | -72.9 | -141.0 | -144.7 | 2.62 |
| 328 | -170.3 | 33.8 | -33.8 | -204.1 | -205.3 | 0.60 |
| 329 | 170.3 | 33.8 | -33.8 | 204.1 | 205.3 | 0.60 |
| 330 | 68.1 | 72.9 | -72.9 | 141.0 | 144.7 | 2.62 |
| 331 | -68.1 | 72.9 | -72.9 | -141.0 | -144.7 | 2.62 |
| 332 | -170.3 | 33.8 | -33.8 | -204.1 | -205.3 | 0.60 |
| 333 | 107.0 | 32.9 | -32.9 | 140.0 | 141.1 | 0.82 |
| 334 | 46.7 | 71.2 | -71.2 | 117.9 | 121.5 | 3.04 |
| 335 | -46.7 | 71.2 | -71.2 | -117.9 | -121.5 | 3.04 |
| 336 | -107.0 | 32.9 | -32.9 | -140.0 | -141.1 | 0.82 |
| 337 | 121.0 | 28.0 | -28.0 | 149.0 | 149.7 | 0.50 |
| 338 | 53.4 | 69.5 | -69.5 | 122.9 | 126.0 | 2.58 |
| 339 | -53.4 | 69.5 | -69.5 | -122.9 | -126.0 | 2.58 |
| 340 | -121.0 | 28.0 | -28.0 | -149.0 | -149.7 | 0.50 |
| 341 | 193.5 | 28.7 | -28.7 | 222.3 | 223.0 | 0.35 |
| 342 | 78.4 | 71.2 | -71.2 | 149.6 | 152.9 | 2.19 |
| 343 | -78.4 | 71.2 | -71.2 | -149.6 | -152.9 | 2.19 |
| 344 | -193.5 | 28.7 | -28.7 | -222.3 | -223.0 | 0.35 |
| 345 | 193.5 | 28.7 | -28.7 | 222.3 | 223.0 | 0.35 |
| 346 | 78.4 | 71.2 | -71.2 | 149.6 | 152.9 | 2.19 |
| 347 | -78.4 | 71.2 | -71.2 | -149.6 | -152.9 | 2.19 |
| 348 | -193.5 | 28.7 | -28.7 | -222.3 | -223.0 | 0.35 |
| 349 | 121.0 | 28.0 | -28.0 | 149.0 | 149.7 | 0.50 |
| 350 | 53.4 | 69.5 | -69.5 | 122.9 | 126.0 | 2.58 |
| 351 | -53.4 | 69.5 | -69.5 | -122.9 | -126.0 | 2.58 |
| 352 | -121.0 | 28.0 | -28.0 | -149.0 | -149.7 | 0.50 |
| 353 | 121.5 | 26.4 | -26.4 | 148.0 | 148.5 | 0.37 |
| 354 | 54.7 | 65.1 | -65.1 | 119.8 | 122.4 | 2.18 |
| 355 | -54.7 | 65.1 | -65.1 | -119.8 | -122.4 | 2.18 |
| 356 | -121.5 | 26.4 | -26.4 | -148.0 | -148.5 | 0.37 |
| 357 | 193.4 | 27.2 | -27.2 | 220.6 | 221.2 | 0.27 |
| 358 | 79.9 | 66.7 | -66.7 | 146.6 | 149.3 | 1.84 |
| 359 | -79.9 | 66.7 | -66.7 | -146.6 | -149.3 | 1.84 |
| 360 | -193.4 | 27.2 | -27.2 | -220.6 | -221.2 | 0.27 |
| 361 | 193.4 | 27.2 | -27.2 | 220.6 | 221.2 | 0.27 |
| 362 | 79.9 | 66.7 | -66.7 | 146.6 | 149.3 | 1.84 |
| 363 | -79.9 | 66.7 | -66.7 | -146.6 | -149.3 | 1.84 |
| 364 | -193.4 | 27.2 | -27.2 | -220.6 | -221.2 | 0.27 |
| 365 | 121.5 | 26.4 | -26.4 | 148.0 | 148.5 | 0.37 |
| 366 | 54.7 | 65.1 | -65.1 | 119.8 | 122.4 | 2.18 |
| 367 | -54.7 | 65.1 | -65.1 | -119.8 | -122.4 | 2.18 |
| 368 | -121.5 | 26.4 | -26.4 | -148.0 | -148.5 | 0.37 |
| 369 | 123.2 | 24.9 | -24.9 | 148.1 | 148.4 | 0.25 |
| 370 | 56.2 | 61.6 | -61.6 | 117.8 | 120.0 | 1.83 |
| 371 | -56.2 | 61.6 | -61.6 | -117.8 | -120.0 | 1.83 |

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| 372 | -123.2 | 24.9 | -24.9 | -148.1 | -148.4 | 0.25 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 373 | 195.4 | 25.6 | -25.6 | 221.0 | 221.4 | 0.18 |
| 374 | 81.9 | 63.1 | -63.1 | 145.0 | 147.3 | 1.54 |
| 375 | -81.9 | 63.1 | -63.1 | -145.0 | -147.3 | 1.54 |
| 376 | -195.4 | 25.6 | -25.6 | -221.0 | -221.4 | 0.18 |
| 377 | 195.4 | 25.6 | -25.6 | 221.0 | 221.4 | 0.18 |
| 378 | 81.9 | 63.1 | -63.1 | 145.0 | 147.3 | 1.54 |
| 379 | -81.9 | 63.1 | -63.1 | -145.0 | -147.3 | 1.54 |
| 380 | -195.4 | 25.6 | -25.6 | -221.0 | -221.4 | 0.18 |
| 381 | 123.2 | 24.9 | -24.9 | 148.1 | 148.4 | 0.25 |
| 382 | 56.2 | 61.6 | -61.6 | 117.8 | 120.0 | 1.83 |
| 383 | -56.2 | 61.6 | -61.6 | -117.8 | -120.0 | 1.83 |
| 384 | -123.2 | 24.9 | -24.9 | -148.1 | -148.4 | 0.25 |
| 385 | 127.0 | 22.7 | -22.7 | 149.6 | 149.9 | 0.16 |
| 386 | 58.7 | 59.1 | -59.1 | 117.8 | 119.6 | 1.57 |
| 387 | -58.7 | 59.1 | -59.1 | -117.8 | -119.6 | 1.57 |
| 388 | -127.0 | 22.7 | -22.7 | -149.6 | -149.9 | 0.16 |
| 389 | 202.9 | 23.3 | -23.3 | 226.1 | 226.4 | 0.12 |
| 390 | 86.3 | 60.6 | -60.6 | 146.8 | 148.8 | 1.31 |
| 391 | -86.3 | 60.6 | -60.6 | -146.8 | -148.8 | 1.31 |
| 392 | -202.9 | 23.3 | -23.3 | -226.1 | -226.4 | 0.12 |
| 393 | 202.9 | 23.3 | -23.3 | 226.1 | 226.4 | 0.12 |
| 394 | 86.3 | 60.6 | -60.6 | 146.8 | 148.8 | 1.31 |
| 395 | -86.3 | 60.6 | -60.6 | -146.8 | -148.8 | 1.31 |
| 396 | -202.9 | 23.3 | -23.3 | -226.1 | -226.4 | 0.12 |
| 397 | 127.0 | 22.7 | -22.7 | 149.6 | 149.9 | 0.16 |
| 398 | 58.7 | 59.1 | -59.1 | 117.8 | 119.6 | 1.57 |
| 399 | -58.7 | 59.1 | -59.1 | -117.8 | -119.6 | 1.57 |
| 400 | -127.0 | 22.7 | -22.7 | -149.6 | -149.9 | 0.16 |
| 401 | 101.3 | 25.5 | -25.5 | 126.8 | 127.2 | 0.37 |
| 402 | 45.7 | 51.4 | -51.4 | 97.1 | 98.6 | 1.54 |
| 403 | -45.7 | 51.4 | -51.4 | -97.1 | -98.6 | 1.54 |
| 404 | -101.3 | 25.5 | -25.5 | -126.8 | -127.2 | 0.37 |
| 405 | 157.9 | 26.0 | -26.0 | 184.0 | 184.5 | 0.28 |
| 406 | 65.6 | 52.4 | -52.4 | 118.0 | 119.6 | 1.30 |
| 407 | -65.6 | 52.4 | -52.4 | -118.0 | -119.6 | 1.30 |
| 408 | -157.9 | 26.0 | -26.0 | -184.0 | -184.5 | 0.28 |
| 409 | 157.9 | 26.0 | -26.0 | 184.0 | 184.5 | 0.28 |
| 410 | 65.6 | 52.4 | -52.4 | 118.0 | 119.6 | 1.30 |
| 411 | -65.6 | 52.4 | -52.4 | -118.0 | -119.6 | 1.30 |
| 412 | -157.9 | 26.0 | -26.0 | -184.0 | -184.5 | 0.28 |
| 413 | 101.3 | 25.5 | -25.5 | 126.8 | 127.2 | 0.37 |
| 414 | 45.7 | 51.4 | -51.4 | 97.1 | 98.6 | 1.54 |
| 415 | -45.7 | 51.4 | -51.4 | -97.1 | -98.6 | 1.54 |
| 416 | -101.3 | 25.5 | -25.5 | -126.8 | -127.2 | 0.37 |
| 417 | 125.8 | 17.4 | -17.4 | 143.1 | 143.3 | 0.08 |
| 418 | 56.3 | 47.1 | -47.1 | 103.4 | 104.5 | 1.12 |

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| 419 | -56.3 | 47.1 | -47.1 | -103.4 | -104.5 | 1.12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 420 | -125.8 | 17.4 | -17.4 | -143.1 | -143.3 | 0.08 |
| 421 | 196.8 | 17.8 | -17.8 | 214.7 | 214.8 | 0.06 |
| 422 | 81.1 | 48.1 | -48.1 | 129.2 | 130.4 | 0.92 |
| 423 | -81.1 | 48.1 | -48.1 | -129.2 | -130.4 | 0.92 |
| 424 | -196.8 | 17.8 | -17.8 | -214.7 | -214.8 | 0.06 |
| 425 | 196.8 | 17.8 | -17.8 | 214.7 | 214.8 | 0.06 |
| 426 | 81.1 | 48.1 | -48.1 | 129.2 | 130.4 | 0.92 |
| 427 | -81.1 | 48.1 | -48.1 | -129.2 | -130.4 | 0.92 |
| 428 | -196.8 | 17.8 | -17.8 | -214.7 | -214.8 | 0.06 |
| 429 | 125.8 | 17.4 | -17.4 | 143.1 | 143.3 | 0.08 |
| 430 | 56.3 | 47.1 | -47.1 | 103.4 | 104.5 | 1.12 |
| 431 | -56.3 | 47.1 | -47.1 | -103.4 | -104.5 | 1.12 |
| 432 | -125.8 | 17.4 | -17.4 | -143.1 | -143.3 | 0.08 |
| 433 | 123.5 | 11.3 | -11.3 | 134.9 | 134.8 | -0.04 |
| 434 | 56.1 | 35.8 | -35.8 | 91.9 | 92.6 | 0.77 |
| 435 | -56.1 | 35.8 | -35.8 | -91.9 | -92.6 | 0.77 |
| 436 | -123.5 | 11.3 | -11.3 | -134.9 | -134.8 | -0.04 |
| 437 | 193.0 | 11.7 | -11.7 | 204.7 | 204.6 | -0.02 |
| 438 | 80.7 | 36.6 | -36.6 | 117.3 | 118.1 | 0.62 |
| 439 | -80.7 | 36.6 | -36.6 | -117.3 | -118.1 | 0.62 |
| 440 | -193.0 | 11.7 | -11.7 | -204.7 | -204.6 | -0.02 |
| 441 | 193.0 | 11.7 | -11.7 | 204.7 | 204.6 | -0.02 |
| 442 | 80.7 | 36.6 | -36.6 | 117.3 | 118.1 | 0.62 |
| 443 | -80.7 | 36.6 | -36.6 | -117.3 | -118.1 | 0.62 |
| 444 | -193.0 | 11.7 | -11.7 | -204.7 | -204.6 | -0.02 |
| 445 | 123.5 | 11.3 | -11.3 | 134.9 | 134.8 | -0.04 |
| 446 | 56.1 | 35.8 | -35.8 | 91.9 | 92.6 | 0.77 |
| 447 | -56.1 | 35.8 | -35.8 | -91.9 | -92.6 | 0.77 |
| 448 | -123.5 | 11.3 | -11.3 | -134.9 | -134.8 | -0.04 |
| 449 | 122.4 | 3.4 | -3.4 | 125.9 | 125.7 | -0.15 |
| 450 | 56.7 | 22.6 | -22.6 | 79.4 | 79.7 | 0.47 |
| 451 | -56.7 | 22.6 | -22.6 | -79.4 | -79.7 | 0.47 |
| 452 | -122.4 | 3.4 | -3.4 | -125.9 | -125.7 | -0.15 |
| 453 | 190.1 | 3.7 | -3.7 | 193.8 | 193.6 | -0.10 |
| 454 | 81.1 | 23.2 | -23.2 | 104.4 | 104.7 | 0.37 |
| 455 | -81.1 | 23.2 | -23.2 | -104.4 | -104.7 | 0.37 |
| 456 | -190.1 | 3.7 | -3.7 | -193.8 | -193.6 | -0.10 |
| 457 | 190.1 | 3.7 | -3.7 | 193.8 | 193.6 | -0.10 |
| 458 | 81.1 | 23.2 | -23.2 | 104.4 | 104.7 | 0.37 |
| 459 | -81.1 | 23.2 | -23.2 | -104.4 | -104.7 | 0.37 |
| 460 | -190.1 | 3.7 | -3.7 | -193.8 | -193.6 | -0.10 |
| 461 | 122.4 | 3.4 | -3.4 | 125.9 | 125.7 | -0.15 |
| 462 | 56.7 | 22.6 | -22.6 | 79.4 | 79.7 | 0.47 |
| 463 | -56.7 | 22.6 | -22.6 | -79.4 | -79.7 | 0.47 |
| 464 | -122.4 | 3.4 | -3.4 | -125.9 | -125.7 | -0.15 |
| 465 | 143.8 | -4.6 | 4.6 | 148.4 | 148.7 | 0.19 |

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| 466 | 64.1 | 11.2 | -11.2 | 75.3 | 75.6 | 0.33 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 467 | -64.1 | 11.2 | -11.2 | -75.3 | -75.6 | 0.33 |
| 468 | -143.8 | -4.6 | 4.6 | -148.4 | -148.7 | 0.19 |
| 469 | 226.1 | -4.6 | 4.6 | 230.7 | 231.1 | 0.14 |
| 470 | 93.0 | 11.5 | -11.5 | 104.6 | 104.9 | 0.25 |
| 471 | -93.0 | 11.5 | -11.5 | -104.6 | -104.9 | 0.25 |
| 472 | -226.1 | -4.6 | 4.6 | -230.7 | -231.1 | 0.14 |
| 473 | 226.1 | -4.6 | 4.6 | 230.7 | 231.1 | 0.14 |
| 474 | 93.0 | 11.5 | -11.5 | 104.6 | 104.9 | 0.25 |
| 475 | -93.0 | 11.5 | -11.5 | -104.6 | -104.9 | 0.25 |
| 476 | -226.1 | -4.6 | 4.6 | -230.7 | -231.1 | 0.14 |
| 477 | 143.8 | -4.6 | 4.6 | 148.4 | 148.7 | 0.19 |
| 478 | 64.1 | 11.2 | -11.2 | 75.3 | 75.6 | 0.33 |
| 479 | -64.1 | 11.2 | -11.2 | -75.3 | -75.6 | 0.33 |
| 480 | -143.8 | -4.6 | 4.6 | -148.4 | -148.7 | 0.19 |

Table B. 6 TM5 First and Second-Order Moments

| TM5 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathbf{M}_{\text {SD }+\mathrm{D}+\mathrm{L}}$ | Mpositive <br> ео. | M $_{\text {negative }}$ еQ. | First-Order Moment | SecondOrder Moment | Percent |
| 1 | 23.0 | 129.7 | -129.7 | 152.7 | 157.4 | 3.11 |
| 2 | 0.0 | 136.9 | -136.9 | -136.9 | -141.7 | 3.53 |
| 3 | 0.0 | 136.9 | -136.9 | 136.9 | 141.7 | 3.53 |
| 4 | -23.0 | 129.7 | -129.7 | -152.7 | -157.4 | 3.11 |
| 5 | 41.5 | 131.1 | -131.1 | 172.6 | 177.2 | 2.68 |
| 6 | -0.2 | 138.9 | -138.9 | -139.1 | -143.6 | 3.21 |
| 7 | 0.2 | 138.9 | -138.9 | 139.1 | 143.6 | 3.21 |
| 8 | -41.5 | 131.1 | -131.1 | -172.6 | -177.2 | 2.68 |
| 9 | 41.5 | 131.1 | -131.1 | 172.6 | 177.2 | 2.68 |
| 10 | -0.2 | 138.9 | -138.9 | -139.1 | -143.6 | 3.21 |
| 11 | 0.2 | 138.9 | -138.9 | 139.1 | 143.6 | 3.21 |
| 12 | -41.5 | 131.1 | -131.1 | -172.6 | -177.2 | 2.68 |
| 13 | 23.0 | 129.7 | -129.7 | 152.7 | 157.4 | 3.11 |
| 14 | 0.0 | 136.9 | -136.9 | -136.9 | -141.7 | 3.53 |
| 15 | 0.0 | 136.9 | -136.9 | 136.9 | 141.7 | 3.53 |
| 16 | -23.0 | 129.7 | -129.7 | -152.7 | -157.4 | 3.11 |
| 17 | 62.2 | 61.6 | -61.6 | 123.8 | 127.0 | 2.62 |
| 18 | -0.7 | 83.4 | -83.4 | -84.0 | -88.2 | 4.95 |
| 19 | 0.7 | 83.4 | -83.4 | 84.0 | 88.2 | 4.95 |
| 20 | -62.2 | 61.6 | -61.6 | -123.8 | -127.0 | 2.62 |
| 21 | 114.7 | 64.7 | -64.7 | 179.4 | 182.7 | 1.85 |
| 22 | -0.9 | 89.4 | -89.4 | -90.3 | -94.6 | 4.73 |
| 23 | 0.9 | 89.4 | -89.4 | 90.3 | 94.6 | 4.73 |
| 24 | -114.7 | 64.7 | -64.7 | -179.4 | -182.7 | 1.85 |
| 25 | 114.7 | 64.7 | -64.7 | 179.4 | 182.7 | 1.85 |
| 26 | -0.9 | 89.4 | -89.4 | -90.3 | -94.6 | 4.73 |
| 27 | 0.9 | 89.4 | -89.4 | 90.3 | 94.6 | 4.73 |
| 28 | -114.7 | 64.7 | -64.7 | -179.4 | -182.7 | 1.85 |
| 29 | 62.2 | 61.6 | -61.6 | 123.8 | 127.0 | 2.62 |
| 30 | -0.7 | 83.4 | -83.4 | -84.0 | -88.2 | $\underline{4.95}$ |
| 31 | 0.7 | 83.4 | -83.4 | 84.0 | 88.2 | 4.95 |
| 32 | -62.2 | 61.6 | -61.6 | -123.8 | -127.0 | 2.62 |
| 33 | 54.7 | 26.7 | -26.7 | 81.5 | 82.5 | 1.22 |
| 34 | 0.4 | 50.2 | -50.2 | 50.6 | 52.6 | 4.05 |
| 35 | -0.4 | 50.2 | -50.2 | -50.6 | -52.6 | 4.05 |
| 36 | -54.7 | 26.7 | -26.7 | -81.5 | -82.5 | 1.22 |
| 37 | 100.9 | 30.2 | -30.2 | 131.2 | 132.3 | 0.87 |
| 38 | 0.9 | 56.7 | -56.7 | 57.6 | 59.9 | 4.05 |
| 39 | -0.9 | 56.7 | -56.7 | -57.6 | -59.9 | 4.05 |
| 40 | -100.9 | 30.2 | -30.2 | -131.2 | -132.3 | 0.87 |
| 41 | 100.9 | 30.2 | -30.2 | 131.2 | 132.3 | 0.87 |
| 42 | 0.9 | 56.7 | -56.7 | 57.6 | 59.9 | 4.05 |

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| 43 | -0.9 | 56.7 | -56.7 | -57.6 | -59.9 | 4.05 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -100.9 | 30.2 | -30.2 | -131.2 | -132.3 | 0.87 |
| 45 | 54.7 | 26.7 | -26.7 | 81.5 | 82.5 | 1.22 |
| 46 | 0.4 | 50.2 | -50.2 | 50.6 | 52.6 | 4.05 |
| 47 | -0.4 | 50.2 | -50.2 | -50.6 | -52.6 | 4.05 |
| 48 | -54.7 | 26.7 | -26.7 | -81.5 | -82.5 | 1.22 |
| 49 | 54.1 | 8.6 | -8.6 | 62.8 | 62.4 | -0.58 |
| 50 | 1.0 | 29.0 | -29.0 | 30.0 | 30.5 | 1.57 |
| 51 | -1.0 | 29.0 | -29.0 | -30.0 | -30.5 | 1.57 |
| 52 | -54.1 | 8.6 | -8.6 | -62.8 | -62.4 | -0.58 |
| 53 | 99.5 | 11.8 | -11.8 | 111.4 | 111.2 | -0.21 |
| 54 | 1.8 | 34.8 | -34.8 | 36.6 | 37.3 | 2.01 |
| 55 | -1.8 | 34.8 | -34.8 | -36.6 | -37.3 | 2.01 |
| 56 | -99.5 | 11.8 | -11.8 | -111.4 | -111.2 | -0.21 |
| 57 | 99.5 | 11.8 | -11.8 | 111.4 | 111.2 | -0.21 |
| 58 | 1.8 | 34.8 | -34.8 | 36.6 | 37.3 | 2.01 |
| 59 | -1.8 | 34.8 | -34.8 | -36.6 | -37.3 | 2.01 |
| 60 | -99.5 | 11.8 | -11.8 | -111.4 | -111.2 | -0.21 |
| 61 | 54.1 | 8.6 | -8.6 | 62.8 | 62.4 | -0.58 |
| 62 | 1.0 | 29.0 | -29.0 | 30.0 | 30.5 | 1.57 |
| 63 | -1.0 | 29.0 | -29.0 | -30.0 | -30.5 | 1.57 |
| 64 | -54.1 | 8.6 | -8.6 | -62.8 | -62.4 | -0.58 |
| 65 | 67.7 | -6.4 | 6.4 | 74.1 | 74.8 | 1.00 |
| 66 | 0.0 | 12.5 | -12.5 | -12.5 | -12.4 | -0.82 |
| 67 | 0.0 | 12.5 | -12.5 | 12.5 | 12.4 | -0.82 |
| 68 | -67.7 | -6.4 | 6.4 | -74.1 | -74.8 | 1.00 |
| 69 | 125.2 | -3.7 | 3.7 | 128.9 | 129.6 | 0.51 |
| 70 | 0.8 | 17.6 | -17.6 | 18.5 | 18.6 | 0.52 |
| 71 | -0.8 | 17.6 | -17.6 | -18.5 | -18.6 | 0.52 |
| 72 | -125.2 | -3.7 | 3.7 | -128.9 | -129.6 | 0.51 |
| 73 | 125.2 | -3.7 | 3.7 | 128.9 | 129.6 | 0.51 |
| 74 | 0.8 | 17.6 | -17.6 | 18.5 | 18.6 | 0.52 |
| 75 | -0.8 | 17.6 | -17.6 | -18.5 | -18.6 | 0.52 |
| 76 | -125.2 | -3.7 | 3.7 | -128.9 | -129.6 | 0.51 |
| 77 | 67.7 | -6.4 | 6.4 | 74.1 | 74.8 | 1.00 |
| 78 | 0.0 | 12.5 | -12.5 | -12.5 | -12.4 | -0.82 |
| 79 | 0.0 | 12.5 | -12.5 | 12.5 | 12.4 | -0.82 |
| 80 | -67.7 | -6.4 | 6.4 | -74.1 | -74.8 | 1.00 |

* All calculations are on the bottom of columns

Table B. 7 TM10 First and Second-Order Moments

| TM10 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathbf{M}_{\text {SD }+\mathrm{D}+\mathrm{L}}$ | $\begin{gathered} \mathbf{M}_{\text {Positive }} \\ \text { EQ. } \end{gathered}$ | M $_{\text {Negative }}$ <br> EQ. | First-Order Moment | SecondOrder <br> Moment | Percent |
| 1 | 23.6 | 191.8 | -191.8 | 215.4 | 231.0 | 7.25 |
| 2 | 0.5 | 198.5 | -198.5 | 199.0 | 214.8 | 7.95 |
| 3 | -0.5 | 198.5 | -198.5 | -199.0 | -214.8 | 7.95 |
| 4 | -23.6 | 191.8 | -191.8 | -215.4 | -231.0 | 7.25 |
| 5 | 42.8 | 193.2 | -193.2 | 236.0 | 251.3 | 6.47 |
| 6 | 0.6 | 200.5 | -200.5 | 201.1 | 216.2 | 7.53 |
| 7 | -0.6 | 200.5 | -200.5 | -201.1 | -216.2 | 7.53 |
| 8 | -42.8 | 193.2 | -193.2 | -236.0 | -251.3 | 6.47 |
| 9 | 42.8 | 193.2 | -193.2 | 236.0 | 251.3 | 6.47 |
| 10 | 0.6 | 200.5 | -200.5 | 201.1 | 216.2 | 7.53 |
| 11 | -0.6 | 200.5 | -200.5 | -201.1 | -216.2 | 7.53 |
| 12 | -42.8 | 193.2 | -193.2 | -236.0 | -251.3 | 6.47 |
| 13 | 23.6 | 191.8 | -191.8 | 215.4 | 231.0 | 7.25 |
| 14 | 0.5 | 198.5 | -198.5 | 199.0 | 214.8 | 7.95 |
| 15 | -0.5 | 198.5 | -198.5 | -199.0 | -214.8 | 7.95 |
| 16 | -23.6 | 191.8 | -191.8 | -215.4 | -231.0 | 7.25 |
| 17 | 62.5 | 108.7 | -108.7 | 171.1 | 184.5 | 7.84 |
| 18 | 0.0 | 130.0 | -130.0 | -130.0 | -145.5 | 11.88 |
| 19 | 0.0 | 130.0 | -130.0 | 130.0 | 145.5 | 11.88 |
| 20 | -62.5 | 108.7 | -108.7 | -171.1 | -184.5 | 7.84 |
| 21 | 116.9 | 111.7 | -111.7 | 228.6 | 242.2 | 5.95 |
| 22 | 0.4 | 136.1 | -136.1 | 136.5 | 152.3 | 11.52 |
| 23 | -0.4 | 136.1 | -136.1 | -136.5 | -152.3 | 11.52 |
| 24 | -116.9 | 111.7 | -111.7 | -228.6 | -242.2 | 5.95 |
| 25 | 116.9 | 111.7 | -111.7 | 228.6 | 242.2 | 5.95 |
| 26 | 0.4 | 136.1 | -136.1 | 136.5 | 152.3 | 11.52 |
| 27 | -0.4 | 136.1 | -136.1 | -136.5 | -152.3 | 11.52 |
| 28 | -116.9 | 111.7 | -111.7 | -228.6 | -242.2 | 5.95 |
| 29 | 62.5 | 108.7 | -108.7 | 171.1 | 184.5 | 7.84 |
| 30 | 0.0 | 130.0 | -130.0 | -130.0 | -145.5 | 11.88 |
| 31 | 0.0 | 130.0 | -130.0 | 130.0 | 145.5 | 11.88 |
| 32 | -62.5 | 108.7 | -108.7 | -171.1 | -184.5 | 7.84 |
| 33 | 56.0 | 62.1 | -62.1 | 118.1 | 126.8 | 7.35 |
| 34 | 1.6 | 89.1 | -89.1 | 90.7 | 102.5 | 12.91 |
| 35 | -1.6 | 89.1 | -89.1 | -90.7 | -102.5 | $\underline{12.91}$ |
| 36 | -56.0 | 62.1 | -62.1 | -118.1 | -126.8 | 7.35 |
| 37 | 103.8 | 66.2 | -66.2 | 169.9 | 179.0 | 5.32 |
| 38 | 3.1 | 96.9 | -96.9 | 100.0 | 112.4 | 12.45 |
| 39 | -3.1 | 96.9 | -96.9 | -100.0 | -112.4 | 12.45 |
| 40 | -103.8 | 66.2 | -66.2 | -169.9 | -179.0 | 5.32 |
| 41 | 103.8 | 66.2 | -66.2 | 169.9 | 179.0 | 5.32 |

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| 42 | 3.1 | 96.9 | -96.9 | 100.0 | 112.4 | 12.45 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 43 | -3.1 | 96.9 | -96.9 | -100.0 | -112.4 | 12.45 |
| 44 | -103.8 | 66.2 | -66.2 | -169.9 | -179.0 | 5.32 |
| 45 | 56.0 | 62.1 | -62.1 | 118.1 | 126.8 | 7.35 |
| 46 | 1.6 | 89.1 | -89.1 | 90.7 | 102.5 | 12.91 |
| 47 | -1.6 | 89.1 | -89.1 | -90.7 | -102.5 | $\underline{12.91}$ |
| 48 | -56.0 | 62.1 | -62.1 | -118.1 | -126.8 | 7.35 |
| 49 | 57.4 | 38.5 | -38.5 | 95.9 | 100.4 | 4.73 |
| 50 | 2.0 | 67.0 | -67.0 | 69.0 | 76.8 | 11.39 |
| 51 | -2.0 | 67.0 | -67.0 | -69.0 | -76.8 | 11.39 |
| 52 | -57.4 | 38.5 | -38.5 | -95.9 | -100.4 | 4.73 |
| 53 | 106.8 | 42.9 | -42.9 | 149.6 | 154.7 | 3.36 |
| 54 | 3.9 | 75.3 | -75.3 | 79.2 | 88.0 | 11.10 |
| 55 | -3.9 | 75.3 | -75.3 | -79.2 | -88.0 | 11.10 |
| 56 | -106.8 | 42.9 | -42.9 | -149.6 | -154.7 | 3.36 |
| 57 | 106.8 | 42.9 | -42.9 | 149.6 | 154.7 | 3.36 |
| 58 | 3.9 | 75.3 | -75.3 | 79.2 | 88.0 | 11.10 |
| 59 | -3.9 | 75.3 | -75.3 | -79.2 | -88.0 | 11.10 |
| 60 | -106.8 | 42.9 | -42.9 | -149.6 | -154.7 | 3.36 |
| 61 | 57.4 | 38.5 | -38.5 | 95.9 | 100.4 | 4.73 |
| 62 | 2.0 | 67.0 | -67.0 | 69.0 | 76.8 | 11.39 |
| 63 | -2.0 | 67.0 | -67.0 | -69.0 | -76.8 | 11.39 |
| 64 | -57.4 | 38.5 | -38.5 | -95.9 | -100.4 | 4.73 |
| 65 | 59.9 | 25.7 | -25.7 | 85.6 | 87.3 | 2.00 |
| 66 | 2.5 | 55.4 | -55.4 | 57.9 | 62.9 | 8.75 |
| 67 | -2.5 | 55.4 | -55.4 | -57.9 | -62.9 | 8.75 |
| 68 | -59.9 | 25.7 | -25.7 | -85.6 | -87.3 | 2.00 |
| 69 | 111.7 | 30.1 | -30.1 | 141.9 | 144.1 | 1.56 |
| 70 | 5.2 | 63.9 | -63.9 | 69.1 | 75.1 | 8.74 |
| 71 | -5.2 | 63.9 | -63.9 | -69.1 | -75.1 | 8.74 |
| 72 | -111.7 | 30.1 | -30.1 | -141.9 | -144.1 | 1.56 |
| 73 | 111.7 | 30.1 | -30.1 | 141.9 | 144.1 | 1.56 |
| 74 | 5.2 | 63.9 | -63.9 | 69.1 | 75.1 | 8.74 |
| 75 | -5.2 | 63.9 | -63.9 | -69.1 | -75.1 | 8.74 |
| 76 | -111.7 | 30.1 | -30.1 | -141.9 | -144.1 | 1.56 |
| 77 | 59.9 | 25.7 | -25.7 | 85.6 | 87.3 | 2.00 |
| 78 | 2.5 | 55.4 | -55.4 | 57.9 | 62.9 | 8.75 |
| 79 | -2.5 | 55.4 | -55.4 | -57.9 | -62.9 | 8.75 |
| 80 | -59.9 | 25.7 | -25.7 | -85.6 | -87.3 | 2.00 |
| 81 | 49.1 | 22.7 | -22.7 | 71.8 | 72.5 | 1.05 |
| 82 | 2.5 | 45.3 | -45.3 | 47.8 | 50.9 | 6.47 |
| 83 | -2.5 | 45.3 | -45.3 | -47.8 | -50.9 | 6.47 |
| 84 | -49.1 | 22.7 | -22.7 | -71.8 | -72.5 | 1.05 |
| 85 | 90.3 | 26.1 | -26.1 | 116.4 | 117.6 | 0.99 |
| 86 | 4.9 | 51.6 | -51.6 | 56.5 | 60.2 | 6.69 |
| 87 | -4.9 | 51.6 | -51.6 | -56.5 | -60.2 | 6.69 |
| 88 | -90.3 | 26.1 | -26.1 | -116.4 | -117.6 | 0.99 |

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| 89 | 90.3 | 26.1 | -26.1 | 116.4 | 117.6 | 0.99 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 90 | 4.9 | 51.6 | -51.6 | 56.5 | 60.2 | 6.69 |
| 91 | -4.9 | 51.6 | -51.6 | -56.5 | -60.2 | 6.69 |
| 92 | -90.3 | 26.1 | -26.1 | -116.4 | -117.6 | 0.99 |
| 93 | 49.1 | 22.7 | -22.7 | 71.8 | 72.5 | 1.05 |
| 94 | 2.5 | 45.3 | -45.3 | 47.8 | 50.9 | 6.47 |
| 95 | -2.5 | 45.3 | -45.3 | -47.8 | -50.9 | 6.47 |
| 96 | -49.1 | 22.7 | -22.7 | -71.8 | -72.5 | 1.05 |
| 97 | 60.6 | 14.4 | -14.4 | 75.0 | 74.3 | -0.83 |
| 98 | 3.1 | 40.0 | -40.0 | 43.1 | 44.7 | 3.87 |
| 99 | -3.1 | 40.0 | -40.0 | -43.1 | -44.7 | 3.87 |
| 100 | -60.6 | 14.4 | -14.4 | -75.0 | -74.3 | -0.83 |
| 101 | 111.6 | 18.0 | -18.0 | 129.7 | 129.4 | -0.21 |
| 102 | 6.3 | 47.0 | -47.0 | 53.2 | 55.6 | 4.39 |
| 103 | -6.3 | 47.0 | -47.0 | -53.2 | -55.6 | 4.39 |
| 104 | -111.6 | 18.0 | -18.0 | -129.7 | -129.4 | -0.21 |
| 105 | 111.6 | 18.0 | -18.0 | 129.7 | 129.4 | -0.21 |
| 106 | 6.3 | 47.0 | -47.0 | 53.2 | 55.6 | 4.39 |
| 107 | -6.3 | 47.0 | -47.0 | -53.2 | -55.6 | 4.39 |
| 108 | -111.6 | 18.0 | -18.0 | -129.7 | -129.4 | -0.21 |
| 109 | 60.6 | 14.4 | -14.4 | 75.0 | 74.3 | -0.83 |
| 110 | 3.1 | 40.0 | -40.0 | 43.1 | 44.7 | 3.87 |
| 111 | -3.1 | 40.0 | -40.0 | -43.1 | -44.7 | 3.87 |
| 112 | -60.6 | 14.4 | -14.4 | -75.0 | -74.3 | -0.83 |
| 113 | 59.3 | 9.3 | -9.3 | 68.6 | 67.5 | -1.55 |
| 114 | 3.5 | 30.9 | -30.9 | 34.5 | 34.9 | 1.35 |
| 115 | -3.5 | 30.9 | -30.9 | -34.5 | -34.9 | 1.35 |
| 116 | -59.3 | 9.3 | -9.3 | -68.6 | -67.5 | -1.55 |
| 117 | 109.2 | 12.4 | -12.4 | 121.7 | 120.8 | -0.67 |
| 118 | 7.1 | 36.9 | -36.9 | 43.9 | 44.9 | 2.09 |
| 119 | -7.1 | 36.9 | -36.9 | -43.9 | -44.9 | 2.09 |
| 120 | -109.2 | 12.4 | -12.4 | -121.7 | -120.8 | -0.67 |
| 121 | 109.2 | 12.4 | -12.4 | 121.7 | 120.8 | -0.67 |
| 122 | 7.1 | 36.9 | -36.9 | 43.9 | 44.9 | 2.09 |
| 123 | -7.1 | 36.9 | -36.9 | -43.9 | -44.9 | 2.09 |
| 124 | -109.2 | 12.4 | -12.4 | -121.7 | -120.8 | -0.67 |
| 125 | 59.3 | 9.3 | -9.3 | 68.6 | 67.5 | -1.55 |
| 126 | 3.5 | 30.9 | -30.9 | 34.5 | 34.9 | 1.35 |
| 127 | -3.5 | 30.9 | -30.9 | -34.5 | -34.9 | 1.35 |
| 128 | -59.3 | 9.3 | -9.3 | -68.6 | -67.5 | -1.55 |
| 129 | 57.4 | 3.2 | -3.2 | 60.6 | 59.4 | -2.03 |
| 130 | 4.2 | 21.0 | -21.0 | 25.2 | 25.0 | -0.85 |
| 131 | -4.2 | 21.0 | -21.0 | -25.2 | -25.0 | -0.85 |
| 132 | -57.4 | 3.2 | -3.2 | -60.6 | -59.4 | -2.03 |
| 133 | 105.3 | 5.8 | -5.8 | 111.2 | 110.1 | -0.98 |
| 134 | 8.0 | 25.9 | -25.9 | 33.9 | 34.0 | 0.27 |
| 135 | -8.0 | 25.9 | -25.9 | -33.9 | -34.0 | 0.27 |

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| 136 | -105.3 | 5.8 | -5.8 | -111.2 | -110.1 | -0.98 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 137 | 105.3 | 5.8 | -5.8 | 111.2 | 110.1 | -0.98 |
| 138 | 8.0 | 25.9 | -25.9 | 33.9 | 34.0 | 0.27 |
| 139 | -8.0 | 25.9 | -25.9 | -33.9 | -34.0 | 0.27 |
| 140 | -105.3 | 5.8 | -5.8 | -111.2 | -110.1 | -0.98 |
| 141 | 57.4 | 3.2 | -3.2 | 60.6 | 59.4 | -2.03 |
| 142 | 4.2 | 21.0 | -21.0 | 25.2 | 25.0 | -0.85 |
| 143 | -4.2 | 21.0 | -21.0 | -25.2 | -25.0 | -0.85 |
| 144 | -57.4 | 3.2 | -3.2 | -60.6 | -59.4 | -2.03 |
| 145 | 71.9 | -6.2 | 6.2 | 78.1 | 79.1 | 1.30 |
| 146 | 4.0 | 10.2 | -10.2 | 14.1 | 13.8 | -2.04 |
| 147 | -4.0 | 10.2 | -10.2 | -14.1 | -13.8 | -2.04 |
| 148 | -71.9 | -6.2 | 6.2 | -78.1 | -79.1 | 1.30 |
| 149 | 132.9 | -4.0 | 4.0 | 136.9 | 137.8 | 0.68 |
| 150 | 8.5 | 14.5 | -14.5 | 23.0 | 22.9 | -0.34 |
| 151 | -8.5 | 14.5 | -14.5 | -23.0 | -22.9 | -0.34 |
| 152 | -132.9 | -4.0 | 4.0 | -136.9 | -137.8 | 0.68 |
| 153 | 132.9 | -4.0 | 4.0 | 136.9 | 137.8 | 0.68 |
| 154 | 8.5 | 14.5 | -14.5 | 23.0 | 22.9 | -0.34 |
| 155 | -8.5 | 14.5 | -14.5 | -23.0 | -22.9 | -0.34 |
| 156 | -132.9 | -4.0 | 4.0 | -136.9 | -137.8 | 0.68 |
| 157 | 71.9 | -6.2 | 6.2 | 78.1 | 79.1 | 1.30 |
| 158 | 4.0 | 10.2 | -10.2 | 14.1 | 13.8 | -2.04 |
| 159 | -4.0 | 10.2 | -10.2 | -14.1 | -13.8 | -2.04 |
| 160 | -71.9 | -6.2 | 6.2 | -78.1 | -79.1 | 1.30 |

Table B. 8 TM15 First and Second-Order Moments

| TM15 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Column | $\mathbf{M S D}_{\text {d }+\mathrm{D}+\mathrm{L}}$ | $\begin{gathered} \mathbf{M}_{\text {Positive }} \\ \text { eq. } \end{gathered}$ | M $_{\text {negative }}$ е E . | First-Order Moment | SecondOrder Moment | Percent |
| 1 | 24.8 | 333.4 | -333.4 | 358.2 | 398.5 | 11.26 |
| 2 | 1.1 | 338.6 | -338.6 | 339.8 | 380.3 | 11.92 |
| 3 | -1.1 | 338.6 | -338.6 | -339.8 | -380.3 | 11.92 |
| 4 | -24.8 | 333.4 | -333.4 | -358.2 | -398.5 | 11.26 |
| 5 | 44.6 | 334.8 | -334.8 | 379.3 | 419.1 | 10.47 |
| 6 | 1.4 | 340.4 | -340.4 | 341.7 | 381.3 | 11.57 |
| 7 | -1.4 | 340.4 | -340.4 | -341.7 | -381.3 | 11.57 |
| 8 | -44.6 | 334.8 | -334.8 | -379.3 | -419.1 | 10.47 |
| 9 | 44.6 | 334.8 | -334.8 | 379.3 | 419.1 | 10.47 |
| 10 | 1.4 | 340.4 | -340.4 | 341.7 | 381.3 | 11.57 |
| 11 | -1.4 | 340.4 | -340.4 | -341.7 | -381.3 | 11.57 |
| 12 | -44.6 | 334.8 | -334.8 | -379.3 | -419.1 | 10.47 |
| 13 | 24.8 | 333.4 | -333.4 | 358.2 | 398.5 | 11.26 |
| 14 | 1.1 | 338.6 | -338.6 | 339.8 | 380.3 | 11.92 |
| 15 | -1.1 | 338.6 | -338.6 | -339.8 | -380.3 | 11.92 |
| 16 | -24.8 | 333.4 | -333.4 | -358.2 | -398.5 | 11.26 |
| 17 | 60.7 | 225.3 | -225.3 | 286.0 | 323.4 | 13.06 |
| 18 | -0.2 | 241.3 | -241.3 | -241.5 | -281.0 | 16.36 |
| 19 | 0.2 | 241.3 | -241.3 | 241.5 | 281.0 | 16.36 |
| 20 | -60.7 | 225.3 | -225.3 | -286.0 | -323.4 | 13.06 |
| 21 | 115.5 | 227.5 | -227.5 | 343.0 | 380.6 | 10.96 |
| 22 | 0.4 | 246.1 | -246.1 | 246.4 | 286.3 | 16.17 |
| 23 | -0.4 | 246.1 | -246.1 | -246.4 | -286.3 | 16.17 |
| 24 | -115.5 | 227.5 | -227.5 | -343.0 | -380.6 | 10.96 |
| 25 | 115.5 | 227.5 | -227.5 | 343.0 | 380.6 | 10.96 |
| 26 | 0.4 | 246.1 | -246.1 | 246.4 | 286.3 | 16.17 |
| 27 | -0.4 | 246.1 | -246.1 | -246.4 | -286.3 | 16.17 |
| 28 | -115.5 | 227.5 | -227.5 | -343.0 | -380.6 | 10.96 |
| 29 | 60.7 | 225.3 | -225.3 | 286.0 | 323.4 | 13.06 |
| 30 | -0.2 | 241.3 | -241.3 | -241.5 | -281.0 | 16.36 |
| 31 | 0.2 | 241.3 | -241.3 | 241.5 | 281.0 | 16.36 |
| 32 | -60.7 | 225.3 | -225.3 | -286.0 | -323.4 | 13.06 |
| 33 | 57.3 | 148.5 | -148.5 | 205.8 | 235.8 | 14.57 |
| 34 | 2.0 | 171.9 | -171.9 | 173.9 | 207.5 | 19.33 |
| 35 | -2.0 | 171.9 | -171.9 | -173.9 | -207.5 | 19.33 |
| 36 | -57.3 | 148.5 | -148.5 | -205.8 | -235.8 | 14.57 |
| 37 | 106.4 | 152.1 | -152.1 | 258.5 | 288.8 | 11.75 |
| 38 | 3.5 | 178.9 | -178.9 | 182.4 | 216.8 | 18.86 |
| 39 | -3.5 | 178.9 | -178.9 | -182.4 | -216.8 | 18.86 |
| 40 | -106.4 | 152.1 | -152.1 | -258.5 | -288.8 | 11.75 |
| 41 | 106.4 | 152.1 | -152.1 | 258.5 | 288.8 | 11.75 |
| 42 | 3.5 | 178.9 | -178.9 | 182.4 | 216.8 | 18.86 |

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| 43 | -3.5 | 178.9 | -178.9 | -182.4 | -216.8 | 18.86 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 44 | -106.4 | 152.1 | -152.1 | -258.5 | -288.8 | 11.75 |
| 45 | 57.3 | 148.5 | -148.5 | 205.8 | 235.8 | 14.57 |
| 46 | 2.0 | 171.9 | -171.9 | 173.9 | 207.5 | 19.33 |
| 47 | -2.0 | 171.9 | -171.9 | -173.9 | -207.5 | 19.33 |
| 48 | -57.3 | 148.5 | -148.5 | -205.8 | -235.8 | 14.57 |
| 49 | 57.8 | 97.1 | -97.1 | 154.9 | 176.5 | 13.94 |
| 50 | 2.3 | 124.3 | -124.3 | 126.5 | 152.7 | $\underline{20.72}$ |
| 51 | -2.3 | 124.3 | -124.3 | -126.5 | -152.7 | $\underline{20.72}$ |
| 52 | -57.8 | 97.1 | -97.1 | -154.9 | -176.5 | 13.94 |
| 53 | 108.0 | 101.3 | -101.3 | 209.3 | 231.5 | 10.62 |
| 54 | 4.3 | 132.3 | -132.3 | 136.7 | 164.1 | 20.07 |
| 55 | -4.3 | 132.3 | -132.3 | -136.7 | -164.1 | 20.07 |
| 56 | -108.0 | 101.3 | -101.3 | -209.3 | -231.5 | 10.62 |
| 57 | 108.0 | 101.3 | -101.3 | 209.3 | 231.5 | 10.62 |
| 58 | 4.3 | 132.3 | -132.3 | 136.7 | 164.1 | 20.07 |
| 59 | -4.3 | 132.3 | -132.3 | -136.7 | -164.1 | 20.07 |
| 60 | -108.0 | 101.3 | -101.3 | -209.3 | -231.5 | 10.62 |
| 61 | 57.8 | 97.1 | -97.1 | 154.9 | 176.5 | 13.94 |
| 62 | 2.3 | 124.3 | -124.3 | 126.5 | 152.7 | $\underline{20.72}$ |
| 63 | -2.3 | 124.3 | -124.3 | -126.5 | -152.7 | $\underline{\underline{20.72}}$ |
| 64 | -57.8 | 97.1 | -97.1 | -154.9 | -176.5 | 13.94 |
| 65 | 59.6 | 64.1 | -64.1 | 123.8 | 137.9 | 11.40 |
| 66 | 2.3 | 94.5 | -94.5 | 96.8 | 116.3 | 20.15 |
| 67 | -2.3 | 94.5 | -94.5 | -96.8 | -116.3 | 20.15 |
| 68 | -59.6 | 64.1 | -64.1 | -123.8 | -137.9 | 11.40 |
| 69 | 112.8 | 68.4 | -68.4 | 181.2 | 196.1 | 8.18 |
| 70 | 5.0 | 103.6 | -103.6 | 108.6 | 129.6 | 19.38 |
| 71 | -5.0 | 103.6 | -103.6 | -108.6 | -129.6 | 19.38 |
| 72 | -112.8 | 68.4 | -68.4 | -181.2 | -196.1 | 8.18 |
| 73 | 112.8 | 68.4 | -68.4 | 181.2 | 196.1 | 8.18 |
| 74 | 5.0 | 103.6 | -103.6 | 108.6 | 129.6 | 19.38 |
| 75 | -5.0 | 103.6 | -103.6 | -108.6 | -129.6 | 19.38 |
| 76 | -112.8 | 68.4 | -68.4 | -181.2 | -196.1 | 8.18 |
| 77 | 59.6 | 64.1 | -64.1 | 123.8 | 137.9 | 11.40 |
| 78 | 2.3 | 94.5 | -94.5 | 96.8 | 116.3 | 20.15 |
| 79 | -2.3 | 94.5 | -94.5 | -96.8 | -116.3 | 20.15 |
| 80 | -59.6 | 64.1 | -64.1 | -123.8 | -137.9 | 11.40 |
| 81 | 45.2 | 49.5 | -49.5 | 94.8 | 104.2 | 9.98 |
| 82 | 3.0 | 72.4 | -72.4 | 75.4 | 88.9 | 17.95 |
| 83 | -3.0 | 72.4 | -72.4 | -75.4 | -88.9 | 17.95 |
| 84 | -45.2 | 49.5 | -49.5 | -94.8 | -104.2 | 9.98 |
| 85 | 83.0 | 53.4 | -53.4 | 136.4 | 146.5 | 7.44 |
| 86 | 5.3 | 79.1 | -79.1 | 84.4 | 99.1 | 17.40 |
| 87 | -5.3 | 79.1 | -79.1 | -84.4 | -99.1 | 17.40 |
| 88 | -83.0 | 53.4 | -53.4 | -136.4 | -146.5 | 7.44 |
| 89 | 83.0 | 53.4 | -53.4 | 136.4 | 146.5 | 7.44 |

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| 90 | 5.3 | 79.1 | -79.1 | 84.4 | 99.1 | 17.40 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 91 | -5.3 | 79.1 | -79.1 | -84.4 | -99.1 | 17.40 |
| 92 | -83.0 | 53.4 | -53.4 | -136.4 | -146.5 | 7.44 |
| 93 | 45.2 | 49.5 | -49.5 | 94.8 | 104.2 | 9.98 |
| 94 | 3.0 | 72.4 | -72.4 | 75.4 | 88.9 | 17.95 |
| 95 | -3.0 | 72.4 | -72.4 | -75.4 | -88.9 | 17.95 |
| 96 | -45.2 | 49.5 | -49.5 | -94.8 | -104.2 | 9.98 |
| 97 | 62.0 | 32.1 | -32.1 | 94.1 | 98.3 | 4.50 |
| 98 | 3.6 | 64.4 | -64.4 | 68.1 | 78.0 | 14.61 |
| 99 | -3.6 | 64.4 | -64.4 | -68.1 | -78.0 | 14.61 |
| 100 | -62.0 | 32.1 | -32.1 | -94.1 | -98.3 | 4.50 |
| 101 | 115.6 | 36.7 | -36.7 | 152.3 | 157.3 | 3.32 |
| 102 | 7.4 | 73.7 | -73.7 | 81.0 | 92.6 | 14.29 |
| 103 | -7.4 | 73.7 | -73.7 | -81.0 | -92.6 | 14.29 |
| 104 | -115.6 | 36.7 | -36.7 | -152.3 | -157.3 | 3.32 |
| 105 | 115.6 | 36.7 | -36.7 | 152.3 | 157.3 | 3.32 |
| 106 | 7.4 | 73.7 | -73.7 | 81.0 | 92.6 | 14.29 |
| 107 | -7.4 | 73.7 | -73.7 | -81.0 | -92.6 | 14.29 |
| 108 | -115.6 | 36.7 | -36.7 | -152.3 | -157.3 | 3.32 |
| 109 | 62.0 | 32.1 | -32.1 | 94.1 | 98.3 | 4.50 |
| 110 | 3.6 | 64.4 | -64.4 | 68.1 | 78.0 | 14.61 |
| 111 | -3.6 | 64.4 | -64.4 | -68.1 | -78.0 | 14.61 |
| 112 | -62.0 | 32.1 | -32.1 | -94.1 | -98.3 | 4.50 |
| 113 | 60.1 | 24.5 | -24.5 | 84.6 | 86.1 | 1.81 |
| 114 | 4.7 | 54.5 | -54.5 | 59.2 | 65.7 | 11.01 |
| 115 | -4.7 | 54.5 | -54.5 | -59.2 | -65.7 | 11.01 |
| 116 | -60.1 | 24.5 | -24.5 | -84.6 | -86.1 | 1.81 |
| 117 | 111.2 | 29.0 | -29.0 | 140.2 | 142.5 | 1.64 |
| 118 | 9.1 | 63.0 | -63.0 | 72.1 | 80.1 | 11.06 |
| 119 | -9.1 | 63.0 | -63.0 | -72.1 | -80.1 | 11.06 |
| 120 | -111.2 | 29.0 | -29.0 | -140.2 | -142.5 | 1.64 |
| 121 | 111.2 | 29.0 | -29.0 | 140.2 | 142.5 | 1.64 |
| 122 | 9.1 | 63.0 | -63.0 | 72.1 | 80.1 | 11.06 |
| 123 | -9.1 | 63.0 | -63.0 | -72.1 | -80.1 | 11.06 |
| 124 | -111.2 | 29.0 | -29.0 | -140.2 | -142.5 | 1.64 |
| 125 | 60.1 | 24.5 | -24.5 | 84.6 | 86.1 | 1.81 |
| 126 | 4.7 | 54.5 | -54.5 | 59.2 | 65.7 | 11.01 |
| 127 | -4.7 | 54.5 | -54.5 | -59.2 | -65.7 | 11.01 |
| 128 | -60.1 | 24.5 | -24.5 | -84.6 | -86.1 | 1.81 |
| 129 | 60.8 | 19.1 | -19.1 | 80.0 | 79.7 | -0.30 |
| 130 | 5.3 | 48.0 | -48.0 | 53.2 | 57.4 | 7.88 |
| 131 | -5.3 | 48.0 | -48.0 | -53.2 | -57.4 | 7.88 |
| 132 | -60.8 | 19.1 | -19.1 | -80.0 | -79.7 | -0.30 |
| 133 | 112.7 | 23.4 | -23.4 | 136.0 | 136.5 | 0.33 |
| 134 | 10.2 | 56.2 | -56.2 | 66.3 | 71.9 | 8.30 |
| 135 | -10.2 | 56.2 | -56.2 | -66.3 | -71.9 | 8.30 |
| 136 | -112.7 | 23.4 | -23.4 | -136.0 | -136.5 | 0.33 |

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| 137 | 112.7 | 23.4 | -23.4 | 136.0 | 136.5 | 0.33 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 10.2 | 56.2 | -56.2 | 66.3 | 71.9 | 8.30 |
| 139 | -10.2 | 56.2 | -56.2 | -66.3 | -71.9 | 8.30 |
| 140 | -112.7 | 23.4 | -23.4 | -136.0 | -136.5 | 0.33 |
| 141 | 60.8 | 19.1 | -19.1 | 80.0 | 79.7 | -0.30 |
| 142 | 5.3 | 48.0 | -48.0 | 53.2 | 57.4 | 7.88 |
| 143 | -5.3 | 48.0 | -48.0 | -53.2 | -57.4 | 7.88 |
| 144 | -60.8 | 19.1 | -19.1 | -80.0 | -79.7 | -0.30 |
| 145 | 63.4 | 15.8 | -15.8 | 79.2 | 78.1 | -1.39 |
| 146 | 5.8 | 44.3 | -44.3 | 50.1 | 53.0 | 5.68 |
| 147 | -5.8 | 44.3 | -44.3 | -50.1 | -53.0 | 5.68 |
| 148 | -63.4 | 15.8 | -15.8 | -79.2 | -78.1 | -1.39 |
| 149 | 118.0 | 19.8 | -19.8 | 137.9 | 137.4 | -0.37 |
| 150 | 11.5 | 52.4 | -52.4 | 63.9 | 67.9 | 6.28 |
| 151 | -11.5 | 52.4 | -52.4 | -63.9 | -67.9 | 6.28 |
| 152 | -118.0 | 19.8 | -19.8 | -137.9 | -137.4 | -0.37 |
| 153 | 118.0 | 19.8 | -19.8 | 137.9 | 137.4 | -0.37 |
| 154 | 11.5 | 52.4 | -52.4 | 63.9 | 67.9 | 6.28 |
| 155 | -11.5 | 52.4 | -52.4 | -63.9 | -67.9 | 6.28 |
| 156 | -118.0 | 19.8 | -19.8 | -137.9 | -137.4 | -0.37 |
| 157 | 63.4 | 15.8 | -15.8 | 79.2 | 78.1 | -1.39 |
| 158 | 5.8 | 44.3 | -44.3 | 50.1 | 53.0 | 5.68 |
| 159 | -5.8 | 44.3 | -44.3 | -50.1 | -53.0 | 5.68 |
| 160 | -63.4 | 15.8 | -15.8 | -79.2 | -78.1 | -1.39 |
| 161 | 52.0 | 17.4 | -17.4 | 69.4 | 68.8 | -0.86 |
| 162 | 5.2 | 38.7 | -38.7 | 43.9 | 45.9 | 4.50 |
| 163 | -5.2 | 38.7 | -38.7 | -43.9 | -45.9 | 4.50 |
| 164 | -52.0 | 17.4 | -17.4 | -69.4 | -68.8 | -0.86 |
| 165 | 95.2 | 20.6 | -20.6 | 115.8 | 115.7 | -0.15 |
| 166 | 9.8 | 44.5 | -44.5 | 54.4 | 57.1 | 5.01 |
| 167 | -9.8 | 44.5 | -44.5 | -54.4 | -57.1 | 5.01 |
| 168 | -95.2 | 20.6 | -20.6 | -115.8 | -115.7 | -0.15 |
| 169 | 95.2 | 20.6 | -20.6 | 115.8 | 115.7 | -0.15 |
| 170 | 9.8 | 44.5 | -44.5 | 54.4 | 57.1 | 5.01 |
| 171 | -9.8 | 44.5 | -44.5 | -54.4 | -57.1 | 5.01 |
| 172 | -95.2 | 20.6 | -20.6 | -115.8 | -115.7 | -0.15 |
| 173 | 52.0 | 17.4 | -17.4 | 69.4 | 68.8 | -0.86 |
| 174 | 5.2 | 38.7 | -38.7 | 43.9 | 45.9 | 4.50 |
| 175 | -5.2 | 38.7 | -38.7 | -43.9 | -45.9 | 4.50 |
| 176 | -52.0 | 17.4 | -17.4 | -69.4 | -68.8 | -0.86 |
| 177 | 64.1 | 11.6 | -11.6 | 75.7 | 74.5 | -1.62 |
| 178 | 6.3 | 35.3 | -35.3 | 41.6 | 42.8 | 2.76 |
| 179 | -6.3 | 35.3 | -35.3 | -41.6 | -42.8 | 2.76 |
| 180 | -64.1 | 11.6 | -11.6 | -75.7 | -74.5 | -1.62 |
| 181 | 117.6 | 14.8 | -14.8 | 132.4 | 131.6 | -0.67 |
| 182 | 12.3 | 41.6 | -41.6 | 53.9 | 55.7 | 3.38 |
| 183 | -12.3 | 41.6 | -41.6 | -53.9 | -55.7 | 3.38 |

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| 184 | -117.6 | 14.8 | -14.8 | -132.4 | -131.6 | -0.67 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 185 | 117.6 | 14.8 | -14.8 | 132.4 | 131.6 | -0.67 |
| 186 | 12.3 | 41.6 | -41.6 | 53.9 | 55.7 | 3.38 |
| 187 | -12.3 | 41.6 | -41.6 | -53.9 | -55.7 | 3.38 |
| 188 | -117.6 | 14.8 | -14.8 | -132.4 | -131.6 | -0.67 |
| 189 | 64.1 | 11.6 | -11.6 | 75.7 | 74.5 | -1.62 |
| 190 | 6.3 | 35.3 | -35.3 | 41.6 | 42.8 | 2.76 |
| 191 | -6.3 | 35.3 | -35.3 | -41.6 | -42.8 | 2.76 |
| 192 | -64.1 | 11.6 | -11.6 | -75.7 | -74.5 | -1.62 |
| 193 | 62.7 | 8.9 | -8.9 | 71.5 | 70.3 | -1.77 |
| 194 | 6.7 | 28.9 | -28.9 | 35.6 | 35.9 | 0.78 |
| 195 | -6.7 | 28.9 | -28.9 | -35.6 | -35.9 | 0.78 |
| 196 | -62.7 | 8.9 | -8.9 | -71.5 | -70.3 | -1.77 |
| 197 | 115.0 | 11.6 | -11.6 | 126.6 | 125.6 | -0.82 |
| 198 | 12.9 | 34.3 | -34.3 | 47.2 | 47.9 | 1.51 |
| 199 | -12.9 | 34.3 | -34.3 | -47.2 | -47.9 | 1.51 |
| 200 | -115.0 | 11.6 | -11.6 | -126.6 | -125.6 | -0.82 |
| 201 | 115.0 | 11.6 | -11.6 | 126.6 | 125.6 | -0.82 |
| 202 | 12.9 | 34.3 | -34.3 | 47.2 | 47.9 | 1.51 |
| 203 | -12.9 | 34.3 | -34.3 | -47.2 | -47.9 | 1.51 |
| 204 | -115.0 | 11.6 | -11.6 | -126.6 | -125.6 | -0.82 |
| 205 | 62.7 | 8.9 | -8.9 | 71.5 | 70.3 | -1.77 |
| 206 | 6.7 | 28.9 | -28.9 | 35.6 | 35.9 | 0.78 |
| 207 | -6.7 | 28.9 | -28.9 | -35.6 | -35.9 | 0.78 |
| 208 | -62.7 | 8.9 | -8.9 | -71.5 | -70.3 | -1.77 |
| 209 | 60.7 | 3.1 | -3.1 | 63.8 | 62.5 | -1.99 |
| 210 | 7.2 | 19.8 | -19.8 | 27.0 | 26.8 | -0.91 |
| 211 | -7.2 | 19.8 | -19.8 | -27.0 | -26.8 | -0.91 |
| 212 | -60.7 | 3.1 | -3.1 | -63.8 | -62.5 | -1.99 |
| 213 | 110.9 | 5.5 | -5.5 | 116.4 | 115.2 | -0.99 |
| 214 | 13.6 | 24.3 | -24.3 | 37.8 | 37.9 | 0.11 |
| 215 | -13.6 | 24.3 | -24.3 | -37.8 | -37.9 | 0.11 |
| 216 | -110.9 | 5.5 | -5.5 | -116.4 | -115.2 | -0.99 |
| 217 | 110.9 | 5.5 | -5.5 | 116.4 | 115.2 | -0.99 |
| 218 | 13.6 | 24.3 | -24.3 | 37.8 | 37.9 | 0.11 |
| 219 | -13.6 | 24.3 | -24.3 | -37.8 | -37.9 | 0.11 |
| 220 | -110.9 | 5.5 | -5.5 | -116.4 | -115.2 | -0.99 |
| 221 | 60.7 | 3.1 | -3.1 | 63.8 | 62.5 | -1.99 |
| 222 | 7.2 | 19.8 | -19.8 | 27.0 | 26.8 | -0.91 |
| 223 | -7.2 | 19.8 | -19.8 | -27.0 | -26.8 | -0.91 |
| 224 | -60.7 | 3.1 | -3.1 | -63.8 | -62.5 | -1.99 |
| 225 | 75.9 | -6.8 | 6.8 | 82.7 | 83.7 | 1.23 |
| 226 | 7.7 | 8.4 | -8.4 | 16.1 | 15.9 | -1.61 |
| 227 | -7.7 | 8.4 | -8.4 | -16.1 | -15.9 | -1.61 |
| 228 | -75.9 | -6.8 | 6.8 | -82.7 | -83.7 | 1.23 |
| 229 | 139.8 | -4.9 | 4.9 | 144.7 | 145.6 | 0.66 |
| 230 | 15.4 | 12.3 | -12.3 | 27.7 | 27.6 | -0.23 |

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| 231 | -15.4 | 12.3 | -12.3 | -27.7 | -27.6 | -0.23 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 232 | -139.8 | -4.9 | 4.9 | -144.7 | -145.6 | 0.66 |
| 233 | 139.8 | -4.9 | 4.9 | 144.7 | 145.6 | 0.66 |
| 234 | 15.4 | 12.3 | -12.3 | 27.7 | 27.6 | -0.23 |
| 235 | -15.4 | 12.3 | -12.3 | -27.7 | -27.6 | -0.23 |
| 236 | -139.8 | -4.9 | 4.9 | -144.7 | -145.6 | 0.66 |
| 237 | 75.9 | -6.8 | 6.8 | 82.7 | 83.7 | 1.23 |
| 238 | 7.7 | 8.4 | -8.4 | 16.1 | 15.9 | -1.61 |
| 239 | -7.7 | 8.4 | -8.4 | -16.1 | -15.9 | -1.61 |
| 240 | -75.9 | -6.8 | 6.8 | -82.7 | -83.7 | 1.23 |

## Appendix C

## Verification of Second-Order Analysis

Table C. 1 M10-First-Order Analysis Moment on Column Number Four

|  | Moment on Base (kN.m) | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -5.85 | 12.27 |
| Super Imposed Load | -8.77 | 18.41 |
| Dead Load | -13.84 | 29.14 |
| Earthquake Load | -227.2 | -46.3 |

Table C. 2 M10-Verification on Second-Order Analysis on Column Number Four

| Parameter | Value |
| :---: | :---: |
| $\sum \mathrm{P}$ | 60633 kN |
| $\mathrm{I}_{\text {Column }}$ | $0.04374 \mathrm{~m}^{4}$ |
| $\mathrm{I}_{\text {Beam }}$ | $8.58 \times 10^{-3} \mathrm{~m}^{4}$ |
| $\mathrm{E}_{\text {Column }}$ | $26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{E}_{\text {Beam }}$ | $23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{G}_{\mathrm{A}}$ (middle) | 10.4 |
| $\mathrm{~K}^{(\text {unbraced })}$ | 1.68 |
| $\mathrm{P}_{\mathrm{ek}}$ (middle) | 351952 kN |
| $\mathrm{G}_{\mathrm{A}}$ (edge) | 20.8 |
| K (unbraced) | 1.8 |
| $\mathrm{P}_{\text {ek }}$ (edge) | 306590 kN |
| $\sum_{\mathrm{ek}}$ | 5268336 kN |
| $\mathrm{B}_{2}$ | 1.012 |
| $C_{m}$ | 0.41 |
| P | 2330 kN |
| K (braced) | 0.70 |
| $\mathrm{P}_{\mathrm{e}}$ | 2027245 kN |
| $\mathrm{B}_{1}$ | $0.41\left(\mathrm{Take} \mathrm{B}_{1}=1\right)$ |
| $\mathrm{M}_{\mathrm{u}}$ | $-260.2 \mathrm{kN} . \mathrm{m}$ |
| $\mathrm{M}_{\mathrm{u}}$ (SAP2000) | $-260.3 \mathrm{kN} . \mathrm{m}$ |
| $\%$ error | $0.04 \%$ |
| Acceptance Criteria | $10 \%$ |
| Comment | OK |

Table C. 3 M15-First-Order Analysis Moment on Column Number Four

|  | Moment on Base (kN.m) | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -6.1 | 12.7 |
| Super Imposed Load | -9.1 | 19.1 |
| Dead Load | -14.4 | 30.3 |
| Earthquake Load | -289 | -96.8 |

Table C. 4 M15-Verification on Second-Order Analysis on Column Number Four

| Parameter | Value |
| :---: | :---: |
| $\sum \mathrm{P}$ | 93083 kN |
| $\mathrm{I}_{\text {Column }}$ | $0.066 \mathrm{~m}^{4}$ |
| $\mathrm{I}_{\text {Beam }}$ | $8.58 \times 10^{-3} \mathrm{~m}^{4}$ |
| $\mathrm{E}_{\text {Column }}$ | $26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $E_{\text {Beam }}$ | $23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{G}_{\mathrm{A}}$ (middle) | 15.7 |
| K(unbraced) | 1.75 |
| $\mathrm{P}_{\text {ek }}$ (middle) | 489431 kN |
| $\mathrm{G}_{\mathrm{A}}$ (edge) | 31.4 |
| K (unbraced) | 1.88 |
| $\mathrm{P}_{\text {ek }}$ (edge) | 424084 kN |
| $\sum \mathrm{P}_{\text {ek }}$ | 7308120 kN |
| $\mathrm{B}_{2}$ | 1.013 |
| $C_{m}$ | 0.41 |
| P | 3803 kN |
| K (braced) | 0.7 |
| $\mathrm{P}_{\mathrm{e}}$ | 3058943 kN |
| $\mathrm{B}_{1}$ | 0.41 (Take $\mathrm{B}_{1}=1$ ) |
| $\mathrm{M}_{\mathrm{u}}$ | -322.4kN.m |
| $\mathrm{M}_{\mathrm{u}}$ (SAP2000) | -327.4kN.m |
| \%error | 1.5 \% |
| Acceptance Criteria | 10\% |
| Comment | OK |

Table C. 5 M20-First-Order Analysis Moment on Column Number Four

|  | Moment on Base (kN.m) | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -6.3 | 13.1 |
| Super Imposed Load | -9.4 | 19.7 |
| Dead Load | -14.8 | 31.2 |
| Earthquake Load | -352.9 | -154.5 |

Table B. 6 M20-Verification on Second-Order Analysis on Column Number Four

| Parameter | Value |
| :---: | :---: |
| $\sum \mathrm{P}$ | 126961 kN |
| $\mathrm{I}_{\text {Column }}$ | $0.097 \mathrm{~m}^{4}$ |
| $\mathrm{I}_{\text {Beam }}$ | $8.58 \times 10^{-3} \mathrm{~m}^{4}$ |
| $\mathrm{E}_{\text {Column }}$ | $26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{E}_{\text {Beam }}$ | $23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{G}_{\mathrm{A}}$ (middle) | 23 |
| $\mathrm{~K}($ unbraced $)$ | 1.82 |
| $\mathrm{P}_{\mathrm{ek}}$ (middle) | 665047 kN |
| $\mathrm{G}_{\mathrm{A}}$ (edge) | 46 |
| K (unbraced) | 1.91 |
| $\mathrm{P}_{\mathrm{ek}}$ (edge) | 603849 kN |
| $\sum \mathrm{P}_{\text {ek }}$ | 10151168 kN |
| $\mathrm{B}_{2}$ | 1.013 |
| $C_{m}$ | 0.41 |
| P | 5461 kN |
| K (braced) | 0.7 |
| $\mathrm{P}_{\mathrm{e}}$ | 4495719 kN |
| $\mathrm{B}_{1}$ | 0.41 (Take $\left.\mathrm{B}_{1}=1\right)$ |
| $\mathrm{M}_{\mathrm{u}}$ | $-389 \mathrm{kN} . \mathrm{m}$ |
| $\mathrm{M}_{\mathrm{u}}($ SAP2000 $)$ | $-399 \mathrm{kN.m}$ |
| $\%$ orror | $2.5 \%$ |
| Acceptance Criteria | $10 \%$ |
| Comment | OK |

Table C. 7 M25-First-Order Analysis Moment on Column Number Four

|  | Moment on Base (kN.m) | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -6.4 | 13.4 |
| Super Imposed Load | -9.6 | 20.1 |
| Dead Load | -15.2 | 31.9 |
| Earthquake Load | -422.8 | -219.7 |

TableC. 8 M25-Verification on Second-Order Analysis on Column Number Four

| Parameter | Value |
| :---: | :---: |
| $\sum \mathrm{P}$ | 162403 kN |
| $\mathrm{I}_{\text {Column }}$ | $0.138 \mathrm{~m}^{4}$ |
| $\mathrm{I}_{\text {Beam }}$ | $8.58 \times 10^{-3} \mathrm{~m}^{4}$ |
| Column | $26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{E}_{\text {Beam }}$ | $23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{G}_{\mathrm{A}}$ (middle) | 32.8 |
| $\mathrm{~K}^{(\text {unbraced })}$ | 1.88 |
| $\mathrm{P}_{\mathrm{ek}}$ (middle) | 886720 kN |
| $\mathrm{G}_{\mathrm{A}}$ (edge) | 65.6 |
| K (unbraced) | 1.92 |
| $\mathrm{P}_{\mathrm{ek}}$ (edge) | 850159 kN |
| $\sum \mathrm{P}_{\text {ek }}$ | 13895032 kN |
| $\mathrm{B}_{2}$ | 1.012 |
| $C_{m}$ | 0.41 |
| P | 7284 kN |
| K (braced) | 0.7 |
| $\mathrm{P}_{\mathrm{e}}$ | 6395972 kN |
| $\mathrm{B}_{1}$ | $0.41\left(\mathrm{Take} \mathrm{B}{ }_{\mathrm{l}}=1\right)$ |
| $\mathrm{M}_{\mathrm{u}}$ | -459 kNN |
| $\mathrm{M}_{\mathrm{u}}$ (SAP2000) | $-478.3 \mathrm{kN} . \mathrm{m}$ |
| $\%$ \%error | $4 \%$ |
| Acceptance Criteria | $10 \%$ |
| Comment | OK |

Table C. 9 M30-First-Order Analysis Moment on Column Number Four

|  | Moment on Base (kN.m) | Moment on Top (kN.m) |
| :---: | :---: | :---: |
| Live Load | -6.57 | 13.7 |
| Super Imposed Load | -9.86 | 20.6 |
| Dead Load | -15.5 | 32.7 |
| Earthquake Load | -474 | -277 |

Table C. 10 M30-Verification on Second-Order Analysis on Column Number Four

| Parameter | Value |
| :---: | :---: |
| $\sum \mathrm{P}$ | 199545kN |
| $\mathrm{I}_{\text {Column }}$ | $0.19 \mathrm{~m}^{4}$ |
| $\mathrm{I}_{\text {Beam }}$ | $8.58 \times 10^{-3} \mathrm{~m}^{4}$ |
| EColumn | $26.6 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{E}_{\text {Beam }}$ | $23 \times 10^{6} \mathrm{kN} / \mathrm{m}^{2}$ |
| $\mathrm{G}_{\mathrm{A}}$ (middle) | 45.2 |
| K(unbraced) | 1.9 |
| $\mathrm{P}_{\text {ek }}$ (middle) | 1195280 kN |
| $\mathrm{G}_{\mathrm{A}}$ (edge) | 90.4 |
| K (unbraced) | 1.95 |
| $\mathrm{P}_{\text {ek }}$ (edge) | 1134770 kN |
| $\sum \mathrm{P}_{\mathrm{ek}}$ | 18640401 kN |
| $\mathrm{B}_{2}$ | 1.011 |
| $C_{m}$ | 0.41 |
| P | 9260 kN |
| K (braced) | 0.7 |
| $\mathrm{P}_{\mathrm{e}}$ | 8806048 kN |
| $\mathrm{B}_{1}$ | 0.41 (Take $\mathrm{B}_{1}=1$ ) |
| $\mathrm{M}_{\mathrm{u}}$ | -511kN.m |
| $\mathrm{M}_{\mathrm{u}}$ (SAP2000) | -539kN.m |
| \%error | 5.2 \% |
| Acceptance Criteria | 10\% |
| Comment | OK |

جامعة النجاح الوطنية كلية الدراسات العليا

## اثار P - دلتا في الخرسانة المسلحة لاطارات البوابة الزلزلالية

إعداد<br>صهيب محمد (بو قبيطة<br>إشراف<br>د. عبد الرزق طوقان

قـمت هذه الأطروحة استكمالا لمتطلبات الحصول على درجة الماجستير في هنسة الإنشاءات بكلية الدراسات العليا في جامعة النجاح الوطنية في نابلس، فلسطين. 2019

# اثار P - دلتا في الخرسانة المسلحة لاطارات البوابة الزلزللية 

إعداد

## صهيب محمد ابو قبيطة

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## الملخص

تأثير P-Delta هو تأثير غير خطي من الارجة الثانية والذي يتواجد في أية منشأ عندما يتعرض الى أحمال محورية و عزوم ويكون تأثيره أكبر في المباني المرتفعة. المباني متعددة الطوابق تتنشر بشكل كبير لإيواء العدد الكبير من السكان اللذين يحتاجون الى سكن خاصة في العواصم والمدن الكبيرة. لذلك، لدراسة تأثير عدد الطوابق على تحليل P-Delta ، تأثير P-Delta تم دراسته على ستة نماذج متعددة الطوابق من البوابات الخرسانية المسلحة والتي تتدرج من خمسة طوابق الى ثلاثين طابق والتي تخضع لحد الفترة الأساسية المنصوص عليها في ASCE 7 مقابل دراسة نفس الستة النماذج السابقة غير أنها لا تخضع لحد الفترة الأساسية. تقع فلسطين في منطقة نشطة زلزاليا حيث أنها تقع على امتداد صدع البحر الميت والذي يعد من أنشط الصدوع الزلزالية في منطقة الثرق الأوسط. لذلك، تم تطبيق أحمال زلزالية على جميع النماذج والتي تم حساب قيمها باستخدام تحليل طيف الإستجابة بالتوافق مع IBC 2015 و (العزم من تحليل الدرجة الأولى و تحليل الدرجة الثانية تم حسابه على العمود الحرج لجميع النماذج ونسبة الزيادة في العزم من تحليل الدرجة الأولى الى تحليل الدرجة الثانية تم حسابه. تم إستخدام برنامج التحليل الإنشائي 19.0 SAP2000 بالتوافق مع الحسابات اليدوية. كنتيجة نهائية، على الأرجح، المهندسين الإنشائيين لن يواجهوا مشاكل مع P-Delta اذا تم بناء منشاءات إطارات بوابات خرسانية مسلحة حتى 25 طابق والتي تخضع للكود وحتى 8 طوابق والتي لاتخضع للكود.


[^0]:    1 Modal Response Spectrum Analysis

[^1]:    * Assumed forces

[^2]:    * Acceptance limit is $10 \%$

[^3]:    * Acceptance limit is $10 \%$

[^4]:    * Acceptance limit is $10 \%$

[^5]:    * acceptance limit is $10 \%$

[^6]:    * acceptance limit is $10 \%$

